TECHNICAL DEVELOPMENT REPORT NO. 350

A PILOT QUESTIONNAIRE STUDY
OF COCKPIT VISIBILITY REQUIREMENTS
FOR ARMY HELICOPTERS

FOR LIMITED DISTRIBUTION

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Rollin E. Farrand Aircraft Division

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SUMMARY

A questionnaire concerning problems of visibility from helicopter cockpits was distributed to 1,341 military helicopter pilots as a part of a broad study relating to helicopter cockpit windscreen outlines, 535 of these questionnaires were completed and returned. The questionnaire included both specific evaluation of visual characteristics of present helicopters for various maneuvers and differing directions of sight, and a general evaluation of different visual problems. The results of the questionnaire study seem to indicate that the pilots have not developed a general awareness or concern for the visual characteristics of helicopter cockpits. This probably is due, at least in part, to the generally good visibility from present helicopters. Most pilots rate the visibility from the helicopter they fly as satisfactory.

Analysis of the questionnaire data shows that visual cutoff angles for the different directions of sight may be defined in terms of windscreen outlines which correspond to pilot ratings of adequate or excellent. In the determination of these visual cutoff angles, the attitude of the helicopter in various maneuvers must be considered. Reasonable accuracy appears to be obtained by this method.

The results of the analysis of the questionnaire data are presented in relation to various specific visual problems and are combined into a general set of requirements which include visual cutoff angles and post arrangements. The results are not considered to be a final solution to the problem of visibility from helicopter cockpits, but offer a useful guide to the amount and quality of visibility which will be considered satisfactory by pilots.

INTRODUCTION

During the past several years, the problem of visibility from aircraft cockpits, as determined by window size, type, and location, has been receiving increased attention. Such attention has arisen as a result of midair collisions in which the pilots involved did not see the colliding aircraft until the instant of impact and also as a result of large variations in visual characteristics of modern aircraft due to a lack of basic standards.

The Department of the Army, cognizant of the problem already existing in fixed-wing aircraft and realizing that this same problem easily could become a serious defect in helicopters, decided to sponsor a study of the cockpit visual characteristics of helicopters. During 1955, the Civil Aeronautics Administration (CAA) Technical Development Center (TDC) commenced this investigation for the Department of the Army. Its purpose was to establish, upon some reasonable basis, definite standards which would insure adequate visibility for the safe operation of helicopters during the performance of all normal flight and military maneuvers. In this broad investigation, several different lines of approach are under way. The pilot questionnaire and supporting data presented in this report represent one line of approach, supplying contributory data but not a final solution to the problem.

While the evaluation of specific windscreen outlines is based on pilot opinion as expressed in the questionnaire, the angular measurements discussed herein were made with a binocular cockpit visibility cameral developed at TDC for this purpose. Some information derived from a study of helicopter flight path performances is presented here in support of various data derived from the questionnaire.

HELICOPTER DATA

Cockpit Visibility.

The quality of existing visibility from the H-13G, H-19, H-21, H-23, H-25A, H-31, H-34, H-37, XH-40, HOK-1, HRS-3, and CH-1 helicopters was recorded with the Center's binocular cockpit visibility camera. The photographs, as well as front and profile photographs of each helicopter, are presented in Appendix I. Data contained in the photographs were reduced further to tabular form and are presented in Table I with supplementary information concerning lineal relationships of the pilot to the cockpit and the ground. The eye-level position in a cockpit will vary with each individual pilot, depending on his height and the type of seat. This variation will be greater in those helicopters which do not have adjustable seats. All binocular vision photographs were obtained at the eye-level location of a pilot of average height, namely, 5 feet 10 1/2 inches. In reducing the photographic data to tabular form, any obstruction to vision over an arc equal to or greater than 10° was considered significant, and cutoff angles were measured to this point. When the copilot obscured portions of the cockpit

Thomas M. Edwards, "Development of an Instrument for Measuring Aircraft Cockpit Visibility Limits," CAA Technical Development Report No. 153, January 1952.

windows or openings, the cutoff angles were measured to the copilot's body. All binocular cockpit visibility measurements were made with the copilot's seat occupied so that actual, rather than theoretical, openings were recorded.

True appreciation of the binocular cockpit visibility photographs shown in Appendix I requires a knowledge of the limitations of the instrumentation. The binocular cockpit visibility camera records the true field of view, including binocular effect, of a pilot scanning the horizon with simple head rotation. The resulting outlines can be used to describe cockpit visibility limitations based on simple head rotation in a horizontal plane and eye movement. The binocular cockpit visibility photographs in Appendix I should not be used to determine absolute limits of foveal vision in any direction since the pilot's viewpoint and visual cutoff angle will change with vertical head rotation as shown in Fig. 1. Normally, a pilot can be expected to use eye movement only to 15° in any direction from the centroveal position. Beyond 15°, head movement must be considered. difference between the true and apparent visual cutoff angles is a function of the elevation angle from the horizon and the distance from the pilot's viewpoint to the windsheeld. The true cutoff angle will always be smaller than the angles given in the binocular cockpit visibility photographs. When judging areas of visibility, it must be remembered that the spherical limits are projected on a grid similar to a Mercator projection. The true solid angle of visibility therefore decreases with increasing elevation angle above and below the horizon.

Another factor which influences the visibility limits is head movement. The increase in visibility resulting from movement of the head toward a particular window is inversely proportional to the distance from the pilot's eyes to the window. The closer the window, the greater is the increase in visibility with a given amount of head movement. This is particularly important during use of the side window, which generally is very close to the pilot.

Flight Characteristics.

A knowledge of the flight characteristics of helicopters, that is, flight-path angles, helicopter attitudes, and so forth, is essential in the consideration of cockpit visibility requirements. This information was obtained through a photographic study of various helicopter maneuvers. Three special-purpose cameras were located as shown in Fig. 2 to record the necessary data. A Fairchild flight analyzer, Fig. 3A, was used to record a straight flight path, Fig. 3B, of approximately 1,200 feet including the touchdown point. A Varitron camera, Fig. 4A, operating at 2 frames per second, was used to record the touchdown area and approximately 200 feet of the

adjacent flight path, as shown in Fig. 4B. A Traid camera, Fig. 5A, remotely controlled to expose one frame, Fig. 5B, each time the Fairchild flight analyzer exposed a segment, was used to detect crab or yaw.

The flight analyzer must track a subject aircraft along a portion of the flight path equal to one-fortieth of the offset distance before the shutter will trip and expose a strip of the negative. Originally, this caused some concern lest critical attitudes occur during the period of generally vertical flight that occurs after the flare but before touchdown and remain undetected. The Veritron sequence camera was used to study this period of flight. This phase of the study indicated that critical attitudes exceeding those being recorded with the flight analyzer were not occurring in this area, therefore, the data obtained with the Fairchild flight analyzer as presented here are considered complete.

The flight analyzer was used to record the performance of five maneuvers. cruise, normal approach, normal takeoff, steep approach, and autorotation. Due to the characteristics of the camera described above and the offset distance as indicated in Fig. 2, this camera was not used to secure confined area data or maneuvers involving near-vertical performance. A summary of the data obtained is presented in Appendix II.

QUESTIONNAIRE STUDY

Procedure.

The questionnaire was developed from a similar one used by this Center to study problems of visibility from the cockpits of transport aircraft.² This questionnaire was discussed with individual pilots at Fort Rucker, Ala., to determine whether or not. (a) it could be easily understood, (b) the pilots' answers could be easily interpreted, and (c) the questionnaire included all information these pilots believed pertinent to this study. As a result of these discussions with pilots, some minor changes were made in the original questionnaire.

Later, 20 pilots completed the questionnaire during five group interviews. At this time, Question 9, dealing with visual references, was added. The questionnaire, as it was distributed to the entire group of pilots, is reproduced in Appendix III. The questionnaire was distributed to pilots in the field directly and also by mail through Post and various

George L. Pigman and Thomas M. Edwards, "Airline Pilot Questionnaire Study on Cockpit Visibility Problems," CAA Technical Development Report No. 123, September 1950.

Continental Army Commands. An addressed return envelope was included for mailing the completed questionnaire directly to the CAA Technical Development Center, Indianapolis, Ind.

Five hundred and thirty-five completed questionnaires were returned, representing 39.9 per cent of the original distribution. The data contained in these questionnaires were transferred to punchcards, and the final tabulation, classification, and totalization of data were carried out by machine. Prior to the final machine analysis, a preliminary investigation of various groupings was conducted with 135 completed questionnaires. Answers to Questions 4 to 9, inclusive, were grouped according to the pilots' experience as determined by their answers to Questions 2 and 3. Those groupings which evidenced no definte pattern were eliminated. The remaining groupings then were used as the basis for the machine analysis of the 535 completed questionnaires that were returned.

General Presentation and Limitations of Basic Data.

Answers to the questionnaire were received for each of the eight helicopters listed in Table II. The number of pilots answering the questionnaire for each of these helicopters varied considerably. The number of completed questionnaires returned for the H-31 and H-37 helicopters was considered, in each case, to be too small to be significant. The answers of the H-25 pilots were included, however, the small sampling must be considered when applying these data.

Most of the helicopters covered by this study have visual cutoff angles that the pilots as a group consider adequate. Because of this fact, it is difficult to determine accurately, based on pilot opinion, the exact areas in which visual cutoff angles become critical.

The data obtained for the various fixed-choice questions are presented either in terms of the percentage of all pilots or the percentage of each group of pilots answering. Where quantitative estimates were on an unrestricted basis, the mean value was used.

DISCUSSION OF QUESTIONNAIRE DATA

Question 1 - Pilot Height.

Early field work indicated a possible correlation between pilot height and pilot opinion. Subsequent work has not supported this belief and these data are not presented here.

Question 2 - Pilot Experience - General.

Here, as in Question 1, early work indicated a possible relationship between various kinds of pilot experience and pilot opinions. The questionnaire results and the flight-path studies have not revealed any definite relationship between the number of years a pilot has been flying or the total number of hours flying time in various types of aircraft and his opinions of the quality of visibility from the helicopter; therefore, these data are not presented here.

A trend was evidenced in Question 6 concerning the amount of obstruction acceptable when compared to the pilots' total helicopter experience; therefore, the distribution of total flying experience in helicopters has been included, and is given in Table III.

Pilots who had 50 hours more in all aircraft than their total hours in helicopters were considered to have both fixed-wing and helicopter experience. This definition is based on the fact that the minimum requirement for a private pilot's license is 40 hours' total flying time. The pilot distribution according to type of experience is indicated in Table IV. A further discussion of data where type of experience is a consideration is found under Question 7.

Question 3 - Pilot Experience - Specific.

Considerable use was made of a breakdown of answers based upon the type of helicopter considered. A distribution of pilots according to the type of helicopter most frequently used is contained in Table II. The preliminary analysis indicated that the experience in parts "b" and "c" of Question 3 had no influence on the remaining questionnaire data, therefore, this information is not presented here.

Question 4 - External Portions of Aircraft Visible from Cockpit.

The replies to this question show that 88 per cent of the pilots answering the questionnaire desire to see some external portion of the helicopter. These results are shown graphically in Fig. 6. Eighty-three per cent of the pilots wish to see the main rotor tips, 71 per cent of the pilots wish to see the main wheels or skids, and 20 per cent of the pilots wish to see the tail rotor. A relatively small number of pilots desire to see the main rotor head. This does not necessarily mean that this location is unimportant, when compared with the other two locations. The main rotor tips and main wheels or skids were listed in the questionnaire and merely needed a check mark to answer, whereas the other locations were suggested spontaneously by the pilots.

The fact that the particular model helicopter in which a pilot currently does most of his flying has a strong influence on the pilot's desire to see the wheels or skids is shown in Fig. 7. Pilots of H-21 and H-25 helicopters who cannot see the wheels readily are much less desirous of seeing this visual clue than pilots flying helicopters where the wheels are more easily seen.

Several other portions of the aircraft, including the tail wheel, running lights, position lights, passenger door, swash plate, litter pods, transmission, and the cooling fan, were specified by pilots under Question 4, but in each case, these represented less than one per cent of the pilots answering.

Question 5 - Maneuvers for Which Maximum Visibility is Required.

The data from this question are presented in Fig. 8. It can be seen that the pilots believe that the maneuver in which good visibility is most urgently required is landing in a confined area. The autorotation landing is their next choice, while hovering for rescue or similar ground work placed third in their preference.

Question 6 - Obstructions to Visibility.

The results of the replies to Question 6 are shown in Fig. 9, which gives the arithmetical mean of numerical ratings for each azimuthal segment specified in the question. It was assumed that the pilot was seated on the right side of the helicopter.

The mean values can vary from 1.00 to 3.00. If it is assumed that a mean value of 1.50 or less corresponds to Condition (1) on the rating scale, indicating a desire for no visual obstructions, then it may be concluded that the pilots desire to have clear vision over a range of azimuth angles extending from 30° left to 40° right. Similarly, if it is assumed that a mean rating value between 1.50 and 2.50 corresponds to Condition (2) on the rating scale, and a mean rating value greater than 2.50 corresponds to Condition (3) on the rating scale, then it may be concluded that the pilots desire no obstruction greater than 1 1/2 inches in projected width between the azimuth angles 30° left to 105° left and 40° right to 110° right, but will accept greater obstructions beyond these limits. The greater importance of visibility on the near side of the helicopter is apparent.

Question 7 - Downward Visibility.

This question was included as an additional means of securing data relating to downward cutoff angles. The pilots were requested to give the desired minimum horizontal distances along four directions of sight under two flight conditions, coming to a hover prior to landing and at the instant

of touchdown. The four directions of sight are shown in Fig. 10. The mean values of the distances in each direction, by helicopter of prime use, are contained in Table V, with a comparison of available and desired downward cutoff angles.

The results also are presented in Table VI, with relation to the type of flight experience recorded by the pilots. Pilots with only helicopter experience generally desired greater visibility than pilots with fixed-wing aircraft experience. The only answers of smaller mean values for fixed-wing aircraft experience were those of H-23 pilots. In this case, the sampling is subject to question, since only ten answers, or two per cent of the total response, were from H-23 pilots. A larger and more representative group of answers might not have supported this variation.

Question 8 - Evaluation of Specific Helicopter Being Flown by Pilot.

The distribution of responses to Question 8 by helicopter of prime use is shown in Table VII, expressed as a percentage of the total response for the group judging the particular aircraft. Since the pilots in the H-25 are seated to the left as opposed to the normal right-side seating for helicopters, the columns for response to the quality of visibility sideward are reversed in Table VII to reflect the normal near- and far-side comparison. As specified in the question, Class 1 signifies excellent visibility, Class 2 signifies adequate visibility, and Class 3 signifies inadequate visibility. There also is shown in Table VII the mean estimation value (M) for each helicopter and for each maneuver. This is the weighted mean value for the three numerical ratings. These mean values can vary from 1.00 to 3.00. A mean value of 2.00 would correspond to adequate visibility, and the amount of variation of the mean value from 2.00 corresponds to the degree of superiority or inferiority of visibility compared to adequate visibility.

None of the helicopters considered were rated inadequate in visibility on an average rating basis although visibility was rated more critically in connection with specific maneuvers.

The proximity of the pilot to the windshield in a given direction appears to be reflected in the pilots' ratings. For example, the H-25 helicopter has a smaller cutoff angle forward and up then the H-19, yet it is rated superior to the H-19 in this respect. The design of the windscreen apparently permits the pilot to increase materially the available visibility in the H-25 with acceptable head and body movement.

Question 9 - External References During Maneuvers.

Insufficient replies were received to Question 9 to warrant inclusion of these data.

Palot Comment on Miscellaneous Vasabality Problems.

A total of 56 pilot comments, received with the questionnaires, were of such a nature that they could not be included under any specific question. These comments dealt with specific visibility problems, and, in some instances, the pilots proposed correction. Some of these comments are presented below.

Fifteen H-34 pilots commented on poor visibility during formation flying. Through interviews with H-34 pilots, it was determined that they were concerned with the visibility obstruction caused by the magnetic compass during turns while flying echelon right formations.

Thirteen pilots mentioned visibility to the rear. Two of these pilots specifically recommended rear-view mirrors. During July 1957, H-34, H-19, and XH-40 helicopters were observed operating at Fort Rucker, Ala., with rear-view mirrors.

Three H-13 pilots and three H-21 pilots wanted the instrument console lowered. Two H-13 pilots recommended that the console be set at a 45° angle to reduce its height.

Two H-23 pilots noted a need for windshield wipers.

A need for improvement in defroster equipment also was noted.

GENERAL DISCUSSION

Visibility Forward and Downward.

Three questions, Nos. 4, 7, and 8, relate to visibility forward and downward from the helicopter.

In the discussion of Question 4, it was stated that the pilots consider good visibility is most urgently required for landing in a confined area. The maximum restriction for a confined area would require a vertical descent from a hover to accomplish a successful landing. It is therefore assumed that the helicopter would be in an essentially level attitude at the instant of touchdown.

In Fig. 11, the maximum mean rating for all conditions forward and downward in each helicopter, as determined by replies to Question 8 and shown in Table VII, has been plotted against the available visual angles below the horizon shown in Table I. A smooth curve has been drawn to represent the average value of these ratings. This curve crosses the adequate rating line discussed in Question 8 at a cutoff angle of 29°.

In Table V, where the actual visual angles forward and downward available from a level attitude at instant of touchdown were compared with the desired angles, the H-21, H-23, and H-34 helicopters were listed as having less than the desired downward vision. These same helicopters were rated inadequate in Fig. 11.

Visibility Forward and Upward.

In Fig. 12, the maximum mean rating for the visual cutoff angle forward and upward under various conditions has been plotted against the actual cutoff angle. Since all of these points fall far below the minimum adequate rating line, it is apparent that the pilots do not consider any of the helicopters listed as having a visual problem forward and upward. A smooth curve drawn through these points intersects the minimum adequate rating line at a cutoff angle of 26°.

Visibility in the Left Sector.

The visibility in the left sector was rated as a whole in the questionnaire rather than being separated into individual components. For this reason, it is not possible to make the direct comparison such as was made for visibility forward. The over-all rating for visibility to the left, however, has been plotted against the individual cutoff angles that comprise visibility in the left sector.

Figure 13 gives a comparison of the maximum mean rating for visibility to the left with the upward cutoff angle at 90° to the left. A smooth curve representing the average of these points crosses the minimum adequate rating line at a cutoff angle of 15°.

In the discussion of Question 4, it was pointed out that 71 per cent of the pilots wanted to see the main rotor tips. Table VIII lists the cutoff angle 90° to the left and upward that is required to the tip plane path at that point. These angles all are less than the 15° determined in Fig. 13. The 15° angle leaves some latitude in all the helicopters listed in Appendix I.

Table VIII was computed without considering the coning angle of the various helicopters. The pilot opinion expressed in replies to Question 4, however, places sufficient emphasis on seeing the rotor tips to warrant reevaluation of this angle to meet the specific need if the coning action of the blade makes it impossible to see the rotor tips.

Figure 14 shows a comparison of the maximum mean rating for visibility to the left with the downward cutoff angle 90° to the left for

various helicopters. A smooth curve representing the average of these points intersects the minimum adequate rating line at a point representing a cutoff angle of 29°.

In Fig. 15, the maximum mean ratings for visibility to the left are plotted against the azimuth cutoff angle to the left. It is apparent that the H-13 data are inconsistent with the other data. This rating may reflect the influence of the greater visibility available to the H-13 pilot a short distance above the horizontal reference plane used to measure these cutoff angles. If the H-13 data are disregarded, a straight line representing the average of the remaining points crosses the minimum adequate rating line at a point representing a visual cutoff angle of 90.5°.

Visibility in the Right Sector.

Figure 16 gives a comparison of the maximum mean ratings for visibility to the right with the upward cutoff angle 90° to the right. The distribution indicated by this graph varies so slightly from a common horizontal line that no conclusion can be made from these data concerning the minimum adequate cutoff angle.

If, however, this angle is designed to provide an adequate visibility area to comply with the pilots' stated desire to see the rotor tips, Table VIII can be used to determine this minimum adequate visual angle. If the coning action of the rotor blade is disregarded, the minimum adequate angle to satisfy this requirement is 15°. In three instances, however, this angle would permit no blade coning. This condition is unrealistic. To provide for an average coning angle of 5° to 7°, 3 the cutoff angle 90° to the right and upward should be 20°.

In Fig. 17, the ratings for visibility to the right are plotted against the downward visual cutoff angles to the right. Again, a curve has been used to represent the average of these points. This curve intersects the minimum adequate rating line at a point representing a cutoff angle of 40.5°.

The maximum mean ratings for visibility to the right have been plotted against the azimuth cutoff angle to the right in Fig. 18. A smooth curve representing the average of these points intersects the minimum adequate rating line at a point representing an azimuth cutoff angle of 100°.

Raymond A. Young, "Helicopter Engineering," The Ronald Press Company, New York, N. Y., 1949, p. 177.

Combined Vision Requirements.

The conclusions that have been drawn from the pilot estimates made in answering the different questions concerning visibility from the cockpit along various directions of sight have been combined into one general and coordinated set of requirements in Fig. 19. This windscreen outline should be acceptable throughout the range of cockpit measurements listed in Table I, with the possible exception of the H-37. Helicopters of this larger size were not in general use at the time the questionnaire was distributed; therefore, sufficient data are not available to permit reliable predictions concerning the applicability of this windscreen outline to helicopters of this and larger sizes.

The vertical post arrangements presented in Fig. 19 have been translated from the reference planes of Fig. 9 to those of Fig. 19 based on the cockpit measurements listed there. The data in Fig. 9 are applicable throughout the range of cockpit measurements presented in Table I because they have been derived independently from the cockpit measurements.

The data presented here are based on the requirements of the several helicopters. Obvious differences in flight characteristics are shown in Appendix II. Helicopters with such a wide variation in flight characteristics will have varying visual requirements however slight these differences might be. To provide one universal solution, the extreme case has been considered in each instance.

CONCLUSIONS

- 1. The data secured from the 535 questionnaires completed by helicopter pilots are reasonably consistent. In general, very little difference exists between pilots having different amounts of experience. More disagreement on general problems is found between pilots flying different models of helicopters. This difference of opinion probably is associated with visibility and operating characteristics of the helicopter to which the pilot is accustomed.
- 2. Visibility in a forward and downward direction, which is critical during all landing meneuvers, is adequate if it extends 29° below the horizon.
- 3. Visibility in a forward and upward direction is critical during maximum performance takeoffs and, to a lesser extent, in other takeoff maneuvers. Visibility in this area is adequate if it extends 26° above the horizon.

4. Visibility to the side is the most critical during confined area operations. To be rated adequate, visibility to the left must be provided through an azimuth angle of 92°. The azimuth angle to the right should be 100°.

Visibility upward at 90° to the left should extend 15° above the horizontal, whereas visibility at 90° to the right should extend 20° above the horizontal.

Visibility downward at 90° to the left should extend 30° below the horizontal, and visibility downward at 90° to the right should extend 40° below the horizontal.

- 5. A majority of pilots desire to see the rotor tips and the main wheels or skids. They want to see the rotor tips with moderate head and eye movement but are almost evenly divided concerning the amount of movement required to see the wheels or skids.
- 6. The requirements illustrated in Fig. 19 are not considered in any sense a final answer to the helicopter cockpit visibility problem. The questionnaire method of investigation has basic weaknesses which lead to lack of precision and need for arbitrary interpretation. This method cannot reveal basic principles which are necessary for complete understanding of the visual cues unconsciously used by the pilot in the performance of various maneuvers. It is believed that the results of the questionnaire, however, have strong provisional validity, and, in particular, that a helicopter cockpit windscreen outline, in agreement with that presented in Fig. 19, would be classified by the average pilot as providing satisfactory visibility from the cockpit.

ACKNOWLEDGMENT

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TABLE I
GENERAL DATA RELATING TO HELICOPTERS

Walaganton	H-13	н-19	Wide Console H-21	Narrow Console H-21	
Helicopter	11-13	11-17	11-21		_
Normal Pilot Seating	Right	Right	Right	Right	Center
Total Visibility (steridians)	8.1	4.9	5•5	58	7-4
Per Cent of Total Steridians	65	39	7+1+	46	59
Monocular Visibility, in Steridians	1.0	1.3	1.7	19	1.1
Binocular Visibility, in Steridians	7.2	3.6	3.8	3.9	6.3
Up and Forward Visibility Cutoff Angle, in Degrees	158	70	85	82	113
Down and Forward Visibility Cutoff Angle, in Degrees	50	21	1.6	42	20
Left-Side Azimuth Cutoff Angle, in Degrees	85	91 ¹	107	80	83 ¹
Right-Side Azimuth Cutoff Angle, in Begrees	1 51	172	136	141	128
Left Upper Vision Cutoff Angle at 90°	Unlimit	ođ*4	31	31	Unlimited
Left Lower Vision Cutoff Angle at 900	37 *	28*	30*	25 *	42 *
Right Upper Vision Cutoff Angle at 900	Unlimit	ed 22	51	61	Unlimited*
Right Lower Vision Cutoff Angle at 90°	66	69	77	72	47
Pilot Eye Level Above Ground in Inches	62.5	122.25	115.5	115.5	60.75
Pilot Eye Distance to Windscreen at 90° Left in Inches	8	15	20	20	25.5
Pilot Eye Distance to Windscreen at $0^{\rm O}$ in Inches	42.5	25.5	25.5	25.5	39.75
Pilot Eye Distance to Windscreen at 90° Right in Inches	38	42	52.5	52,5	35. 5

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TABLE I (continued)
GENERAL DATA RELATING TO HELICOPTERS

Helicopter	H-25	H-31	H-34	H-37	XH-40	HOK-⊥
Normal Pilot Scating	Left	Right	Right	Right	Right	Right
Total Visibility (steridians)	3.7	5.5	5-5	5.0	4.9	5•~
Per Cent of Total Steridians	30	44	Jtpr	40	39	45
Monocular Visibility, in Steridians	0.9	1.6	1.6	12	13	1.0
Binocular Visibility, in Steridians	2.8	3.9	3 . 9	3.8	3,6	4•-
Up and Forward Visibility Cutoff Angle, in Degices	38	88	70	126	35	43
Down and Forward Visibility Cutoff Angle: in Degrees	43	42	24	17	55	55
Left-Side Azimuth Cutoff Angle, in Degrees	130	87	112	123	125	136
Right-Side Azimuth Cutoff Angle, in Degrees	843	98	148	162	113	162
Left Upper Vision Cutoff Angle at 900	38	112	20	16 ¹	22	32
Left Lower Vision Cutoff Angle at 900	58 *	30 *	33 *	20 *	22 *	26*
Right Upper Vision Cutoff Angle at700	9	42	1	3.4	22	42
Right Lower Vision Cutoff Angle at 900	24	66	70	51	58	60
Pilot Eye Level Above Ground in Inches	72.25	94	130 5	135 5		
Pilot Eye Distance to Windscreen at 90° Left in Inches	43 5	13.5	15 25	21 25		
Pilot Eye Distance to Windscreen at OO in Inches	41 75	39•75	20.5	214		
Pilot Eye Distance to Windscreen at 90° Right in Inches	14.5	40.5	48.25	67 25		

GENERAL DATA RELATING TO HELICOPTERS

Helicopter	HRS-3	CH-1
Normal Pilot Seating	Right	R_E h t
Total Visibility (steridians)	4.3	5 8
Per Cent of Total Steridians	34	46
Monocular Visibility, in Steridians	1.1	14
Binocular Visibility, in Steridians	3 2	4.4
Vp and Forward Visibility Cutoff Angle, in Degrees	60	53
Down and Forward Visibility Cutoff Angle, in Degrees	27	21
Left-Side Azimuth Cutoff Angle, in Degrees	88 ³	80
Right-Side Azimuth Cutoff Angles in Degrees	180	269 ²
Left Upper Vision Cutoff Angle at 900	1	8 **
Left Lower Vision Cutoff Angle at 900	30 *	25 *
Right Upper Vision Cutoff Angle at 900	9	54
Right Lower Vision Cutoff Angle at 900	69	77
Pilot Eye Level Above Ground in Inches		
Pilot Eye Distance to Windscreen at 90° Left in Inches		
Pilot Eye Distance to Windscreen at 0° in Inches		
Pilot Eye Distance to Windscreen at 90° Right in Inches		

Any obstruction 10° in width or greater is considered the visual limit. 1 - Vision obstructed between 10° and 5° above horizontal plane. Note

- 2 Copilot is not included.
- 3 Copilot is considered an obstruction

^{* - 900} obstruction by copilot. angle measured to lower window edge immediately in front of copilot or copilot's legs.

^{** -} Angle measured to front of rotor shaft.

TABLE II

DISTRIBUTION OF PILOTS PARTICIPATING IN THE QUESTIONNAIRE STUDY ACCORDING TO THE TYPE OF HILLICOPTER MOST FREQUENTLY USED

Type of Helicopter	No. of Pilots
Not Specified	6
H - 13	171
H - 19	48
H - 21	65
н - 23	87
н - 25	11
H - 31	1
H - 3 ¹ 4	143
н - 37	3

TABLE III

DISTRIBUTION OF RETURNED QUESTIONNAIRES COMPARED WITH TOTAL HELICOPTER EXPERIENCE

Helicopter Hours	No of Pilots
Not Specified	5
0 to 299	154
300 to 499	131
500 th 999	148
1000 to 1999	72
2000 to 2999	<u>1</u> 4
3000 to 4999	5

TABLE IV

PILOTS ANSWERING THE QUESTIONNAIRE GROUPED ACCORDING TO TYPE OF EXPERIENCE

Type of Experience	No. of Pilots
Not Specified	6
Helicopter Experience Only	158
Helicopter and Fixed-Wing Experience	365

TABLE V

A COMPARISON OF DESIRED VISUAL CUTOFF ANGLES WITH AVAILABLE CUTOFF ANGLES

Maneuver		Minimum Desired Ground Distance (feet)				ound	Cutoff Angle Downwa					rd (degrees)			
							Desired				Available				
Di	irection of		A	В	C	D									
Coming to a Hover		H-13	24	22	16	22	Ext	reme	ly v	arioble.	The	samo	. as	below.	
		H-19	23	22	17	19									
		H-57	34	20	17	17									
		H-23	17	12	14	17									
		II-25	56	30	23	30									
		H-34	19	18	11	15									
т- 6		g 1.		_	~		0	-	a	_	-	.		T.	
D.	rection of	-	A	В	C	D	A	B	C	D	A	_	C	D	
During Landing at		H-13	15	14	10	13	18	19	25	19	50	37	66	55	
Instant of		н-19	20	20	16	18	21	22	24	22	32		83	58	
Touchdown		H-21	19	15	13	17	24	28	30	25	16*	_	77	65	
		H-23	11	8	9	11	27	36	32	27			47	47	
		II-25	12	13	14	13	27	25	23	25	43	38		-	
		H-34	15	14	9	11	38	33	47	47	24 *	33	70	5 2	

^{*} Available downward angle less than desired.

Note. Directions of sight are shown in Fig. 10.

TABLE VI

A COMPARISON OF DESIRED VISIBLE GROUND DISTANCES BY TYPE OF EXPERIENCE

Direction of Sight	н Н-13	1	H-19		H-21		H-23		H-25 ²		H-3 ¹ +	
Direction	Helicopter Only	Both	Helicopter Ouly	Both	Helicopter Only	Both	Hellcopter Only	Both	Helicopter Only	B7 th	Halooplar Osly	DOTA
of Sight			Mar	ieuve	r - Wh	en C	oming	to a	Hover			
A B C D	14 14 10 14	27 24 19 25	14 13 6 8	30 28 24 26	19 15 14 15	46 22 19 19	58 21 23 22	15 11 13 16		56 30 23 30	19 18 12 13	18 17 10 18
Direction of Sight					uver - nstant		-		g at			
A B C D	12 10 8 11	16 15 11 15	12 12 5 7	26 25 23 25	17 12 10 12	22 18 17 21	24 15 16 22	10 7 8 10		12 13 14 13	15 13 8 11	14 17 10 10

- Note: 1. Type of surcraft flown helicopter only or helicopter and fixed-wing surcraft.
 - 2. H-25 pilots participating in this study had fixed-wing experience, therefore, no comparsson was possible.
 - 3. Directions of sight are shown in Fig. 10.

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TABLE VII

EVALUATION OF VISIBILITY AS PROVIDED BY SPECIFIC HELICOPTERS

Visibility Upward to Front

		H-1	-3			H-2	19		H-23				
Takeoff Run	1 97	2	3	M 1.04	1 65	2 33	3 2	M 1.38	1 84	2 13	3 3	M 1.19	
Straight Climb	98	2	0	1.02	63	31	6	1.44	88	11	1	1.14	
Cruise	98	1	ı	1.02	75	21	<u>}</u>	1.29	89	9	2	1.13	
Level Turns	97	2	1	1.01	63	33	4	1.42	84	14	2	1.17	
Final Approach	98	2	0	1.02	79	19	2	1.23	75	17	8	1.33	
Hover	98	1	1	1.03	75	23	2	1.27	88	10	2	1.37	
Landing	96	3	1	1.04	81	17	ı	1.21	84	8	8	1.23	
Autorotation Glide	99	1	0	1.01	79	19	2	1.22	84	11	5	1.20	
Autorotation Landing	98	1	1	1.02	77	23	0	1.23	75	16	9	1.34	
Maximum Performance Takeoff	95	5	0	1.03	61	17	12	1.51	83	14	3	1.20	
Landing in a Confined Area	98	1	1	1.03	73	25	2	1.29	72	15	13	1.41	
Average	97	2	ı	1.03	72	24	14	1.32	82	13	5	1.31	

Note: In terms of the percentage of pilots who rated in each of three categories.

- 1 ~ Excellent
- 2 Adequate
- 3 Inadequate

Evaluation (M) is a weighted average of categories 1, 2, and 3

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EVALUATION OF VISIBILITY AS PROVIDED BY SPECIFIC HELICOPTERS

Visibility Upward to Front

		H-	23			H-	25			H-34					
Takeoff Run	1 80	2 17	3 3	M 1.22	1 91	2 9	3	M 1.09	1 82	2 15	3 3	M 1.20			
Streight Climb	81	18	1	1.21	91	9	0	1.09	80	18	2	1.23			
Cruise	78	21	ı	1.24	91	9	0	1.09	82	16	2	1.19			
Level Turns	84	16	0	1.16	80	20	0	1.20	74	23	3	1.30			
Final Approach	79	21	0	1.21	82	18	0	1.18	80	20	0	1.20			
Hover	81	17	2	1.20	91	9	0	1.09	84	15	ı	1 18			
Landing	81	19	0	1.19	91	9	0	1.09	87	13	0	1.13			
Autorotation Glide	85	15	0	1.20	91	9	0	1.09	85	13	2	1.18			
Autorotation Landing	83	17	0	1.17	91	9	0	1.09	84	15	1	1.18			
Maximum Performance T keoff	80	17	3	1.22	73	18	9	1.36	74	19	7	1.33			
Lending in a Confined Area	83	17	0	1.17	100	0	0	1.00	82	15	3	1.21			
Average	81	18	1	1.20	38	11	1	1.12	81	17	2	1.21			

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EVALUATION OF VISIBILITY AS PROVIDED BY SPECIFIC HELICOPTERS

Visibility Downward to Front

		Н	-13			H	-19		H-21				
Takeoff Run	1 87	2 13	3 -	M 1.13	1 52	2 42	3 6	M 1.54	1 41	2 25	3 3 ⁴	M 1.95	
Straight Climb	90	9	ı	1 11	58	38	14	1.46	49	28	23	1.74	
Cruise	94	6	0	1.07	71	25	4	1.33	52	31	17	1.65	
Level Turns	91	8	1	1.09	60	35	5	1.44	5 ¹ 4	29	17	1.63	
Final Approach	76	21	2	1.28	38	44	18	1.81	17	25	58	2.42	
Hover	85	14	1	1.16	50	33	17	1.67	37	35	28	1.91	
Landin;	82	17	l	1.18	31	50	19	1.88	31	20	49	2 17	
Autorotation Glide	78	21	1	1.22	44	43	13	1.72	34	23	43	2.09	
Autorotation Landing	77	20	3	1,26	23	38	39	2 17	17	23	60	1.83	
Maximum Ferformance Takeoff	91	9	0	1.09	58	33	9	1.50	50	25	25	1.73	
Landin, in a Confined Area	71	23	6	1.36	73	25	2	1.29	72	15	13	1.41	
Avera_e	84	15	ı	1.19	51	37	12	1.62	41	25	33	1.87	

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EVALUATION OF VISIBILITY AS PROVIDED BY SPECIFIC HELICOPTERS

Visibility Downward to Front

			H - 23	ŀ		H	I - 25	j	H-34					
Takeoff Run	1 33	2 49	3 18	м 1.85	1 91	2 9	3 0	M 1.09	1 63	2 34	3 3	M 1.47		
Straight Climb	<u>4</u> О	52	8	1.68	91	9	0	1.09	65	32	3	1.37		
Cruise	49	44	7	1.58	91	9	0	1.09	73	26	1	1.27		
Level Turns	52	41	7	1 55	90	10	0	1.10	69	30	1	1.32		
Final Approach	22	33	45	2 21	73	27	0	1.27	ħΟ	51	9	1.69		
Hover	29	51	20	1.92	100	0	0	1.00	48	42	10	1.63		
Kanding	20	50	30	2,10	100	0	0	1.00	39	46	15	1.61		
Autorotation Glile	27	46	27	2.00	91	9	0	1.09	54	37	9	1.55		
Autorotation Landing	14	44	42	2.28	82	18	0	1.18	27	45	28	1.23		
Maximum Performance Takeoff	l+1	45	14	1.70	91	0	9	1.18	61	33	6	1.47		
Landing in a Confined Area	20	32	48	2.27	82	18	0	1.18	32	<u> 1</u> 14	24	1.91		
Average	32	44	24	1.92	89	9	1	1.12	52	38	10	1 65		

EVALUATION OF VISIBILITY AS PROVIDED BY SPECIFIC HELICOPTERS

Visibility in Left Sector*

		H	[-13	3			II - 19)	H-21			
Takeoff Run	1 94	2 6	3	M 1.06	1 46	2 48	3	M 1.60	1 69	2 2 <i>5</i>	3 6	M 1.38
Straight Climb	96	3	1	1.05	46	50	4	1 58	75	19	6	1.31
Cruise	96	4	0	1.04	54	44	2	1.48	70	25	5	1.34
Level Turns	88	10	2	1.14	29	50	21	1.92	59	33	8	1.48
Final Approach	95	5	0	1.05	44	48	8	1.65	58	38	Ħ	1.52
Hover	90	9	1	1.10	42	46	12	1.71	63	30	7	1.45
Landin	96	3	1	1.05	42	42	ló	1.75	63	25	12	1 42
Autorotation Glide	95	14	ı	1.06	1414	42	14	1.71	59	39	2	1.42
Autorotation Landing	94	5	1	1.06	27	44	29	2.02	58	36	6	1.48
Maximum Performance Takeoff	94	5	ı	1.06	46	48	6	1.60	69	30	1	1.33
Landin, in a Confined Area	87	11	2	1.15	29	38	33	2 04	47	41	12	1.66
Avera _e e	93	6	1	1.08	41	45	13	1.73	63	31	6	1.44

^{*}Evaluations apply to visibility requirements in both the horizontal and vertical lirections.

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EVALUATION OF VISIBILITY AS PROVIDED BY SPECIFIC HELICOPTERS

Visibility in Left Sector

		E	I - 23			H-	25			I	1-34	
Takeoff Run	1 59	2 36	3 5	M 1.46	1 18	2 64	3 18	M 2.00	1 54	2 42	3 4	M 1.51
Straight Climb	79	21	0	1.21	36	55	9	1 73	51	42	7	1.55
Cruise	82	18	0	1.18	82	18	0	1.18	59	35	6	1.47
Level Turns	56	34	10	1.53	20	70	10	1.90	37	43	20	1.52
Final Approach	73	26	1	1.28	36	5 5	9	1.73	53	41	6	1 54
Hover	73	24	3	1.31	36	64	0	1 64	45	47	8	1.62
Landin _o	75	2 2	3	1.27	36	64	0	1.64	50	43	7	1 57
Autorotation Glide	72	28	0	1.29	36	56	8	1.72	48	44	8	1.60
Autorotation Landing	40	35	25	1.85	36	0	64	2.24	40	48	12	1.70
Maximum Performance Takeoff	76	23	1	1 25	46	54	0	1.55	48	46	6	1.58
Landing in a Confined Area	41	34	25	1.27	18	55	27	2.09	3 <u>1</u> 1	45	21	1.87
Avera _b e	66	27	7	1.50	36	50	14	1.77	47	43	10	1.59

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EVALUATION OF VISIBILITY AS PROVIDED BY SPECIFIC MELICOPTERS

Visibility in Right Sector

	H-13				Η-	19			H-21				
Takeoff Run	1 64	2 31	3 5	M 1.41	1 79	2 17	3 4	M 1.29	1 95	2 5	3 0	M 1.05	
Strai_ht Climb	68	29	3	1.35	81	15	4	1.19	9 <i>5</i>	5	0	1.05	
Cruise	72	24	1+	1.31	90	10	C	1.10	95	5	0	1.05	
Level Turns	5 <u>1</u> 4	35	11	1.53	71	27	2	1.31	89	11	0	1,11	
Final approach	62	34	4	1.34	81	19	0	1.19	86	13	ı	1.21	
Hover	57	35	8	1.58	79	21	0	1.21	91	8	1	1.11	
Landins	66	28	6	1.39	79	21	0	1 21	91	9	0	1.09	
Autorotation Glide	61	33	б	1.46	85	13	2	1.17	92	6	2	1.09	
Autorotation Landing	60	33	7	1.47	79	19	2	1.23	88	9	3	1.10	
Maximum Periormance Takeoff	61	35	4	1.44	92	8	0	1.08	92	8	0	1.08	
Landing in a Confined Area	46	40	14	1.67	77	23	0	1.23	80	14	6	1 27	
Average	61	32	7	1.45	81	18	ı	1,20	90	9	1	1.11	

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EVALUATION OF VISIBILITY AS PROVIDED BY SPECIFIC HELICOPTERS

Visibility in Right Sector

		H	- 23			H	-25			H-3	4	
Takeoff Run	1 68	2 31	3 1	M 1.33	1 73	2 17	3	M 1.27	1 85	2 13	3 2	M 1.18
Straight Climb	49	44	7	1.58	82	18	0	1.18	82	17	1	1.18
Crulse	76	21	3	1.26	82	18	0	1.18	59	35	6	1.47
Level Turns	53	36	11	1 58	40	50	10	1.70	73	23	4	1.31
Final Approach	72	26	2	1.31	82	18	0	1.18	84	15	1,	1.17
Hover	69	27	4	1.34	82	18	0	1.18	84	14	2	1.16
Landing	73	25	2	1.17	91	9	0	1.09	85	14	ı	1.17
Autorotation Glile	73	24	3	1 30	91	9	0	1.09	85	13	2	1.16
Autorotation Landin	71	25	14	1,34	91	9	0	1.09	84	14	2	1.18
Maximum Performance Takeoff	75	23	2	1.27	91	9	0	1 09	87	13	0	1.14
Landing in a Confined Area	62	27	11	1.48	73	27	0	1.27	78	15	5	1.29
Average	67	28	5	1.36	80	18	2	1.21	81	17	2	1.22

T.BLE VIII

A COMPARISON OF AVAILABLE AND NECESSARY
VISION CUTOFF ANGLES SIDEWARD AND UPWARD
TO SEE THE MAIN OR FRONT ROTOR TIPS

	Required Angle (deg.)	Available Angle (deg.)	Required Angle (deg.)	Available fingle (deg.)
Helicopter	Near S	Side*	Far S	ide*
H-13	14.8	Unlimited	12.9	Unlimited
H-19	4.9	24 4	4.4	7.8
H-21	14.2	51.3	12.6	28.8
H-23	14.0	81.1	14.0	81.1
H-25	9•3	9.8	10.1	34.1
H - 34	8.8	1.0	8.0	19.8

^{*} Due to side-by-side seating arrangement where applicable

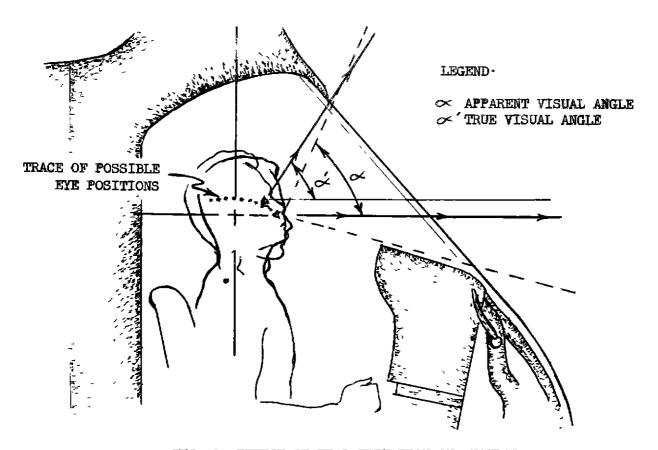


FIG. 1 EFFECT OF HEAD ELEVATION ON ANGULAR VISIBILITY LIMITS

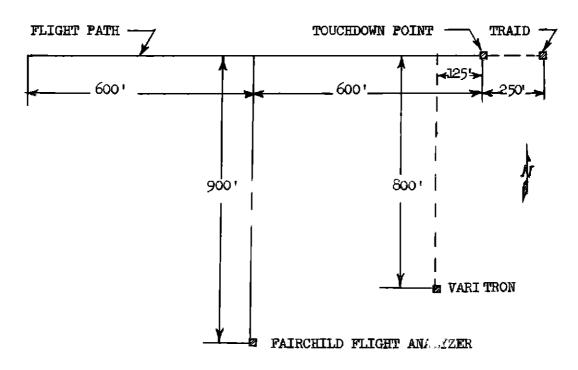


FIG. 2 CAMERA ARRANGEMENT FOR FLIGHT PATH PHOTOGRAPHS



FIG. 3A THE FAIRCHILD FLIGHT ANALYZER

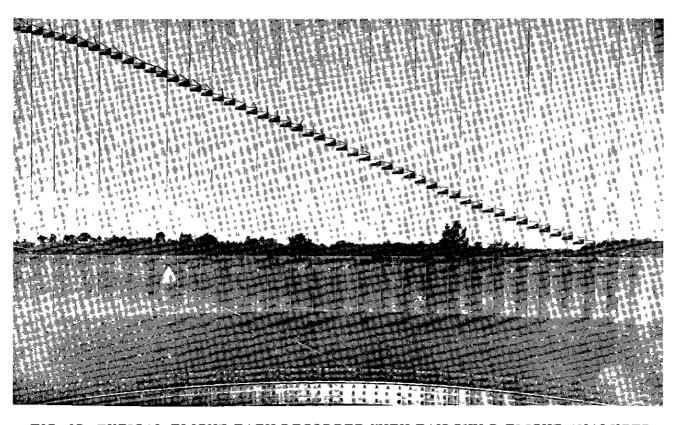


FIG. 3B TYPICAL FLIGHT PATH RECORDED WITH FAIRCHILD FLIGHT ANALYZER

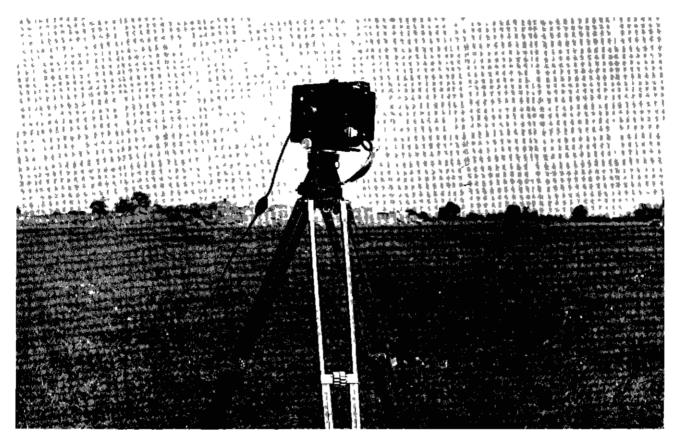


FIG 4A THE VARITRON CAMERA

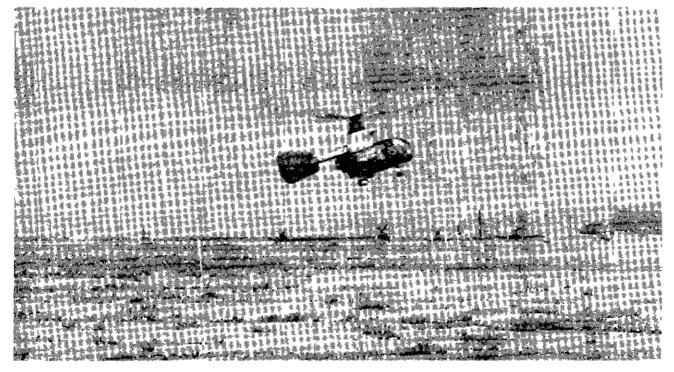
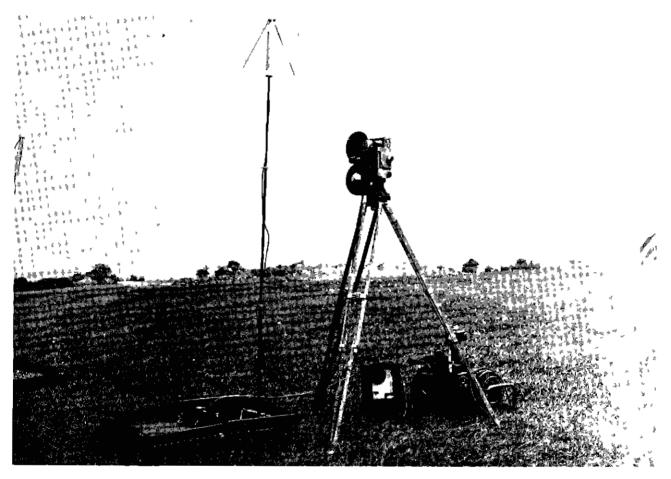


FIG. 4B TYPICAL PICTURE RECORDED WITH VARITRON CAMERA



FIG, 5A THE TRAID CAMERA

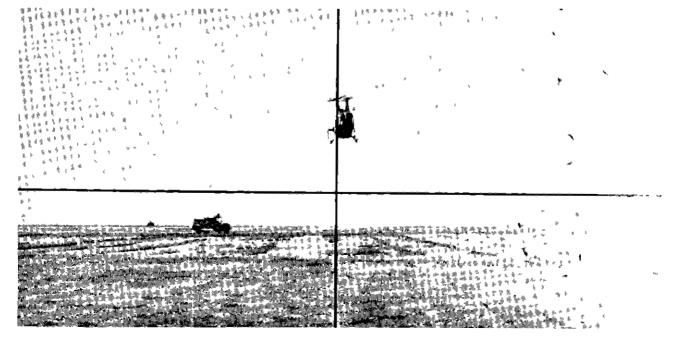


FIG 5B TYPICAL PICTURE TAKEN WITH TRAID CAMERA

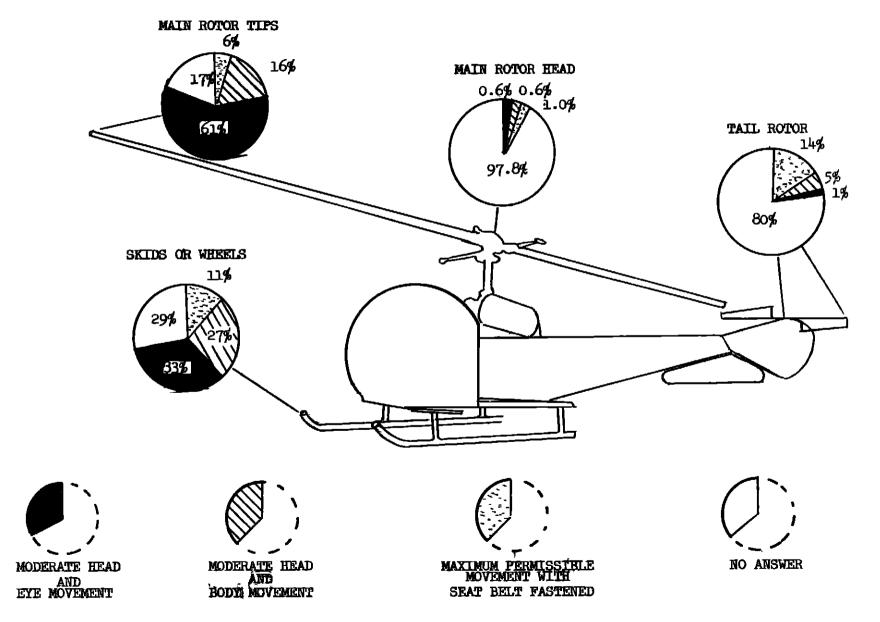


FIG. 6 PERCENTAGE OF FILOTS DESIRING TO SEE EXTERNAL PORTIONS OF THE HELICOPTER

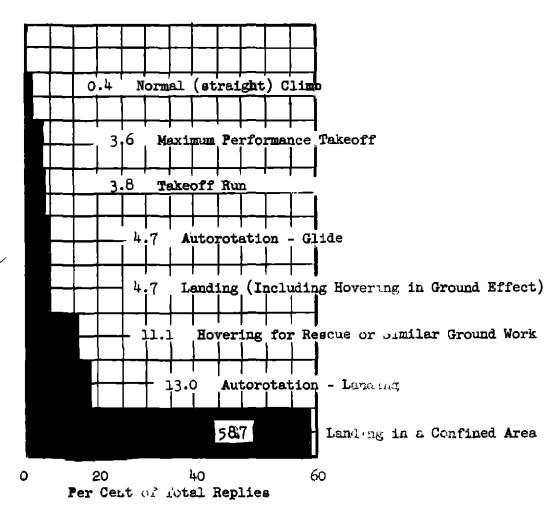


FIG. 8 RELATIVE IN TANCE OF VISIBILITY DURING COMMON MANEUVERS SHOWN IN PERCENTAGE OF TOTAL REPLIES

ESTIMATION RATING SCALE.

- 1. NO OBSTRUCTION PERMISSIBLE
- 2. MINOR OBSTRUCTION PERMISSIBLE \leq 1 1/2" WIDE 3. MORE OBSTRUCTION PERMISSIBLE > 1 1/2" WIDE

Permissible Obstruction

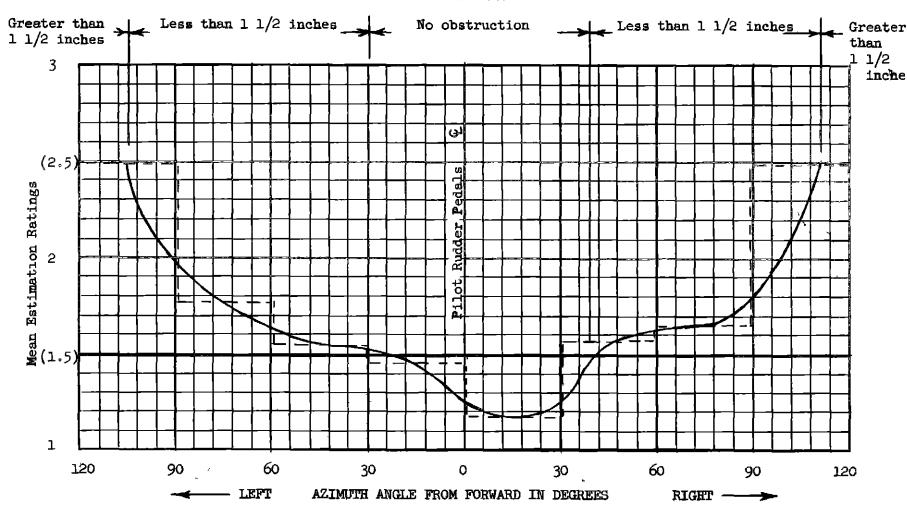
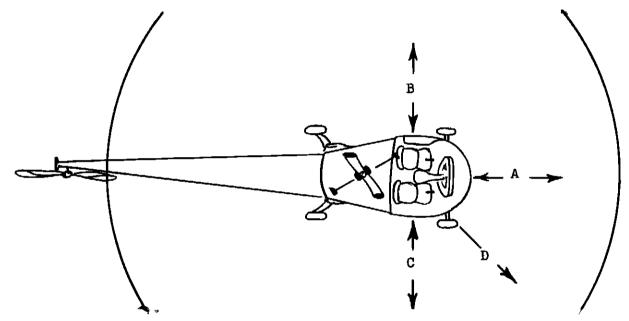


FIG. 9 VARIATION OF MEAN ESTIMATION VALUES OF OBSTRUCTION RATINGS WITH AZIMUTH ANGLE



NOTE THIS IS A PLAN VIEW OF A HELICOPTER. DISTANCES ARE TO BE GROUND (OR HORIZONTAL) DISTANCES.

NOTE: IF THE HELICOPTER YOU ARE CONSIDERING HAS THE NORMAL PILOT'S SEAT ON THE LEFT, CONSIDER DIMENSION D AS BEING BETWEEN A AND B.

NOTE. DISTANCE "C" IS CONSIDERED AS THE EXTERNAL DISTANCE IMMEDIATELY ADJACENT TO THE PILOT AND ALL QUESTIONNAIRES WERE INTERPRETED IN THIS MANNER CAUSING "B" AND "C" TO BE TRANSPOSED FOR PILOTS SEATED LEFT.

FIG. 10 LINES OF SIGHT FOR QUESTION 7

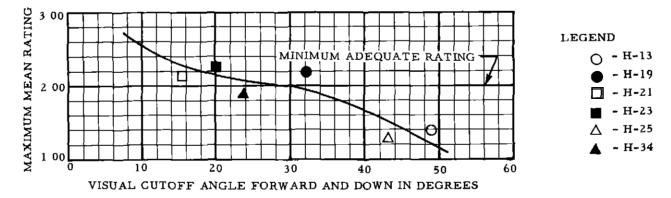


FIG 11 A COMPARISON OF THE MAXIMUM MEAN RATINGS (TABLE VII) FOR VISIBILITY FORWARD AND DOWN WITH THE VISUAL CUTOFF ANGLE FORWARD AND DOWN (TABLE I)

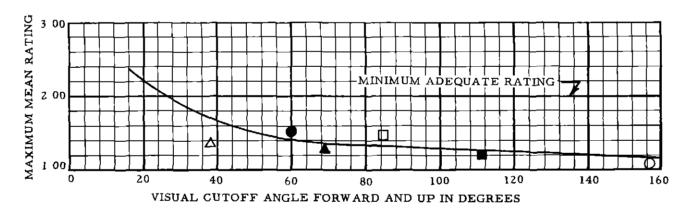


FIG 12 A COMPARISON OF THE MAXIMUM MEAN RATINGS (TABLE VII) FOR VISIBILITY FORWARD AND UP WITH THE VISUAL CUTOFF ANGLE FORWARD AND UP

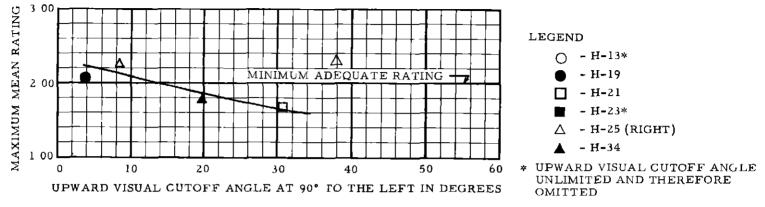


FIG 13 A COMPARISON OF THE MAXIMUM MEAN RATING FOR VISIBILITY TO THE LEFT WITH THE UPWARD CUTOFF ANGLE AT 90° TO THE LEFT

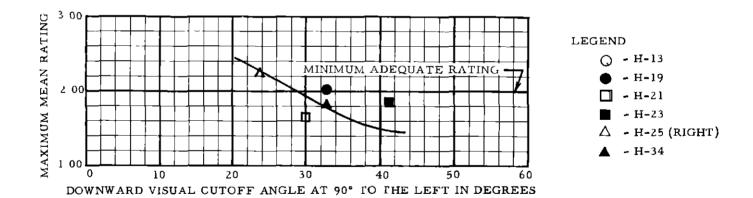


FIG 14 A COMPARISON OF THE MAXIMUM MEAN RATING FOR VISIBILITY TO THE LEFT WITH THE DOWNWARD VISUAL CUTOFF ANGLE AT 90° TO THE LEFT

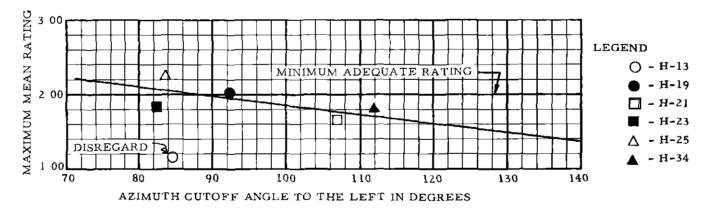


FIG 15 A COMPARISON OF THE MAXIMUM MEAN RATING FOR VISIBILITY TO THE LEFT WITH THE AZIMUTH CUTOFF ANGLE TO THE LEFT

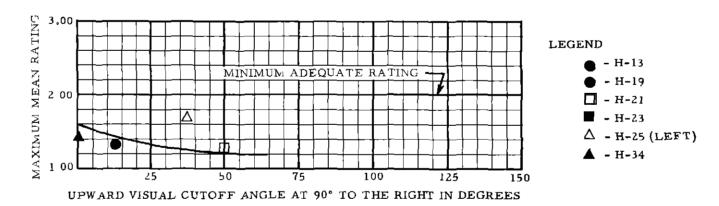


FIG 16 A COMPARISON OF THE MAXIMUM MEAN RATING FOR VISIBILITY TO THE RIGHT WITH THE UPWARD CUTOFF ANGLE AT 90° TO THE RIGHT

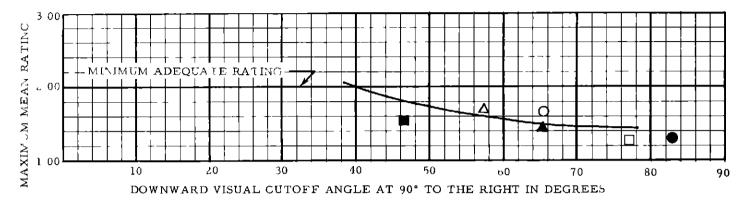


FIG 17 A COMPARISON OF THE MAXIMUM MEAN RATING FOR VISIBILITY TO THE RIGHT WITH THE DOWNWARD CUTOFF 90° TO THE RIGHT

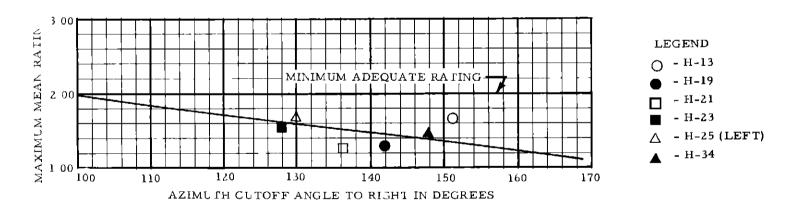


FIG 18 A COMPARISON OF THE MAXIMUM MEAN RATING FOR VISIBILITY
TO THE RIGHT WITH THE AZIMUTH CUTOFF ANGLE TO THE RIGHT

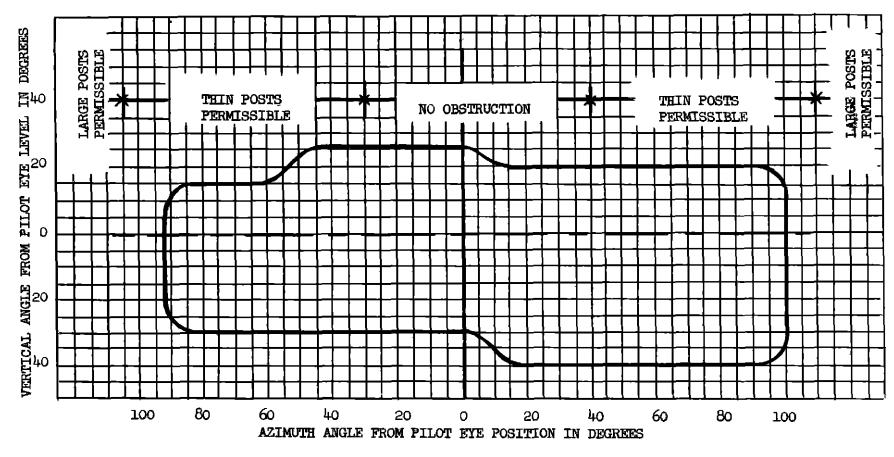


FIG. 19 RECOMMENDED HELICOPTER MINIMUM WINDSCREEN OUTLINE AS DETERMINED FROM PILOT QUESTIONNAIRE STUDY

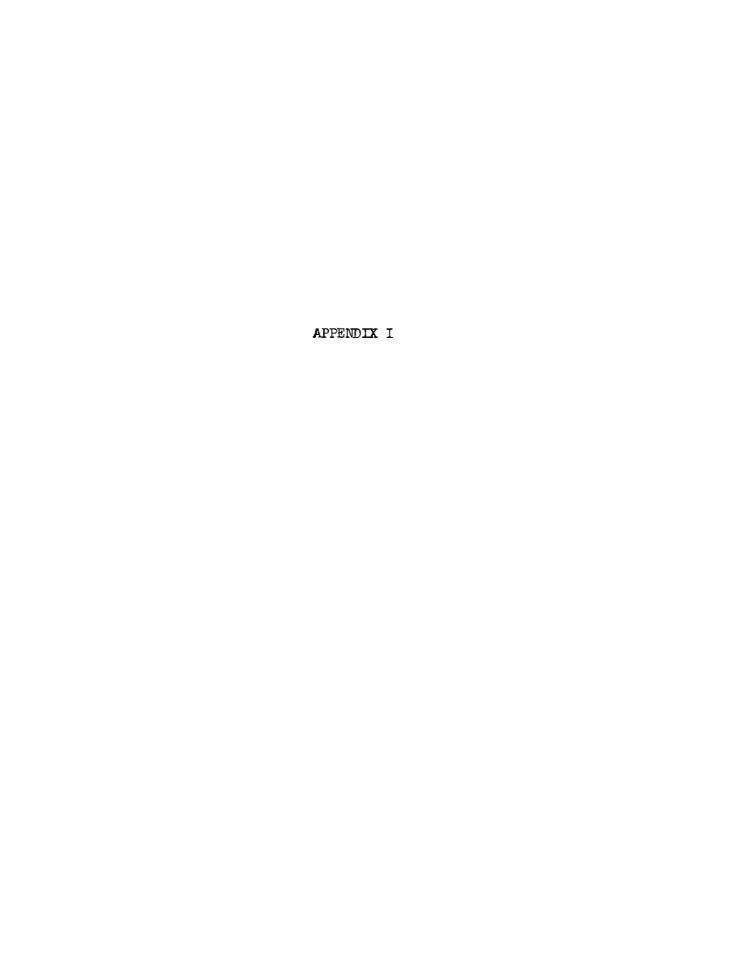


FIG 20 BELL H-13G BINOCULAR COCKPIT VISIBILITY PHOTOGRAPH

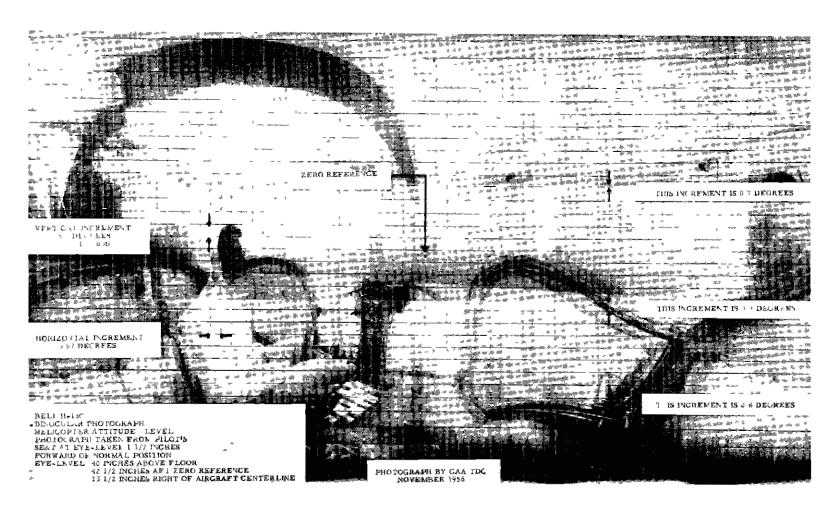




FIG 21 BELL H-13G FRONT VIEW



FIG 22 BELL H-13G SIDE VIEW

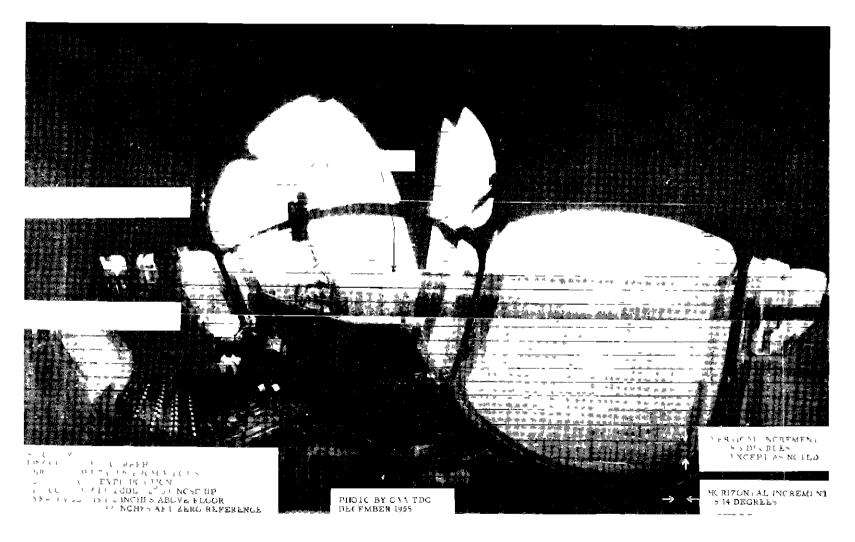


FIG 23 SIKORSKY H-19 BINOCULAR COCKPIT VISIBILITY PHOTOGRAPH

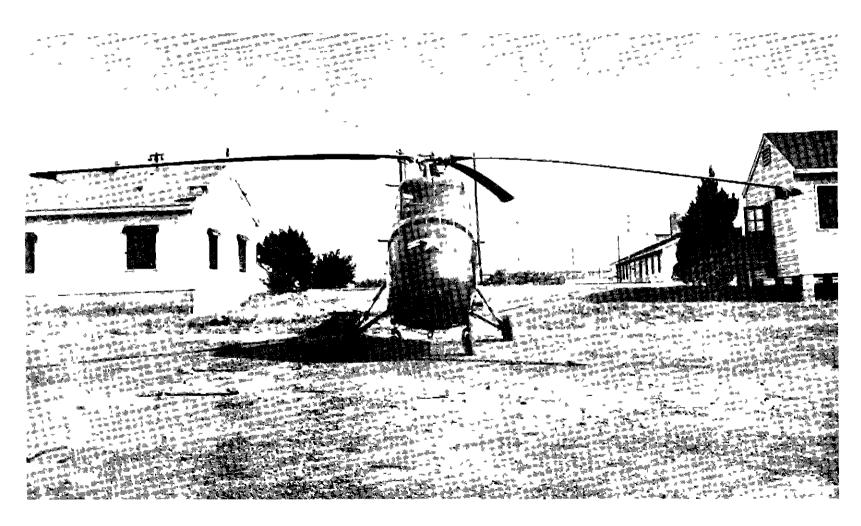


FIG. 24 SIKORSKY H-19 FRONT VIEW



FIG 25 SIKORSKY H-19 SIDE VIEW

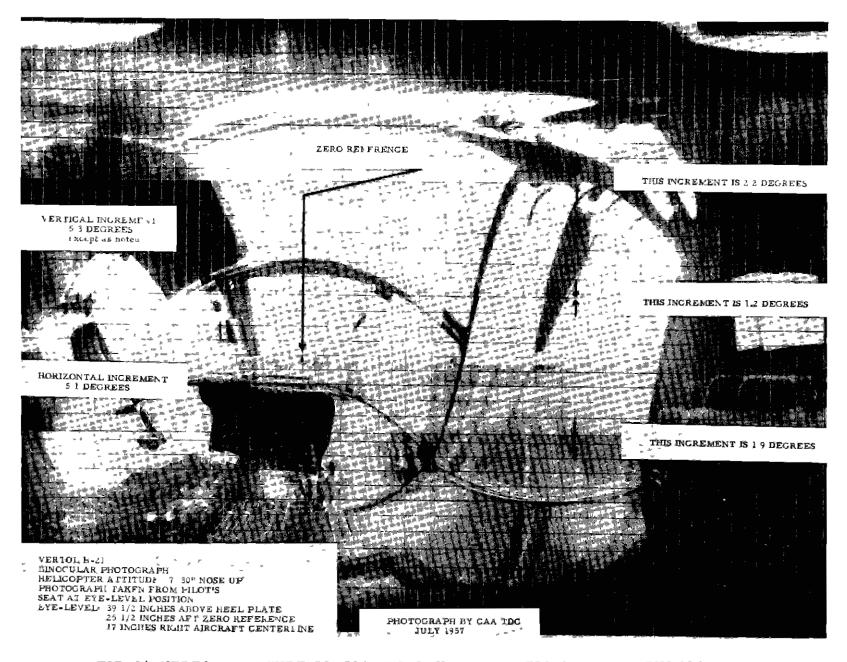


FIG 26 VERTOL H-21 WIDE CONSOLE BINOCULAR COCKPIT VISIBILITY PHOTOGRAPH



FIG. 27 VERTOL H-21 FRONT VIEW

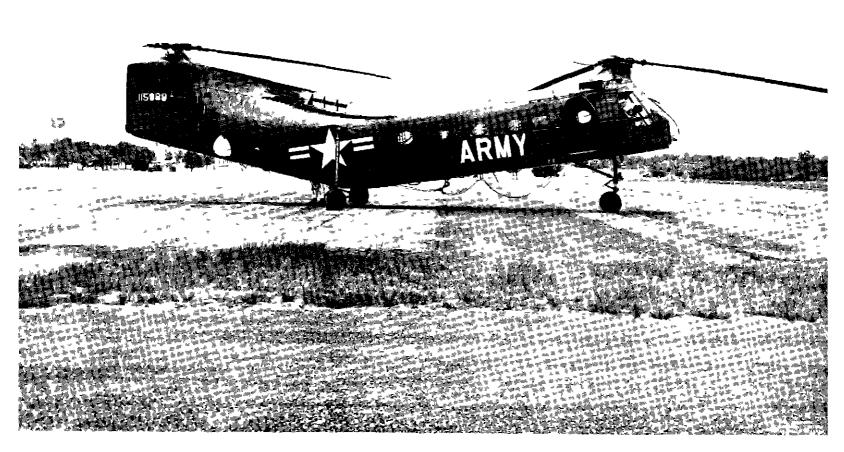


FIG 28 VERTOL H-21 SIDE VIEW

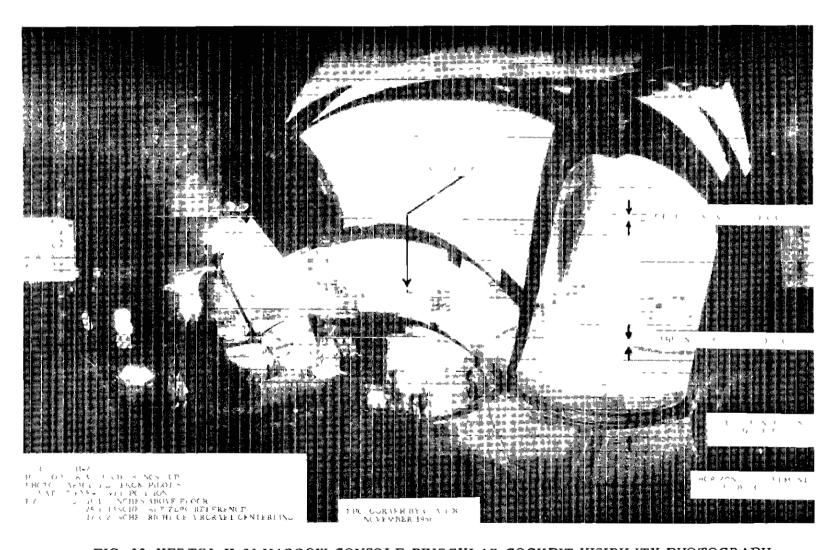


FIG 29 VERTOL H-21 NARROW CONSOLE BINOCULAR COCKPIT VISIBILITY PHOTOGRAPH

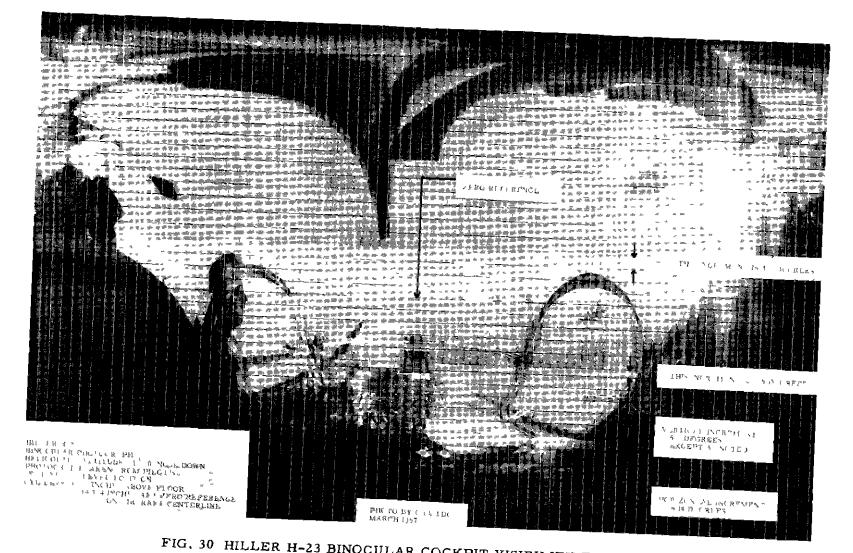


FIG. 30 HILLER H-23 BINOCULAR COCKPIT VISIBILITY PHOTOGRAPH

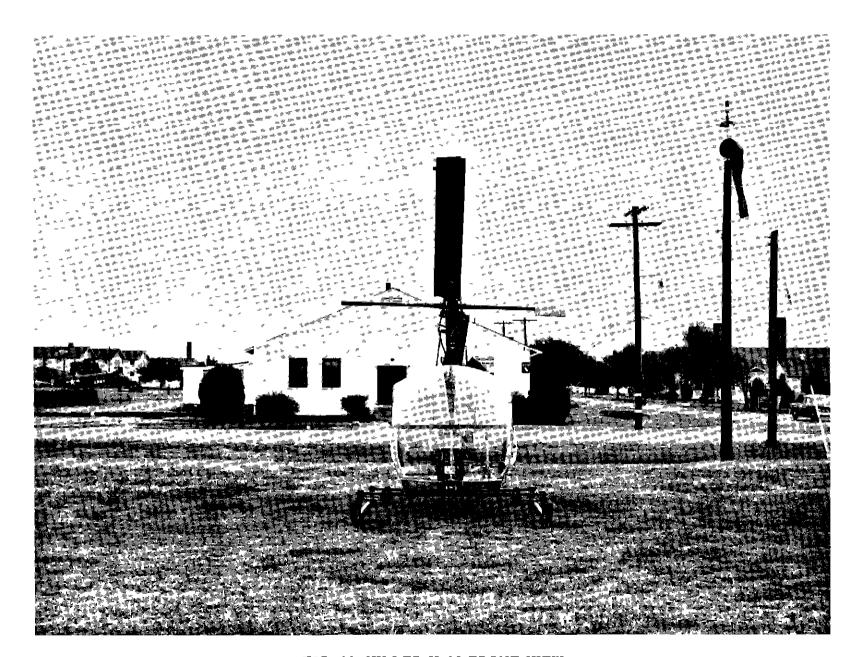


FIG 31 HILLER H-23 FRONT VIEW



FIG. 32 HILLER H-23 SIDE VIEW

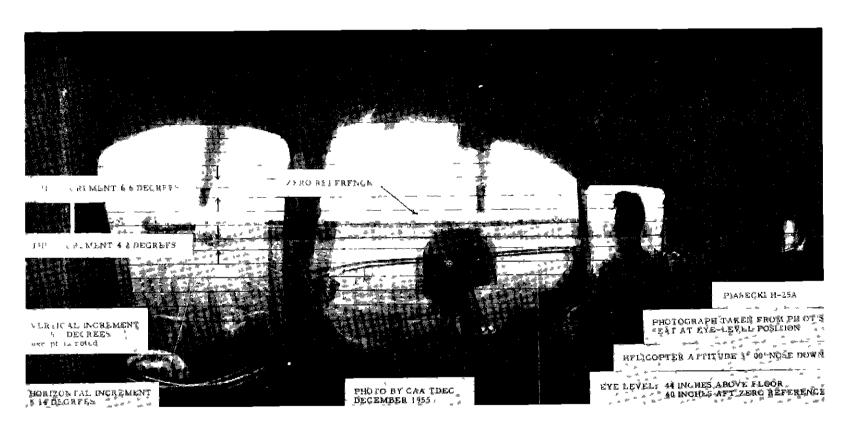


FIG 33 PIASECKI H-25A BINOCULAR COCKPIT VISIBILITY PHOTOGRAPH

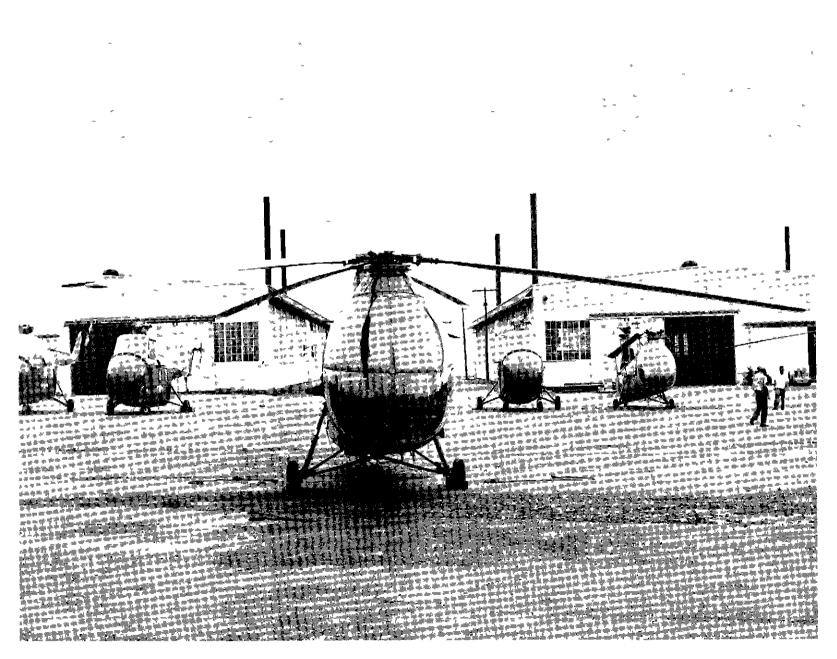


FIG 34 PIASECKI H-25A FRONT VIEW

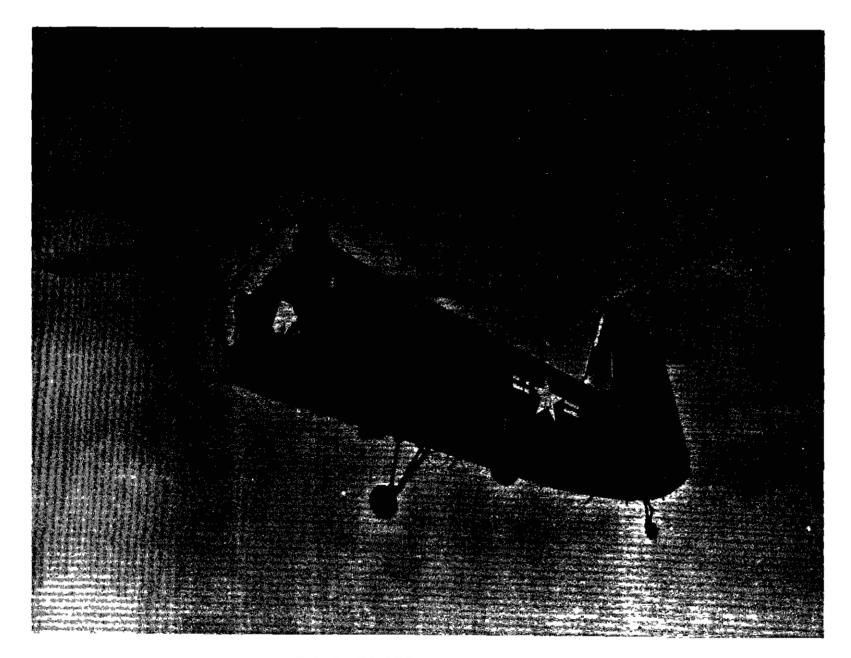


FIG. 35 PIASECKI H-25A SIDE VIEW

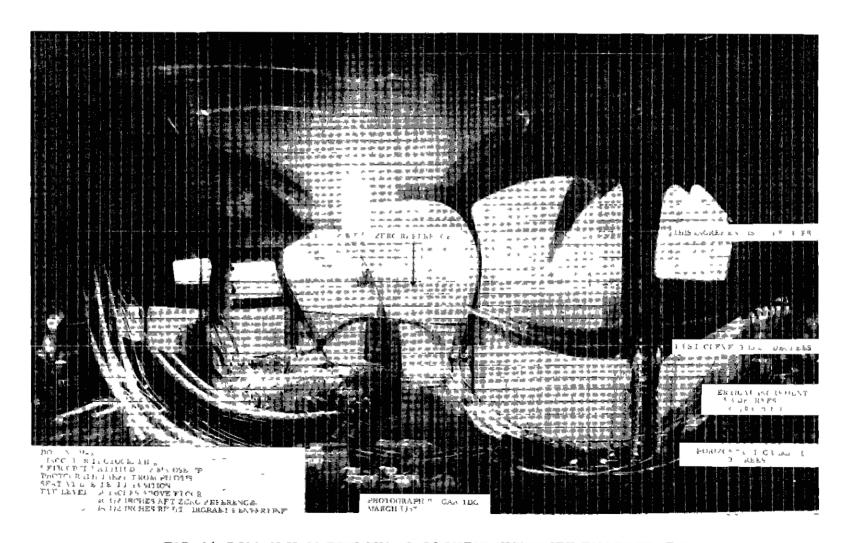


FIG. 36 DOMAN H-31 BINOCULAR COCKPIT VISIBILITY PHOTOGRAPH

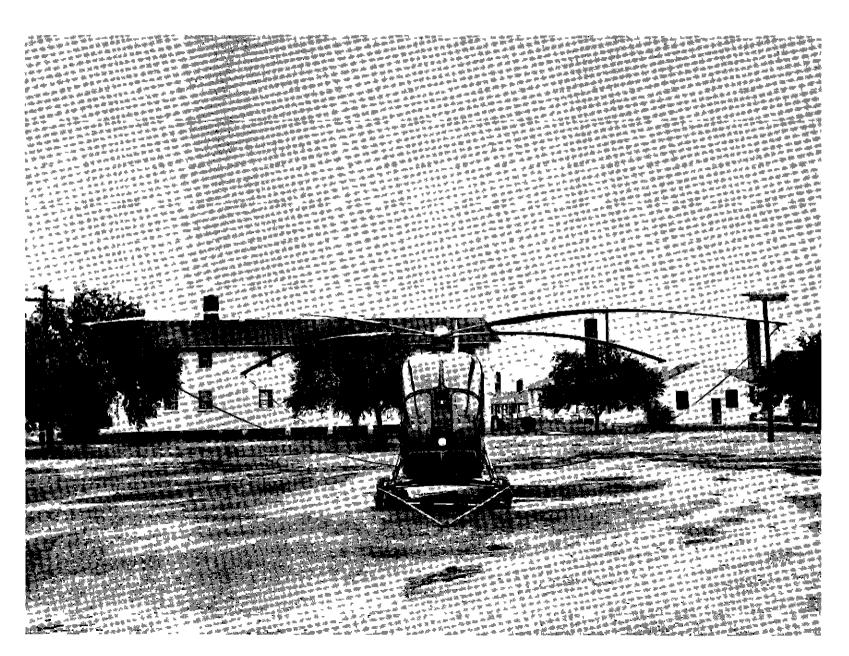


FIG. 37 DOMAN H-31 FRONT VIEW

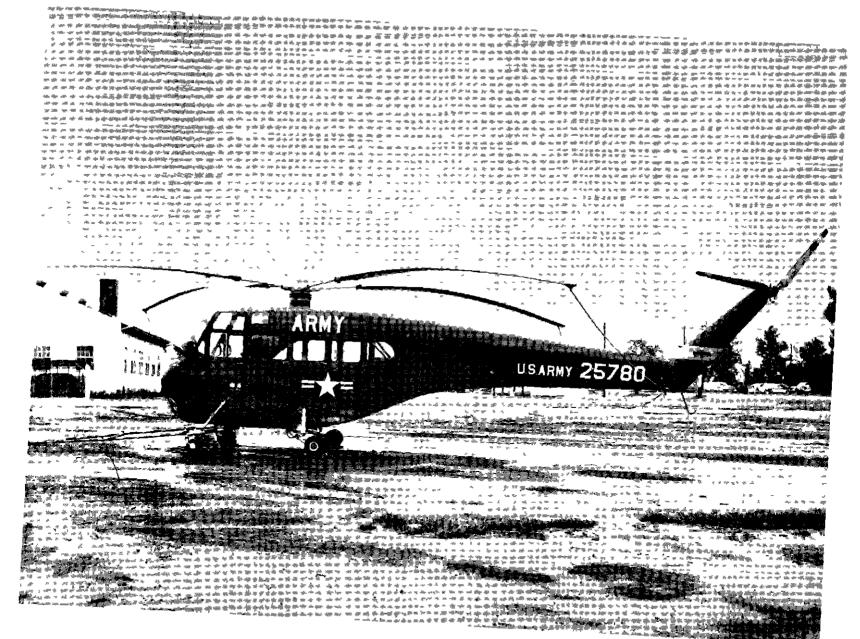


FIG 38 DOMAN H-31 SIDE VIEW

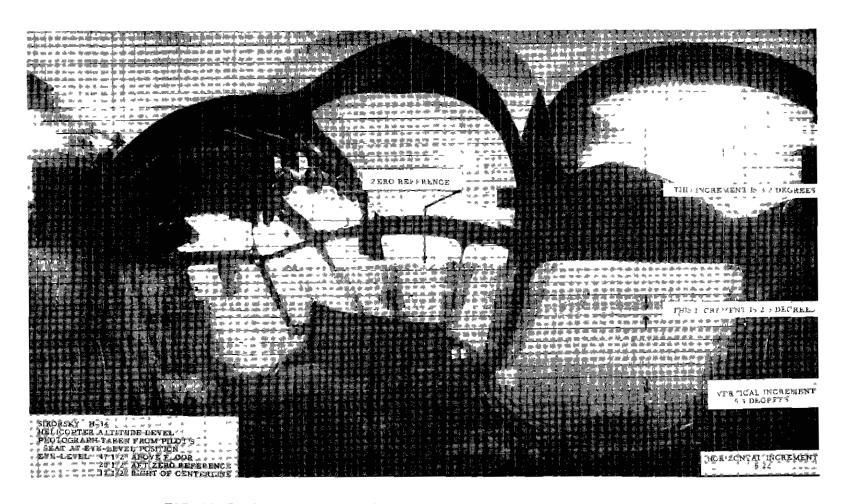
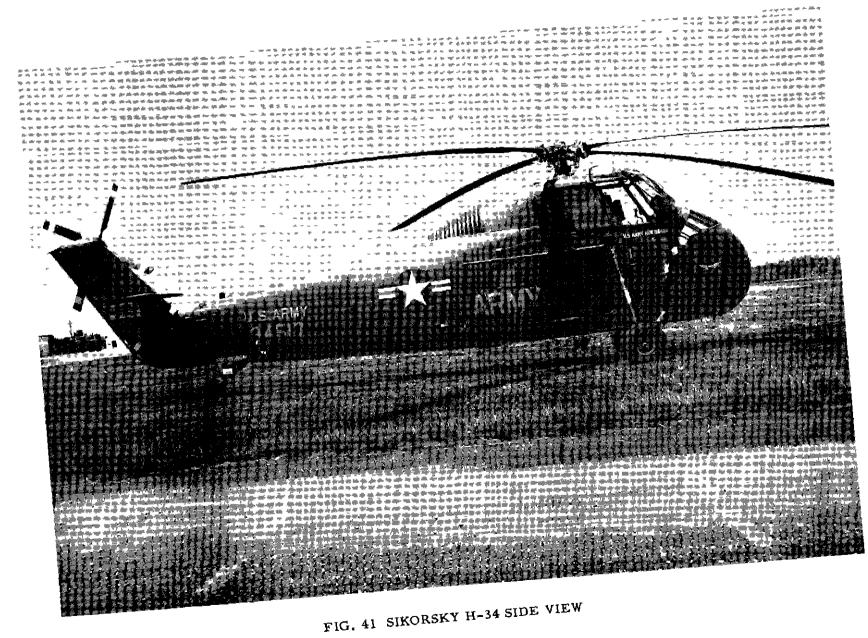


FIG 39 SIKORSKY H-34 BINOCULAR COCKPIT VISIBILITY PHOTOGRAPH



FIG 40 SIKORSKY H-34 FRONT VIEW



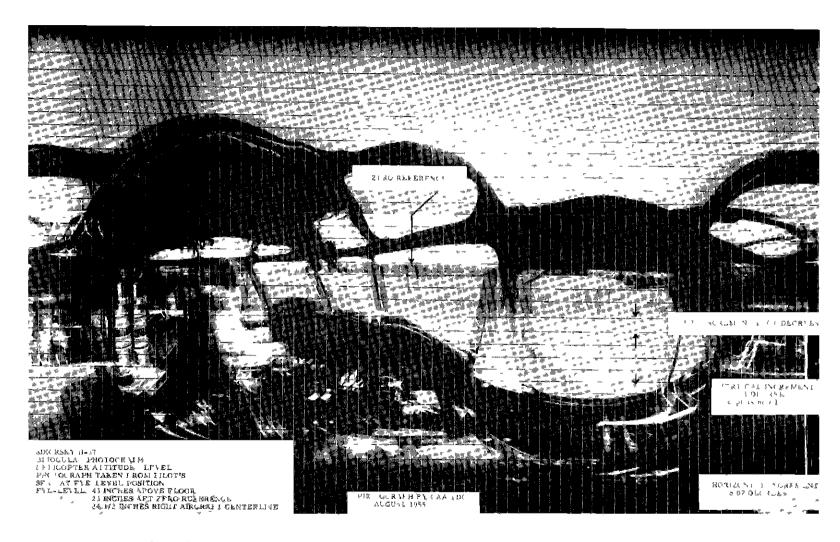


FIG. 42 SIKORSKY H-37 BINOCULAR COCKPIT VISIBILITY PHOTOGRAPH

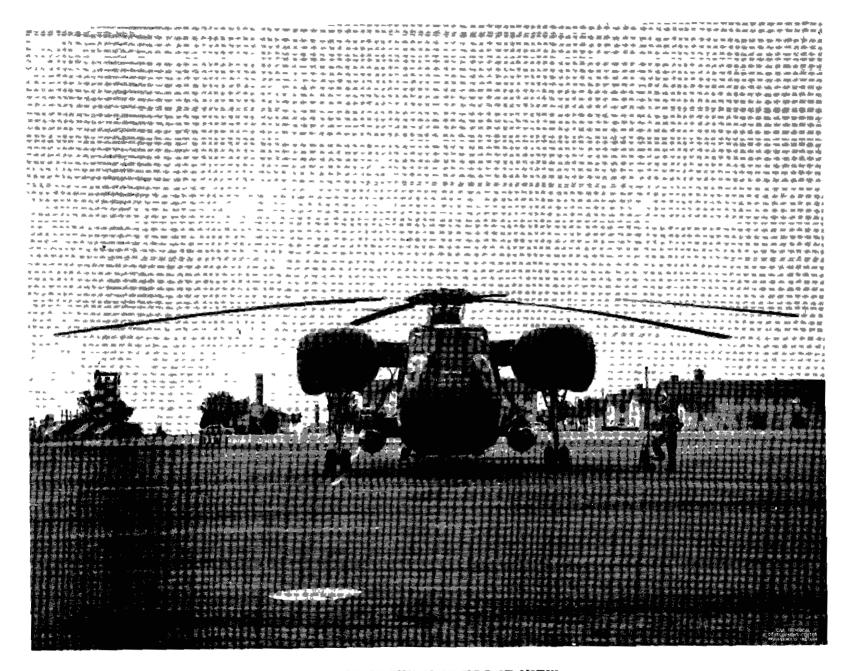


FIG 43 SIKORSKY H-37 FRONT VIEW



FIG 44 SIKORSKY H-37 SIDE VIEW



FIG 45 BELL XH-40 BINOCULAR COCKPIT VISIBILITY PHOTOGRAPH



FIG. 46 BELL XH-40 FRONT VIEW

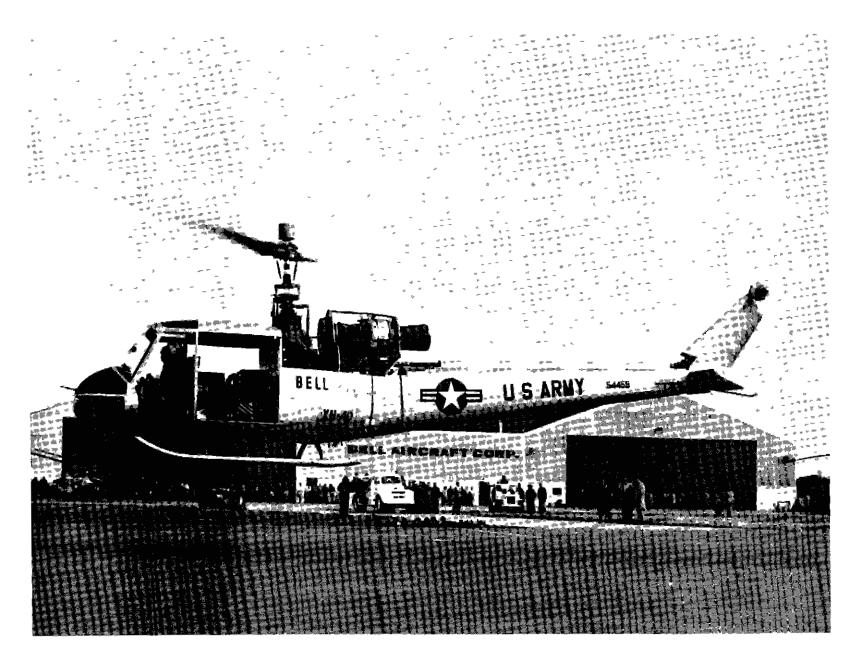


FIG 47 BELL XH-40 SIDE VIEW

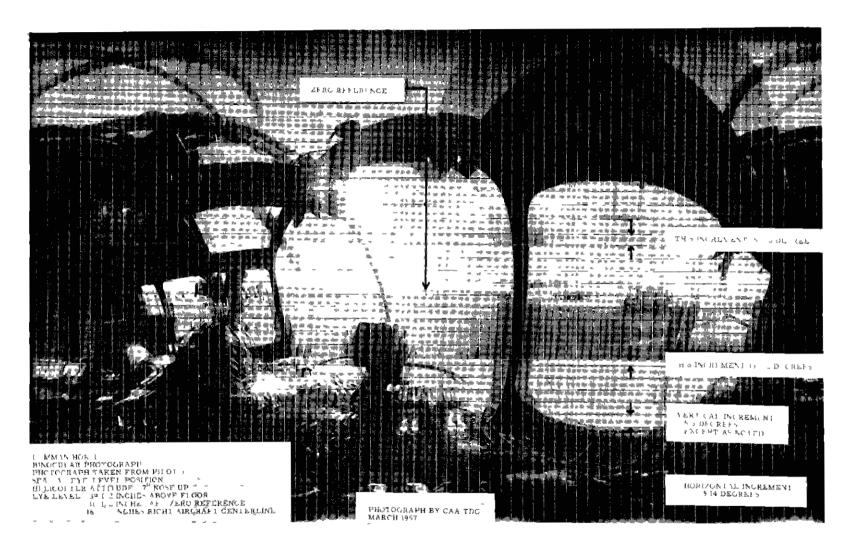


FIG 48 KAMMAN HOK-1 BINOCULAR COCKPIT VISIBILITY PHOTOGRAPH



FIG 49 KAMMAN HOK-1 SIDE VIEW

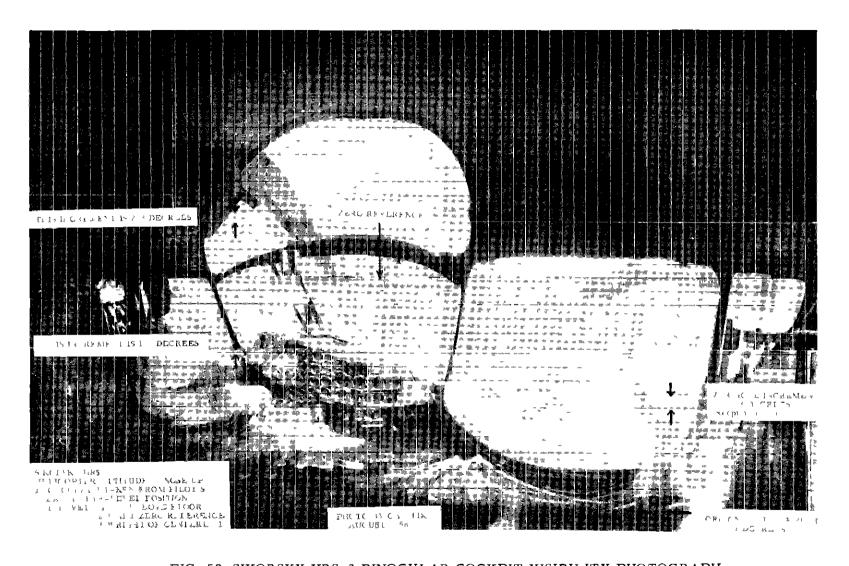


FIG 50 SIKORSKY HRS-3 BINOCULAR COCKPIT VISIBILITY PHOTOGRAPH



FIG 51 SIKORSKY HRS-3 SIDE VIEW

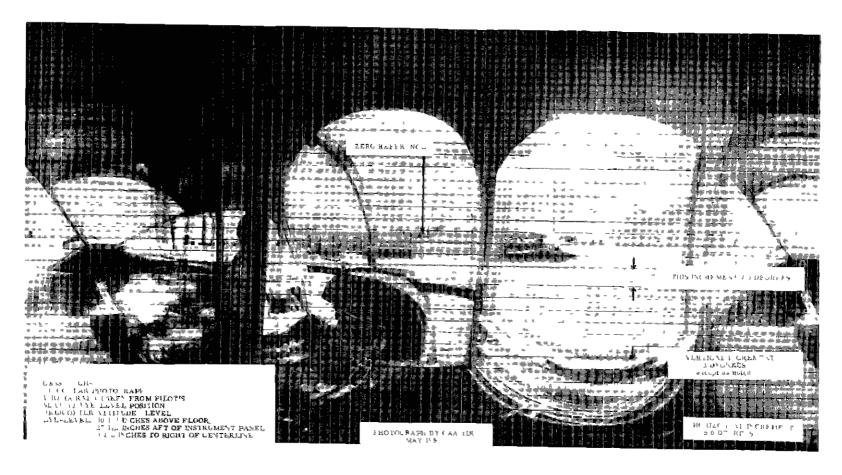


FIG 52 CESSNA CH-1 BINOCULAR COCKPIT VISIBILITY PHOTOGRAPH



FIG 53 CESSNA CH-1 SIDE VIEW



FIG 54 CESSNA CH-1 FRONT VIEW

APPENDIX II

HELICOPTER FLIGHT CHARACTERISTICS

Data herein relate to the flight characteristics of the HOK-1, HRS-3, H-13, H-21, and H-34 helicopters. All data presented have been reduced from photographic records obtained with the Fairchild flight analyzer, as discussed in this report. Typical flight-path pictures for each type of maneuver recorded are shown in Figs. 55 to 59 of this appendix. The manner in which pertinent information was derived is indicated on each figure.

The length of flight path being considered is determined by the formula

 $L = \frac{nd}{40}$

where

L = length of flight path (feet)

n = number of frames on the picture to the touchdown point

d = perpendicular distance from the camera to the flight path (feet).

Wherever flight attitudes are referred to in this appendix, positive values indicate nose up and negative values indicate nose down. Positive flight-path angles indicate that the helicopter is ascending and negative values, descending.

In a normal takeoff maneuver, a nose-down attitude is experienced during the first portion of the recorded flight path, while a nose-up attitude occurs during the latter portion. For this reason, the "average attitude" for takeoff maneuvers has been broken down into "average nose-up attitude" and "average nose-down attitude."

The sight angle is the visual angle required to see a touchdown target during approach. This angle is defined by the intersection of a line parallel to the helicopter centerline extended from and forward of the piloteye position with a line through the piloteye position and the touchdown target. This angle is determined by adding the target-sight angle indicated in Figs. 55, 57, and 58, and the helicopter attitude. The target-sight angle, in turn, is the angle formed by the intersection of a line through the pilots' eye position and the touchdown target with the artificial horizon (horizontal markings placed on the photograph by the camera).

HOK-1 NORMAL TAKEOFF

1 FLIGHT NO	2 HEADING (deg)	3 WI VELOCIT		5 LENGTH OF) FLIGHT PATH	6 MAXIMUM ATTITUDE	7 distance col 6 after takeoff	8 minimum aititude	9 DISTANCE COL 8 AFTER TAKBOFF	10 AVERAGE NOSE-UP ATTITUDE	11 AVERAGE NOSE-DOWN ATTITUDE	12 AVERAGE FLIGHT PATH
1	200			945	3 000'	790'	- 6º30'	115'	1°30'	-2 ⁰ 301	3°301
2	200			12601	50301	1060'	-16°00'	70'	3 ⁰ 00 '	-6°00•	6 ⁰ 301
3	200			1285	10301	1035	-13 ⁰ 30'	451	3 ⁰ 001	-6°30•	5 ⁰ 00'
4	200			13051	4°00'	1105'	-10 ⁰ 00'	315'	2º30'	-3 ⁰ 001	4°001
5	200			1305	4°301	9701	- 8°00'	70"	2º30"	-3°30'	4°30'
6	200			1240'	5º30'	700	- 9°30'	115'	3 ⁰ 00 '	-2°30'	7°∞′
7	200			13051	3°30'	1240'	- 6º30'	45'	1 ⁰ 30'	-3 ⁰ 001	60001
8	200			1240'	40301	8801	- 3 ⁰ 00'	7 0'	2 ⁰ 301	-1°00'	י ⁰ 00י
9	200			1215'	3°30'	10801	- 5 ⁰ 30 -	405'	10301	-3°00'	20301
10	200			1170	30001	655	- 9°∞'	901	2 ⁰ 001	ن ⁶⁰ 00ء	2 ⁰ 301
11	200			1240'	3°30"	225'	-12 ⁰ 301	25	20001	-2 ⁰ 50 ا	ن30°5
12	200			1285	60001	745'	0030	205'	3°30'	None	8°00'
13	200			1215'	9 º00'	6551	-10 ⁰ 00'	70'	4°00'	-5°30'	9 ⁰ 001
14	200			1285'	7 ⁰ 30 '	12401	-15 ⁰ 30'	7 0'	2°30'	-3°00'	5°30'
15	200			1195 i	40301	610'	- 7°30'	70'	2 ⁰ 00'	-2 ⁰ 001	5 ⁰ 00'
16	200			1215'	4°30'	630'	-11 ⁰ 30'	135'	2 ⁰ 30'	-4°301	4°00'
17	200	9 mph	180	13051	8º301	1285 ن	- 8° 6 01	45"	3 ⁰ 30'	-3°30'	3°30'
18	200	9 mph	180	13051	7 ⁰ 30′	י 2011	- 2 ⁰ 00'	700'	4°00'	-1°00'	ر 000 8
19	200	9 mph	180	12601	7 ⁰ 30 °	1150'	- 5°00°	45'	4°30'	-3°00'	8°∞′
20	200	9 mph	180	1240"	40301	810'	- 4 ⁰ 00'	3601	3 ⁰ 001	-2°∞¹	4°00'
21	200	9 ապրի	180	1305'	3°30"	1285'	- 4 ⁰ 301	2951	s ₀ 001	-1 [°] 30'	3 ⁰ 30'
AVERAGE					5 ⁰ 00'	965'	- 8°∞'	180'	3 ⁰ 001	-3 ⁰ ∞¹	5 ⁰ 30'

-

HOK-1 AUTOROTATION TO A PANEL WITH POWER RECOVERY

1 FLIGHT NO	2 HEADING (deg)	AETOCITA MIN		5 LENGTH OF FLIGHT PATH	6 DISTANCE TOUCHED DOWN BEFORE TOUCHDOWN PANEL	7 MAXIMUM ATTITUDE	8 DISTANCE COL 7 BEFORE ACTUAL TOUCHDOWN	9 MINIMUM ATTITUDE	10 DISTANCE COL 9 BEFORE ACTUAL TOUCHDOWN	11 AVERAGE ATTITUDE	12 AVERAGE FLIGHT PATH	13 MAXIMUM SIGHT ANGLE	14 DISTANCE COL 13 BEFORE ACTUAL TOUCHDOWN	15 MINIMUM SIGHT ANGLE	16/ DISTANCE COL 15 BEFORE ACTUAL TOUCHDOWN	17 AVERAGE SIGHT ANGLE
1	020			1240 '	251	23 ⁰ 001	225'	- 3°00'	945'	20001	-16 ⁰ 00'	300001	2251	80301	630'	13°00'
2	020			12401	701	200001	270'	-10 ⁰ 00'	11951	10001	-16 ⁰ 30'	240301	2701	30301	9701	100001
3	020		w	1215'	٥٠	200301	295'	- 00301	-B35'	5°30 '	-15 ⁰ 00'	36 ⁰ 00 '	295'	14°30'	835'	20 ° 30'
4	020			1170'	451	190301	225'	o ⁰ 00 '	1105'	1,0301	-13 ⁰ 00'	30 ⁰ 001	225'	12 °3 0'	790'	16°00'
5	020			10351	٥١	23 ⁰ 00'	315'	- 5 ⁰ 00'	720 I	4 ⁰ 00 '	-20 ⁰ 30'	47 ⁰ 00 °	315'	17 ⁰ 00'	7201	ر 00°72
6	020			1215	25'	28 ⁰ 30 '	1801	- 6 ⁰ 30'	1080'	7°00'	-10 ⁰ 30'	36 ⁰ 00 '	180'	3 ⁰ 301	10801	14°00′
7	020			1240'	251	17 ⁰ 00'	2051	- 3 ⁰ 30'	1105'	2 ⁰ 00'	-19 ⁰ 08	27 ⁰ 00 ′	205'	12°30'	4951	17 ⁰ 00'
8	020			745	45'	23 ⁰ 301	2501	- 2º00	6551	7 °0 0 '	-14°00°	30°30'	250'	16º00'	655'	20 ⁰ 301
9	020			1125'	0,	ر 16 ⁰ 00	2501	2 ⁰ 00'	7001	7 ⁰ 00 '	-22 ⁰ 001	34°00'	790'	17 ⁰ 30'	4751	25 ⁰ 301
10	020			1215'	0'	24 ⁰ 00 '	270'	- 3°00'	9251	5 ⁰ 301	19 ⁰ 30،	37 ⁰ 301	270'	14°00'	925'	21000
11	020	9 mph	180	12601	0'	15 ⁰ 00'	360'	- 2 ⁰ 30'	1105'	⁴⁰ 30 ،	-14 ⁰ 30'	42 ⁰ 00 '	2951	15 ⁰ 30'	1240'	27 ⁰ 301
12	020	9 առջև	180	1215'	25'	180001	135'	- 5 ₀ 001	4301	3 ⁰ ∞'	-19 ⁰ 00'	38 ⁰ 00'	135'	19 ⁰ 00'	4301	5110001
13	020	9 առթե	180	1215'	25'	18 ⁰ 30'	160'	- 0 ⁰ 30'	8551	4°30'	-18 ⁰ 001	36°00'	160'	16 ⁰ 301	8551	21 ⁰ 00'
14	020	9 ապրհ	180	1330	0'	27 ⁰ 00'	250'	- 2 ⁰ 30'	12851	5 ⁰ 00 1	-15 ⁰ 30'	38 ⁰ 00′	2 50'	12 ⁰ 00 ا	1125'	18°00'
Average					20'	51 ₀ 00 i	235'	- 3 ⁰ 00'	9351	4°30'	-16 ⁰ 30'	35 ⁰ 00'	2751	13 ⁰ 00'	8501	19 ⁰ 30'

HOK-1	STREE	APPROACH	щО	Δ	PANET.

		HOK			OACH TO A P	ANEL										
I FLIGHT NO	2 HEADING (deg)	WIN VELOCITY	FROM	5 LENGTH OF FLIGHT PATH	6 DISTANCE TOUCHED DOWN BEFORE TOUCHDOWN PAREL	7 MAXIMUM ATTITUDE	8 DISTANCE COL 7 BEFORE ACTUAL TOUCHDOWN	9 MINIMUM ATTITUDE	10 DISTANCE COL 9 BEFORE ACTUAL TOUCHDOWN	11 AVERAGE ATTITUDE	12 AVERAGE FLIGHT PATH	13 MAXIMUM SIGHT ANGLE	14 DISTANCE COL 13 HEFORE ACTUAL TOUCHDOWN	15 MINIMUM SIGHT ANGLE	16 DISTANCE COL 15 BEFORE ACTUAL TOUCHDOWN	17 AVERAGE SIGHT ANGLE
ı	020			1215'	251	ا 16 ⁰ 00	1251	o°∞•	900'	400	-16 ⁰ 00'	32°00'	115'	16 ⁰ 301	9001	20°30'
2	020			1170'	45'	23000	701	رە00°1-	1080'	6°00'	-20°00'	100°00'	701	18°30'	1080'	24 ⁰ 00 '
3	020			1240'	45'	17 ⁰ 00'	2701	00001	9901	8°00'	- 7°∞'	33 ⁰ 301	251	8°001	9901	18°30'
4	020		-	1105'	25'	210001	901	3 ⁰ 30 '	8801	8°00'	-17 ⁰ 00'	39°30'	901	20°30'	880	25 ⁰ 301
5	020			1240'	45'	20°30'	70'	3 ⁰ 00'	1080'	9°001	-14°30'	38°30'	70'	19°00'	1080	28°00'
6	020			1215'	701	19001	25'	5°00'	9451	9 ⁰ 331	- 9°30'	31°00'	70'	16 ⁰ ∞'	9451	22 ⁰ 301
7	020			11251	25'	21 ⁰ 00'	901	50∞•	945	8°30'	-12 ⁰ 30'	32 ⁰ 00'	901	18 ⁰ 00'	9451	210001
8	020		-	1105'	45'	18°30'	901	o ^o oo•	1060'	7°30'	-15 ⁰ 001	29 ⁰ 00 '	901	13°30'	10601	21000'
9	020			11251	45 *	140301	251	5°00'	3601	7°30′	-13 ⁰ 00'	270301	70'	19 ⁰ 001	1035'	22 ⁰ 301
10	050			1125'	451	18 ⁰ 00'	451	3 ⁰ 301	9001	7°001	-13 ⁰ 25'	30°00'	451	16 ⁰ 30'	1035'	210001
11	050			10801	45'	140001	701	<i>6</i> °∞∙	3851	8°30°	-17 ⁰ 001	300301	70'	18 ⁰ 301	251	28º001
12	020			1060'	25'	13 ⁰ 30'	270	6°30′	765'	9 °00 ′	-19 ⁰ 301	37 ⁰ ∞¹	270'	28 ⁰ 30'	810'	31°30'
13	020			10801	70"	13°30'	5401	5 ⁰ ∞¹	7651	9 ⁰ 301	-19 ⁰ 00'	3 5°3 0'	340'	22 ⁰ 301	251	30°00'
14	020			1150'	45	14 ⁰ 00 '	251	10301	745'	8º001	-14 ⁰ 30'	33 ⁰ 301	2701	ا∞17	7651	25 ⁰ 001
15	020	9 шры	180	1240	0'	ر ⁰ 00 د	2701	4°30'	855	7⁰∞¹	-19 ⁰ 00'	۴۲۰ ₀ 00 ،	701	25 ⁰ 00'	1215'	33°30'
16	020	9 mph	180	1170'	70'	00000	451	00001	970'	70001	-15 ⁰ 30'	34°30'	2051	16°∞'	9901	24°301
17	020	9 жұры	180	1240'	0'	17 ⁰ 00'	251	2 ⁰ 30'	855'	8 ⁰ 30¹	-15 ⁰ 30'	36 ⁰ 001	251	18°00'	1215'	26 ⁰ 00 '
18	020	9 mp h	180	1240'	25'	140301	451	4030'	790'	90001	-16°30'	300001	270'	19 03 01	251	25 ⁰ 00'
19	020	9 mp h	180	12401	70'	18°30'	1601	3°00'	1195'	8°30'	- 7 ⁰ 301	28 °3 01	1601	n₀∞ı	י1195י	18°30'
20	020	9 жүр	180	1285'	25'	16000	6751	4°30"	11951	9°30'	- 9°30'	28 °3 0'	5851	14 ⁰ 30'	1195'	22°30'
AVERAGE	S				40'	16°30'	145'	30∞	8901	8°00'	-14 ⁰ 30'	33 ⁰ 30'	2651	180001	9501	240301

HOK - 1 NORMAL LANDING TO A PANEL

		2	1.	-	6	7	8	9	10	11	12	13	14	15	16	17
1 FLIGHT NO	2 HEADING (deg)	WINE VELOCITY		5 LENGTH OF FLIGHT PATH	6 DISTANCE TOUCHED DOWN BEFORE TOUCHDOWN	MAXIMUM ATTITUDE	DISTANCE COL 7 BEFORE ACTUAL TOUCHDOWN	MINIMUM ATTITUDE	DISTANCE COL 9 BEFORE ACTUAL TOUCHDOWN	AVERAGE ATTITUDE	AVERAGE FLIGHT PATH	MAXIMUM SIGHT ANGLE	DISTANCE COL 13 BEFORE ACTUAL TOUCEDOWN	MINIMUM SIGHT ANGLE	DISTANCE COL 15 BEFORE ACTUAL TOUCHDOWN	AVERAGE SIGHT ANGLE
					PANEL											
ı	020			1170'	451	30°001	70 '	00001	970	9 ⁰ 001	- 7 ⁰ 00'	43 ⁰ 301	70'	6°00'	970'	16 ⁰ 30'
2	020			1170'	45•	22 ⁰ 301	901	00001	9001	8°30'	- 8°30'	35 ⁰ 301	90'	7°30'	1125'	169301
3	020			10601	451	1 1 00'	2701	o°30'	1035'	7°00'	- 7°30'	25 ⁰ 30 '	270'	6º30'	1035'	74000°
4	υ20			1240	451	51 ₀ 00,	701	0°30'	1215'	9001	- 7°30'	33º∞'	701	7°30'	1215'	15°30'
5	020			11251	251	230301	115'	1°00'	1060'	10°30'	-11°00'	39°00'	115'	12 º 00'	1060'	550001
6	020			1150'	701	21°30'	135'	1 30'	1060'	90301	-13 ⁰ 00'	33 ⁰ 30'	135'	14 ⁰ 00'	1060'	20°30'
7	020			12401	0'	23 ⁰ 00'	160'	0 ⁰ 00'	970'	8 ° 30'	-15 ⁰ 30'	fr}t-500 i	135'	14°30'	720 '	230301
8	020			1215'	251	26 ° 30'	180'	-2°00'	1060'	ن‰بت	- 4°30'	31°30'	180'	20301	1060 י	15 º 00'
9	020			9901	45'	19 ⁰ 001	901	1°30'	9001	10 ⁰ 00 '	- 7 ⁰ 30'	31°00'	45'	8°30'	9001	180001
10	020			11951	70'	190001	90'	3°00'	10151	90301	- 40301	25°30'	90'	7°30'	1015'	140301
11	020			1240	25'	12°30'	115'	3°00'	990'	80301	-11º00'	250301	135'	140001	990'	200001
12	020			1170'	701	23 ⁰ 001	25"	0°30'	1035'	10 ⁰ 301	- 4º00'	27 0 001	25'	3°30'	1035'	140001
13	020			10351	901	14°30'	יפבנ	3 ⁰ 00'	9901	90001	- 9°30'	24°30'	135'	11030'	9901	170001
14	020	9 m ph	180	י1195	451	23 ⁰ 001	901	3°00'	9901	90001	- 8º 30'	30°00′	135'	100301	990'	150301
15	020	9 mph	180	12851	45'	22°30'	45'	00001	945'	80001	-12°30'	33 ⁰ 001	451	110001	9251	170001
16	020	9 ապրա	180	12851	451	200001	45'	4°30'	1240'	10°30'	- 5°30'	• 270∞	451	10 ⁰ 301	1240'	17000 ا
17	020	9 жерь	180	12401	251	18 ⁰ 30'	340'	o°301	1080'	9000	-11 ₀ 00,	30 ⁰ 001	70'	10 ⁰ 30 '	1035'	180301
18	020	9 mph	180	1195'	70'	15°30'	2701	3º00'	1170'	¹ഗ ് ശ∙	- 6°30'	20 ⁰ 00 '	250'	9°00'	1170'	140301
19	020	9 mph	180	1305'	251	170001	160'	3 ⁰ 00 °	11501	100001	- 7°30'	25 ⁰ 00 '	160'	10 ⁰ 00'	1150'	17 ⁰ 30'
AVERAG	E				45.	20°30'	155	1°30′	10901	9°30'	- 8 ⁰ 30'	30°30'	125'	9 ⁰ 301	1065	17 ⁰ 301

HRS-3 AUTOROTATION TOWARD A PANEL WITH POWER RECOVERY

1 FLICHT WO.	EADING (deg.)	VELOCITY]	H HROM Geg.)	5 LENOTH OF FLIGHT PATH	DISTANCE TOUCHED DOWN BEFORE TOUCHDOWN PANEL	7 MAXIMUM ATTITUDE	8 DISTANCE COL. 7 BEFORE ACTUAL TOUCHDOWN	MINIMUM ATTITUDE	DISTANCE COL. 9 BEFORE ACTUAL TOUCHDOWN	11 AVERAGE ATTITUDE	12 AVERAGE FLIGHT PATH	13 MAXIMUM SIGRET ANGLE	14 DISTANCE COL.13 BEFORE ACTUAL TOUCHDOWN	15 MINIMUM SIGHT ANGLE	16 DISTANCE COL.15 BEFORE ACTUAL TOUCHDOWN	17 AVERAGE SIGHT ANGLE
1	020	8-10 kmots	310	1125	251	140001	1601	-8°00'	4501	۰ <mark>۰</mark> 30 ن	-21 ⁰ 00'	31°00'	1601	12°30'	3851	21°00'
s	020	8-10 knots	310	1195'	45'	15 ⁰ 30'	2701	-2 ⁰ 30'	1801	10301	-14 ⁰ 00 '	54°00'	270'	7°001	180'	12°30'
3	020	8-10 kmots	310	1240'	01	57 ₀ 00 •	115'	ن ⁰ 00-	5651	2°∞'	-14°30°	55 [°] ∞'	251	9 ⁰ 001	· 9401	18 ⁰ 00'
4	020	8-10 kmots	310	1150'	0'	14 ⁰ 30'	2251	-6°301	1060'	2 ⁰ 30 '	-15 ⁰ 001	30°30'	2251	8°30'	1060	16°30'
5	020	8-10 kmots	310	12151	45.	12°30'	¥301	-7 ⁰ 30۱	10351	1°30 ن	-14 ⁰ 001	17 ⁰ 30'	251	ن ⁰ 00°	10351	110001
AVERAGE					23'	ن‰16°	240'	-6°00'	7101	1030	-15 ⁰ 30'	31 °3 0'	140'	8 ⁰ 001	595 •	16 ⁰ ∞'

HRS-3 NORMAL TAKEOFF

1 FLIO ET	2 HEADI (deg.		3 WII ELOCIT		5 LENGTH OF FLIGHT PATH	6 NAXIMIN ATFITUDE	7 DISTANCE COL. 6 AFTER TOUCHDOWN	8 MINIMUM ATTITUUE	9 DISTANCE COL. 8 AFTER TAKEOFF	10 AVERAGE NOSE-UP ATTITUDE	11 AVERAGE NOSE-DOWN ATTITUDE	12 AVERAGE FLICET PATH
1	200	8-10	knots	310	12601	19001	9001	-ന₀∞∙	901	10001	⁻ 4°∞'	6 ⁰ ∞'
2	200	8-10	knots	310	9701	-3 ⁰ 00 1	4951	-13 ⁰ 00'	135'	None	-6°30'	<i>6</i> ⁰∞י
3	200	8-10	mots	310	12401	-2 ⁰ 001	יפפננ	-12 ⁰ 30 '	251	None	-8 ⁰ 301	3 ⁰ 301
Į.	200	8-10	knots	310	13051	°000	12851	-14 ⁰ 301	115'	None	-6°30'	6 ⁰ 00'
5	200	8-10	lmota	310	1260'	o°∞′	9001	- 5 ⁰ 301	2051	None	-3 ⁰ 00'	7 ⁰ 30'
6	200	8-10	knots	310	1305'	-3°001	2951	- 7°301	9251	None	-5 ⁰ 30'	3 ⁰ 301
7	200	8-10	knots	310	1285'	-3°30'	10801	-11°30'	701	None	-5°301	20001
8	200	8-10	knots	310	13051	_ro30 i	י 1260	-11°∞'	25'	Hone	-7°001	1 ⁰ 00'
AVERAGE	;					-2 ⁰ 001	925'	-11 ₀ 00,	3651	None	-6º∞'	1 ₀ 301

HRS-3 NORMAL LANDING TO A PANEL

l Flight No.	2 HRADIN (deg.)	VELOCITY	FROM	5 LENGTH OF) FLIGHT PATH	6 DISTANCE TOUCHED DOWN BEFORE TOUCHDOWN PANEL	7 MUNITAN ATTITUDE	8 DISTANCE COL. 7 BEFORE ACTUAL TOUCHDOWN	9 MINIMUM ATTITUDE	10 DISTANCE COL. 9 BEFORE ACTUAL TOUCHDOWN	11 AVERAGE ATTITUDE	12 AVERAGE FLIGHT PAIH	13 MAXIMUM SIGHT ANGLE	14 DISTANCE COL.13 BEFORE ACTUAL TOUCHDOWN	15 Minimum Signi Angle	16 DISTANCE COL.15 HEFORE ACTUAL TOUCHDOWN	17 AVERAGE SIGHT ANGLE
1	020	8-10 knots	310	11701	701	7°30'	610'	-2 ⁰ 301	10801	3 ⁰ 30 '	- 9 ⁰ 00 -	17 ⁰ 001	6101	6 ⁰ 301	10801	12 ⁰ 30' .
2	020	8-10 knots	310	11951	451	7 ⁰ 30'	115'	0 ⁰ 00'	יסדננ	2 ⁰ 30 '	-10 ⁰ 00'	19 ⁰ 00'	115'	ن30°م11	5851	14°001
3	020	8-10 knots	310	1125'	451	150301	2051	-2 ⁰ 30'	10151	5 ⁰ 00 °	- 8 ⁰ 00'	90°∞	01	ە ⁰ 30 ر	1015'	15 ⁰ 00 '
14.	020	8-10 kmots	310	1195'	451	8º301	225'	-1°30'	9901	3 ⁰ ∞¹	- 7 ⁰ 30'	17°∞'	451	6 ⁰ 301	1035'	11°00'
5	020	8-10 kmots	310	1170'	45'	80001	3601	-2°30'	1035	3 ⁰ 001	- 8 ⁰ 30 -	16°30'	115'	6000°	10351	110301
6	620	8-10 kmots	310	11501	701	6 ⁰ 301	3151	00001	1035*	3°00'	- 4°00°	16 ⁰ 001	251	ا ⁰ 30ء	1035	9 ⁰ 001
7	020	8-10 knots	310	1195'	01	10 ⁰ 00'	451	-0°30'	1125'	5 ⁰ 001	- 1°30'	27 ⁰ 00'	451	1°30'	1125'	9 ⁰ 00 r
AVERAGE					451	9 ⁰ ∞¹	2751	-1°30'	1085'	3 ⁰ 301	- 7º001	29 ⁰ 001	125'	ر 6 ⁰ 00	1040	12 ⁰ 00'
		HRS	-3 S7	EEP APPE	ROACH TO A F	PANEL										
1	020	8-10 knots	310	1215'	251	5 ⁰ 001	701	-5 ₀ 00 ,	5201	o ^o 301	-17°001	27 ⁰ 301	251	16 ⁰ 30 ا	1170'	210001
2	020	8-10 knots	310	12601	0'	4°301	251	-5°30'	1240'	°000	-15 ⁰ 301	29°00'	251	ا 30°00	12401	18°30'
3	020	8-10 kmots	. 310	1240'	451	9 ⁰ 001	901	-o°30′	585'	3 ⁰ ∞′	-13 ⁰ 00'	5/t ₀ 30 t	251	13 ⁰ 30'	5851	170001
4	020	8-10 knots	330	1105'	160'	3 ⁰ 00 '	970'	-10301	8101	1°00'	-17 ⁰ 00'	19 ⁰ 30†	970'	14°30'	2951	17 ⁰ 00'
5	020	8-10 knots	310	1195'	45'	5°30'	25"	-3 ⁰ 301	1060'	0°301	-16 ⁰ 301	57 ₀ 30 i	251	13 ⁰ 301	10601	18 ⁰ 00'
6	020	8-10 knots	310	1215'	יס?	ر‱6	901	-4°00'	970'	0°30'	-16 ⁰ 00'	21 0 00'	6101	ن9 ⁰ 00 و	25'	16 ⁰ 00'
7	020	8-10 knots	310	12401	451	გ _ი იი ,	160'	- ¹⁴⁰ 30 ¹	540°	ا⊙30	-11°30'	26 ⁰ 301	115'	12 ⁰ 301	5401	ن 16 ⁰ 00
ខ	020	8-10 kmots	310	יסקבנ	70*	ή ₀ 00 ι	701	-2 ⁰ 001	6101	o °3 0'	-13 ⁰ 30'	18 ⁰ 001	701	12 ⁰ 301	1125'	16 ₀ 001
9	020	8-10 knots	310	1240'	0'	3°30'	70'	-3 ⁰ 001	7001	-1°001	-14 ⁰ 301	38 ⁰ 301	25'	13 ⁰ 001	1215'	18 ⁰ 00'
AVERAGE	3				55'	50301	175'	-3 ⁰ 001	7751	ە°30،	-15 ⁰ 00'	25 ⁰ 001	2101	13 ⁰ 00'	885'	17 ⁰ 301

H-13 CRUISE

,	0			SED - 40		_		_
l Flight No.	2 HEADING (deg.)	AETOCITA AII	FROM	5 LENGTH OF)FLICHT PATH	6 MAXIMUM ATTITUDE	7 MINIMUM ATTITUDE	8 Average Attitude	9 AVERAGE FLIGHT PATH
1	180	10-12	247.5	12851	-0°301	- გ ^ი იი 1	- 5°00°	-0°30'
2	180	0-5	1.80	1285	0 ⁰ 001	- 5 ⁰ 30 ا	- 3°00'	-2 ⁰ 00'
3	180	0-5	225	1305'	-3 ⁰ 001	- 7 ⁰ 30'	- 5 ⁰ 30'	-1 ⁰ 00'
4	180	0-5	225	12401	-0°301	- 6 ⁰ 30'	- 3°00'	-2°30'
5	180	0-5	225	12601	-3 ⁰ 301	- 9 ⁰ 301	- 6 ⁰ 001	-1°00'
б	180	0-5	225	1260'	2 0 001	-11°00'	- 6 ⁰ 301	-0°301
AVERAGE					-1 ⁰ 001	- 8°00°	- 5 ⁰ 001	1°30'
		CRU	JISE SPE	SED - 60	KNOTS			
1	180	10-12	247.5	1260'	20001	- 7 ⁰ 301	- 3°30'	-3 ⁰ 301
2	180	0-5	180	11951	-7t0301	-11°00'	- 8°001	-10001
3	180	0-5	225	12851	-2 ⁰ 001	-12 ⁰ 00′	- 8 ⁰ 301	o°30*
4	180	0-5	225	10601	o ⁰ 301	- 9 ⁰ 001	- 4 ⁰ 301	-2°00'
5	180	0-5	225	1125'	-4°301	-12 ⁰ 00'	- 8 ⁰ 30'	-0°30'
6	180	0-5	225	1260'	-7 ⁰ ∞¹	-130001	-10 ⁰ 00 '	-1°30'
AVERAGE					-20301	10 ⁰ 30'	- 7 ⁰ 001	-1°30'
		CRU	ISE SPE	ED - 80	KNOTS			
1	180	10-12	247.5	1215'	-6 ⁰ 001	-12 ⁰ 30′	- 9 ⁰ 30'	0°30'
2	180	0-5	180	12601	-3 ⁰ 301	-12 ⁰ 30†	- 8°00'	1°30'
3	180	0-5	225	1305'	-8°30;	-13°00'	-11°00'	-0°301
4	180	0-5	225	12401	-3°00°	- 9 ⁰ 00'	- 6°00°	-1°301
5	180	0-5	225	12601	-8 ⁰ 001	-13 ⁰ 00†	-10 ⁰ 001	-1°301
6	180	0-5	225	1240'	00001	-12 ⁰ 00 '	- 9 ⁰ 301	-1°00'
AVERAGE					-5°00'	-12 ⁰ 00'	- 9º00°	-0°301

R-21 CRUISE

CRUISE SPEED - 70 KNOTS

1 10-12	l FLICHT NO.	2 HEADING (deg.)	3 VELOCITY	FROM	5 LENOTH OF)FLIGHT PATH	6 MAXIMUM ATTITUDE	7 MINIMUM ATTITUDE	8 AVERAGE ATTITUDE	9 AVERAGE FLIGHT PATH
180	1		10-12	180	1305′	ر 5 ⁰ 00	-1°00'	3 ⁹ 301	00301
180	2		10-12	180	1305'	5 ⁰ 00 '	09001	4°00'	۱ 30 ⁰ 0
5 180 0-8 225 1305 6°00 2°00 3°30 2°00 6 180 0-8 225 1305 5°30 0°00 3°30 2°30 7 180 0-8 225 1305 5°30 -0°30 2°00 0°30 8 180 0-8 225 1305 9°30 2°30 5°00 0°00 9 180 0-8 225 1260 7°30 3°00 5°00 2°30 CRUSERS SPEEDS - 90 1°00 1°00 4°00 0°30 CRUSERS SPEEDS - 90 2°00 1°00 4°00 0°30 CRUSERS SPEEDS - 90 2°00 2°00 3°30 0°00 0°00 CRUSERS SPEEDS - 100 1°30 1°30 1°30 1°30 1°30 0°00 1°30 0°00 0°00 1°30 0°00 1°30 0°00 1°30 0°00 1°30 0°00 1°30 1°30 1°30	3	180	0-8	225	13051	7 ⁰ 001	0°301	14000 i	00001
6 180 0-8 225 1305' 5°30' 0°00' 3°30' 2°30' 0°30' 6°3	4	180	o-8	225	1305'	12°00′	1900'	6°00'	-1 ⁰ 00'
7 180 0-8 225 1300 4°30' -0°30' 2°00' 0°00' 8 180 0-8 225 1305' 9°30' 2°30' 5°00' 0°00' 9 180 0-8 225 1260' 7°30' 3°00' 5°00' 2°30' AVERAGE TOTAL 180 1305' 6°00' 1°00' 4°00' 0°30' CRUTETE SPIECT - 90' 1°00' 4°00' 0°30' 0°30' 0°30' 0°30' 0°00' 0°30' 0°00' 0°30' 0°00'	5	180	0-8	225	13051	6 ⁰ 001	2 ⁰ 00'	3°30'	2 ⁰ 00 '
8 180 0-8 225 1305' 9°30' 2°30' 5°00' 0°00' 9 180 0-8 225 1260' 7°30' 3°00' 5°00' 2°30' AVERACE ***********************************	6	180	0-8	225	13051	5°301	00001	3 ⁰ 301	20301
9 180 0-8 225 1260' 7030' 300' 500' 2030' 10 180 0-8 225 1305' 600' 100' 300' -0030' AVERACE TCRUISE SPEED - 90 NOTS 1 10-12 180 1305' 500' 200' 100' 100' 100' 2 10-12 180 1305' 500' 200' 100' 100' 100' 3 180 0-8 225 1305' 4030' -0030' 100' 100' 4 180 0-8 225 1305' 200' -0030' 100' 100' 100' 5 180 0-8 225 1305' 200' -0030' 100' 100' 100' 6 180 0-8 225 1305' 200' -0030' 100' 100' 100' 7 180 0-8 225 1305' 200' -0030' 100' 100' 9 180 0-8 225 1305' 200' -0030' 100' 100' 10 180 0-8 225 1260' 200' 000' 100' 100' AVERACE TCRUISE SPEED - 110 KNOTS 1 10-12 180 1305' 200' -0030' 000' 00' 100 100 100' 100' 100' 100 100 100 0-8 225 1305' 200' 000' 100' 100' 100 100 100 0-8 225 1305' 200' 000' 100' 100' 100 100 100 0-8 225 1305' 200' 000' 100' 100' 100 100 100' 100' 100	7	180	8-0	225	1330'	[†] °30•	-0°301	2 ⁰ 00 1	0 ⁰ 301
10 180 0-8 225 1305' 6°00' 1°00' 3°00' -0°30' 0°30' 1°40' 1°40' 0°30' 1°40' 1°40' 0°30' 1°40' 1	8	180	0-8	225	1305'	9°30'	2 ⁰ 301	5 ⁰ 00 '	o ^o ∞′
CRUISE SPEED - 90 KNOTS 1°00' 4°00' 0°30' 0°00' 0°30' 0°	9	180	0-8	225	1260'	7 ⁰ 301	3 ⁰ 00'	5 ⁰ ∞¹	2 0 301
1	10	180	0-8	225	1305'	6 ⁰ ∞¹	10001	30001	-0°30'
1 10-12 180 1305' 5900' 200' 3030' 0000' 2 10-12 180 1305' 2030' -100' 1030' -0030' 3 180 0-8 225 1305' 4030' -0030' 1030' 1000' 4 180 0-8 225 1305' 2030' -0030' 100' -1030' 5 180 0-8 225 1305' 200' -0030' 100' -0030' 6 180 0-8 225 1305' 200' -0030' 100' 100' 7 180 0-8 225 1305' 200' -0030' 100' 100' 8 180 0-8 225 1305' 1030' -0030' 000' 100' 9 180 0-8 225 1305' 4030' 200' 100' 100' 10 180 0-8 225 1305' 200' 000' 100' 100' AVERAGE CRUISE SPEED - 110 KNOTS 1 10-12 180 1305' 000' -0030' 000' 000' 2 10-12 180 1305' 000' -0030' 000' 000' 3 180 0-8 225 1305' 200' -0030' 000' 000' 1 100' 000' 1 100' 000' 1 100' 000' 1 100' 000' 0	AVERAGE					70001	10001	4°00'	0°30"
2			CRU	ise sp	1000 - 90	KNOTS			
3 180 0-8 225 1305; 4-90; 0-90; 1-90; 1-90; 1-90; 4 180; 4 180 0-8 225 1305; 2-90; -0-90; 1-90; -1-90; 5 180 0-8 225 1305; 2-90; 0-930; 0-90; 1-90; 1-90; 6 180 0-8 225 1305; 2-90; 0-930; 0-90; 1-90;	1		10-12	180	13051	ا 5 0 00	2900	3 ⁰ 301	00001
4 180 0-8 225 1305' 2°30' -0°30' 1°00' -1°30' 5 180 0-8 225 1305' 1°30' -0°30' 0°30' -0°30' 6 180 0-8 225 2305' 2°00' 0°30' 1°30' 1°00' γ 180 0-8 225 1305' 1°30' -0°30' 0°00' 1°00' 8 180 0-8 225 1305' 1°30' 2°00' 3°00' 0°00' 1°00' 9 180 0-8 225 1260' 2°00' 0°00' 1°30' -1°00' 10 180 0-8 225 1305' 2°30' 0°30' 1°30' 0°00' AVERAGE T 10-12 180 1305' 0°00' -0°30' 0°00' 0°00' 2 10-12 180 1305' 0°00' -0°30' 0°00' 0°00' 0°00' 1°00' 0°30' 0°00' <td>5</td> <td></td> <td>10-12</td> <td>180</td> <td>13051</td> <td>2⁰30'</td> <td>-1901</td> <td>1°30'</td> <td>-0⁰301</td>	5		10-12	180	13051	2 ⁰ 30'	-1901	1°30'	-0 ⁰ 301
5 180 0-8 225 1305' 1 30' -0 30' 0 30' -0 30' 6 180 0-8 225 2305' 290' 0 30' 1 30' 1 00' 7 180 0-8 225 1305' 1 30' -0 30' 0 00' 1 00' 8 180 0-8 225 1305' 4 30' 2 90' 3 00' 0 00' -1 00' 9 180 0-8 225 1260' 2 00' 0 30' 1 30' 0 00' 10 180 0-8 225 1305' 2 90' 0 30' 1 30' 0 00' AVERAGE 10-12 180 1305' 0 00' -0 30' 0 00' 0 00' 1 10-12 180 1305' 0 00' -0 30' 0 00' 0 00' 2 10-12 180 1305' 2 00' -0 30' 0 00' 0 00' 3 180 0-8	3	180	0-8	225	1305'	1t 030,	0°30'	1°30'	10001
6 180 0-8 225 1305' 290' 0'30' 1'30' 1'00' 7 180 0-8 225 1305' 1'30' -0'30' 0'00' 1'00' 8 180 0-8 225 1305' 1'30' 2'00' 3'00' 0'00' 9 180 0-8 225 1260' 2'00' 0'00' 1'30' 0'00' AVERAGE CRUISE SPEED - 110 KNVTS CRUISE SPEED - 110 KNVTS 1 10-12 180 1305' 0'00' -0'30' 0'00' 0'00' 2 10-12 180 1305' 0'00' -0'30' 0'00' 0'00' 3 180 0-8 225 1305' 2'00' -0'30' 0'30' 0'00' 4 180 0-8 225 1305' 2'00' -0'30' 0'30' 0'00' 5 180 0-8 225 1305' -1'00' -3'30' 0'30' 1'00' 6 180 0-8 225 1305' -1'00' -5'30' -2'00' 10'00' 7 180 0-8 225 1305' -1'00' -5'30' -2'00' -0'30' 0'00' 7 180 0-8 225 1305' -1'00' -5'30' -2'00' -1'00' 8 180 0-8 225 1305' -1'00' -5'30' -2'00' -1'00'	4	180	0-8	225	1305'	2 °30'	-0°301	19001	-1°30'
180	5	180	9-8	225	13051	1°30'	۰۵ ₀ 30،	ن 30 ⁰ 0	-0°30'
8 180 0-8 225 1305' 4 30' 2 90' 3 90' 0 90' 1 90' 1 90' 10' 10 180 0-8 225 1260' 2 90' 0 90' 1 9	6	180	0-8	225	13051	29001	o°3o'	1°30'	1,00
9 180 0-8 225 1260' 2°00' 0°00' 1°30' 0°00' 10 180 0-8 225 1305' 2°30' 0°30' 1°30' 0°00' AVERAGE CRUISE SPEED - 110 KNOTS 1 10-12 180 1305' 0°00' -0°30' 0°00' 0°00' 2 10-12 180 1305' 0°00' -2°00' 1°00' 0°00' 3 180 0-8 225 1305' 2°00' -0°30' 0°30' 0°00' 0°00' 4 180 0-8 225 1305' -1°30' -3°30' -2°30' 1°00' 5 180 0-8 225 1305' -1°00' -5°30' -3°00' 0°00' 6 180 0-8 225 1305' -1°00' -5°30' -2°30' 0°00' 7 180 0-8 225 1305' -1°00' -3°00' -2°00' -1°00' 8 180 0-8 225 1305' -1°00' -3°00' -2°00' -1°00'	7	180	0-8	225	1305'	1 030 '	-0°30'	ooor	י 1900
10 180 0-8 225 1305' 2°30' 0°30' 1°30' 0°00' AVERACIE CRUISE SPEED - 110 KNOTS 1 10-12 180 1305' 0°00' -0°30' 0°00' 0°00' 0°00' 2 10-12 180 1305' 0°00' -2°00' 1°00' 0°30	8	180	0-8	225	1305'	4 °30 '	2901	3000	0901
3001 0001 1001	9	180	0-8	225	1260'	200'	100°0	1 ⁰ 30 '	-1001
CRUISE SPRED - 110 KNOTS 1 10-12 180 1305' 0°00' -0°30' 0°00' 0°00' 2 10-12 180 1305' 0°00' -2°00' -1°00' 0°30' 3 180 0-8 225 1305' 2°00' -0°30' 0°30' 0°00' 4 180 0-8 225 1305' -1°30' -3°30' -2°30' 1°00' 5 180 0-8 225 1305' -1°00' -5°30' -3°30' 0°00' 6 180 0-8 225 1285' 0°00' -4°30' -2°30' -0°30' 7 180 0-8 225 1305' -1°00' -3°00' -2°00' -1°00' 8 180 0-8 225 1305' 2°00' -2°00' 0°30' 1°00'	10	180	0-8	225	1305'	2 ° 30 '	o ⁰ 30 '	1 3 0'	0,000,
1 10-12 180 1305' 0°00' -0°30' 0°00' 0°00' 2 10-12 180 1305' 0°00' -2°00' -1°00' 0°30' 3 180 0-8 225 1305' 2°00' -0°30' 0°30' 0°00' 4 180 0-8 225 1305' -1°30' -3°30' -2°30' 1°00' 5 180 0-8 225 1305' -1°00' -5°30' -3°00' 0°00' 6 180 0-8 225 1285' 0°00' -4°30' -2°30' -0°30' 7 180 0-8 225 1305' -1°00' -3°00' -2°00' -1°00' 8 180 0-8 225 1305' 2°00' -2°00' 0°30' 1°00'	AVERAGE					3°00'	0 ⁰ 30 ¹	2 ⁰ 30'	o⁰oo¹
2 10-12 180 1305' 0°00' -2°00' -1°00' 0°30' 3 180 0-8 225 1305' 2°00' -0°30' 0°30' 0°00' 4 180 0-8 225 1305' -1°30' -3°30' -2°30' 1°00' 5 180 0-8 225 1305' -1°00' -5°30' -3°00' 0°00' 6 180 0-8 225 1285' 0°00' -4°30' -2°30' -0°30' 7 180 0-8 225 1305' -1°00' -3°00' -2°00' -1°00' 8 180 0-8 225 1305' 2°00' -2°00' 0°30' 1°00'			CRUISE SI	PRED -	- 110 KNO	TS			
2 10-12 180 1305' 0°00' -2°00' -1°00' 0°30' 3 180 0-8 225 1305' 2°00' -0°30' 0°30' 0°00' 4 180 0-8 225 1305' -1°30' -3°30' -2°30' 1°00' 5 180 0-8 225 1305' -1°00' -5°30' -3°00' 0°00' 6 180 0-8 225 1285' 0°00' -4°30' -2°30' -0°30' 7 180 0-8 225 1305' -1°00' -3°00' -2°00' -1°00' 8 180 0-8 225 1305' 2°00' -2°00' 0°30' 1°00'	J		10-12	180	1305'	09001	-0°30'	0 ⁰ 001	00001
4 180 0-8 225 1305' -1°30' -3°30' -2°30' 1°00' 5 180 0-8 225 1305' -1°00' -5°30' -3°00' 0°00' 6 180 0-8 225 1285' 0°00' -4°30' -2°30' -0°30' 7 180 0-8 225 1305' -1°00' -3°00' -2°00' -1°00' 8 180 0-8 225 1305' 2°00' -2°00' 0°30' 1°00'	2		10-12			09001		-1°∞¹	00301
4 180 0-8 225 1305' -1°30' -3°30' -2°30' 1°00' 5 180 0-8 225 1305' -1°00' -5°30' -3°00' 0°00' 6 180 0-8 225 1285' 0°00' -4°30' -2°30' -0°30' 7 180 0-8 225 1305' -1°00' -3°00' -2°00' -1°00' 8 180 0-8 225 1305' 2°00' -2°00' 0°30' 1°00'	3	180	0-8	225	13051	2 ⁰ 001	-a°30'	ە ⁶ 30 ئ	، 20 ⁰ 0
5 180 0-8 225 1305' -1°00' -5°30' -3°00' 0°00' 6 180 0-8 225 1285' 0°00' -4°30' -2°30' -0°30' 7 180 0-8 225 1305' -1°00' -3°00' -2°00' -1°00' 8 180 0-8 225 1305' 2°00' -2°00' 0°30' 1°00'	4	180	0-8	225	1305	-1°30'			1 ⁰ 001
6 180 0-8 225 1285' 0°00' -4°30' -2°30' -0°30' 7 180 0-8 225 1305' -1°00' -3°00' -2°00' -1°00' 8 180 0-8 225 1305' 2°00' -2°00' 0°30' 1°00'	5	180	0-8	225				-3°∞'	0°00'
7 180 0-8 225 1305' -1°00' -3°00' -2°00' -1°00' 8 180 0-8 225 1305' 2°00' -2°00' 0°30' 1°00'		180	0-B			0 ⁰ 001			-0°30'
8 180 0-8 225 1305' 2°00' -2°00' 0°30' 1°00'				225					
		,	0-8						10001
	9								
AVERAGE 0°00' -3°00' -1°30' 0°00'									

H-34 CRUISE CRUISE SPEED - 70 KNOTS

1 FLICHT NO.	2 HEADING (deg.)	3 WIN VELOCITY	FROM	5 LENGTH OF)FLIGHT PATH	6 maximum attitude	7 NINIMUM ATTITUIR	8 AVERACE ATTITUDE	9 AVERAGE FLIGHT PATH
1		10-12	180	13051	5 ₀ 00 t	0°301	1°30'	0 ⁰ 00 •
2		10-12	180	13051	3 ⁰ 301	0 ⁰ 00*	2 ⁰ 301	1°001
3		10-12	180	1215'	19301	-1 ⁰ 00'	0 ⁰ 301	10301
4		10-12	180	1305'	3 ⁰ 301	0 <mark>0</mark> 001	2 ⁰ 001	2 ⁰ 301
5		10-12	180	13051	ħο30+	ا2 ⁰ 30	3 ⁰ 301	1°30'
6	180	o – 8	225	1305'	19301	-1°001	0°30°	10001
7	180	o - 8	225	13051	ا ²⁰ 00	ا 30°0	1°30'	0°301
8	180	0-8	225	1305'	3 ⁰ 301	-0°301	2 ⁰ 001	2 ⁰ 001
9	180	0-8	225	10801	2 ⁰ 301	0 <mark>0</mark> 001	7 ₀ 301	o ⁰ 001
AVERACIE					2 ⁰ 301	00001	1 ₀ 30,	1 ⁰ 001
		CRU	ise sp	EED - 90				
1		10-12	180	1305'	1 30'	0 ⁰ 30'	1000,	2°00′
2		10-12	180	1305'	1 ⁰ 30'	-0°301	10001	3 ⁰ 001
3		10-12	180	13051	1 ⁰ 00'	0 ⁰ 001	10000	1 ⁰ 00'
ΙĻ		10-12	180	13051	2 ⁰ 301	-1°30'	19001	2 ⁰ 00'
5		10-1 2	180	1305'	29001	0901	0°301	3 ⁰ 00'
6		10-12	180	13051	4°00°	o ² co⁺	2 00'	3 ⁰ 00'
7	180	0–8	225	13051	10000	-3°001	100'	۱06 ⁰ 0
8	180	8-0	225	13051	19001	-3°30'	-1°00°	ا‱ا
9	180	0-8	225	1305'	י ⁰ 30י	-1°30'	1000	3 ⁰ 00'
10	180	0-8	225	13051	ا ⁰ 30 ا	-2 ⁰ 00'	o°30¹	0°301
AVERACE					1°30'	-1901	o°301	100,8
		CRUI	SE SPE	ED - 11	O KINOTIS			
ı		10-12	180	1240'	10301	-3 ⁰ 001	-0°301	11 ₀ 00 t
2		10-12	180	1305'	29001	-3°∞¹	-1 ₀ 00;	3 ⁰ 00 •
3		10-12	180	13051	-1°00'	-3°∞¹	-2 <mark>0</mark> 001	°000
4		10-12	180	12851	1°30'	-1°00†	°000	3 ⁰ 001
5		10-12	180	1195'	-1 ⁰ 00'	-2°30'	-1°30'	1 ⁰ 301
6		10-12	180	1305'	00301	-2 ⁰ 301	-0°30'	1°00'
7	180	o - 8	225	1305'	-1°30'	_†o00↓	-3°00'	-0°301
8	180	o - 8	225	13051	00001	-2 ⁰ 001	-1°00'	-1°001
9	180	0-8	225	12401	-1°00'	-3 ⁰ 30†	-2°001	o ⁰ 301
10	180	0-8	225	13051	-19301	-4°301.	-3°001	-0°301
AVERACE					0000	-3°∞¹	-1°301	1°00′

H-13 NORMAL LANDING TO A PANEL

1 FLIGHT NO	2 HEADING (deg)	VELOCITY		5 LENGTH OF FLIGHT PATH	6 DISTANCE TOUCHED DOWN BEFORE TOUCHDOWN PANEL	7 MAXIMUM ATTITUDE	8 DISTANCE COE 7 BEFORE ACTUAL TOUCHDOWN	9 MINIMUM ATTITUUS	10 DISTANCE COL 9 BEFORE ACTUAL TOUCHDOWN	ll AVERAGE ATTITUDE	12 AVERAGE FLIGHT PATH	13 Maximum Sight Angle	14 , DISTANCE COL 13 BEFORE ACTUAL TOUCHDOWN	15 Minimum Sight Angle	16 DISTANCE COL 15 BEFORE ACTUAL TOUCHDOWN	17 Averade Sight Angle
1	180	1 8/g-3 6	330	1285'	0,	29001	4501	-2 ⁰ 001	9251	₀ 030•	-14 ⁰ 30'	23 ⁰ 301	901	13 ⁰ 00'	925'	ر ₀ 30 ا
2	360	18/g-36	330	12401	01	7 ⁰ 00'	י80י	-4°30"	855'	-1 ⁰ 00'	-19 ⁰ ∞¹	19 ⁰ 301	1215'	7 ₀ 30,	ևց։	14 ⁰ 30 ن
3	180	18/g-36	330	9451	01	2 ⁰ 30'	115'	-4 ⁰ 001	6751	-1°00°	-11°30'	14 ⁰ 30'	901	6 ⁰ 30'	6751	9 ° 30'
Ĭţ	160	18/g-36	330	1105*	יס	4°00'	901	ر ⁰ 00 و	970'	و2 ⁰ 00 و	-12 ⁰ 001	21°00'	251	12 ⁰ 301	9901	ن16 ⁰ 00
5	360	18/g-36	330	9251	01	5°30'	701	-1 ⁰ 30'	6551	1°00'	-13°00'	47 ⁰ 301	451	13 ⁰ 00'	6551	18 ⁰ 001
6	•	0-5	360	י 700	0'	4 ⁰ 301	701	-2 ⁰ 001	610'	o ^o 301	-15 ⁰ 30'	ا 26 ⁰ 00	451	13 ⁰ 001	360'	15°00'
7		0-5	360	790	25'	2 ⁰ 00'	45'	-3°00'	360 I	۰۰ ⁰ 30 -	-22 ⁰ 00 °	28 ⁰ 00 r	451	21°30'	360"	24 ⁰ 301
8		0-8	180	675'	0'	5 ⁰ 00'	901	-2 ⁰ 301	3601	0°30•	-15 ⁰ 301	11 ₀ 001	25'	13 ⁰ 30'	2501	17 ° 00'
9		0-2	360	12401	01	17₀00,	70'	-0°30'	810'	2 ⁰ 001	-10°30'	82°30'	25'	10 ⁰ 301	8101	15 ⁰ 30'
10		0-2	360	1015	0'	3 ⁰ 00'	135'	-3 ⁰ 00'	6301	-0°30°	-12 ⁰ 00'	23 ⁰ 001	45.	ا 30°8	2501	י‰בו
n		0-2	360	12401	01	8°30'	45'	0 ⁰ 301	475'	2 ⁹ 00 ·	-13 ⁰ 301	90 ⁰ 001	251	15°∞'	1035'	20 ⁰ 00 '
12		Wind steady toward 10 v guets to 19	with	1195'	25'	6°30'	701	1 ⁰ 00'	6301	2 ⁰ 30'	-11°30'	17 ⁰ 301	430	π ₀ ‰י	115'	140001
13	180	0-5	225	1305'	01	2 ⁰ 001	901	-3 ⁰ 301	1801	-1 ⁰ 301	-13 ⁰ 001	90 ⁰ 001	25'	10 ⁰ 00 '	1080'	14°00'
14	180	0-5	247	י 1260	01	2000	251	-6°00'	1060'	-2°30'	-12°30'	90 ⁰ 001	251	7 ⁰ 001	1080'	14°00'
15	180	0-5	225	1215'	25'	2 ⁰ 301	451	-5°30'	5851	-39001	-18 ⁰ 001	19 ⁰ 301	6551	10 ⁰ 301	70'	17 ⁰ 00'
AVERAGE					05'	^{դ 0} 30 '	1051	-2 ⁰ 301	6501	، 00 ⁰ 0	-14 ⁰ 00'	42 0 001	185'	π _ο 30,	5801	15 ⁰ 30'

H-13 STEEP APPROACH TO A PANEL

1 FLIGHT NO	2 HEADING (deg)	3 WIND VELOCITY (FROM	5 LENCTH OF FLIGHT PATH	6 DISTANCE TOUCHED DOWN BEFORE TOUCHDOWN PANEL	7 Munikam Auvitita	8 DISTANCE COL. 7 BEFORE ACTUAL TOUCHDOWN	MINIMIM ATTITUDE	10 DISTANCE COL 9 BEFORE ACTUAL TOUCHDOWN	11 AVERAGE ATTITUDE	12 AVERAGE FLICET PATH	13 MAXIMUM SICHT ANGLE	14 DISTANCE COL 13 BEFORE ACTUAL TOUCHDOWN	15 MINIMUM SIGET ANGLE	16 DISTANCE COL 15 BEFORE ACTUAL TOUCHDOWN	17 AVERAGE SIGHT ANGLE
1		0-5	3 6 0	1170'	01	50001	701	-3 ⁰ 001	10801	-0°301	-14°001	33°00'	25'	130001	10801	18º30'
2		0-5	360	925	01	7°30°	901	-2 ⁰ 30 ا	7651	00001	-14°30°	21 <mark>0</mark> 30।	70'	120001	9251	15°00'
3		0-5	360	8101	0'	4000	451	-3 ⁰ 001	4501	-0°301	-23°∞'	32 ⁰ 001	451	22°301	4501	240301
14		0-5	360	7451	01	5 ₀ 00,	70*	-2 ⁰ 001	6551	00001	-20°30'	30°00°	451	19 ⁰ 30'	655'	24°001
5		0-2	360	יסקבנ	01	70301	451	-2 ⁰ 30*	6751	00001	-17°301	87°00°	251	16°30'	1601	<u>21</u> 030।
6		0-2	360	10151	יס	6°00•	451	00001	5851	2 ⁰ 301	ن ⁰ 00،	74 ⁰ 301	251	18°30'	970'	24°301
7	-	0-2	360	10351	01	5 ⁰ 00'	70'	00301	4951	20301	-16º30'	32°30'	701	19 ⁰ 001	5851	21°30°
8	Wind 10	steadying with guste	toward to 19	ייסקבנ	01	30001	70'	10301	2251	5 ₀ 001	-14 ⁰ 001	31°30′	45'	16°30'	1035'	19 ⁰ 001
9	180	10-12	247 5	1260	01	2 ⁰ 00'	7201	-5 ⁰ 301	10801	-2 ⁰ 30*	-18 ⁰ 301	89°001	251	16°00'	1240"	29°001
10	180	0-5	180	12601	0'	0000	251	ب°30	7001	-3°001	-15 ⁰ 301	ا ⁰ 00	251	12 ⁰ 301	12401	200001
11	180	0-5	247 5	700	0'	00301	251	-5°001	610'	-2°301	-17°00'	90 ⁰ 001	25'	13 ⁰ 30'	610'	21°30'
12	180	0-5	225	1215'	251	10001	45'	4°00°	4751	-2 ⁰ 30'	-18 ⁰ 30'	17 ⁰ 30'	565	12030	45'	16º001
13	180	0-5	225	11951	70'	00001	2501	-60001	945	-3 ⁰ 001	-18°00°	20°30'	250	6°00'	451	16 ⁰ 30'
AVERAG	E				10'	3 ⁰ 301	120'	-3 ⁰ 001	6751	-0°30'	-17 ⁰ 001	50°004	95'	15°00'	695 i	21 ⁰ 00'

H-13 AUTOROTATION TO A TOUCHDOWN

1 FLICHT NO	2 HEADING (deg)	VELOCITY VELOCITY	4 FROM (deg)	5 LENGTH OF FLIGHT PATH	6 DISTANCE TOUCHED DOWN BEFORE TOUCHDOWN PAMEL	7 MAXIMUM ATTITUDE	8 DISTANCE COL 7 REFORE ACTUAL TOUCHDOWN	9 MINIMUM ATTITUDE	10 DISTANCE COL 9 BEFORE ACTUAL TOUCHDOWN	11 AVERAGE ATTITUDE	12 AVERAGE FLIGHT PATH	13 MAXIMUM SIGHT ANGLE	14 DISTANCE COL 13 BEFORE ACTUAL TOUCHDOWN	15 MINIMUM SIGET ANGLE	16 DISTANCE COL 15 HEFORE ACTUAL TOUCHDOWN	17 AVERAGE SIGHT ANGLE
ı	L-360	18/g-36	330	9451	01	2 ⁰ 00 '	1801	- 3 ⁰ 30'	360'	-0°301	-23°301	880001	25'	30001	701	180301
2	L-360	18/g-36	330	12401	25'	2 ⁰ 001	7901	- 6 ⁰ 30י	1215'	-1 ⁰ 00'	-21 ⁰ 001	27 ⁰ 00 •	6751	14 ⁰ 30'	12151	23 ⁰ 00
3	L-360	18/g-36	330	10351	01	3 ⁰ 00'	2051	- 1000°	6101	0°301	-160301	170001	9901	2 ⁰ 301	135'	וססינו
Ţŧ		0-5	360	יסקננ	251	6º00'	701	- 2 ⁰ 30'	6751	10001	-18 ⁰ 30'	220001	340'	120001	115'	18 ⁰ 00'
5		0-5	360	5201	0'	4°30'	451	- 2 ⁰ 00'	180'	00001	-25 ⁰ 001	900001	25'	20°30'	701	27 ⁰ 30°
6		Q - 5	360	1240'	01	60∞₁	160*	- 11 ₀ 001	2501	-1°00'	-19 ⁰ 001	27 ⁰ 00 '	451	8º00'	115'	18 ⁰ 30'
7		0-2	360	1080'	70'	6°30'	1601	- 1º00'	450*	10001	-21°301	23 ⁰ 00'	9001	5°30'	45 •	18 0 301
8		0-2	360	11251	701	5°30'	451	- 60301	9701	-1°00°	-190001	17 ⁰ 30'	585	7°30'	251	ر3 ⁰ 00
9		10-12	247	5 925	0'	0001	901	-17030ء	9251	-3 ⁰ 001	-25°∞'	89°30'	251	14 ⁰ 001	9251	24°00"
10		0-5	180	9901	01	2 ⁰ 00 '	251	-12 ⁰ 30'	8351	-4°00'	-24°30°	90 ⁰ 001	25'	،30 ،	8101	19 ⁰ 001
AVERAG	E				201	3 ⁰ 301	180'	- 5 ⁰ 001	6301	-1°00'	-21°00'	49 ⁰ 001	365'	10 ⁰ 001	3551	18 ⁰ 301

H-13 NORMAL TAKEOFF

l FLIGHT NO.	2 HEADING (deg.)	VELOCITY		5 LENGTH OF FLIGHT PATH	6 MAXIMUM ATTITUDE	7 DISTANCE COL. 6 AFTER TAKEOFF	8 minimum attitude	9 DISTANCE COL. 8 AFTER TAKEOFF	10 AVERAGE NOSE-UP ATTITUDE	11 AVERAGE NOSE-DOWN ATTITUDE	12 AVERAGE FLIGHT PATH
1	R-180	18/g-36	330	1305'	3 ⁰ 001	13051	-12 ⁰ 301	451	2 ⁰ 001	- 3°30'	7 °00''
2	R-180	80/g-18	330	1215'	0 ⁰ 301	4051	- 4 ⁰ 001	451	0 ⁰ 301	- 2 ⁰ 001	40301
3	R-180			13051	ο ⁰ 301	4951	- 8º001	901	0 ⁰ 30†	- 2 ⁰ 30'	6°00'
4	R-180	80/g-18	330	1125	20001	1060'	- 4°001	2951	1 ⁰ 00'	- 2 ⁰ 301	7 ⁰ 00'
5		0-5	360	1260'	-3 ⁰ 30'	451	-10 ⁰ 00'	4051		- 7 ⁰ 301	7 ⁰ 30'
6		0-5	360	8801	-2°001	، 511	-12 ⁰ 30'	405		- 9 ⁰ 00'	8 [°] ∞'
7		0-8	180	675	-8 ⁰ 00'	675	-13 ⁰ 00'	2251		-11°00'	3 ⁰ ∞'
8		0-2	360	1080'	-6 ⁰ 001	1601	-12 ⁰ 30'	810		- 9 ⁰ 00'	9 ⁰ 001
9		Q - 2	360	1195'	-7°30'	2701	-12 ⁰ 30'	115'		- 9°00'	6 ⁰ ∞'
10		0-2	360	1285	-6°30'	451	-14°30'	5401		-10 ⁰ 00'	7 ⁰ 30'
11		0-2	360	1260'	0 ⁰ 001	1215'	- 9°30'	70'		- 5 ⁰ 00'	10°30'
12		0-2	360	11951	-1 ₀ 00,	1105'	-12 ⁰ 30'	1801		- 6 ⁰ 30'	16°30'
13		steady 10 gusts to 19		1195'	-2°00'	1125	-10 ⁰ 30'	430•		- 7°00'	11°30'
AVERAG	æ			1150'	-2 ⁰ 30'	615'	-10°30'	2801	10001	- 69301	8º00'

H-21 NORMAL TAKEOFF

l FLIGHT NO.	2 HEADING (deg.)	yeloc:		FROM deg.)	5 LENGTH OF FLIGHT PATH	6 MAXIMUM ATTITUDE	7 DISTANCE COL. 6 AFTER TAKEOFF	8 MINIMUM ATTITUDE	9 DISTANCE COL. 8 AFTER TAKEOFF	10 AVERAGE NOSE-UP ATTITUDE	11 AVERAGE NOSE-DOWN ATTITUDE	12 AVERAGE FLIGHT PATH
1	285	15-20 1	uiby	275	7001	140301	610'	-2 ⁰ 00'	315'	2 ⁰ 301	-1°00'	4°30'
2	285	15-20 r	mph	275	585 '	4°00'	4301	-3 ⁰ 001	160'	3 ⁰ 001	-1°30'	3 ⁰ 301
3	285	15 - 20 i	mph	275	810'	8 ⁰ 301	<u> بويد</u>	00001	475	3°30'	None	1°30'
4	285	15-20 1	шрр	275	565'	7 ⁰ 00°	01	00001	2701	2 ⁰ 001	None	1 ⁰ 50'
5	285	15-20 1	mph	275	475'	10001	4301	-3 ⁰ 001	451	1 ⁰ 00'	-1°30'	1 ⁰ 00'
6	285	15 - 20 1	mph	275	5401	1 ⁰ 001	5201	-3°301	2051	10001	-2°00 •	2º001
7	285	5 - 10 i	mph	275	630'	¹⁴ 000 a	451	-1°00°	2951	2 ⁰ 001	-0°30'	1°30'
8	285	5 - 10 :	mph	275	765 '	10°30'	701	-3 ⁰ 001	3851	4 ⁰ 301	-1°30'	10001
9	285	5 -1 0 1	mph	275	700'	2 ⁰ 00'	6751	-3°00'	430	ە°30 ن	-1°30'	1°30'
10	285	5 - 10 1	mph	275	745	7 ⁰ 001	01	-3°00'	4051	2 ⁰ 301	-1°30'	2 ⁰ 001
11	285	5 - 10 :	mph	275	6751	5 ⁰ 001	0"	-2 ⁰ 301	495'	2 ⁰ 301	-1°30'	2 ⁰ 001
12	285	5 -1 0 :	шБр	275	7651	2 ⁰ 00 °	01	-1°30°	2051	10001	-0°30°	2 ⁰ 00'
13	285	5-10	шЪр	275	765	3 ⁰ 301	01	-0°301	2051	10001	-0°30 •	2 ⁰ 00'
AVERAGE	ı					4 ⁰ 301	2801	-2°00'	315'	20001	-1°30'	2 ⁰ 001

H-21 AUTOROTATION LANDING TO TOUCHDOWN

1 FLIGHT NO	2 READING (deg.)	VELOCITY :	l _t FROM deg)	5 LENGTH OF FLIGHT PATH	6 DISTANCE TOUCHED DOWN BEFORE TOUCHDOWN PANEL	7 MAXIMUM ATTITUDE	8 DISTANCE COL 7 BEFORE ACTUAL TOUCHDOWN	9 MINIMUM ATTITUDE	10 DISTANCE COL 9 BEFORE ACTUAL TOUCHDOWN	11 AVERAGE ATTITUDE	12 AVERAGE FLIGHT PATH	13 MAXIMUM SIGHT ANGLE	14 DISTANCE COL 13 HEFORE ACTUAL TOUCHDOWN	15 MINIMUM SIGHT ANGLE	16 DISTANCE COL 15 BEFORE ACTUAL TOUCHDOWN	17 AVERAGE SIGHT ANGLE
ı	285	15-20 ապրհ	275	13501	451	220001	201	170301	12401	17 ⁰ 30'	-14 ⁰ 301	46°30'	201	25 030'	1285'	37 ⁰ 00 •
2	285	15-20 mph	275	12151	45'	20 ⁰ 00 '	901	ا∞ەتت	450	15 ⁰ 30'	-13 ⁰ 30'	35 ⁰ ∞¹	0'	25 ⁰ 001	4501	29 ⁰ 001
3	285	15-20 mph	275	12151	25'	23 ⁰ 301	451	7 ⁰ 301	יס2דד	13 ⁰ 301	-14 ⁰ 30"	58 ⁰ 301	0'	മ <mark>ം</mark> 30 പ	1088	27 ⁰ 001
4	285	15-20 mph	275	י05ננ	25'	20 ⁰ 001	1151	،30متر	1015'	14 ⁰ 001	-17 ⁰ 001	40 ⁰ 00'	0'	28 ⁰ 30'	10351、	32 ⁰ 301
5	285	5-10 mph	275	1305	01	5 ₇₀ 00 ₁	360'	9 ⁰ 30'	1285	16°30'	- 9°001	73 ⁰ 00 '	45*	20 ⁰ 00 r	1215'	31°30'
6	285	5-10 mph	275	1305	0'	18 ⁰ 30'	675'	7 ⁰ 301	9701	14 ⁰ 301	-14 ⁰ 30'	90 ⁰ 001	0'	24°00"	970'	37 ⁰ ∞¹
7	285	5-10 mph	275	1240	901	18°00'	451	5 ⁰ 001	1150'	13 ⁰ 301	-10 ⁰ 00'	27 000'	45'	140001	1150'	മ°∞∙
в	285	5-10 mph	275	11051	0"	28°00'	45•	1 ⁰ 001	7001	8°30'	-22 ⁰ 30'	90°001	01	2 ¹ 1 ⁰ 001	700 '	35°30'
9	285	5-10 mph	275	יסקננ	0"	21 º 00'	901	1000°	10801	9 ⁰ 30	-19 ⁰ 30'	54°301	901	2 ¹ 4°00 •	1080	ن30 ⁰ 30
10	285	5-10 mph	275	1215'	45"	23 ⁰ 301	451	20001	1060	9 ⁰ 30 '	-20 ⁰ 30'	_{የተ} 000 •	201	22 ⁰ 001	1060'	28 ⁰ 00'
11	285	5-10 ագրհ	275	11501	25'	21°30'	90'	7 ₀ 00,	1035'	13°30'	-14 ⁰ 001	38 ⁰ 00'	451	18 ⁰ 00'	1035'	26°00'
12	265	5-10 mph	275	1195'	115'	28 ⁰ 00 '	451	6 ⁰ 301	5201	ייססינו	-1 ₇₀ 00.	39 ⁰ 001	45"	18 ⁰ 301	5201	23 ⁰ 001
13	285	5-10 mgh	275	11951	יפננ	22 ⁰ 30 •	20'	8 ⁰ 301	810'	13 ⁰ 301	-110001	33°30'	201	17 ⁰ 301	4951	ن 22 ⁰ 00
AVERAGE	:				41'	ن ₃₀ 30	120'	7°00'	9201	13 ⁰ 001	-15 ⁰ 00 ا	51°30'	251	21 ⁰ 30'	930"	و29 ⁰ 30

H-21 NORMAL LANDING TO A PANEL

1 FLIGHT NO	2 HEADING (deg)	3 WIND VELOCITY :		5 LENGTH OF FLIGHT PATH	6 DISTANCE TOUCHED DOWN BEFORE TOUGHDOWN PANEL	7 MAXIMUM ATTITUDE	8 DISTANCE COL 7 HEFORE ACTUAL TOUCHDOWN	9 MINIMUM ATTITUDE	10 DISTANCE COL 9 REPORE ACTUAL TOUCHDOWN	11. AVERAGE ATTITUDE	12 AVERAGE FLIGHT PATE	13 Maximum Sigett Angle	14 DISTANCE COL 13 BEFORE ACTUAL TOUCHDOWN	15 MINIMUM SIGRT ANGLE	16 DISTANCE COL 15 BEFORE ACTUAL TOUCHDOWN	17 AVERAGE SIGHT ANGLE
ı	285	15-25 ապրհ	275	12851	45"	22 ⁰ 30*	0,	9 ⁰ 00'	1035'	13 ⁰ 301	- 8º001	46°301	0'	17 ⁰ 301	10801	24°30'
2	285	15-25 mph	275	12601	25 '	17000	0'	7 ⁰ 001	1190'	12 ⁰ 301	- 9 º 00'	70 0 001	01	16°30′	11901	23 ⁰ 001
3	285	15-25 արև	275	12851	251	199001	201	6°00'	12601	12 ⁰ 301	- 7 ⁰ 001	57 ⁰ 301	0'	13 ⁰ 00'	1220'	23 ⁰ 00 •
4	285	5 -10 ազմ ահ	275	1215'	25'	22 ⁰ 001	0'	13 ⁰ 001	10101	15°30'	- 7°30'	90 ⁰ 001	0'	22 ⁰ 001	1010'	29 ⁰ 001
5	285	5-10 mph	275	11501	701	200001	70"	7°30'	י25ננ	13 ⁰ 001	- 7 ⁰ 001	32 ⁰ 001	70'	14 ⁰ 301	1125'	21 ⁰ 301
6	285	5-10 mph	275	11951	01	19 ⁰ 30'	901	8º30'	1170	13 ⁰ 00'	- 7 ⁰ 001	86 ⁰ 001	201	16°30′	1170'	26°00°
7	285	5-10 maph	275	9251	01	23 ⁰ 301	70'	7 ⁰ 00'	9001	770001	-17 ₀ 001	70 ⁰ 301	201	19 ⁰ 301	9001	31 ₀ 001
8	285	5-10 առաջե	275	1305'	0*	17 ⁰ 00 ن	50'	8°30'	11001	12 ⁰ 00'	،30°متد-	90 ⁰ 001	01	20°30"	1100'	28 ⁰ 30
average					244	50 ₀ 00,	40°	8°301	1135،	13 ⁰ 00'	- 8 ⁰ 301	68 ⁰ 001	15'	18º001	17001	25 ⁰ 301
		H-21	STE	EP APPRO	ACH TO A PA	NEL.										
ı	285	10-15 mph	275	9701	451	15°30'	90'	$\pi_o \infty$.	945'	13 ⁰ 30'	-13 ⁰ 301	37 ⁰ 001	90'	25 ⁰ 30'	9451	32 ⁰ 001
2	285	10-15 mph	275	1285'	01	20 ⁰ 301	1190'	11°00'	540"	14 ⁰ 301	-15 ⁰ 001	90 ⁰ 00 r	0'	27 ⁰ 00'	5201	33°00'
3	265	10-15 mmph	275	ا1260	251	16°30′	201	9°30'	1035'	13 ⁰ 30'	-13 ⁰ 001	ن30°15	0"	23 ⁰ 001	1035'	28°00'
4	285	10-15 mgh	275	1305'	0*	18 <u>0</u> 001	500,	90001	75f0,	13 ⁰ 00'	-12 ⁰ 00'	⁹⁰ 00₁	0,	5J ₀ 001	12401	30 ₀ 00 i
5	285	5-10 mph	275	12601	25'	21°30'	90'	10 ⁰ 001	1080'	14°00'	-14 ⁰ 30'	39°00'	45'	250301	1125'	30°001
6	285	5-10 mph	275	12401	0'	22 ⁰ 001	1215'	ەن∞،	5851	15°30'	-12°30'	90°00'	0'	27 ⁰ 301	5851	34°30'
7	285	5-10 mph	275	12401	251	ن16 ⁰ 00	610'	100001	י05ננ	13 ⁰ 00'	ا 9 ⁰ 00	49°30'	0'	19 ⁰ 30'	י071נ	26 ⁰ 00'
8	285	5-10 mph	275	9701	01	17 ⁰ 001	201	5°30'	8551	ı‱ı	-16 ⁰ 00'	90°00'	0'	22 <i>0</i> 301	8551	33°83'
9	285	5-10 mph	275	יסגת	0'	17 ⁰ 001	901	5 ⁰ 301	11251	70 <u>,</u> 00,	-19 ⁰ 30*	83°00'	יס	26 ⁰ 301	1125'	35°30°
10	285	5-10 ஜூ	275	1305'	0'	20°00'	701	9 °00'	1010'	12 ⁰ 001	-17 ⁰ 001	67 ⁰ 001	20'	28 ⁰ 00 °	1010'	33 ⁰ 30'
11	285	5-10 ար հ	275	13051	01	15 9 001	1351	90301	900*	12900	-16°30'	909001	0'	2მ ^ი იი '	900*	35°30'
- AVERAGE	;				111	189001	325'	9 °00¹	10201	139001	-14°30*	70 ° 301	15'	25 ⁰ 00'	9601	35 ₆ 00,

H-34 STEEP APPROACH TO A PANEL

1 F1 IGHT NO	2 HEADING (deg)	AETOCILA MIV		5 LENCTII OF FLIGHT PATH	6 DISTANCE TOUCHED DOWN BEFORE TOUCHDOWN PANEL	7 maximum attitude	8 DISTANCE COL 7 BEFORE ACTUAL TOUCHDOWN	9 MINIMUM MUTITUAE	10 DISTANCE COL 9 BEFORE ACTUAL TOUCHDOWN	ll Average Attitude	12 AVERAGE FLIGHT PATH	13 MAXIMUM SIGHT ANGLE	14 DISTANCE COL 13 BEFORE ACTUAL TOUCHDOWN	15 MINIMUM SIGHT ANGLE	16 DISTANCE COL 15 BEFORE ACTUAL TOUCHDOWN	17 AVERAGE SIGHT ANGLE
1	180	0		1125'	ە09	9 ⁰ 301	יכננ	4°301	700'	7 ⁰ 001	-11 ₀ 001	19000	2251	14 ⁰ 301	6751	16 ⁰ 30'
- 2	180	٥		1125	135"	16°00'	135'	6 ⁰ 001	7201	9 °30'	- 7 ⁰ 001	220301	135'	110301	7201	15°30'
3	180	0	~ -	1450"	יפננ	110301	ا کتد	5°30'	990•	6°30'	- 8º001	17 ⁰ 301	135'	120301	9901	15 ⁰ 30'
ήt	180	٥		1720،	70'	ن∞مت	451	10301	475	7 ⁰ 00'	-14 ⁰ 001	22 ⁰ 001	205	17 ⁰ 00•	11251	19 ⁰ 301
5	180	0		11951	70'	13°30'	901	7°30'	745"	99001	- 7º301	25 ⁰ 001	25'	15 ⁰ 30,	ن 990	18 ⁰ 30
6	1.80	٥		1125'	- 70'	11000'	2051	6°30'	251	8º30'	-10°00'	21000'	160'	16 ⁰ 30+	9451	19 ⁰ 00'
7	180	0		11501	ילננ	9 ⁰ 50'	901	יססיק	610 t	8°00'	- 7 ⁰ 301	16°00'	4051	130301	8101	140301
8	180	0		1150'	901	12 ⁰ 301	8551	4 ⁰ 30'	1125	8 ° 30'	-13°00'	26000	855'	17 ⁰ 001	י25י	22 ⁰ 00'
9	180	0		1015'	2051	14 ⁰ 00'	2051	5°30'	9901	10°30'	- 9 ⁰ 001	19 ° 30'	2051	14 ⁰ 001	9901	170001
10	280	a		1015'	901	15°30'	901	60001	8551	ر00 ⁰ 00ء	-12 ⁶ 001	26 ⁰ 301	901	18 ⁰ 001	855 '	22 ⁰ 001
11	180	0	- -	יס7גנ	115'	110001	251	6°30'	4301	80001	-12 ⁰ 301	210001	900'	17 ⁰ 001	430 r	ا 19 ⁰ 00
12	ە80	0		1195'	701	12 ⁰ 00'	1601	1000°	7201	8 ⁰ 00'	-13 ⁰ 301	23 ⁰ 301	540'	16°30•	251	20°30'
13	180	0		1080'	45"	10 ⁰ 301	1601	4°30'	451	ر ₃₀ 0	- 8º001	ا 30°19	160'	13 ⁰ 301	8551	16°00'
14	180	0		10601	901	9°30'	01	3°30'	701	5°30'	-17 ⁰ 301	23°30'	4051	16°∞•	701	20 ⁰ 301
15	780	O		יסקבב	451	9 ⁰ 00 ⊓	251	4°00'	1035'	6°30'	-100001	ن 30°25	25'	140001	1035'	176301
16	180	0		11951	251	90301	160 ،	2 ⁰ 30'	11251	7 ⁰ 001	-18º30'	և8 ⁰ 30 ^լ	01	و21 ⁰ 00	1125'	28 0 30•
17	180	0		1215'	25"	90001	107	ħ _O OO i	9251	6º30'	-19 ⁰ 001	28°301	701	5ή _ο 00 ι	270'	260301
18	160	0		1080'	701	17 ₀ 001	901	80001	7001	96301	- 9 ⁰ 001	21°301	901	17 ⁰ 001	700'	19 ⁰ 00'
19	180	0		1205 ن	45"	10 ⁰ 301	9001	6 ⁰ 301	901	9000'	-14°301	2 6º10 '	0'	21 ⁰ 00 •	5651	230301
20	180	o		1060	205"	12 ⁰ 301	2251	4º30"	4501	80001	-10 ⁰ 00'	18°30'	8351	110301	4501	15 ⁰ 301
21	180	0		1215'	45'	12 ⁰ 00'	5151	6°301	2251	8 ⁰ 301	-11°30'	25°30'	251	16 ⁰ 301	4301	200001
AVERAG	E				85'	11 ₀ 301	2501	5°30'	6901	8°00'	-11 ⁰ 301	23 ⁰ 301	4201	16 ⁰ 001	7151	19 ⁰ 301

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H-34 LANDING OVER 60-FOOT TREES WITHIN 250 FEET OF THESE TREES

l FLIGHT NO	2 3 4 HEADING WIND VELOCITY FROM	5 MAXIMUM ATTITUDE	6 MINIMUM ATTITUDE	7 AVERAGE ATTITUDE	8 AVERAGE FLIGHT PATH	9 MAXIMUM SIGHT ANGLE	10 MINIMUM SIGHT ANGLE	ll Average Siget Angle
1	Communications failed	л _о 301	30301	8º001	140 0 01	26 ⁰ 001	160301	21°∞¹
2	between operator and aircraft	140301	ر ₀₃ 0ء	100301	12 ⁰ 00'	27 ⁰ 001	18 ⁰ 30'	22 0 301
3		17 ₀ 30،	11 ₀ 00 1	7 ⁰ 001	140301	26 ⁰ 301	17 ⁰ 301	21°∞י
4		13 ⁰ 001	8 ⁰ 001	10 ⁰ 001	12 ⁰ 301	25°30'	19 °3 0'	22 ⁰ 301
5		120301	3° 3 01	7 ⁰ 001	140301	27 ⁰ 301	17 ⁰ 30'	21 ° 00'
6		13 ⁰ 301	8000 i	17 ₀ 00,	140001	27 ⁰ 001	21°30'	24°001
7		10 ⁰ 001	20301	7 ⁰ 00'	13 ⁰ 001	24 ⁰ 00 '	14 ⁰ 301	20°00'
8		110001	<i>6</i> 0301	8000°	10 ⁰ 301	23 ⁰ 30'	17 ⁰ 30'	19 ° 30°
9		10 ⁰ 00'	6°001	8 ⁹ 001	130001	24°30'	19 ⁰ 301	21 °30'
10		10 ⁰ 301	6 ⁰ 301	გ _ი 00•	9 º00'	35°30'	ن30°301	23 ⁰ 301
11		14 ⁰ 00'	2 ⁰ 001	9 ⁰ 00'	100001	58 ⁰ 001	18 ₀ 001	34°301
12		100301	6°00'	9 ⁰ 001	14 ⁰ 001	54°00'	22 ⁰ 00 '	32 0 301
13		π _ο ωι	4°30'	7 ⁰ 301	10 ⁰ 301	36 ⁰ 301	18 <mark>0</mark> 30 ا	23 ⁰ 901
14		13 ⁰ 301	5 ⁰ ∞'	9 ⁰ 301	13 ⁰ 001	50°00'	20°30'	29°30'
15		9 0 001	7°30'	80301	13 ⁰ 00'	27 ⁰ 301	21°00'	24 ⁰ 00'
16		9 ⁰ 001	20001	5 ⁰ 001	17 ₀ 00,	23 °00'	14°00'	17 ⁰ 30 ا
17		ن‰تت	י∞י7	9 ⁰ 00'	12 ⁰ 00'	61°00'	23 ⁰ 301	31 ° 30'
18		ا ⁰ 00 ن	6 ⁰ 001	7 ⁰ 301	י _י פסית	52 ⁰ 30'	21°30'	27 ⁰ 30 '
19		100301	3°30'	6 ⁰ 301	15 ⁰ 00'	22°00'	15 ⁰ 301	18 ⁰ 24"
20		120301	გ <mark>ი</mark> ვი I	100301	160301	45°30'	28 ⁰ 30 I	35 ⁰ 00 '
21		12 ⁰ 001	7 ⁰ 00'	в <mark>о</mark> 00 г	13 ⁰ 301	24 ⁰ 301	20 ⁰ 301	21°30'
22		9 ⁰ 301	5 ⁰ 301	7°30'	14 ⁰ 301	25 ⁰ 001	19 ⁰ 00 '	22 ⁰ 30 '
23		ار‱00	6°30'	ხ ⁰ 301	12 ⁰ 30'	25 ⁰ 301	20 ⁰ 301	23 ⁰ 001
24		10 ⁰ 301	7 ⁰ 30'	9 000'	12 ⁰ 00 י	66 ⁰ 301	220001	יי ₃₁ 0
25		13 ⁰ 00'	ا ⁰ 00	п _о ю,	12 ⁰ 00 '	90 ⁰ 001	24°301	39 ⁰ 00 ¹
26		10 ⁰ 30'	6°30'	გ <mark>ი</mark> 30 -	13 ⁰ 001	26 ⁰ 301	19 ° 30'	ا <u>22⁰00</u>
AVERAGE		יס3סידו	6 ⁰ 00 '	გ <mark>ი</mark> 30 г	13°∞'	37 ⁰ 00 '	19 ⁰ 30'	25 ⁰ 00 '

H-34 AUTOROTATION TO A PANEL

NO FLIGHT I	2 HEADING (deg)	AETOCIAA AIU	5 LENGTH OF FLIGHT PATH	6 DISTANCE TOUCHED DOWN BEFORE TOUCHDOWN PANEL	7 MAXIMUM ATTITUDE	8 DISTANCE COL 7 BEFORE ACTUAL TOUCHDOWN	9 MINIMUM ATTITUDE	10 DISTANCE COL 9 REFORE ACTUAL TOUCEDOWN	11 AVERAGE ATTITUDE	12 AVERAGE FLIGHT PATH	13 MAXIMUM SIGHT ANGLE	14 DISTANCE COL 13 HEFORE ACTUAL TOUCHDOWN	15 MINIMUM SIGHT ANGLE	16 DISTANCE COL 15 BEFORE ACTUAL TOUCHDOWN	17 AVERAGE SIGET ANGLE
1	180	0	 יסקבנ	90'	16 ° 30′	1150	6 ⁰ 001	3 ⁴⁰ '	، 30م	-14°001	30°30'	9901	20 ° 00 '	70 '	ر ₂₅ °30 و
2	180	o	 1240'	0"	23 °30'	2701	3 001	1170	10°00'	- 9 ⁰ 30'	50 ⁰ 301	251	12 ⁰ 30′	1170'	57,001
3	180	o	 י1215	25'	17°30'	901	5°30'	2951	10°301	ن‱ <u>س</u> ۔	55° 30'	0"	19 ⁰ 30'	1080'	24 ⁰ 30′
4	180	0	 1105'	1351	19 ⁰ 001	451	7 ⁰ 30¹	4951	ن30° تت	-12°30'	28°001	9901	170301	1801	23 00'
5	180	o	 1215	45"	18 ⁰ 00 •	251	6 ⁰ 30'	9451	ന₀∞₁	- 9°30'	45 ⁰ 301	01	ن ⁰ 30 ن	9451	22 301
6	180	0	 12601	0'	15°00'	10151	0000	225'	7 ⁰ 301	-15 ⁰ 30'	38 ⁰ 301	451	16 ⁰ 30'	1215'	26 ⁰ 001
7	180	0	 י1035	2251	18 ⁰ 30'	45"	4°301	810'	9 ⁰ 00 •	- 7º301	24 ⁰ 001	45	100301	655'	14 ⁰ 301
8	180	0	 1170'	701	ن19 ⁰ 00	1080'	3°00'	4301	π _ο ∞,	-17 ⁰ 00'	38°30'	901	25 ⁰ 001	430°	37°30'
9	180	0	 11951	70"	22 ° ∞'	01	7 ⁰ 00'	5201	170001	- 7°30'	40 ⁰ 00 ¹	0"	13 ⁰ 30'	520"	18°30'
10	180	0	 11951	701	18 ⁰ 00'	90'	4°301	1195'	9°30'	-10 ⁰ 00'	29 ⁰ 001	901	140301	9701	19 ° 30°
11	180	0	 10801	180'	18 ⁰ 301	25'	3 ⁰ 301	و855	90301	- 7º00'	26 ⁰ 001	25'	10 ⁰ 00′	855'	16 ⁰ 001
AVERAGE	:			851	ارو ⁶ 30	3601	4° 30 '	745	10°30'	-17 ₀ 00،	37°00'	2851	160001	8001	220301

H-34 NORMAL LANDING TO A PANEL

1 FLIGHT NO	2 HEADING (deg)	3 WIND VELOCITY	5 LENGTH OF FLICHT PATH	6 DISTANCE TOUCHED DOWN BEFORE TOUCHDOWN PANEL	7 MAXIMUM ATTITUDE	8 DISTANCE COL 7 BEFORE ACTUAL TOUCHDOWN	9 MINIMUM ATTITUDE	10 DISTANCE COL 9 BEFORE ACTUAL TOUCHDOWN	11 AVERAGE ATTITUDE	12 AVERAGE FLIGHT PATH	13 MAXIMUM SIGHT ANGLE	14 DISTANCE COL 13 BEFORE ACTUAL TOUCHDOWN	15 MINIMUM SIGHT ANGLE	16 DISTANCE COL 15 BEFORE ACTUAL TOUCHDOWN	l7 AVERAGE SIGHT ANGLE
1	180	0	 11501	901	12000	901	5°301	45'	8°00'	- 3º30'	28°301	901	10 ⁰ 301	5651	13000'
2	180	o	 1195	701	13 ⁰ 00'	160'	5°30'	1080'	9 ° 00'	- 6º30'	22 ° 00'	ا 160	120001	990 '	16 ⁰ 00
3	180	o	 1015'	135"	15°301	160'	8°00'	520'	17 ₀ 00 •	- 4º30'	Not Ava	ilable	Not Ava	ilable	
14	180	0	 1150'	901	12 ⁰ 001	135'	5 ⁰ 301	451	8 ⁰ 00 •	- 6º30'	18º30'	135'	6 ⁰ 001	25 '	14 ⁰ 301
5	180	0	 יסקבנ	45'	15 ⁰ 00'	135'	5 ⁰ 001	1080'	9 °00 '	- 7º00'	25 ⁰ 001	135'	120301	9251	18°00'
6	180	0	 1035	205'	139001	225'	7 ⁰ 30•	7901	10 ⁰ 30	- 4º00'	17 ⁰ 30۱	2251	12 ⁰ 00'	1015'	15 ⁰ 001
7	180	o	 11051	135'	14°00'	25'	6 ⁰ 001	970'	10 ⁰ 001	- 6 ⁰ 001	ار000ع	251	12 ⁰ ∞'	970'	16°00'
8	180	0	 10801	901	л ₀ 00 і	1601	ن30°5	1060'	9 ⁰ 00 r	- 5 ⁰ 001	17 ⁰ 30	70°	110001	1060'	14°30'
9	180	٥	 1150 ن	90'	13 ⁰ 00'	225'	3°30'	1105'	8º001	- 5º30'	19 0 301	2501	9 ⁰ 301	1105'	14°30'
10	180	0	 990'	160'	160001	2051	7⁰∞'	251	ب30°متت	- 4 ⁰ 301	20 ⁰ 30 •	1801	140301	5401	16 ⁰ 00′
11	180	0	 1125'	90'	13 ⁰ 00'	2501	6º301	1060'	90301	- 7 ⁰ 00'	21000 ല	2501	14°00'	1060'	17 ⁰ 30'
12	180	o	 11501	901	17 ₀ 30.	2501	2 ⁰ 001	1060'	8°00'	- 7 ⁰ 301	20 ⁰ 00 •	270'	10 ⁰ 00 '	1060'	16°30'
13	180	o	 1080'	2051	14°00'	115'	7 000'	900'	9 ⁰ 301	- 6°30'	18 ⁰ 00'	1601	13 ⁰ 001	1105'	15 ⁰ 00'
14	180	٥	 1125'	901	80401	315'	5°00'	251	7 ⁰ 00 '	- 7°30'	16 ⁰ 00 ا	45'	12 ⁰ 30 '	1060'	14º00'
15	180	0	 10601	2051	10°30'	135'	4º00'	9451	6°30'	- 4°30'	14°30'	135'	8 ⁰ 301	945'	10 ° 30'
16	180	0	 1105'	45'	10 ⁰ 30'	1801	6°00'	3401	9 °00'	- 7º30'	36 ⁰ 301	251	15 ⁰ 30'	340*	19 ⁰ 30'
17	180	0	 1150،	451	יסניינו	901	5 ⁰ 001	8351	י‱י	-12 ⁰ 301	22 ⁰ 301	901	17 ⁰ 00 ا	1835 و	19 ⁰ 001
18	180	0	 יס7ננ	45'	10°30'	225'	40001	12500	7 ⁰ 30'	-12 ⁰ 30'	22 ⁰ 30'	2251	16º00'	1250 ن	19 ⁰ 30'
19	180	o	 1170	901	،30متت	701	3°30'	1080'	8º001	- 4º30'	19 ⁰ 30'	יס	8º001	10801	13 ⁰ 301
20	180	o	 1215'	45'	160001	180'	1 ₀ 30,	11501	10 ₀ 001	- 9 ⁰ 00'	25°00'	1801	140001	11501	19 ⁰ 30'
ല	180	0	 1015	205"	8º00"	115'	4 ⁰ 001	6551	5°30'	-10 ⁰ 00'	14°00'	9251	7 ⁰ 301	1 251	12°30'
AVERAG	E			108	1 <i>2</i> 030	170'	5 ⁰ 301	8051	8º301	- 7º00'	മ⁰യ∙	215'	12 ⁰ 301	8601	ن30°15

H-34 NORMAL TAKEOFF

l FLIGHT NO	2 HEADING (deg)		3 4 WIND CITY FROM	5 Length of Flight Path	6 MAXIMUM ATTITUDE	7 DISTANCE COL 6 AFTER TAKEOFF	8 MINIMUM AUTITUDE	9 DISTANCE COL 8 AFTER TAKEOFF	10 AVERAGE NOSE-UP ATTITUDE	11 AVERAGE NOSE-DOWN ATTITUDE	12 AVERAGE FLIGHT PATH
1	360	(1260'	5°00'	1240	- 4º00°	125'	10001	- 10301	5°∞'
2	360	C		1260'	4 ⁰ 001	3851	- 5º301	7901	10301	- 2º301	4º00"
3	360	C		1260'	60301	1240'	- 9º001	451	700°	- 4º301	6°00'
4	360	c		1260'	5000	1240'	- 8000	701	2 ⁰ 301	- 5°00'	6∞'
5	360	c	· ~-	1260י	0000	1105'	- 8 ⁰ 30י	2251	None	- 4º301	3°30'
6	360	C		1260'	Jo004	8551	- 9°30'	1151	00301	- 3°30'	5°30'
7	360	C		12401	00001	10151	رە30° -	901	None	- 2°30'	5 ⁰ 001
8	360	C	·	12601	00301	י25י	- 6°30'	251	00001	- 3°00'	3°30'
9	360	Ç		1215'	00001	01	- 8 ⁰ 30 -	70'	None	- 3°30'	3°30'
10	3 6 0	î c)	1305'	0000	610"	-10 ⁰ 001	251	None	- 3°∞¹	4°30'
17	360	C	ı	1260'	4°00'	7451	-16 ⁰ 30	ا رکند	3°00'	-17 ₀ ∞າ	6°30'
12	360	C)	1260'	00001	451	- 7 ⁰ 00	901	None	- 3°30'	^{4°} ∞'
13	360	()	1260'	1°30'	7001	-13 ⁰ 30'	135'	10001	- 5°301	ن 30°6
14	360	()	12401	10001	1125'	ار ₀ 30ء	115'	10001	- 5°001	6°00'
15	360	(11951	1 ⁰ 301	855'	۰۳ ₀ ∞،	1601	10001	- 6°001	8°00'
16	360	C)	12601	40301	1060'	- 9 ⁰ 301	251	2°30'	- 6º001	6°30'
18	360		0	7901	-2°301	540	- 80001	180'	None	- 5°00'	7°00*
19	360		0	1240'	1°30'	9701	-10°30'	701	a°301	- 6°30'	5°∞¹
20	360		0	12601	00001	251	~15°301	115'	None	- 7 ⁰ 001	7 ⁰ ∞¹
21	360		0	12601	1°30'	9251	-16°30'	901	1°30'	- 6°301	6 ⁰ 00'
22	360		0	12601	-1°30'	810'	-130301	2701	None	- 5°30'	50001
AVERAGE					1°30'	8001	-10 ⁰ 001	145'	5 ₀ 001	- 4°301	5°30'

NOTE WHERE NO INFORMATION IS GIVEN IN COLUMNS 2, 3, AND 4, COMMUNICATIONS FAILURE BETWEEN THE FAIRCRILD OPERATOR AND PILOT OF THE AIRCRAFT CONCERNED OCCURRED

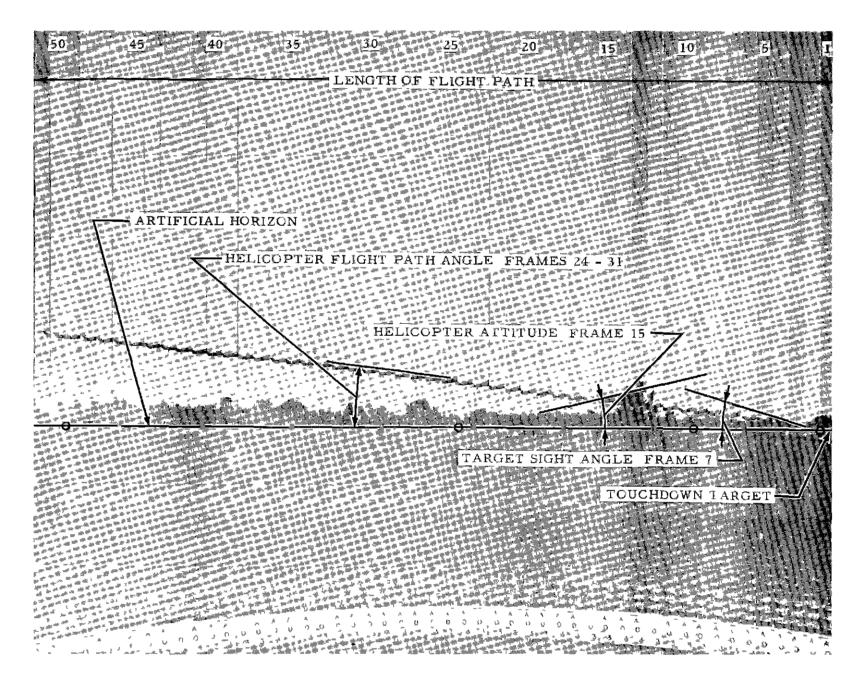


FIG 55 METHOD OF ANALYSIS FOR A NORMAL LANDING H-34

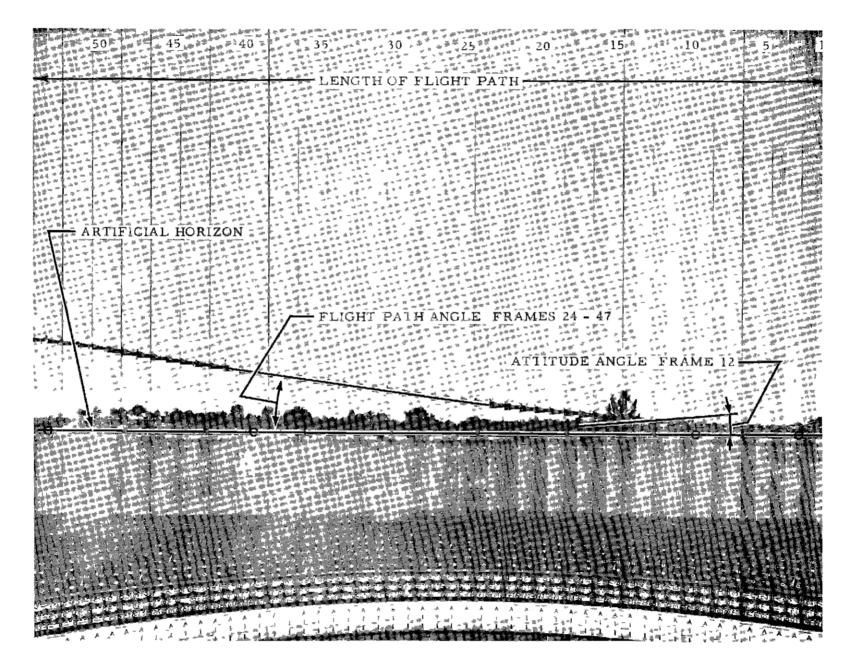


FIG 56 METHOD OF ANALYSIS FOR A NORMAL TAKEOFF H-34

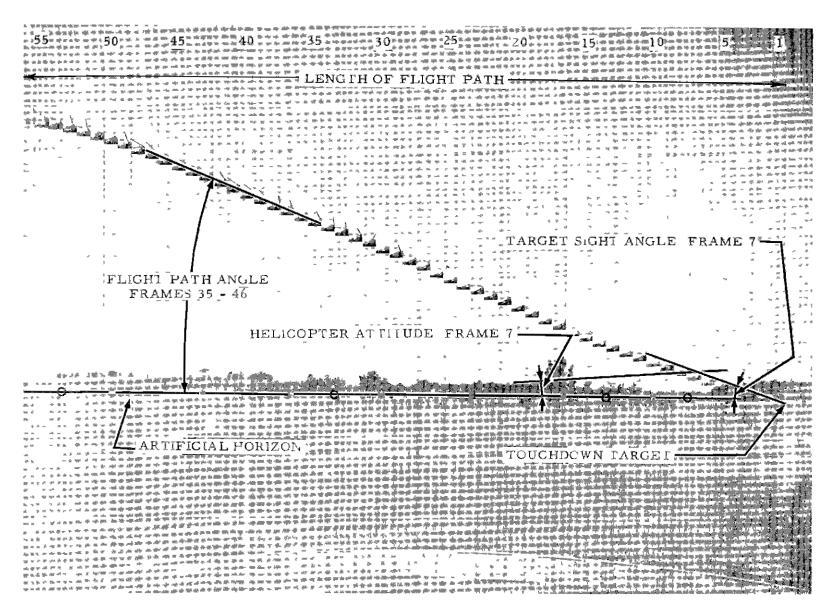


FIG 57 METHOD OF ANALYSIS FOR A STEEP APPROACH H-34

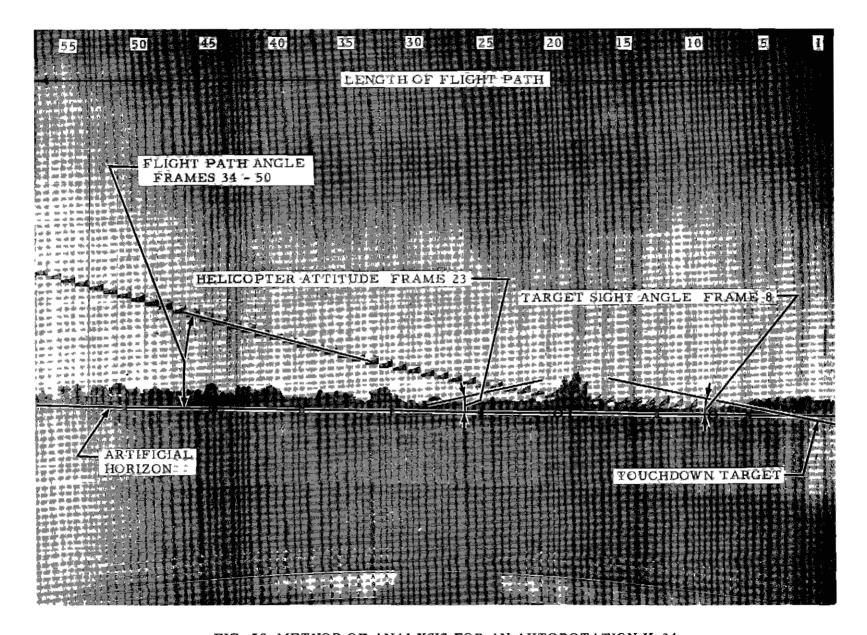


FIG. 58 METHOD OF ANALYSIS FOR AN AUTOROTATION H-34

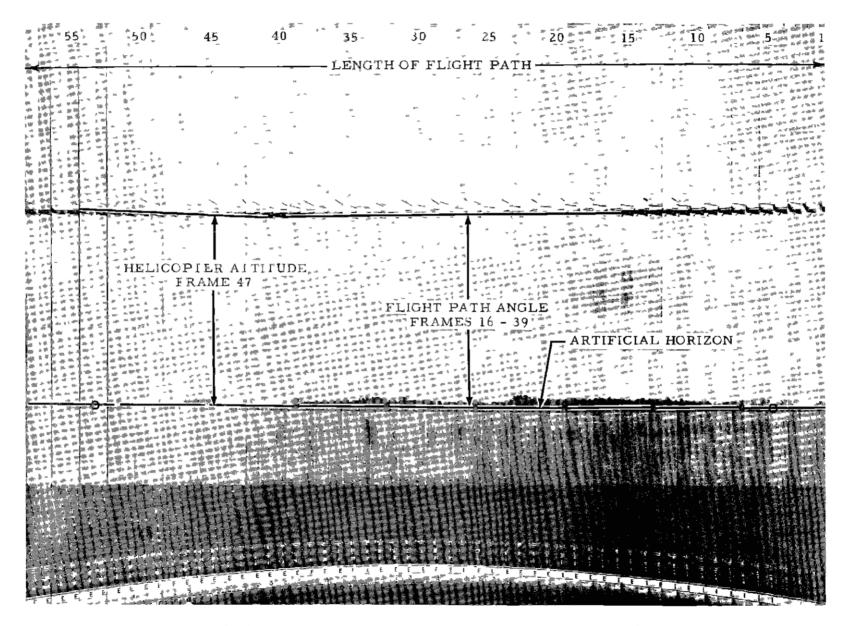


FIG 59 METHOD OF ANALYSIS FOR CRUISE FLIGHT H-34

AFPENDIX III

U. S. DEPARTMENT OF COINCRCE
CIVIL AERONAUTICS ADMINISTRATION
Technical Development Center
P. O. Box 5767
Indianapolis, Indiana

PILOT QUESTIONNAIRE

ON

COCKPIT VISIBILITY IN HELICOPTERS

The CAA Technical Development Center is conducting a project to determine the cockpit visibility requirements for helicopters. The ultimate objective of the project is to establish standards of cockpit visibility which will be consistent with adequate operating safety.

The present phase of the program calls for the determination of helicopter operating characteristics during certain maneuvers, particularly landings and takeoffs, and the visibility requirements necessary for the safe and satisfactory accomplishment of these maneuvers.

Cockpit visibility requirements are directly related to the field of vision required to perform various maneuvers. This field of vision is being investigated not only to determine how it affects the limits of the windshield but also to determine how much obstruction is acceptable within the windshield area.

Some helicopters have excellent visibility. As the helicopter is developed and adapted to a wide range of uses, instruments will be added and structural requirements increased. Both of these requirements will tend to reduce visibility. In some instances this is already a fact.

Since the best judges of the cockpit visibility needed for safe flight are the pilots themselves, we are asking you to supply us with the necessary basic information from which an analysis can be made. Such an analysis will permit a quantitative evaluation of the present designs and the establishment of quantitative standards for future designs.

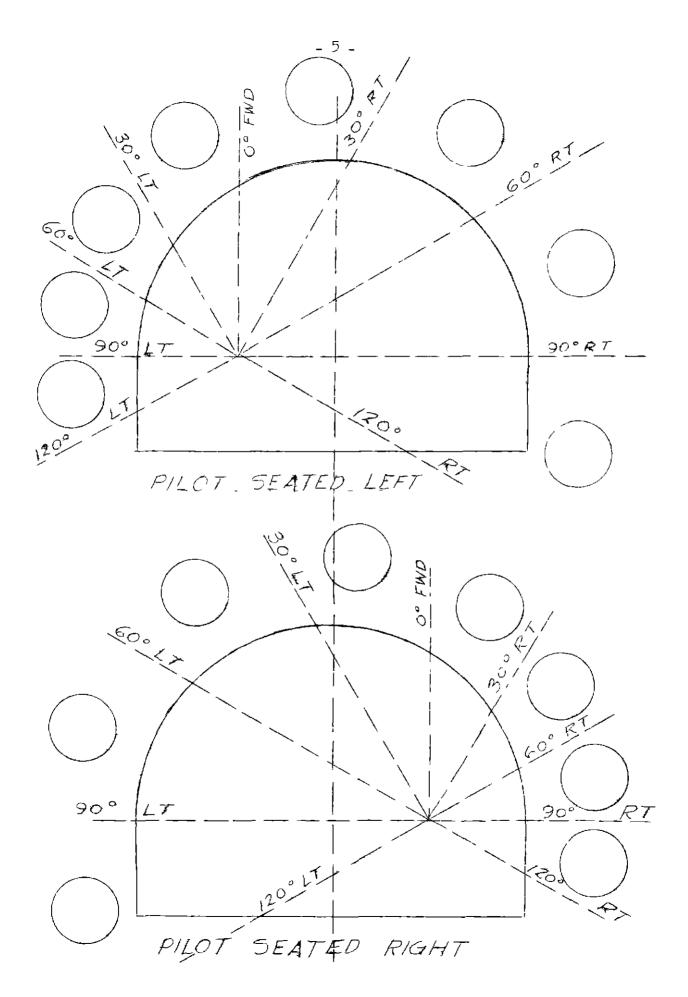
The success of this cockpit visibility study is dependent upon your returning this questionnaire as soon as possible with all questions answered. Your response will be strictly confidential and will be used only for analytical purposes by the CAA. No one else is authorized to see your answers or to use the information for any other purpose.

Name	e					
ı.	Hel	ght _				
2.	a.	How	many	years	have you been flying aircraft? _	
	b.	How	many	total	hours do you have in aircraft?	
	c.	How	many	hours	do you have in helicopters?	

3•	a,	In what model helicopter are y	ou now doing most	or all of your		
	b.	How many hours do you have in	this helicopter? _			
	с.	How many hours flying time in per month?	this helicopter do	you now get		
4.	a.	Do you consider it necessary, from an over-all safety standpoint, to be able to see any external portion of the aircraft from the cockpit? Yes No				
	ъ.	If you answered "Yes" to part a, check the boxes below indicating the portions of the helicopter which you feel must be seen and the amount of head and body movement which you feel is permissible in each case.				
		Moderate Head & Eye Movement	Moderate Head & Body Movement	Maximum Permissible Movement with Seat Belt Fastened		
Skida or Wheels						
	n Ro Tıps	l l				
Oth (Sp	e r e c if	у)				
5•	men urg	ce a check mark in one of the belower for which you feel maximum ently required. Do not considerations.	m visibility from	the cockpit is most		
		l. Takeoff run				
		3. Landing (including hovering in ground effect)				
		4. Autorotation - glid	de			

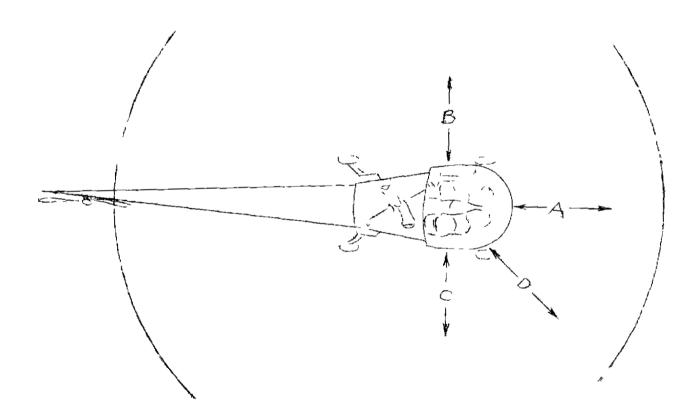
<u></u> 5.	Autorotation - landing
6.	Maximum performance takeoff
7.	Landing in a confined area
8.	Hovering for rescue purposes or similar ground operations.

- 6. Two charts to be used for estimating the obstruction to vision at different lateral directions of sight which you feel is permissible for safe operation of the aircraft during all maneuvers are shown below. One chart provides for the pilot seated in the left seat, the other chart provides for the pilot in the right seat. These charts are provided with radial lines emitting every 30 degrees from the indicated cockpit seat. Between each radial line is a box in which your evaluation of that region should be entered. Rating scales to be used for your evaluation follow
 - 1. No obstruction permissible.
 - 2. Minor obstructions permissible, such as thin posts (1-1/2) wide or smaller, etc.
 - 3. More obstruction permissible, such as wide posts (greater than 1-1/2"), etc.



7. How close to the helicopter (in feet) do you desire to see the ground along each of the sight planes indicated under the given conditions.

a.	When coming to a Distance A		Distance C	Distance D
ď		at anstant of tou Distance B		Distance D
c.		where such visio		Distance D



Note: This is a plan view of a helicopter. Distances are to be ground (or horizontal) distances.

Note: If the helicopter you are considering has the normal pilot's seat on the left, consider dimension D as being between A and B.

- 8. For the model you have indicated in Question 3, please classify the visibility for the maneuvers and directions shown in the following table. This classification should be made on the basis of adequacy of windshield area and angles of sight from the normal pilot seat in the cockpit. The numbers of classifications to be used are:
 - 1. Visibility excellent, no improvement desired.
 - 2. Visibility adequate, some improvement desirable, but not mandatory.
 - 3. Visibility not adequate, improvement strongly desired.

Fill in every space in the table below with the number you think classifies the visibility of this aircraft for each maneuver and direction indicated.

MANEUVER	VISIBILITY UPWARD	VISIBILITY DOWNWARD	VISIBILITY	SIDEWARD
_	TO FRONT	TO FRONT	TO LEFT	TO RIGHT
Takeoff run				
Straight clim	·			
Cruising				
Level turns				
Final approach	1			
Hovering				
Landing				-
Autorotation - glide	- 			
Autorotation -	-			
Other (specify	<u> </u>			
(opo 0 11)				
Maximum perfor ance takeoff				
Landing in a confined are	ea			

9.	Please indicate the external reference, or references, you utilize	ın
	flying a helicopter during each of the maneuvers indicated below.	(If
	a portion of the cockpit is referenced to some external reference,	
	please indicate this fact.)	

- A. Cruising
- B. Normal takeoff
- C. Maximum performance takeoff
- D. Normal approach
- E. Steep approach
- F. Touch down to a point
- G. Autorotation