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# **AN ANALYSIS OF THE CANDLEPOWER DISTRIBUTION REQUIREMENTS OF RUNWAY LIGHTS**

By  
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Airport Division

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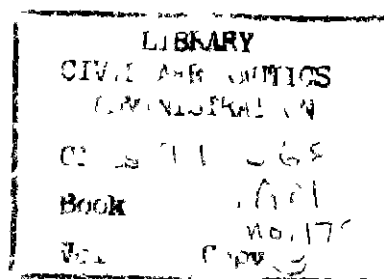
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## AN ANALYSIS OF THE CANDLEPOWER DISTRIBUTION REQUIREMENTS OF RUNWAY LIGHTS

### SUMMARY

This report discusses the runway light candlepower distribution based on the functional requirements for adequate guidance of various types of commercial aircraft during circling, approaching, landing, taxiing, and take-off operations conducted in accordance with Civil Air Regulations.

Individual aircraft characteristics were considered in establishing positional limits between which different types of aircraft might be expected to operate during various maneuvers. After consultation with a number of pilots, assumptions were made concerning which lights a pilot should be able to see for guidance during each operation. The directions and distances between the lights and the successive aircraft positions were then calculated and tabulated.

Critical atmospheric transmissivities were assumed in conformity with existing regulations governing operations under various weather minimums, and the corresponding candlepower requirements were then determined. These candlepower requirements were then evaluated in order to determine their practicability and in order to determine whether intensities needed for some airplane positions in space would cause glare at other positions.

Finally, acceptable candlepower (cp) values were plotted as isocandle curves which can serve as a basis for performance specifications for improved runway lights.

### INTRODUCTION

The need for runway lights to outline runways as an aid to air navigation was recognized many years ago. Many types of lights embodying the ideas of different designers have been developed. Those in the high-intensity category are generally classified on the basis of their candlepower distribution as wide beam, narrow beam, or movable beam units, and considerable uncertainty and confusion exist concerning their relative effectiveness and proper application.

The essential function of the runway lights is to mark the position and limits of the runway. In order to be useful these lights must be made visible to the prospective user at various stages of landing and take-off operations. It would appear then that a logical approach to the problem of runway lighting would be to study aircraft operations

with a view to establishing approximately the paths described by aircraft during various maneuvers and to calculate the candlepower distribution required, in order that pilots can see the necessary lights from aircraft following those paths.

Since operations and procedures are more or less standard, being governed by regulations as well as by aircraft characteristics and limitations, it is not an impossible task to determine the general direction and distance from any runway light to an airplane during the course of any specific operational procedure. The required candlepower of the light in the direction of the pilot can then be determined mathematically.

In this study use was made of the characteristics of typical commercial airplanes whose maneuvering paths lie farther out from the runway, whose turning radii are greater, and whose demands on the lights are therefore more critical than those of smaller aircraft. Any lights designed for commercial airplanes should be satisfactory for most of their military counterparts and for smaller aircraft.

The authors wish to emphasize that this analysis has involved consideration of certain intangible factors which cannot be defined exactly. In dealing with such factors an attempt was made to arrive at reasonable assumptions through discussion with qualified persons. The study is intended to serve as a guide in the development and design of new lighting units.

### AIRPLANE PATHS DURING APPROACH, LANDING, AND TAKE-OFF OPERATIONS

Runway lights are used during approaches and landings to provide visual guidance to a pilot for a period which begins after he has started his approach and ends as he leaves the runway after completing his landing. The area within which guidance is required depends largely upon the type of approach he is making. The runway light intensity required depends upon the distance from which the lights must be seen and upon the visibility condition prevailing. Runway lights are also used during take-offs while the airplane is on or above the runway.

The types of approaches that may be made under certain limiting visibility conditions are governed by Civil Air Regulations. Minimum operating conditions vary somewhat at different airports chiefly because of

TABLE I

## MINIMUMS GOVERNING APPROACHES AND TAKE-OFFS AT INDIANAPOLIS INDIANA\*

Class of Operator	Type of Aircraft	Landing Minimums Day or Night				Take-Off Minimums Day or Night			Alternate Landing Minimums
		Circling Approach**		Straight-in Approach**		Radio Range or Automatic Direction Finder	ILS		
		Radio Range or Automatic Direction Finder	ILS	Radio Range (Runway 9) or Automatic Direction Finder (Runway 4)	ILS Runway 4		Runways 4 or 22	Other Runways	
Scheduled Air Carrier	DC-3 or Smaller	400-1***	400-1	400-1	200-1/2	300-1	200-1/2	300-1	800-2
	CV-240 M-202	500-1 1/2	500-1 1/2	400-1	200-1/2	300-1	200-1/2	300-1	800-2
	DC-6 DC-4 L-49	500-1 1/2	500-1 1/2	400-1	200-1/2	Runways 4 or 22 200-1/2 Others 300-1	200-1/2	300-1	800-2
All Others Except Military and Users to Whom the CAA has Specifically Authorized Lower Minimums	Aircraft Having Stall Speeds Above 75 mph	500-1 1/2	500-1 1/2	500-1	400-3/4	300-1	300-3/4	300-1	1000-3
	Aircraft Having Stall Speeds Below 75 mph	400-1	400-1	500-1	400-3/4	300-1	300-3/4	300-1	1000-3

## NOTES ON TABLE I

\*While minimums in this chart apply specifically to operations at Weir Cook Municipal Airport Indianapolis Indiana they are applicable to most other airports where the same terrain and obstruction conditions prevail

\*\*Circling approach minimums (known as regular landing minimums) are authorized when it is necessary to circle or maneuver the aircraft during an approach for landing. Straight-in landing minimums are authorized only when landing can be accomplished straight in from the navigational facility being used to the near end of the runway without exceeding 500 feet per minute rate of descent and without change of direction of more than 30°

\*\*\*In columns 3 to 10, numbers indicated thus 400-1 means a ceiling of 400 feet and visibility of 1 mile

Sources of information CAA Flight Information Manual, Vol 5 No 2 June 1951

differences in obstructions. Representative minimums are summarized in Table I which shows those in effect at Weir Cook Airport, Indianapolis, Indiana.

It can be seen from Table I that circling approach minimums applicable to commercial aircraft generally vary from 400-and-1 to 500-and-1 1/2. This means from a 400-foot ceiling and 1-mile visibility to a 500-foot ceiling and 1 1/2-mile visibility. Daylight visibilities are reported on the basis of the observed visual range for large solid objects. Night visibilities are based on observations of a 25-candlepower lamp. The most commonly applied minimums for straight-in ILS approaches and landings are 300-and-3/4, although at one specially equipped airport qualified commercial pilots

have been permitted to make instrument approaches and landings with minimums of 100-and-1/4. Straight-in approaches from a radio facility are generally permitted at 400-and-1. Both circling and straight-in approaches are also made under unrestricted visibility conditions. Take-off minimums vary generally from 200-and-1/2 to 300-and-1.

The several types of approaches mentioned in the regulations do not make identical demands upon the lights. For the circling approach, which may be started from any direction, guidance over a wide area is required from the lights. For the straight-in approach from a radio facility, the horizontal path limits are narrowed since straight-in approaches are permitted

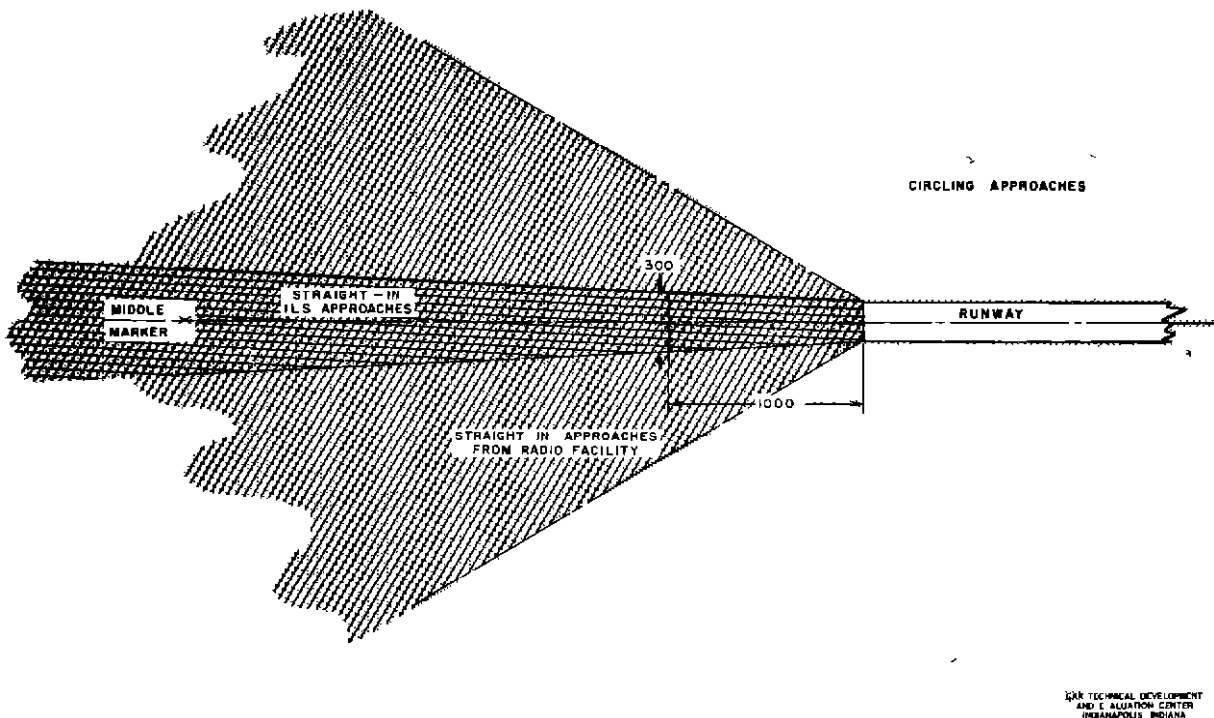


Fig. 1 Plan View of Runway and Approach Showing Directional Guidance Areas Required from Runway Marker Lights

only when the angular difference between the runway direction and a course from the navigation aid to the approach end of the runway is  $30^\circ$  or less. For the straight-in ILS approach the limits are further narrowed to a region relatively close to the localizer course which the pilot is following in his let-down. For this type of approach there are approach lights to guide him before he reaches the area where the limited coverage of the runway lights will be visible. The directional guidance areas to be served by runway lights are shown in Fig. 1.

#### The Circling Approach

In order to determine distances through which the light must travel to reach the pilot's eye, consideration must be given to the circling procedure before establishing the limits of the possible paths of an approaching airplane. This procedure is required when a pilot first establishes visual contact with the airport from a point not suitable for his final approach. It may also be necessitated by traffic conditions.

During the down-wind leg of the ap-

proach, the altitude of the airplane will be limited by two factors: (1) the minimum clearance that must be maintained above the tallest obstructions in the airport vicinity, and (2) the maximum height from which a pilot can descend in comfort and safety. At most airports the minimum height above ground for circling is established at 500 feet. A let-down rate of 500 feet per minute may be considered a maximum for aircraft not pressurized, although this rate could be increased in an emergency. On this basis, a height of 2,000 feet may be considered a maximum for normal approaches and might be increased to 4,000 feet for emergency conditions.

From a series of actual flight measurements conducted by TDEC engineers, it had been previously determined that during the circling approach the maneuvering speed was approximately  $16/9 V_s$ , where  $V_s$  = stall speed. For the several types of airplanes listed, the values would be approximately those shown in Table II. There are many variables which affect these speeds, but for the purposes of this determination the values

TABLE II  
SPEED AND VISIBILITY CHARACTERISTICS OF VARIOUS AIRLINERS

Aircraft		Characteristic Speed* (mph)		Typical Circling Approach (15° Bank)		Minimum Forward Visibility on Ground** (Feet)	
Manufacturer	Model	Approach	Stall	Selected Speed (mph)	Corresponding Turn Radius (Feet)	Landing	Taxing
Boeing	377	150	100	175	7600	146	73
Consolidated-Vultee	Convair 240	115 - 130	91	160	6400	62	59
Douglas	DC-6	125 - 135	89	160	6400	104	98
Douglas	DC-4	120 - 135	83	150	5600	152	98
Douglas	DC-3	95 - 110	69	120	3600	104	331
Lockheed	Constellation	120 - 135	88	160	6400	155	123
Martin	202	115 - 130	83	150	5600	69	66

\*These figures were supplied by the respective manufacturers. They will vary under different conditions of altitude, loading, winds, and reserve fuel carried. However, for lighting requirements and estimations, these values will be satisfactory.

\*\*Some of these figures were obtained from manufacturers and the others from actual measurement at this Center.

given are sufficiently accurate.

At the end of the down-wind leg, the pilot makes a turn of approximately 180°. This turn is so executed that at its completion the airplane will be at a height of approximately 500 feet above the runway level and in position to begin a final approach from a point approximately 3,500 feet distant from the runway end. During the down-wind leg the pilot maintains the necessary horizontal distance from the runway to permit him to execute the turn properly. This distance is about twice the radius of turn, and is dependent upon the degree of bank and the maneuvering speed. These factors are interrelated as indicated by the equation

$$R = \frac{V^2}{g \tan \phi} \quad (1)$$

where

R = radius of turn (feet)

V = maneuvering speed prior to landing (feet per second)

g = acceleration of gravity (32.2 feet per second per second)

φ = degree of bank. For turns made by passenger-carrying aircraft, 15° is a normal and comfortable bank.

Because of their greater speeds the large airplanes require a larger turning radius, and pilots normally execute the down-wind legs at a greater distance from the runway

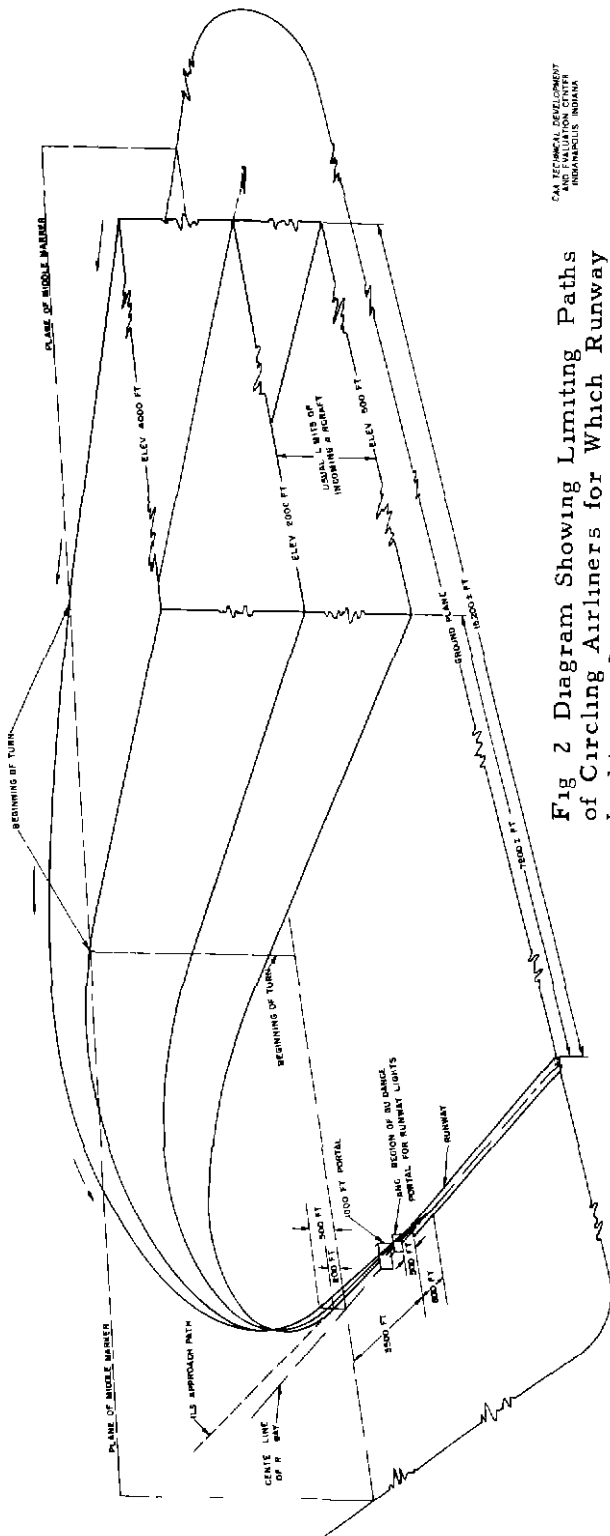
For this reason, Civil Air Regulations require higher minimum visibility conditions for large aircraft.

From the data shown in Table II and the application of Equation (1), the theoretical limits of circling paths shown in Fig. 2 were determined. Figs. 3A and 3B show some typical circling approaches plotted from actual field observations under good visibility conditions. These are preliminary results from a collision course study now being conducted at this Center as one phase of an investigation concerning the development of cockpit visibility standards. The results of this investigation will be published in a subsequent report. While there is considerable variation between paths for the same model of airplane, the paths shown are generally consistent with the theoretical computations.

#### ASSUMPTIONS AND CALCULATIONS

Before an optimum candlepower distribution for runway lights can be determined it is necessary to arrive at a practical assumption concerning the number of feet of runway a pilot should be able to see from various points along his path. It is understandable that most pilots might prefer to be able to see all the runway, or all the lights outlining the runway, at all times after starting an approach. Such a requirement would not be realistic, however, because it would involve excessive candlepower for low visibility conditions and would produce glare.





**Fig 2 Diagram Showing Limiting Paths of Circling Airliners for Which Runway Lighting is Required to Provide Guidance**

In arriving at a tentative decision on this point, discussions were held with a group of pilots who were asked to state their opinions about the minimum length of runway necessary to be seen during each part of the approach. The consensus is listed in Table III. The pilots stated that for all operations it is desirable to see the lights on both sides of the runway in order that the runway may be more readily recognized.

Having assumed which lights it is desirable for a pilot to see in order to keep himself properly oriented during the different operations, the next step was to calculate the distances and angles from these lights to the various airplane positions. In making the calculations it was assumed that the glide angle is  $2.65^\circ$ , the touchdown point is 800 feet from the runway end, and the runway width (when necessary to be considered) is 200 feet.

After the calculations of angle and distance were made, the candlepower values required to make the lights visible from the maneuvering path were read from nomographs

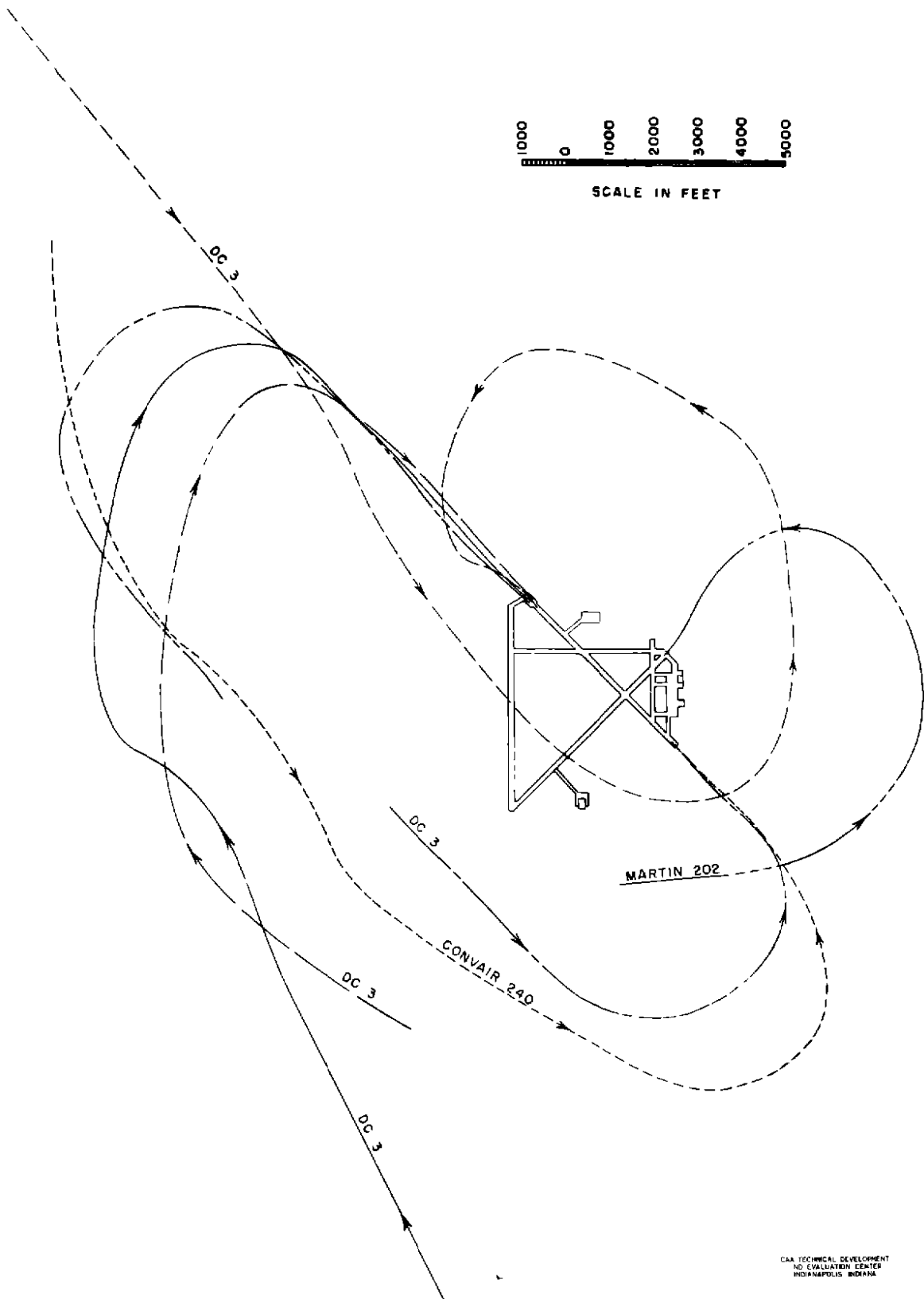


Fig 3A Plan View of Circling Approach Patterns Followed by Various Airliners

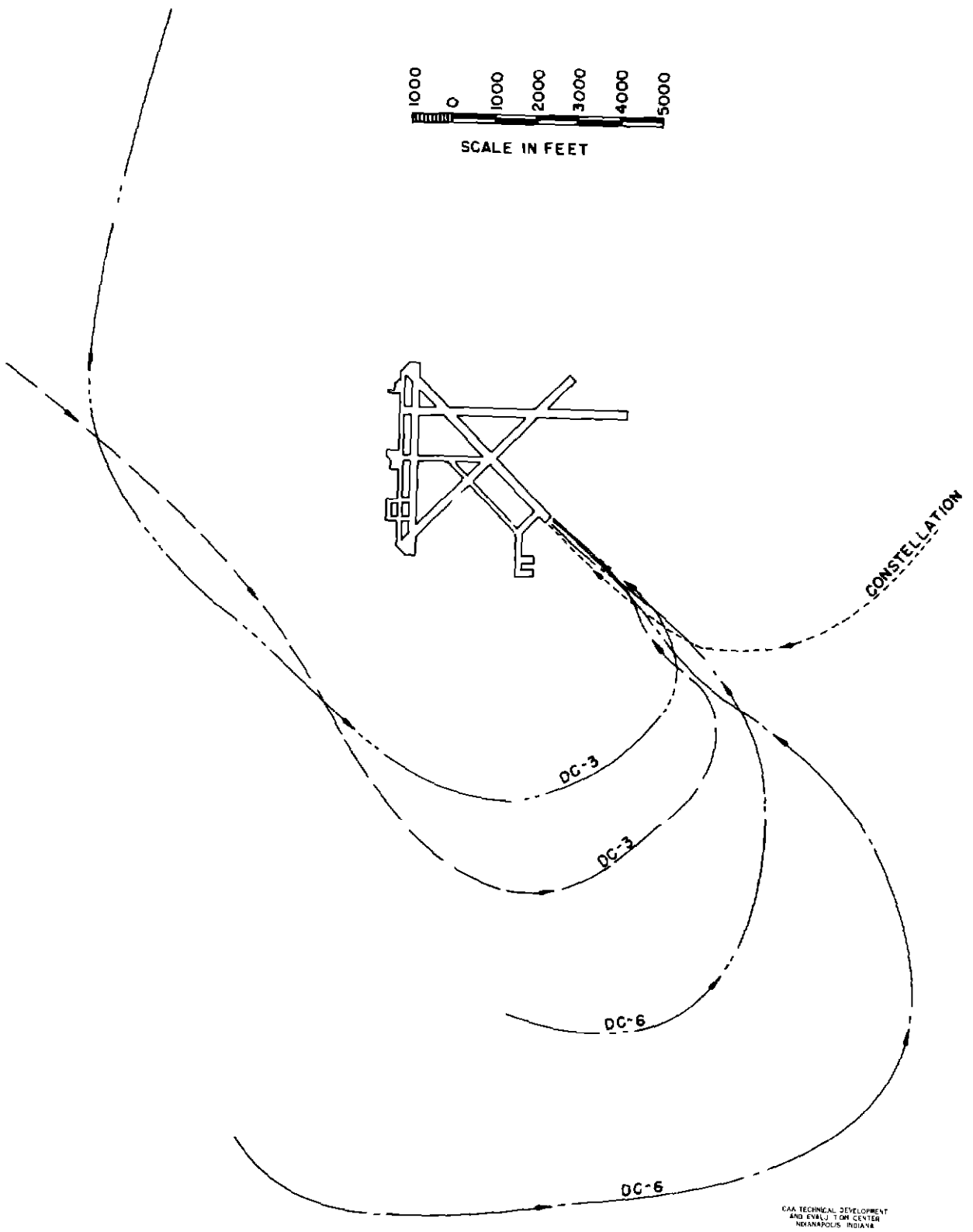


Fig 3B Plan View of Circling Approach Patterns Followed by Various Airliners

TABLE III  
LENGTH OF RUNWAY DESIRABLE FOR PILOTS TO SEE  
DURING AIRPLANE OPERATIONS UNDER RESTRICTED VISIBILITY

Visibility	Operation	Position of Airplane	Height Above Runway Level (Feet)	Minimum Length of Runway to be Seen (Feet)
Down to 1 Mile (Night)	Circling on Path as Near as Possible to Runway	On Down-Wind Leg	500 to 4000	5000
		At 90 Point on 180 Turn	350 to 2000	1000
	Crossing Runway	Above Runway	500 to 4000	3000 In Either Direction
		1/4-Mile From Either Side of Runway		
		1/2-Mile From Either Side of Runway		
	Straight-in Approach From Radio Facility	8000 Feet to 10000 Feet Beyond Runway End	400 to 525	200
		3500 Feet Beyond Runway End	200 to 600	Entire Runway (5000-8000)
Down to 1/4 or 1/2-Mile (Day or Night)	Straight-in ILS Approach	3500 Feet Out - Not More than 300 Feet Off Course	On Glide Path Above Glide Path Below Glide Path	1000
		1000 Feet Out - Not More than 150 Feet Off Course	On Glide Path Above Glide Path	1000
		500 Feet Out - Not More than 125 Feet Off Course	On Glide Path Above Glide Path	1000
Down to 1/4-Mile (Day or Night)	Touchdown Take-Off		On Runway	1000
	Taxiing			500

published previously<sup>1</sup> These values, listed in Tables IVa and IVb, were used to plot isocandle curves The isocandle curves shown in Fig 4 cover the requirements for circling approaches Figs 5 and 6 show the candlepower requirements for straight-in approaches under selected conditions

Intensities which are only high enough for threshold visibility from one point in the maneuvering space may be glaring to the same pilot or to other pilots when viewed from some other point in the same direction but at a shorter distance For example, a light which was barely visible to the pilot on the down-wind leg of his approach might be glaring during the landing roll While it is possible to vary the intensity of lights to some extent for varying weather conditions, it is not practicable to vary it for different stages of an approach, because several pilots at different points may be using the lights simultaneously

<sup>1</sup>M S Gilbert and H J Cory Pearson, "A Study of the Visibility and Glare Ranges of Slope-Line Approach Lights," CAA Technical Development Report No 150, November 1950, pp 6-7

Calculations of glare ranges for various values of light intensity were based on the assumption that glare can be produced by candlepower values approximately 1,000 times those producing threshold illumination at a given distance Glare diagrams shown in Figs 7 and 8 are based on intensities required for seeing 1,000 feet of runway from the middle marker and from the 1,000-foot portal during a 1/4-mile night visibility condition The glare diagrams in Figs 9 and 10 are based on intensities required for seeing 1,000 feet of runway from the 1,000-foot portal during 1/4- and 1/2-mile day visibility conditions As a result of field tests, all candlepower values of 200 or less were considered to be non-glaring for short-range viewing regardless of the visibility condition

## DISCUSSION

A study of the tabulated candlepower data, isocandle curves, and glare diagrams indicates that the candlepower values required to satisfy the various conditions outlined in Table III are within practical limits

<sup>2</sup>Ibid, pp 5 and 8

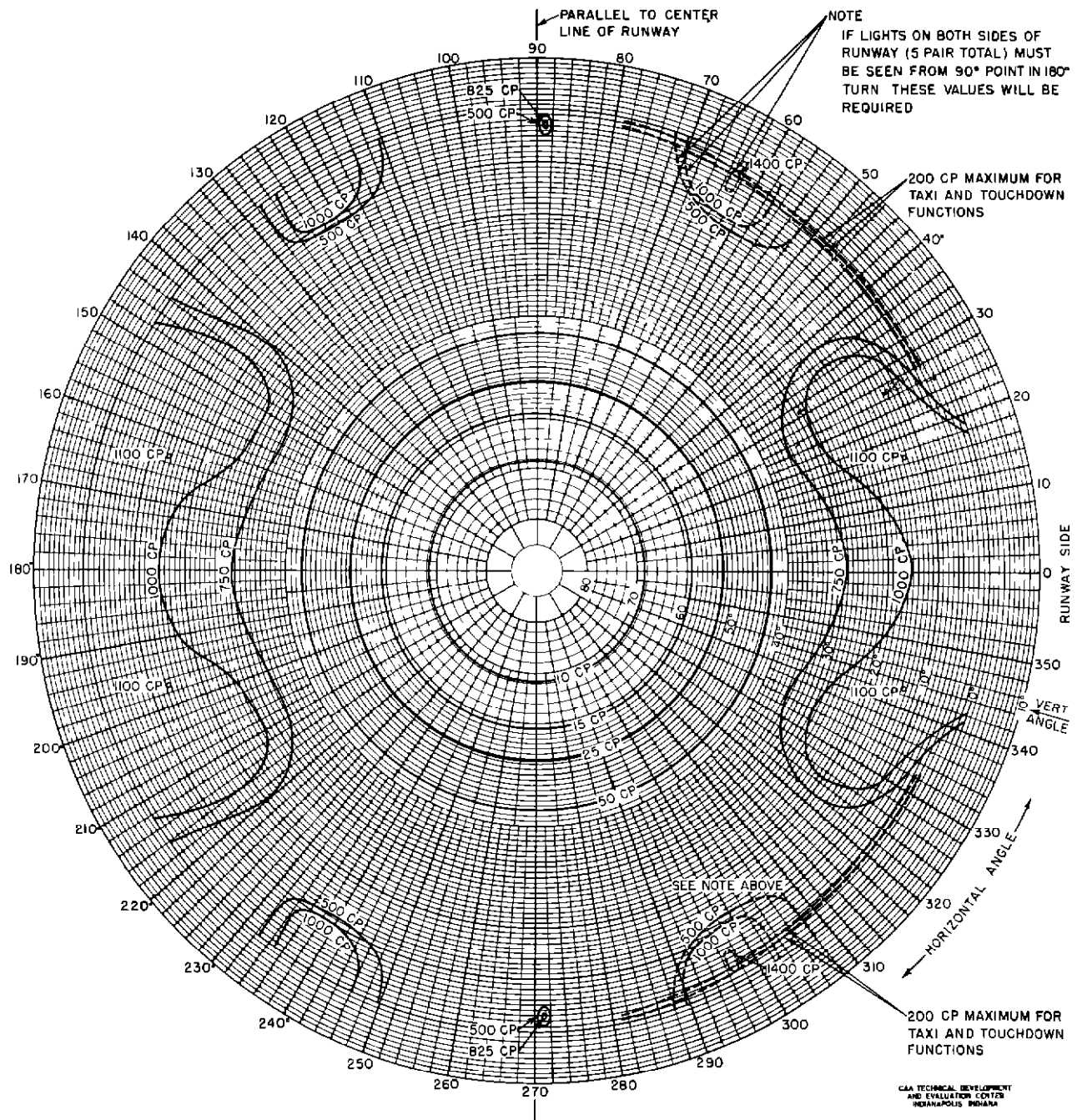


Fig 4 Isocandle Curves Showing Minimum Candlepower Requirements of Runway Lights for Night Operations Under Circling Minimums

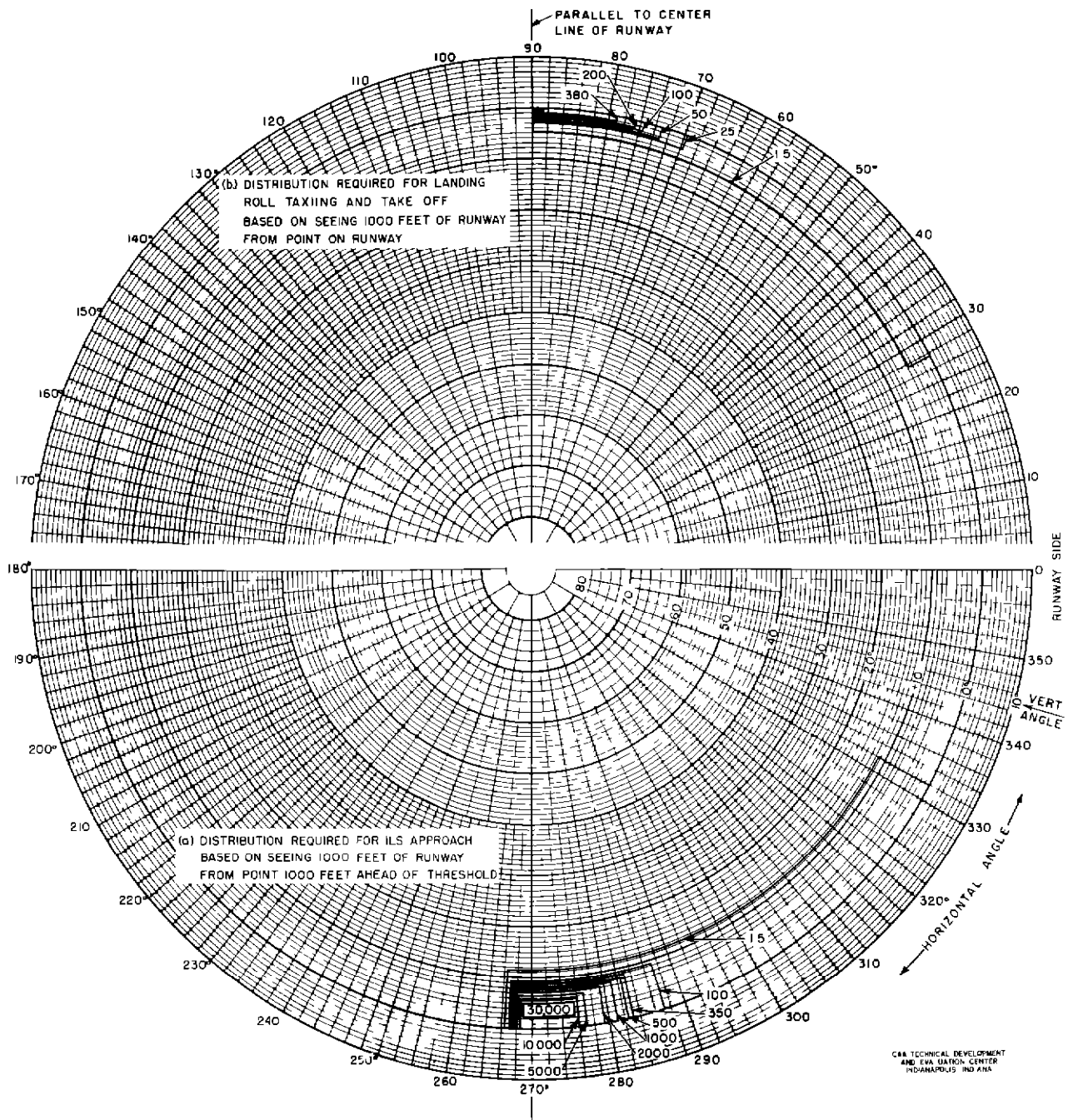


Fig 5 Isocandle Curves Showing Minimum Candlepower Requirements for Day Operations During 1/4-Mile Visibility Condition

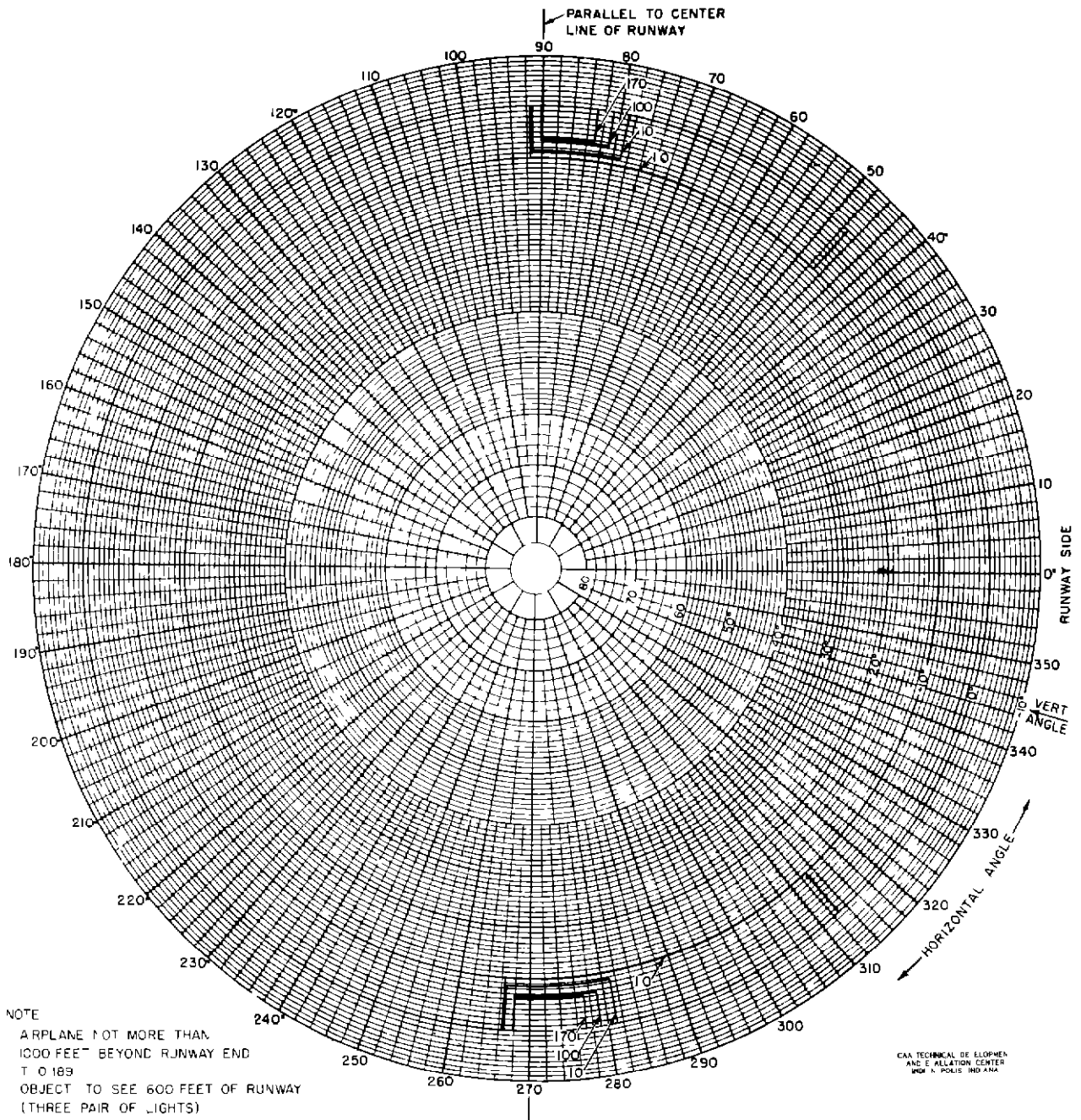


Fig 6 Isocandle Curves Showing Minimum Candlepower Requirements for Night Operations During 1/4-Mile Visibility Condition

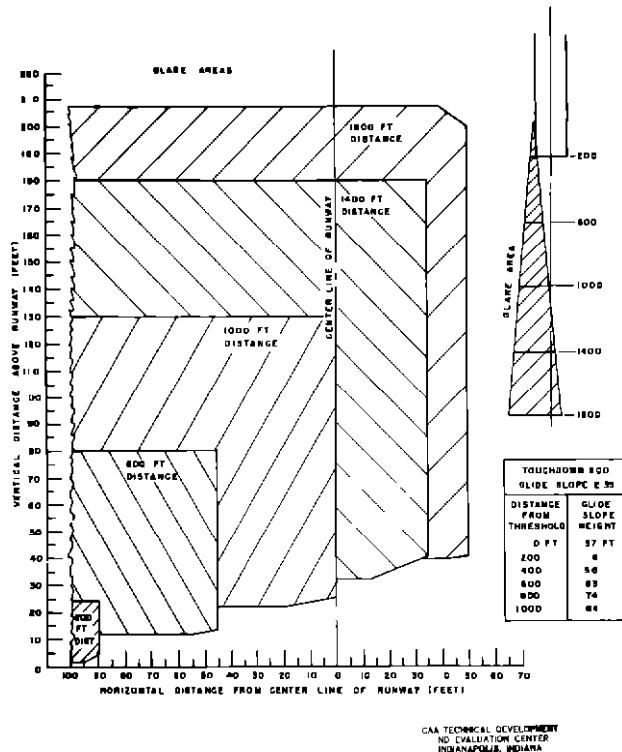


Fig 7 Outlines of Glare Area, at Designated Distances from Airplane, of a Runway Light Based on Candlepower Required to See 1,000 Feet of Runway from Middle Marker  
Visibility 1/4-Mile, Night (25 cp)

with the following exceptions

a To make 8,000 feet of runway visible from the middle marker (3,500 feet out from the runway end) for night visibilities below 1 1/2 miles would require impractical candlepower values which would also produce glare areas

b To make 1,000 feet of runway visible from the middle marker during 1/2-mile day visibility would also require excessively high candlepower values To make the same length of runway visible from the same point during 1/2-mile night visibility would involve glare at other positions, although the candlepowers would be relatively low and easy to obtain

c To make 1,000 feet of runway visible from a point 1,000 feet beyond the threshold during 1/4-mile night visibility would also result in glare Reducing to 600 feet the length of runway to be seen would correct this condition

The requirement that the lights on both sides of the runway should be visible from a given point gives rise to some points of conflict The difficulty is particularly ap-

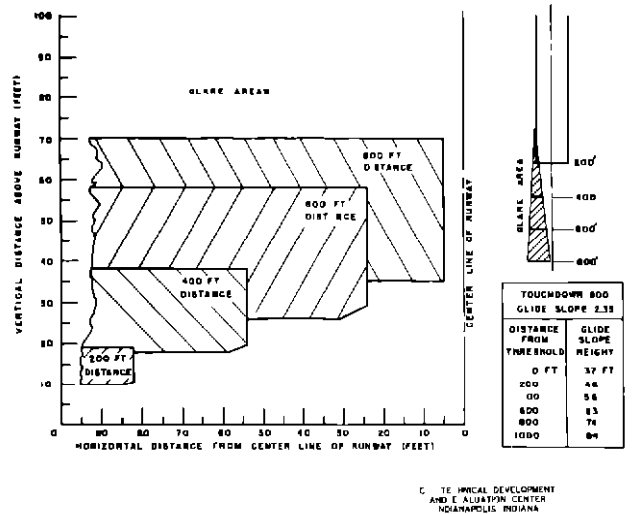


Fig 8 Outlines of Glare Area, at Designated Distances from Airplane, of a Runway Light Based on Candlepower Required to See 1,000 Feet of Runway from the 1,000-Foot Portal.  
Visibility 1/4-Mile, Night (25 cp)

parent in the case of the candlepower distribution required by a pilot on the downwind leg of a circling approach These values are much too high for a taxiing pilot who sees the lights at the same angles If the requirement were modified to make only the lights on the near side of the runway visible to the pilot on the downwind leg, this difficulty would be eliminated

In order to cover various situations and visibility conditions, it appears that several candlepower distribution patterns will be required from the runway lights A distribution throughout a 360° horizontal range and with no high intensity beams, as shown in Fig 4, will be needed on all runways during night visibility conditions of one mile or more in order to provide guidance during (1) circling approaches, (2) approaches from a radio facility, (3) deceleration after touchdown, (4) taxiing, and (5) take-off

For straight-in ILS approaches there will be required from lights in the touchdown area a different pattern consisting of a high intensity beam through relatively narrow limits In order to maintain a margin of safety for a landing airplane which might be somewhat off course prior to crossing the threshold, the beam spread should be sufficient to cover a portal 300 feet wide at a distance 1,000 feet out from the runway end The isocandle diagram Fig 5a, illustrates the minimum candlepower needed for the 1/4-mile visibility condition, the most restricted condition considered in this study



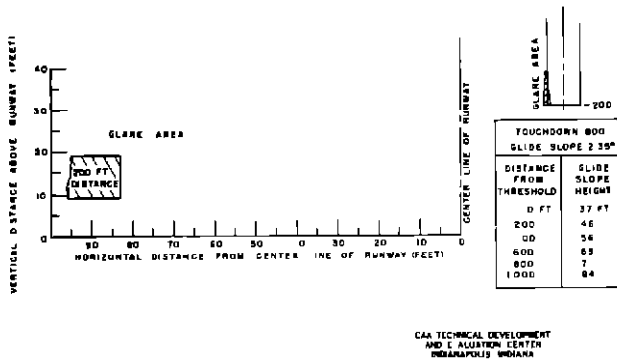


Fig 9 Outlines of Glare Area, at Designated Distances from Airplane, of a Runway Light Based on Candlepower Required to See 1,000 Feet of Runway from the 1,000-Foot Portal  
Visibility 1/4-Mile, Day

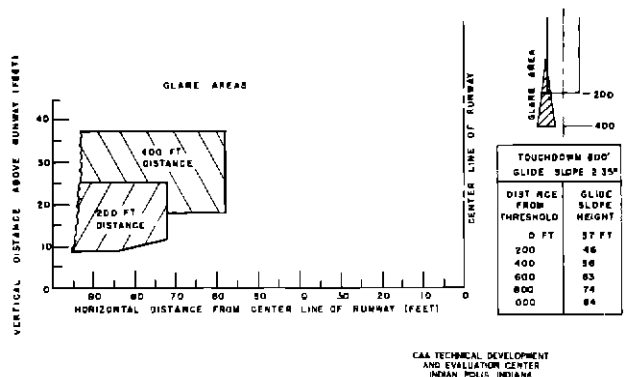


Fig 10 Outlines of Glare Area, at Designated Distances from Airplane, of a Runway Light Based on Candlepower Required to See 1,000 Feet of Runway from the 1,000-Foot Portal.  
Visibility 1/2-Mile, Day.

This high intensity beam of 30,000 cp is based on the assumption of a need for seeing 1,000 feet of runway from a point on the 1,000-foot portal

A narrower beam of lower intensity will suffice for the landing roll, taxiing, and take-off during visibility conditions below one mile. See Fig 5b. The peak of 380 cp in this low intensity beam is based on a requirement for seeing only the 1,000 feet of runway immediately ahead of the pilot after the airplane touches the runway. Should it be considered necessary to extend this latter range of visibility, higher beam candlepower values obviously will be needed. The maximum may be as high as 30,000 cp should it be necessary for a pilot to see 2,000 feet of runway from the touchdown point.

In the event that it is undesirable or impractical to design a high-intensity lighting unit of sufficient beam width for use in the touchdown zone, narrow beam units with the same maximum candlepower can be used in groups of two or three along the touchdown zones so that the total width of the combined beams will be sufficient to cover the portal.

The intensity requirements will be considerably lower for operations at night, when the background brightness is low. See Fig 6. Lights should be designed on the basis of the intensities required for the most exacting combination of transmissivity and background brightness expected to occur at representative airports. Lower intensities for less exacting conditions may be obtained through intensity control, either manual or automatic. Since intensities up to 200 cp can be tolerated from the glare standpoint under the best night visibility conditions, it is not

necessary to use values less than this even though minimum requirements indicate use of lower intensities.

It may be possible to develop a single type of runway light to combine the distributions required for all or several functions. In any event, there is little doubt that combination or multiple units can be used to achieve the desired distributions. Combining units to meet the candlepower requirements may also help in mitigating glare under certain conditions through use of the most suitable unit of the combination for any specific visibility condition.

## CONCLUSIONS

1 On the basis of the assumptions set forth in this report the theoretical candlepower distributions needed to meet various requirements and conditions have been established. The following conditions cannot be met with reasonable candlepower requirements without producing glare:

a To see 8,000 feet of runway from the middle marker in visibility conditions less than the 1 1/2-mile night condition.

b To see 1,000 feet of runway from the middle marker during the 1/2-mile day or night condition.

c To see 1,000 feet of runway from the 1,000-foot portal during 1/4-mile night visibility. To see 600 feet of runway without glare would be theoretically possible.

2 It is not practical to see lights along both sides of the runway during the downwind leg of a circling approach under some visibility conditions, because this will introduce

glare to pilots of other aircraft already on the runway

3 Three types of candlepower distribution are required for runway lights. The first will cover the requirements for circling and other approaches under visibility conditions down to one mile by day or night. The second will be required for the touchdown zone for straight-in ILS approaches. The third will be required for the runway beyond the touch-

down zone when the visibility is below one mile. It may be necessary to use multiple fixtures in order to meet all conditions.

4 Since the distributions developed in this report are based entirely on theoretical analysis, it is recommended that they be given adequate service testing before they are used as a basis for specification requirements.

TABLE IVa

MINIMUM CANDLEPOWER REQUIREMENTS OF RUNWAY LIGHTS FOR NIGHT OPERATIONS UNDER CIRCLING MINIMUMS

Aircraft Position Requirement    Circling on Downwind Leg To See 2500 Feet of Runway From the Path Indicated						
Path - Horizontal Distance Between Runway Edge and Down-Wind Leg (Feet)	Altitude (Feet)	Vertical Angle (Degrees)	Horizontal Angle		Candlepower	
			Runway Side (Degrees)	Off Runway Side (Degrees)	1 1/2-Mile Visibility T = 0.881	1-Mile Visibility T = 0.7854
15200	500	1.9	0	180	1035	
	2000	7.5	0	180	1050	
	4000	14.7	0	180	1075	
	500	1.9	8.6    351.4	171.4    188.6	1055	
	2000	7.5	8.6    351.4	171.4    188.6	1055	
	4000	14.6	8.6    351.4	171.4    188.6	1075	
12000	500	2.39	0	180	250	
	2000	9.46	0	180	250	
	4000	18.4	0	180	375	
	500	2.35	10.85    349.15	169.15    190.85	250	
	2000	9.29	10.85    349.15	169.15    190.85	260	
	4000	18.2	10.85    349.15	169.15    190.85	400	
7200	500	3.99	0	180		280
	2000	15.54	0	180		300
	4000	29.1	0	180		750
	500	3.79	17.75    342.25	162.25    197.75		325
	2000	14.84	17.75    342.25	162.25    197.75		400
	4000	27.9	17.75    342.25	162.25    197.75		925
Requirement    To See an Additional 2500 Feet of Runway - (5000 Feet Total)						
15200	500	1.8	16.8    343.2	163.2    196.8	1080	
15200	2000	7.2	16.8    343.2	163.2    196.8	1090	
15200	4000	14.1	16.8    343.2	163.2    196.8	1100	
12000	500	2.24	20.94    339.06	159.06    200.94	375	
12000	2000	8.81	20.94    339.06	159.06    200.94	420	
12000	4000	17.3	20.94    339.06	159.06    200.94	760	
7190	500	3.35	32.61    327.39	147.39    212.61		1000
7190	2000	13.20	32.61    327.39	147.39    212.61		1030
7190	4000	25.1	32.61    327.39	147.39    212.61		1050

T = Transmissivity

TABLE IVa (Continued)

MINIMUM CANDLEPOWER REQUIREMENTS OF RUNWAY LIGHTS  
FOR NIGHT OPERATIONS UNDER CIRCLING MINIMUMS

Aircraft Position At 90 Point on 180 Turn Requirement To See Pair of Lights Listed							
Path	Altitude (Feet)	Threshold To Pair Of Lights (Feet)	Vertical Angle (Degrees)	Horizontal Angle		Candlepower	
				Runway Side (Degrees)	Off Runway Side (Degrees)	1 1/2-Mile Visibility T = 0 881	1-Mile Visibility T = 0 7854
15200	350	200	1 5		123 8 236 2	925	
	1000		4 2		123 8 236 2	925	
	2000		8 4		123 8 236 2	930	
	350	400	1 5		123 3 236 7	930	
	1000		4 2		123 3 236 7	950	
	2000		8 3		123 3 236 7	1000	
	350	800	1 4		122 5 237 5	1050	
	1000		4 1		122 5 237 5	1050	
	2000		8 1		122 5 237 5	1100	
	350	200	1 5	55 5 304 5		930	
	1000		4 2	55 5 304 5		930	
	2000		8 4	55 5 304 5		930	
	350	400	1 5	56 0 304 0		935	
	1000		4 2	56 0 304 0		935	
	2000		8 3	56 0 304 0		935	
	350	800	1 4	56 9 303 1		1050	
	1000		4 1	56 9 303 1		1050	
	2000		8 1	56 9 303 1		1050	
12000	350	200	1 6		128 7 231 3	330	
	1000		4 7		128 7 231 3	335	
	350	400	1 6		128 1 231 9	375	
	1000		4 6		128 1 231 9	380	
	350	800	1 6		126 9 233 1	560	
	1000		4 5		126 9 233 1	570	
	350	200	1 6	50 6 309 4		375	
	1000		4 6	50 6 309 4		380	
	350	400	1 6	51 2 308 8		390	
	1000		4 6	51 2 308 8		400	
	350	800	1 6	52 3 307 7		580	
	1000		4 5	52 3 307 7		600	
7200	350	200	2 5		117 0 243 0		700
	1000		7 1		117 0 243 0		700
	350	400	2 4		116 4 243 6		920
	1000		6 9		116 4 243 6		925
	350	800	2 3		115 2 244 8		1225
	1000		6 7		115 2 244 8		1250
	350	200	2 5	61 7 298 3			700
	1000		7 1	61 7 298 3			700
	350	400	2 4	62 3 297 7			925
	1000		6 9	62 3 297 7			930
	350	800	2 3	63 6 296 4			1400
	1000		6 6	63 6 296 4			1400

TABLE IVa (Continued)

**MINIMUM CANDLEPOWER REQUIREMENTS OF RUNWAY LIGHTS  
FOR NIGHT OPERATIONS UNDER CIRCLING MINIMUMS**

		Aircraft Position Requirement		Crossing Runway To See 3000 Feet of Runway in Either Direction from Various Altitudes		
Position of Aircraft with Respect to Runway	Altitude (Feet)	Vertical Angle (Degrees)	Horizontal Angle		Candlepower	
			Runway Side (Degrees)	Off Runway Side (Degrees)	1 1/2-Mile Visibility T = 0.681	1-Mile Visibility T = 0.7854
Over	500	78.7	0		1.0	1.0
	2000	87.1	0		1.0	1.0
	4000	88.6	0		1.25	6.5
	500	78.6	84.3		1.0	1.0
	2000	87.1	84.3		1.0	1.0
	4000	88.6	84.3		1.25	6.5
	500	14.0	87.1		1.0	1.0
	2000	45.0	87.1		1.0	1.25
	4000	63.4	87.1		3.0	10.0
	500	9.5	88.1		1.0	1.5
	2000	33.7	88.1		1.0	3.5
	4000	53.1	88.1		4.0	23.0
1/4-Mile Distant	500	20.8		180	1.0	1.0
	2000	56.6		180	1.0	1.0
	4000	71.7		180	2.0	9.0
	500	18.2	0		1.0	1.0
	2000	52.8	0		1.0	1.0
	4000	69.2	0		2.0	10.0
	500	16.8		142.9 217.1	1.0	1.0
	2000	50.4		142.9 217.1	1.0	1.0
	4000	67.5		142.9 217.1	2.0	10.0
	500	15.4	33.3 326.7		1.0	1.0
	2000	47.7	33.3 326.7		1.0	1.0
	4000	64.5	33.3 326.7		2.0	10.0
	500	11.8		123.4 236.6	1.0	1.0
	2000	39.8		123.4 236.6	1.0	2.0
	4000	59.1		123.4 236.6	2.5	15.0
	500	11.3	52.8 307.2		1.0	1.0
	2000	38.5	52.8 307.2		1.0	2.0
	4000	57.8	52.8 307.2		2.5	15.0
	500	8.6		113.7 246.3	1.0	2.5
	2000	31.5		113.7 246.3	1.5	5.0
	4000	50.8		113.7 246.3	5.0	30.0
	500	8.5	63.1 296.9		1.0	3.0
	2000	30.8	63.1 296.9		1.5	5.0
	4000	50.0	63.1 296.9		5.0	30.0
1/2-Mile Distant	500	10.7		180	1.0	1.0
	2000	37.1		180	1.0	2.5
	4000	56.5		180	2.5	15.0
	500	10.0	0		1.0	1.0
	2000	35.2	0		1.0	3.0
	4000	54.7	0		3.0	17.0
	500	10.1		159.3 200.7	1.0	1.5
	2000	35.3		159.3 200.7	1.0	3.0
	4000	54.8		159.3 200.7	3.0	20.0
	500	9.4	19.4 340.6		1.0	2.0
	2000	33.6	19.4 340.6		1.0	4.0
	4000	53.0	19.4 340.6		3.5	25.0
	500	8.6		142.9 217.1	1.0	2.7
	2000	31.2		142.9 217.1	1.0	5.0
	4000	50.4		142.9 217.1	4.5	35.0
	500	8.2	35.2 324.8		1.0	3.0
	2000	29.9	35.2 324.8		1.5	6.5
	4000	49.0	35.2 324.8		5.0	35.0
	500	7.1		131.3 228.7	2.0	7.0
	2000	26.6		131.3 228.7	2.5	10.0
	4000	45.0		131.3 228.7	6.0	45.0

TABLE IVa (Continued)

**MINIMUM CANDLEPOWER REQUIREMENTS OF RUNWAY LIGHTS  
FOR NIGHT OPERATIONS UNDER CIRCLING MINIMUMS**

Aircraft Position On Glide Path Requirement To See First Pair of Lights						
Distance Out (Feet)	Altitude (Feet)	Vertical Angle (Degrees)	Horizontal Angle Runway Side (Degrees)		Candlepower	
					1 1/2-Mile Visibility T = 0 881	1-Mile Visibility T = 0 7854
8000	409	2 9	89 2	270 8	- -	825
10500	526	2 8	89 4	270 6	200	- -

Aircraft Position Over Middle Marker - 3500 Feet Out Requirement To See Pairs of Lights Listed						
200	200	3 1	88 45	271 55	1 0	4 0
	600	7 69	88 9	271 55	1 0	4 0
2400	200	1 9	88 9	271 1	6 5	75 0
	600	5 8	88 9	271 1	7 5	80 0
4600	200	1 4	89 2	270 8	40 0	600 0
	600	4 2	89 2	270 8	45 0	700 0
7800	200	1 0	89 4	270 6	275 0	8000 0
	600	3 0	89 4	270 6	280 0	8500 0

Aircraft Position At Touchdown (Pilot 20 Feet From Runway Edge) Requirement To See 1000 Feet of Runway						
200	16	3 32	46 47	313 53	1 0	1 0
400		2 07	64 60	295 40	1 0	1 0
600	(Convair)	1 45	72 43	287 57	1 0	1 0
800		1 17	76 62	283 38	1 0	1 0
1000		0 9	79 25	280 75	1 0	1 0
200	20 8	4 32	46 47	313 53	1 0	1 0
400		2 68	64 60	295 40	1 0	1 0
600	(Constellation)	1 88	72 43	287 57	1 0	1 0
800		1 45	76 62	283 38	1 0	1 0
1000		1 17	79 25	280 75	1 0	1 0

Aircraft Position Taxiing 20 Feet From Runway Edge Requirement To See 500 Feet of Runway						
100	12 5	3 33	27 77	332 23	1 0	1 0
300	(Convair)	2 02	57 67	302 33	1 0	1 0
500		1 33	69 20	290 80	1 0	1 0
100	16 2	4 32	27 77	33 23	1 0	1 0
300	(Constellation)	2 60	57 67	302 33	1 0	1 0
500		1 73	69 20	290 80	1 0	1 0

TABLE IVb

**MINIMUM CANDLEPOWER REQUIREMENTS OF RUNWAY LIGHTS  
FOR NIGHT AND DAY OPERATIONS UNDER ILS MINIMUMS**

Aircraft Position Requirement		Over Middle Marker (3500 From Runway End) To See 1000 Feet of Runway							
Distance Threshold to Pair of Lights (Feet)	Vertical Angle (Degrees)			Horizontal Angle		Candlepower 1/2-Mile Visibility			
	On Glide Path Altitude (200 Feet)	Above Glide Path Altitude (500 Feet)	Below Glide Path Altitude (100 Feet)	Runway Side (Degrees)	Off Runway Side (Degrees)	T = 0 5201 Night	T = 0 612 Day		
A On Course									
200	3 1	7 7	1 5	88 3	271 7		360	62500	
400	2 9	7 3	1 5	88 4	271 6		660	100000	
600	2 8	6 9	1 4	88 5	271 5		975	130000	
800	2 7	6 6	1 3	88 5	271 5		1600	180000	
1000	2 5	6 3	1 3	88 6	271 4		2000	300000	
B 300 Feet Off Course									
200	3 1	7 7	1 5	83 8	276 2	93 1 266 9	360	62500	
400	2 9	7 3	1 5	84 1	275 9	92 9 267 1	660	100000	
600	2 8	6 9	1 4	84 4	275 6	92 8 267 2	975	130000	
800	2 7	6 6	1 3	84 7	275 3	92 7 267 3	1600	180000	
1000	2 5	6 3	1 3	84 9	275 1	92 5 267 5	2000	300000	
Aircraft Position Requirement		1000 Feet From Runway End To See 1000 Feet of Runway							
Distance Threshold to Pair of Lights (Feet)	Altitude (Feet)	Vertical Angle (Degrees)	Horizontal Angle		Candlepower				
			Runway Side (Degrees)	Off Runway Side (Degrees)	1/2-Mile Visibility		1/4-Mile Visibility		
					T = 0 5201 Night	T = 0 612 Day	T = 0 189 Night	T = 0 378 Day	
A On Glide Path and On Course									
200	84	4 0	84 8	275 2		1 0	160	10	950
400		3 4	85 5	274 5		1 0	280	46	2200
600		3 0	86 1	273 9		1 3	500	150	5500
800		2 7	86 5	273 5		2 3	800	520	12000
1000		2 4	86 8	273 2		4 0	1500	2200	29000
B Above Glide Path On Course									
200	175	8 3	84 8	275 2		1 0	160	11	950
400		7 1	85 5	274 5		1 0	290	50	2300
600		6 2	86 1	273 9		1 3	500	160	5500
800		5 6	86 5	273 5		2 3	800	550	12000
1000		5 0	86 8	273 2		4 0	1500	2200	29000
C At Glide Path Level - 150 Feet Off Course									
200	84	4 0	80 5	279 5		1 0	160	12	980
400		3 4	81 9	278 1		1 0	295	53	2300
600		3 0	82 9	277 1		1 3	500	170	5500
800		2 6	83 7	276 3		2 3	840	575	12200
1000		2 4	84 3	275 7		4 0	1500	2300	30000
200		4 0		92 4	267 6	1 0	150	11	940
400		3 4		92 0	268 0	1 0	275	45	2250
600		3 0		91 8	268 2	1 3	495	158	5500
800		2 7		91 6	268 4	2 3	800	575	12000
1000		2 4		91 4	268 6	4 0	1500	2200	29000
D Above Glide Path - 150 Feet Off Course									
200	175	8 2	80 5	279 5		1 0	170	13	1000
400		7 0	81 9	278 1		1 0	298	55	2500
600		6 2	82 9	277 1		1 3	500	170	5500
800		5 5	83 7	276 3		2 3	840	585	12200
1000		5 0	84 3	275 7		4 0	1500	2300	30000
200		8 3		92 4	267 6	1 0	160	12	975
400		7 1		92 0	268 0	1 0	293	50	2300
600		6 2		91 8	268 2	1 3	500	170	5500
800		5 5		91 6	268 4	2 3	820	575	12100
1000		5 0		91 4	268 6	4 0	1500	2300	30000

TABLE IVb (Continued)

**MINIMUM CANDLEPOWER REQUIREMENTS OF RUNWAY LIGHTS  
FOR NIGHT AND DAY OPERATIONS UNDER ILS MINIMUMS**

		Aircraft Position Requirement		At 500 Foot Portal To See 1000 Feet of Runway					
Distance Threshold to Pair of Lights (Feet)	Altitude (Feet)	Vertical Angle (Degrees)	Horizontal Angle		Candlepower				
			Runway Side (Degrees)	Off Runway Side (Degrees)	1/2-Mile Visibility		1/4-Mile Visibility		
					T = 0 5201 Night	T = 0 612 Day	T = 0 189 Night	T = 0 378 Day	
A On Glide Path and On Course									
200	60	4 8	81 1	278 9		1 0	23	1 0	65
400		3 8	83 0	277 0		1 0	54	1 5	200
600		3 1	84 3	275 7		1 0	100	5 0	500
800		2 6	85 2	274 8		1 0	197	25 0	1300
1000		2 3	85 8	274 2		1 0	370	100 0	3600
B Above Glide Path On Course									
200	125	10 0	81 1	278 9		1 0	25	1 0	70
400		7 9	83 0	277 0		1 0	58	1 5	205
600		6 5	84 3	275 7		1 0	120	6 0	530
800		5 5	85 2	274 8		1 0	220	25 0	1400
1000		4 8	85 8	274 2		1 0	400	100 0	3800
C At Glide Path Level - 125 Feet Off Course									
200	60	4 7	71 4	288 6		1 0	27	1 0	80
400		3 7	75 4	284 6		1 0	56	2 0	230
600		3 0	78 0	282 0		1 0	120	7 0	600
800		2 6	79 8	280 2		1 0	220	27 0	1500
1000		2 3	81 1	278 9		1 0	385	102 0	3750
200		4 9			91 2 268 8	1 0	23	1 0	67
400		3 8			91 0 269 0	1 0	52	1 5	200
600		3 1			90 8 269 2	1 0	100	6 0	500
800		2 6			90 7 269 3	1 0	200	25 0	1350
1000		2 3			90 6 269 4	1 0	380	100 0	3650
D Above Glide Path - 125 Feet Off Course									
200	125	9 6	71 4	288 6		1 0	28	1 0	86
400		7 6	75 4	284 6		1 0	60	2 0	235
600		6 3	78 0	282 0		1 0	125	7 0	620
800		5 4	79 8	280 2		1 0	225	25 0	1560
1000		4 6	81 1	278 9		1 0	420	110 0	4200
200		10 1			91 2 268 8	1 0	24	1 0	70
400		7 9			91 0 269 0	1 0	53	1 5	210
600		6 5			90 8 269 2	1 0	110	6 0	510
800		5 5			90 7 269 3	1 0	205	25 0	1400
1000		4 8			90 6 269 4	1 0	390	102 0	3900
Aircraft Position Requirement At Touchdown Point - 20 Feet From Runway Edge To See 1000 Feet of Runway									
200	On Runway	3 32	46 47	313 53		1 0	2	1 0	3
400		2 07	64 6	295 4		1 0	6	1 0	13
600		1 45	72 43	287 57		1 0	17	1 0	42
800		1 17	76 62	283 38		1 0	37	1 0	125
1000		0 9	79 25	280 75		1 0	85	3 5	380
200		4 32	46 47	313 53		1 0	2	1 0	3
400		2 68	64 6	295 4		1 0	6	1 0	13
600		1 88	72 43	287 57		1 0	17	1 0	42
800		1 45	76 62	283 38		1 0	37	1 0	125
1000		1 17	79 25	280 75		1 0	85	3 5	380
Aircraft Position Requirement Taxiing 20 Feet From Runway Edge To See 500 Feet of Runway									
100		3 33	27 77	332 23		1 0	1 5	1 0	1 5
300		2 02	57 67	302 33		1 0	3 5	1 0	6 0
500		1 33	69 2	290 8		1 0	11 0	1 0	24 0
100		4 32	27 77	332 23		1 0	1 5	1 0	1 5
300		2 6	57 67	302 33		1 0	3 5	1 0	6 0
500		1 73	89 2	290 8		1 0	11 0	1 0	24 0