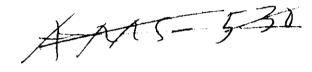
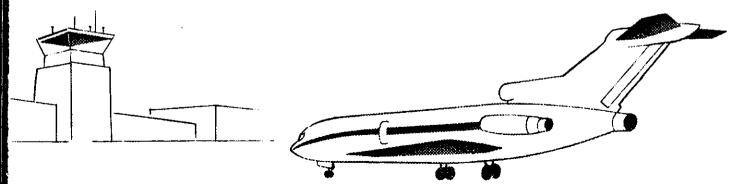
AC 65-4B





AIRCRAFT DISPATCHER written test guide

DEPARTMENT OF TRANSPORTATION FEDERAL AVIATION ADMINISTRATION



AIRCRAFT DISPATCHER WRITTEN TEST GUIDE

REVISED 1972



DEPARTMENT OF TRANSPORTATION FEDERAL AVIATION ADMINISTRATION FLIGHT STANDARDS SERVICE

PREFACE

This test guide was prepared by the Flight Standards Service, Federal Aviation Administration, Department of Transportation as Advisory Circular AC 65-4B to assist applicants who are preparing for the Aircraft Dispatcher Written and Practical Tests. It supersedes the Aircraft Dispatcher Written Test Guide, AC 65-4A issued in 1969.

This guide outlines the type and scope of knowledge covered in the tests, lists reference materials available from the Superintendent of Documents, U.S. Government Printing Office, and presents sample questions. As a convenience to applicants, those portions of the Federal Aviation Regulations concerning the eligibility, knowledge, and experience requirements for the certificate have been included. Applicants should be aware, however, that regulations are subject to amendment. Any question regarding the currency of these quoted excerpts may be checked with the appropriate FAA office.

Comments regarding this publication should be directed to the Department of Transportation, Federal Aviation Administration, Flight Standards Technical Division, P.O. Box 25082, Oklahoma City, Oklahoma 73125.

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INTRODUCTION

The Aircraft Dispatcher is an important member of the airline operation team and must be able to speak the language of the operating crews as well as that of management. He shares responsibility with the pilot for flight planning details that affect the safe conduct of the planned operation. After dispatching the flight, he performs important coordination functions involving the aircraft and other departments of the airline. He also provides the pilot with advisory information affecting the safe progress of the flight.

The Aircraft Dispatcher should, therefore, possess knowledge across the broad spectrum of airline operation as reflected in the section of this guide titled "Aeronautical Knowledge Covered by the Written Test." It is recognized that certain topics concerning domestic flight operations have no counterpart in international flight operations and vice versa; however, the applicant who is fully educated in the subject areas listed will be adequately prepared for the written test.

AIRCRAFT DISPATCHER CERTIFICATE REQUIREMENTS

Certification requirements for the Aircraft Dispatcher Certificate are excerpted from the Federal Aviation Regulations, Part 65.

\$ 65.51 Certificate required.

(a) No person may serve as an aircraft dispatcher (exercising responsibility with the pilot in command in the operational control of a flight) in connection with any civil aircraft in air commerce unless he has in his personal possession a current aircraft dispatcher certificate issued under this subpart.

(b) Each person who holds an aircraft dispatcher certificate shall present it for inspection upon the request of the Administrator or an authorized representative of the National Transportation Safety Board, or of any Federal, State, or local law enforcement officer.

\$ 65.53 Eligibility requirements: general.

To be eligible for an aircraft dispatcher certificate, a person must-

(a) Be at least 23 years of age;

(b) Be able to read, speak, and understand the English language, or have an appropriate limitation placed on his certificate;

(c) Comply with §§ 65.55, 65.57, and 65.59.

§ 65.55 Knowledge requirements.

(a) An applicant for an aircraft dispatcher certificate must pass a written test on-

(1) The regulations of this chapter that apply to the duties of an aircraft dispatcher;

(2) The general system of collecting and disseminating weather information;

(3) Interpreting aviation weather reports, including abbreviations and symbols, as prescribed in [National Weather Service Federal Meteorological Handbook No. 1] as amended;

(4) The fundamentals of meteorology as applied to aircraft operations, particularly as to—

(i) Surface and upper air weather maps and general characteristics of air masses, pressure systems, and frontal systems, including their symbols and nomenclature;

(ii) Cloud forms and their significance; and

(iii) Icing, turbulence, thunderstorms, fog and low ceilings, winds aloft, pressure pattern flying, the influence of terrain on meteorological conditions, and general principles of forecasting and analysis;

(5) Principles of aircraft navigation with particular respect to instrument operation and procedures;

(6) Communications facilities and procedures;

(7) Air navigation facilities and procedures; and

(8) Air traffic control procedures.

(b) A report of the test is sent to the applicant. A passing grade is evidence, for a period of 24 months after the date the test is given, that the applicant has complied with this section.

\$ 65.57 Experience requirements.

An applicant for an aircraft dispatcher certificate must present documentary evidence satisfactory to the Administrator that he has the experience prescribed in any one of the following paragraphs:

(a) A total of at least 2 of the 3 years before the date he applies, in scheduled air carrier operations, scheduled military aviation operations, or any other aircraft operations that the Administrator finds provides equivalent experience—

(1) As a pilot member of a flight crew;

(2) As a flight radio operator or ground radio operator;

(3) As a flight navigator;

(4) As a meteorologist;

(5) Performing the duties of an aircraft dispatcher or his assistant; or

(6) Performing other duties that the Administrator finds provide equivalent experience.

(b) A total of at least 2 of the 3 years before the date he applies, as an air route traffic controller or a certificated air traffic control tower operator.

(c) A total of at least 1 of the 2 years before the date he applies, as an assistant in dispatching scheduled air carrier aircraft performing the duties of an aircraft dispatcher under the direct supervision of a certificated dispatcher.

(d) Within 90 days before the date he applies, successful completion of a course of instruction approved by the Administrator as adequate for the training of an aircraft dispatcher.

An applicant is entitled to credit any combination of experience in paragraph (a), or paragraphs (a) and (b), of this section, if the aggregate of that experience is at least 2 years.

\$ 65.59 Skill requirements.

An applicant for an aircraft dispatcher certificate must pass a practical test-

(a) With respect to any one type of large aircraft used in air carrier operations, on-

(1) Weight and balance limitations;

(2) Performance operating limitations;

(3) Using cruise control charts;

(4) Fuel and oil capacities and rates of consumption; and

(b) Using the operations manual;

(b) On the characteristics of air routes and airports with particular reference to-

(1) Landing areas;

(2) Lighting facilities; and

(3) Approach and landing facilities and procedures;

(c) On the use and limitations of sensitivetype altimeters;

(d) On applying available weather forecasts and reports to determine whether a flight can be made safely;

(e) On using the Airman's Guide and the Flight Information Manual [Editorial note: Now the Airman's Information Manual];

(f) On dispatching and assisting a flight under adverse weather conditions; and

(g) On emergency procedures.

AIRCRAFT DISPATCHER QUALIFICATION TESTS

The Written Test

Nature of the test

The Aircraft Dispatcher Written Test is a single-section type, which permits a practical or operational approach to the problems that arise in planning and conducting transport flight operations.

The multiple choice questions in this guide are developed from typical situations involving preflight, inflight, and postflight duties of the Aircraft Dispatcher.

Material furnished the applicant for use during the test may include the following:

Significant Weather Map; Constant Pressure Charts; Area Forecasts; Terminal Forecasts; Sequence Reports; Segments of En route Charts; Instrument Approach Procedure Charts; excerpts from *Airman's Information Manual;* Aircraft Particulars; Minimum Equipment List for Dispatch; Aircraft Performance Data.

Taking the test

The written test may be taken at FAA General Aviation and Air Carrier District Offices of the Flight Standards Service, and at certain other designated places. After completing the test, the answer sheet and papers used for computations or notations will be surrendered to the proctor before leaving the test room.

Test grades are mailed to applicants on AC Form 8060-37, Airman Written Examination Report. The report also contains coded indicators of the knowledge areas which presented difficulty in the test. These coded indicators are related to an enclosed Written Examination Subject Matter Outline for quick and easy identification of knowledge deficiencies. The study outline contained in this guide is similar, but may not be exactly the same as the outline which the applicant receives with AC Form 8060-37. An applicant who receives a failing grade must present the appropriate AC Form 8060-37 for re-testing.

The applicant should keep in mind the following points when taking the test:

1. Read each question or problem carefully without looking at the possible answers. The applicant should clearly understand the problem before formulating the steps toward its solution.

2. He should then determine which of the alternatives most nearly corresponds with the answer he has formulated. The answer chosen should completely solve the problem.

3. From the answers given it may appear that there is more than one possible answer; however, only one answer is correct and complete. The other answers are either incomplete or derived from popular misconceptions.

4. If a particular test item proves difficult, it is best to proceed to another question. When the less difficult questions have been answered, the others should then be reconsidered.

5. There are no "trick" questions in the test.

THE PRACTICAL TEST

Prior to certification, the applicant must complete the practical test described in FAR 65.59. Whereas the written test is broad and general in scope, the practical test focuses attention on the specifics of the dispatching problem at the local level. The applicant must be thoroughly familiar with the contents of a typical air carrier operations manual for a particular aircraft. He must also know how to use the Airman's Information Manual and be aware of the characteristics of air routes and airports. He may be asked to complete the simulated dispatch of a flight over a route in his dispatch area and should be prepared to provide a thorough weather briefing on the proposed operation.

RETESTING AFTER FAILURE

An applicant who fails the Aircraft Dispatcher Written or Practical Test may apply for retesting after 30 days following the date he failed the test; or upon presenting a statement from a certificated and appropriately rated ground instructor, or a certificated Aircraft Dispatcher certifying that he has given

the applicant at least 5 hours of additional instruction in each of the subjects failed and now considers that the applicant is ready for retesting.

AERONAUTICAL KNOWLEDGE COVERED BY THE WRITTEN TEST

I. FEDERAL AVIATION REGULATIONS

A. FAR 1, 61, and 65

- 1. General definitions (1.1).
- 2. Abbreviations and symbols (1.2).
- 8. Airline Transport Pilots-logging of instrument time and recency of experience (61.41, 61.47).
- 4. Medical certificates—duration (61.43).
- 5. Instruction and privileges (61.163, 61.165).
- 6. Aircraft Dispatcher (65).

B. FAR 91

- 1. Aircraft speed limitations (91.70).
- 2. Compliance with ATC clearances, etc. (91.75).
- 3. Altimeter settings (91.81).
- 4. Operating on or in the vicinity of an airport (91.85 through 91.89).
- 5. Operating in Positive Control and Jet Advisory areas (91.97, 91.99).
- 6. Visual Flight Rules (91.105 through 91.109).
- 7. Takeoff and landing under IFR (91.116, 91.117).
- 8. Minimum and cruising altitudes—IFR operations (91.119, 91.121).
- 9. IFR, radio communications (91.125).
- 10. IFR operations; two-way radio communications failure (91.127).
- 11. Operation under IFR in controlled airspace; malfunction reports (91.129).
- C. FAR 121—Performance, Special Airworthiness, Instrument and Equipment Requirements
 - 1. Manual requirements (121.131 through 121.141).

- 2. Performance, reciprocating engine powered airplanes (121.171 through 121.187).
- 3. Performance, turbine engine powered airplanes (121.189 through 121.197).
- 4. Fire precautions (121.221).
- 5. Cargo location and security (121.285 through 121.287).
- 6. Landing gear, aural warning (121.289).
- 7. Instruments and equipment (121.301 through 121.311 and 121.313 through 121.325).
- 8. Supplemental oxygen (121.327, 121.329, 121.333, 121.337).
- 9. Overwater operations and icing conditions (121.339 through 121.341).
- 10. Recorders, flight and voice (121.343, 121.359).
- 11. Radio equipment and weather radar (121.345 through 121.357).

II. AIRMAN'S INFORMATION MANUAL—BA-SIC FLIGHT MANUAL AND ATC PROCE-DURES

- A. Chapter 1. General.
- B. Chapter 2. Navigation Aids.
- C. Chapter 3. The Airspace.
- D. Chapter 4. Air Traffic Control.
- E. Chapter 5. Safety of Flight.

III. FLIGHT PLANNING AND AIR NAVIGATION

- A. Aviation Weather
 - 1. Elementary meteorology.
 - 2. Air masses and fronts.
 - 3. Thunderstorms.
 - 4. Icing hazards and ice formation.
 - 5. Common "IFR" producers.
 - 6. Aviation weather reports.
 - 7. Aviation weather forecasts.

- 8. Weather charts: Surface, Depiction, Radar, Constant Pressure, Significant Weather, Upper Wind Progs.
- 9. High altitude weather features.
- 10. Pressure, density, and true altitude details.
- B. Computations
 - 1. Flight time en route.
 - 2. Required fuel.
 - 3. Dispatched endurance.
 - 4. Actual and allowable payload determination.
 - 5. Weight and balance-weight limitations.
 - 6. Weight and balance-location of center of gravity (CG).
 - 7. Weight and balance-shifting, adding, or removing weight.
 - 8. Performance charts-graphs.
 - 9. Performance information tabulated.
 - 10. Off-course and return to course.
 - 11. Wind experienced en route-direction and speed.
 - 12. Wind components-head, tail, crosswind.
 - 13. Airspeed adjustments to maintain schedule or arrival.
 - 14. Specific range-turbojet airplanes.
 - 15. Estimated time of arrival (ETA).
 - 16. Pressurized airplane climb—cabin time or rate.
 - 17. Flight progress.
 - 18. Cruise control techniques.
- C. Miscellaneous
 - 1. Definition of Mach number and critical Mach number.
 - 2. Subsonic, transonić, supersonic flight regimes.
 - 3. Determination of Mach number or True Airspeed from given information.
 - 4. Interpretation of En route and Instrument Approach Charts.
- D. FAR 121--Personnel requirements; qualifications and duty time limitations
 - 1. Airman and crewmember requirements (121.381 through 121.395).

- 2. Emergency evacuation duties (121.397).
- 3. Crewmember and dispatcher training program (121.400; 121.411 through 121.419; 121.422, 121.424, 121.427).
- 4. Crewmember qualification (121.431 through 121.434; and 121.437 through 121.447).
- 5. Aircraft dispatcher qualifications and duty limitations (121.461 through 121.465).
- 6. Flight time limitations: domestic air carrier (121.470, 121.471).
- 7. Flight time limitations: flag air carrier (121.480 through 121.491).
- 8. Flight time limitations: supplemental air carrier and commercial operator (121.500 through 121.509; 121.513 through 121.525).
- E. FAR 121—Flight Operations
 - 1. Responsibility for operational control (121.533 through 121.537).
 - 2. Operation: flight deck duty, etc. (121.543 through 121.549).
 - 3. Emergencies: domestic and flag air carriers (121.557).
 - 4. Emergencies: supplemental air carriers (121.559).
 - 5. Reporting conditions in flight (121.561, 121.563).
 - 6. Engine inoperative: landing: reporting (121.565).
 - Briefing of passengers (121.333, 121.571, 121.573).
 - 8. Minimum altitudes for use of the automatic pilot (121.579).
- F. FAR 121-Dispatching and Flight Release Rules
 - 1. Dispatching and flight release authority (121.591 through 121.-597).
 - 2. Familiarity with weather conditions and information to pilot-incommand (121.599, 121.601).
 - 3. Equipment, facilities and service (121.603 through 121.609).
 - 4. Dispatch and flight release (121.-611 through 121.615).
 - 5. Alternate airport for departure (121.617).

- 6. Alternate airports (121.619, 121.-621, 121.623).
- 7. Alternate airport weather minimums (121.625).
- 8. Flight in unsafe conditions (121.-627, 121.629).
- 9. Dispatch rules: original, redispatch, or amendment (121.631 through 121.635).
- 10. Takeoffs from unlisted and alternate airports (121.637).
- Fuel supply: all operations: domestic air carriers (121.639, 121.-647).
- 12. Fuel supply: (turbojet) flag and supplemental air carriers (121.641 through 121.645).

- 13. Takeoff and landing weather minimums (121.649 through 121.655).
- 14. Flight altitude rules (121.657 through 121.661).
- 15. Responsibility for dispatch release, load manifest, and flight plan (121.663 through 121.667).

G. FAR 121-Records and Reports

- 1. Records (121.683, 121.685, 121.-711).
- 2. Releases (121.687, 121.689, 121.-709).
- 3. Load manifests (121.691 through 121.697).
- 4. Reports (121.703, 121.705).

REFERENCE MATERIALS

The following list of publications and materials is provided for the benefit of individuals who wish to prepare for the written test. Except for free *advisory circulars* and *charts*, all of these items are available through the U.S. Government Printing Office.

Textbooks and other reference materials are also available from many commercial publishers. It is the responsibility of each applicant to obtain study materials appropriate to his needs.

Free FAA publications may be obtained from "Department of Transportation, Distribution Unit, TAD-484.8, Washington, D.C. 20590."

Nore.—References listed were available at the time this publication went to press.

FEDERAL AVIATION REGULATIONS (FARs)

The subscription prices listed include automatic revision service to all Parts contained in the Volume ordered. The FAR Parts contained in each Volume are listed in the "Advisory Circular Checklist and Status of Federal Regulations," obtainable free on request from FAA.

	Dulas	Additional for Foreign
Wel I Dent 1 Definitions	Price	Mailing
Vol. I, Part 1, Definitions and Abbreviations Vol. IX, Part 65, Certifica-	\$ 2.50	\$0.75
tion: Airmen Other Than		
Flight Crewmembers	\$ 6.00	\$1.50
Vol. VI, Part 91, General Operating and Flight		
Rules	\$ 9.00	\$2.25
Vol. VII, Part 121, Certifica- tion and Operations: Air Carriers and Commercial		
Operators of Large Air- craft	\$10.50	\$2.75

FLIGHT INFORMATION PUBLICATIONS

Airmen's Information Manual—This publication presents in four parts, information necessary for the planning and conduct of a flight in the National Airspace System. It is designed to be used in the cockpit for preflight and inflight operations by pilots and contains both instructional and procedural information. The subscription consists of:

- Part 1—Basic Flight Manual and ATC Procedures. Issued quarterly (\$7.00; Foreign \$8.75).
- Part 2—Airport Directory. Issued semiannually (\$7.00; Foreign \$8.75).
- Part 3—Operational Data. Issued every 56 days, and
- Part 3A—Notices to Airmen. Issued every 14 days. (\$22.00; Foreign \$27.50).
- Part 4—Graphic Notices and Supplemental Data. Issued semiannually (\$9.50; Foreign \$12.00).

Terminal Air Traffic Control-7110.8C (two-year subscription with changes issued quarterly) \$13.50 domestic; \$17.00 foreign-GPO (TD 4.308: T 27/971).

En Route Air Traffic Control-7110.9C (two-year subscription with changes issued quarterly) \$8.00 domestic; \$10.00 foreign-GPO (TD 4.308 En 1/971).

These FAA Handbooks prescribe air traffic control procedures and phraseology for use by personnel providing terminal and en route air traffic control services. Although written for the air traffic controller, the handbooks are excellent for the study of standard communication procedures by others who need to be familiar with them.

STUDY MANUALS

Aviation Weather, AC 00-6 (\$4.00-GPO Catalog No. FAA 5.8/2:W37). An excellent reference treating all phases of meteorology of interest to the Aircraft Dispatcher. Aviation weather reports and forecasts are also covered in detail with respect to format and content. Air Navigation, AF Manual 51-40, Volume I (\$4.00). This U.S. Air Force publication is an excellent reference for basic navigation.

Aircraft Performance—Reciprocating and Turboprop Engine Aircraft, AF Manual 51-9 (\$1.50). This U.S. Air Force publication contains much material having civil aviation applications.

Pilot's Weight and Balance Handbook, AC 91-23 (\$1.25-GPO Catalog No. TD 4.408: P 64/3). An excellent treatment of the subject from the standpoint of the pilot and aircraft owner or operator.

Civil Use of U.S. Government Instrument Approach Procedure Charts, AC 90-1A (Free from FAA). Describes instrument approach procedure charts.

CHARTS

Instrument Approach Procedure Charts (10¢ per airport set). Individual charts give detailed information on procedure for each type of approach at the airport.

En route Charts (35¢ each). These charts provide the necessary aeronautical information for en route instrument navigation (IFR) in the established airway structure.

Area Charts (10¢ each). These charts supplement the En route Charts by giving departure, arrival, and holding procedures at principal airports.

Checks or money orders for charts should be made payable to "NOS, Dept. of Commerce, C-44" and sent to: Distribution Division (C-44), National Ocean Survey, Washington, D.C. 20235.

HOW TO OBTAIN GPO PUBLICATIONS

(1) Use an order form, not a letter unless absolutely necessary. Order forms, which may be duplicated by the user, are included in the catalog "FAA Publications," sent free upon request from:

> Department of Transportation Distribution Unit, TAD 484.3 Washington, D.C. 20590

(2) Send separate orders for subscription and non-subscription items.

(3) Give the exact name of the publication and in the case of a single publication the GPO catalog number, e.g., TD 4.408:In 7/3 or FAA 5/8:W 37.

(4) Send a check or money order made payable to the Superintendent of Documents. Send the exact amount (no cash). (Include an additional 25 percent of the total order to cover postage for foreign mailing.)

(5) Enclose a self-addressed mailing label if you have no order blank.

(6) Use special delivery when needed.

(7) Use GPO bookstores.

Mail orders may be directed to the Washington headquarters of the Superintendent of Documents, U.S. Government Printing Office, Washington, D.C. 20402.

Several retail bookstores for GPO publications have also been established throughout the country. The GPO bookstores are located at the following addresses:

 GPO Bookstore 2121 8th Ave. North Birmingham, Ala. 35203 GPO Bookstore Federal Building Room 1015 300 N. Los Angeles St. Los Angeles, Calif. 90012 GPO Bookstore Federal Building Room 1023 450 Golden Gate Ave. San Francisco, Calif. 94102 GPO Bookstore Federal Building Room 1421 1961 Stout St. Denver, Colo. 80202 GPO Bookstore Pueblo Memorial Airpark Pueblo, Colo. 81001 	 GPO Bookstore Federal Office Building Room 1463 14th Floor 219 South Dearborn St. Chicago, Ill. 60604 GPO Bookstore Room G-25 J.F.K. Federal Building Government Center Boston, Mass. 02203 GPO Bookstore Federal Building Room 135 601 East 12th Street Kansas City, Mo. 64106 GPO Bookstore 26 Federal Plaza Room 110 New York, N.Y. 10007
GPO Bookstore Federal Building Room 100 275 Peachtree St., N.E. Atlanta, Ga. 30303	GPO Bookstore Federal Building U.S. Courthouse Room 1C46 1100 Commerce St. Dallas, Tex. 75202

SAMPLE TEST

The following sample test is similar in format to the official FAA written test. It is important to remember, however, that these test items do not direct attention to all of the topics on which you will be tested in the official test. For this reason, you should concentrate on the section entitled "Aeronautical Knowledge Covered by the Test." A knowledge of all of the topics presented in the outline—not just the ability to answer these few sample test items—should be your goal as you prepare for the written test.

The increased performance of present day transport category aircraft requires greater emphasis on high-altitude meteorology, highspeed aerodynamics, and turbine equipment. Applicants should, therefore, expect to encounter test items dealing with these areas in the written test.

Answers to the sample test items are given at the end of the test, along with a detailed analysis, or explanation, of each test item.

NOTE—The reader should be aware that the sample test items are based on regulations and procedures in effect at the time of final editing of this guide. Similar test items in the official FAA written tests should always he answered in terms of current regulations and procedures.

SITUATION

You are a certificated Aircraft Dispatcher employed by an airline whose central dispatch office is located at the John F. Kennedy International Airport. The company is an air carrier, authorized to operate in scheduled transportation under pertinent regulations as a domestic and flag carrier.

The domestic route structure provides service between metropolitan areas in the midwest and the east coast, while the international structure provides service to west European terminals. Domestic routes are served by four-engine turbine powered aircraft and three-engine turbine powered aircraft. International routes are served by four-engine turbine powered aircraft. 1. The airplane dispatched has a seating capacity of 120 passengers. What is the minimum number of flight attendants required by FAR 121 if 100 passengers are aboard?

1—Two. 2—Three. 3—Four. 4—Five.

2. Assume that your company operates only turbojet powered airplanes which have been in operation under Part 121 for more than 90 days. As an aircraft dispatcher, your "operating familiarization" must have been accomplished within the preceding 12 calendar months in—

- 1-each of the types of airplanes you will dispatch.
- 2-at least one of the types of airplanes you will dispatch.
- 3-a Group II airplane operated under Part 121 or by observing 5 hours of approved simulator training.
- 4—any Group I or Group II airplane operated under Part 121 or by observing 5 hours of approved simulator training.

3. An aircraft dispatcher must specifically authorize the flight of a flag air carrier airplane from an intermediate airport, if it remains at that airport for more than—

1—1 hour.
 2—2 hours.
 3—4 hours.
 4—6 hours.

4. Assume these conditions: (1) a threeengine turbine powered air carrier airplane; (2) weather conditions at the airport of takeoff below the landing minimums listed in the certificate holder's operations specifications. Under these conditions, an alternate must be specified which is at a distance of not more than—

1-1 hour at normal cruising speed in still air with one engine inoperative.

* * * * *

- 2-1 hour at long range cruising speed with all engines operating.
- 3-2 hours at normal cruising speed with one engine inoperative.
- 4-2 hours at normal cruising speed with all engines operating.

5. The airborne weather radar is inoperative on a passenger-carrying turbojet powered airplane in your company's fleet. Thunderstorms are forecast along the route of flight. You may dispatch this airplane only if the—

- 1-radar is repaired prior to dispatch.
- 2-flight can be completed in VFR conditions, day or night.
- 3-flight can be completed in VFR or IFR conditions during daylight hours.
- 4-flight can be completed in VFR conditions during daylight hours.

6. Assume a cabin altitude of 14,000 feet for 50 minutes. How much supplemental oxygen must be provided for the passengers on a turbine engine powered air carrier airplane?

- 1-Enough oxygen for 10 percent of the passengers for 20 minutes.
- 2—Adequate oxygen for each passenger for the entire 50 minute flight segment.
- 3—Sufficient oxygen for 30 minutes for 30 percent of the passengers.
- 4—One hour oxygen supply for 20 percent of the passengers.

7. An emergency arises which requires immediate decision and action. If the pilot-incommand and dispatcher are in communication, which statement is correct?

- 1—The dispatcher and pilot-in-command shall make a joint decision, share responsibility for any action taken, and file a joint written report with the Administrator within 10 days.
- 2-The pilot-in-command shall make a decision, take whatever action he considers necessary, and send a written report to the Administrator within 10 days after returning to his home base.
- 3-The dispatcher shall make a decision and direct the emergency action to be followed by the pilot-in-command, and 10 days thereafter, file a written report with the air carrier's operations manager.

4---The pilot shall take no action unless the dispatcher approves it, and 10 days subsequent to the emergency, submit a written report to the air carrier's operations manager.

8. One engine of a three-engine turbojet powered air carrier airplane is shut down in flight. The pilot-in-command—

- 1—may continue to the destination only if the dispatcher authorizes this course of action.
- 2-must return to the departure airport if he has not completed his climb to en route altitude.
- 3—may proceed to any airport that he selects if he decides this is as safe as landing at the nearest suitable airport.
- 4—is required by regulations to land at the nearest suitable airport.

9. Instruction in which of the following is *not* required in the initial training of an aircraft dispatcher?

- 1-Emergency assignment of duties of individual crewmembers.
- 2-Provisions of appropriate Federal Aviation Regulations.
- 3—The certificate holder's operating manual.
- 4—The certificate holder's operating specifications.

10. Which of the following is correct in regard to Standard Instrument Departures (SIDs)?

- 1—A SID may not be issued to the pilot of an air carrier flight unless he specifically requests it.
- 2-The pilot of any civil aircraft may be issued a SID whenever ATC considers it appropriate.
- 3-SIDs are published for all airports located in high density traffic areas and having complex departure routes.
- 4—Standard Instrument Departures are published only for those airports having Standard Terminal Arrival Routes.

* * * * * *

You report to the Operations Office at 1100 Eastern Standard Time on February 8 (1600Z) and survey the weather conditions for the area in which the flight will be dispatched.

* * * * *

11. From a review of the JFK Area Forecast (Figure 2, Appendix), you determine that-

- 1-ceilings in southeastern Pennsylvania and southern New Jersey will gradually lower during the forecast period.
- 2-the high pressure area over the upper Hudson Valley will remain stationary.
- 3—ceilings will remain constant in northeastern Pennsylvania and southeastern New York from 0100Z to 1900Z Sunday.
- 4-by 1900Z Sunday, mixed precipitation and ceilings of 200 to 500 feet will be prevalent in western Pennsylvania.

12. Which statement correctly interprets the Terminal Forecasts (Figure 4, Appendix)?

- 1-At ORD, after 1600Z, the ceiling is expected to vary from 800 to 200 feet.
- 2—The surface wind velocity at CLE is expected to decrease during the forecast period.
- 3—At MKE, at the beginning of the forecast period, the visibility is expected to be 5 miles in haze and smoke.
- 4—The visibility at ORD, at the beginning of the forecast period, is expected to be 2 miles.

13. From an inspection of the NMC High Level Weather Prog Chart (Figure 5, Appendix) you determine that—

- 1-occasional light icing can be expected in the vicinity of 35°N./87°W.
- 2-the cloud coverage in the vicinity of 42°N./90°W. is 6/8 cirrus.
- 3—an 8/10 sky coverage exists in southern Illinois.
- 4-the sky condition in southern Wisconsin is 6/10 cirrus coverage with the cloud bases at 36,000 feet.

* * * * *

Your work schedule requires the completion of flight planning arrangements for your company's Flight 105 which is scheduled to depart John F. Kennedy International Airport for Chicago-O'Hare International Airport at 1230 EST.

* * * * *

Complete the flight time analysis in the Appendix, Figure 34. An appropriate Chart segment for this routing between JFK and ORD is also included in the Appendix, Figure 28.

Note.—The flight time analysis form used in Figure 34 is not intended to be an operational form. It is used here for an orderly presentation of flight planning data. Similar forms are made available to applicants when they take the official written test. Applicants may use these forms or any other flight planning form of their selection.

* * * * * *

14. The estimated time en route from JFK to ORD is—

1-1 hour 41 minutes.

2-1 hour 44 minutes.

3-1 hour 47 minutes.

4-1 hour 51 minutes.

15. What is the minimum weight of fuel required for this flight? (Include 1,000 pounds for missed approach.)

1—27,100 pounds. 2—27,700 pounds. 3—28,300 pounds. 4—28,900 pounds.

* * * * *

Test items 16, 17, and 18 are based on information given in the loading schedule below.

Item	Weight/ Pounds	Moment/ 1000
Basic Operating Weight (BOW) _	88,350	80,552.3
Forward Cargo	3,000	. <u></u>
Aft Cargo	4,000	
Forward Passengers (20)	3,400	
Aft Passengers (54)	9,180	
Fuel Tank #1	11,500	
Tank #2	11,500	
Tank #3	11,500	
Total		

* * * * *

16. Compute the ramp CG in percent of MAC (Figures 9 and 10, Appendix).

1-17.6% MAC. 2-17.0% MAC. 3--16.4% MAC. 4-15.8% MAC.

17. You determine the zero fuel weight for this flight to be-

1-121,460 pounds. 2-117,600 pounds. 3-106,740 pounds. 4-107,930 pounds. 18. What is the estimated landing weight at ORD?

1-124,800 pounds. 2-125,050 pounds. 3-125,330 pounds. 4-125,950 pounds.

19. If only structural weight limitations were considered, the maximum payload for this aircraft would be (Figure 9, Appendix).

1-24,500 pounds.

- 2-25,000 pounds.
- 3—29,110 pounds.
- 4-29,650 pounds.

20. The basic operating weight of an aircraft is defined as its—

- 1-maximum authorized weight less disposable fuel.
- 2-weight, ready for flight, including fuel and payload.
- 3-weight, ready for flight, including crew but without payload and fue!.
- 4-empty weight, plus fixed ballast, residual fuel, and residual oil.

21. Assume the following conditions: (1) runway length-9,000 feet; (2) stopway length-2,000 feet; (3) clearway length-5,000 feet. In determining the takeoff weight limitations of a turbine engine powered transport category airplane certificated after August 29, 1959, the "accelerate-stop" distance must not exceed-

1-9,000 feet. 2-10,000 feet. 3-10,500 feet.

4-11,000 feet.

22. Based on the conditions stated in the previous test item, what is the maximum allowable takeoff distance?

1--- 9,000 feet. 2---11,000 feet. 3---13,500 feet. 4---14,000 feet.

23. An airplane is departing on Runway 22 and the tower-reported wind is 190° at 30 knots. What is the crosswind component (Figure 11, Appendix)?

1-18 knots from the left.

2-20 knots from the right.

3—15 knots from the left.

4-23 knots from the right.

24. The temperature at FL-310 is -36° C. What is the relationship of this temperature to International Standard Temperature (ISA) (Figure 12, Appendix)?

1-Standard.

- 2-Five degrees warmer than standard.
- 8-Ten degrees colder than standard.
- 4-Ten degrees warmer than standard.

25. Determine the station pressure if the altimeter setting is 29.60 and the airport elevation is 410 feet (Figure 13, Appendix).

1 - 26.4	in.	Hg.
228.6	in.	Hg.
3-30.0	in.	Hg.
4-29.2	in.	Hg.

26. Determine the stabilizer trim setting in units airplane nose up if the CG is 20% MAC and the flap setting is 25° (Figure 14, Appendix).

 $1-5\frac{1}{2}$ 2-6. $3-6\frac{1}{2}.$ 4-7.

27. Determine the average EPR for takeoff at John F. Kennedy International Airport under these conditions (Figures 13, 14 and 32, Appendix).

Temperature Altimeter setting Engines 1 and 3 Engine 2	29.95. A/C "ON"
1	
2—1.92.	
3—1.94.	
4	
1	NO BLEED.

28. Based on the following conditions, what is V_R and V_2 for a normal takeoff (Figure 14, Appendix)?

29. A close examination of the Takeoff Performance Chart (Figure 16, Appendix) reveals that—

- 1-higher than standard temperatures increase the climb limit weight.
- 2-a headwind will decrease the effective runway length.
- 3—higher than standard temperatures decrease the allowable brake release gross weight.
- 4—with regard to effective runway length, up-slope will tend to balance the effect of tailwind.

30. Determine the runway limit gross weight at brake release under the following conditions (Figure 16, Appendix).

Airport pressure altitude	3,000 feet.
Runway length available	7,000 feet.
Headwind	20 knots.
Runway slope	1% up.
Average EPR	1.90.
Temperature	85° F.
CG	13%.
1—136,000 pounds.	
2—138,500 pounds.	
3-140,000 pounds.	
4—142,000 pounds.	

31. Using the same data given in the previous test item, determine the climb limit gross weight (Figure 16, Appendix).

1—139,250 pounds. 2—142,500 pounds.

- 3-153,000 pounds.
- 4-157,000 pounds.

32. From the Simplified Flight Planning Chart (Figure 17, Appendix), determine the trip time and fuel under the following conditions.

Distance 1,400 Tailwind 50 km	
Cruise altitude FL-31	t0.
Landing weight 126,00	
Average temperature ISA -	+10° C.
1-2 hours 47 minutes and 24,20)0 pounds.
2-3 hours 56 minutes and 25,40	0 pounds.
3-3 hours 04 minutes and 26,60)0 pounds.
4-3 hours 12 minutes and 27,10	0 pounds.

33. Based on the following conditions, what is the weight of an airplane (to the nearest 100 pounds) when it reaches FL-330 (Figures 18 and 19, Appendix)? Departure airport elevation __ Sea level. Brake release weight _____ 150,500 pounds. Average climb temperature __ ISA ~15° C.

1-145,000	pounds.
2—145,200	pounds.
3145,700	pounds.
4-146,000	pounds.

34. Assume these conditions:

Flight	leve	1	 310.	
Mach			 .82.	
OAT			 -40°	С.
Total	fuel	flow	 8,860	lbs./hr.

What is the Nautical Air Miles (NAM)/1,000 pounds of fuel?

1—54.8 NAM/1,000 pounds. 2—52.6 NAM/1,000 pounds. 3—49.7 NAM/1,000 pounds. 4—48.6 NAM/1,000 pounds.

35. Assuming an indicated Mach .82 cruise and a constant gross weight, select the correct general statement regarding fuel flow (see Planning Charts, Figures 21, 22, and 23 in Appendix). A lower fuel flow occurs when—

- 1-altitude is increased or temperature is decreased.
- 2-both altitude and temperature are increased.
- 3-temperature is increased or altitude is decreased.
- 4-both altitude and temperature are decreased.

36. If the true air temperature is -30° C. and an aircraft is cruising at FL-290 at 490 knots true airspeed, what is the Mach number?

1---.84. 2---.83. 3---.82. 4---.81.

37. Assuming the same temperature, flight level, and true airspeed as that given in the previous test item, what is the calibrated airspeed?

1 - 286	knots.
2—298	knots.
3304	knots.
4-312	knots.

38. What is the required holding fuel for a three-engine flag air carrier turbojet under

the conditions listed below (Figure 24, Appendix)?

- Estimated weight upon arrival at alternate _____ 120,000 pounds. Average temperature _____ ISA standard. Alternate airport elevation _ 3,500 feet.
- 1-3,210 pounds. 2-3,340 pounds. 3-3,450 pounds.
- 4---3,680 pounds.

39. Determine the approximate landing weight under these conditions (Figures 22 and 18, Appendix).

Elevation of landing air-	
port	Sea level.
Cruise altitude	FL-290.
Cruise temperature	-30° C.
Airplane weight (cruise)	
at 1600Z	150,000 pounds.
Estimated landing time	1715Z.
1—130,000 pounds.	

2—135,000 pounds. 3—140,000 pounds. 4—145,000 pounds.

40. Under the following conditions, what is the field length limit gross weight (Figure 25, Appendix)?

Runway length available 7,200 feet.
Headwind
component 12 knots.
Pressure altitude 2,000 feet.
CG 17% MAC.
Dispatched weight _ Under 160,000 pounds.
Anti-skid Off.
Nose brake Off.
Runway surface
conditions Dry.
1152,000 pounds.
2-147,000 pounds.
3-140,000 pounds.
4—135,000 pounds.

41. Assume that ATC follows normal practice and conforms to the "hemispheric rule" in assigning a flight level. Which of the following responses include three flight levels appropriate for a westbound IFR flight.

1-FL-280, FL-300, FL-320. 2-FL-280, FL-300, FL-350. 3-FL-260, FL-280, FL-310. 4-FL-260, FL-280, FL-300. 42. Which statement is true regarding the instrument approach chart for Chicago-O'Hare Airport (Figure 33, Appendix)?

- 1—Precision Approach Radar monitor is available for an ILS approach to Runway 14L.
- 2—The distance from the final approach fix (FAF) to the localizer missed approach point (MAP) is 5.0 NM.
- 3—Runway 14L has centerline lighting.
- 4-Visual Approach Slope Indicators are installed on Runways 14L and 14R.

43. At a departure airport, the pressure altitude is 2,000 feet and the temperature is $+20^{\circ}$ C. What is the approximate density altitude?

1-3,000 feet. 2-2,500 feet. 3-3,500 feet. 4-4,000 feet.

* * * * *

The following seven test items apply to a flight from New York (KJFK) to Paris (LFPO).

* * * * *

44. The approximate wind direction and velocity at the 300-millibar level (Figure 6, Appendix) at a position 50°N./43°W. is—

1-008°/88 knots. 2-180°/76 knots. 3-230°/75 knots. 4-280°/80 knots.

45. Select the correct statements regarding the tropopause (boundary between the troposphere and stratosphere).

- A. The strongest jet streams exist in the "break" region between the polar and tropical tropopause.
- B. The average height of the tropopause is greater in the polar regions than in the tropics.
- C. The tropopause is higher in summer than in winter.
- D. The tropopause is generally free of turbulence.
 - 1-A, B, and C only.
 - 2-B, C, and D only.
 - 3-B and D only.
 - 4-A and C only.

46. One of the following is a characteristic of jet streams.

- 1—In middle and high altitudes the strength of jet streams is greater in summer than in winter.
- 2—As a jet stream migrates southward, its core rises and its speed increases.
- 3—Severe clear air turbulence occurs to a maximum degree on the warm side of a jet stream and above the jet core.
- 4—The core of strongest winds in jet streams is found above 40,000 feet at all latitudes.

47. Refer to the Tropopause/Vertical Wind Shear Chart (Figure 7, Appendix). If an aircraft is maintaining FL-330 on the route shown, it would—

- 1-remain below the tropopause for the entire route.
- 2-be below the tropopause at a position 52°N./30°W.
- 3-be below the tropopause at a position 47°N./56°W.
- 4-remain above the tropopause for the entire route.

48. Refer to the Surface Prog. and Significant Weather Chart (700-150 MB) (Figure 8, Appendix). Which of the following statements is correct?

- 1-Severe turbulence can be expected near LFPO (49°N./3°E.).
- 2—Occasional moderate turbulence can be expected on the great circle route near 50°N./45°W.

- 3—An aircraft maintaining FL-310 for the entire route (KJFK to LFPO) will remain above all clouds.
- 4-A low pressure center is expected in the vicinity of 52°N./25°W.

49. Assuming the following conditions, compute the estimated flight time from KJFK to LFPO.

Total distance	2,600 NM.
Cruise altitude	FL-330.
Average temperature	−50° C.
Average cruise speed	Mach .82.
Time and distance	
for climb	30 min./200 NM.
Time and distance	
for descent	20 min./115 NM.
Cruise wind factor	+80 knots.
1—4 hours 32 minutes.	
2-4 hours 41 minutes.	
3-4 hours 58 minutes.	

4-5 hours 12 minutes.

50. Basing your computation on the en route time you determined in the preceding test item and on the data given below, what is the total fuel burn from KJFK to LFPO?

Average fuel flow in climb __ 14,000 lbs./hr. Average fuel flow in cruise __ 11,500 lbs./hr. Average fuel flow in descent _ 6,000 lbs./hr.

1-51,500 pounds.

2-53,200 pounds.

- 3-56,500 pounds.
- 4-59,200 pounds.

ANALYSIS OF ANSWERS TO SAMPLE TEST ITEMS

1-(2)

See FAR 121.391.

2-(3)

See FAR 121.463.

3--(4)

See FAR 121.595.

4-(3)

See FAR 121.617.

5---(4)

See FAR 121.357.

6-(1)

See FAR 121.329.

7—(2)

See FAR 121.557.

8-(3)

See FAR 121.565.

9-(1)

See FAR 121.415.

10---(2)

The pilot of any civil aircraft may be issued, and is expected to accept, a Standard Instrument Departure (SID) unless he specifies "NO SID" either orally or in the "remarks" section of his flight plan. See the Airman's Information Manual, Part I.

11---(1)

Choice 1 – Correct, as stated under "clouds and weather."

Choice 2-Incorrect; the high pressure area will continue moving eastward.

Choice 3 – Incorrect; ceilings are expected to lower rapidly in northeastern Pennsylvania and southeastern New York from 0100Z to 1900Z Sunday.

Choice 4-Incorrect; under "outlook," mixed precipitation should move into southeastern Pennsylvania and extreme southern New Jersey between 0200Z and 0500Z Sunday. 12-(3)

Choice 1-Incorrect; the ceiling at ORD is expected to vary from 800 feet to 500 feet after 1600Z.

Choice 2 – Incorrect; the surface wind velocity is expected to increase by 1700Z.

Choice 3-Correct; refer to the Key to Aviation Weather Reports and Forecasts (Figs. 1a and 1b, Appendix).

Choice 4 – Incorrect; the visibility at ORD at the beginning of the forecast period should be 4 miles.

13-(2)

Choice 1-Incorrect; the symbol on the chart in this area indicates light turbulence.

Choice 2 - Correct.

Choice 3 – Incorrect; the sky coverage is given in eighths, not tenths. The coverage is eight-eighths cirrus and cirrostratus.

Choice 4 – Incorrect; in the area referred to, the sky coverage is six-eighths cirrus with cloud bases at 30,000 feet.

14---(3)

The flight time from JFK to ORD is 1 hour and 47 minutes.

15-(2)

The time and fuel summary is reproduced below:

En route	01:47/17,100 lbs.
Alternate	00:15/2,900 lbs.
Reserve	00:45/6,700 lbs.
Extra for approach and	
missed approach	1,000 lbs.
TOTAL	02:47/27,700 lbs.

16-(3)

Divide total moment (126,751,300 lbs.-in.) by total weight (142,430 lbs.) to get a CG location of 889.9 inches aft of the datum. Subtract LEMAC from the CG to determine its location aft of LEMAC (889.9"-860.2"= 29.7" aft of LEMAC). Dividing this figure (29.7") by MAC (180.7") yields a CG of 16.4% of MAC.

17-(4)

Zero fuel weight is the basic operating weight plus the payload of cargo and passengers. On this flight, the zero fuel weight is:

Basic operating weight	88,350 lbs.
Payload	
Zero fuel weight	107,930 lbs.

18---(3)

The ramp weight reduced by the estimated weight of fuel consumed (including taxi fuel) from JFK to ORD (17,100 lbs.) results in an estimated landing weight of 125,330 lbs.

19-(4)

The difference between basic operating weight (88,350 lbs.) and maximum zero fuel weight (118,000 lbs.) is the maximum payload (29,650 lbs.).

20-(3)

See the FAA Pilot's Weight and Balance Handbook, AC 61-13, for the definition of "basic operating weight" and other aircraft weight terms.

21---(4)

FAR 121.189 states, "The accelerate-stop distance must not exceed the length of the runway plus the length of any stopway." In this case the accelerate-stop distance is 9,000 feet (runway length) plus 2,000 feet (stopway length).

22—(3)

FAR 121.189 states, "The takeoff distance must not exceed the length of the runway plus the length of any clearway except that the length of any clearway included must not be greater than one-half the length of the runway." Since in this case the runway length is 9,000 feet and the clearway length is 5,000 feet, the "length of the runway plus half the length of the runway" figure must be used (9,000 feet plus 4,500 feet).

23-(3)

You may use either a computer, or Figure 11 in the Appendix, to plot the problem. The crosswind component is 15 knots from the left.

24—(4)

The intersection of 31,000 feet and -36° C. falls on the ISA +10° C. line.

25—(4)

Enter the chart on altimeter setting line (29.60); plot a line to the right to station elevation (410 feet); plot a line down to station pressure (29.2 in. Hg).

26-(4)

On the stabilizer trim setting chart, read 7 units airplane nose up at the intersection of 20% MAC and 25° flaps.

27—(1)

From Figure 13, determine that the station pressure at Kennedy (see Figure 32 for field elevation) is approximately 30 inches Hg. In Figure 14, no EPR bleed corrections are required since engines 1 and 3 have air conditioning "ON" and engine 2 has "NO BLEED." Read the average takeoff EPR at the intersection of 45° F. OAT and 30 inches Hg.

28---(1)

Enter the chart at the bottom of the "Pressure Altitude" column (600 feet is in the -1,000 to 1,000 ft. box); move right to first temperature box (-60° F. to $+90^{\circ}$ F.); move down to box opposite 25° flaps; interpolate chart to determine $V_{\rm R}$ and V_2 for 156,000 lbs. $V_{\rm R}$ is 121 knots and V_2 is 137 knots; add 1 knot to each since the CG is forward of 14%.

29---(3)

Choice 1-Incorrect; higher than standard temperatures have no effect on climb limit weight.

Choice 2 – Incorrect; a headwind increases the effective runway length.

Choice 3 - Correct; from an inspection of the "runway limit temperature correction" portion of the chart, it is evident that higher than standard temperatures decrease allowable brake release gross weight.

Choice 4 – Incorrect; the reverse is true a runway down-slope will tend to balance the effect of a tailwind.

30—(1)

Refer to the Explanation of Takeoff Performance Chart, Figure 15, and to the Takeoff Performance Chart, Figure 16, of the Appendix. Note that the runway limit gross weight must be reduced by 1,500 lbs. since the CG is forward of 14%.

31---(3)

Follow the directions given in Explanation of Takeoff Performance Chart. Note that the

climb limit value may be the limiting weight in some cases.

32-(1)

Enter the chart on the "trip distance" base line and use the plotted lines on the chart as a guide. Read trip fuel on the right of the chart and trip time on the upper left, applying the ISA $\pm 10^{\circ}$ C. correction.

33---(4)

From the "TIME AND FUEL FROM BRAKE RELEASE TO CLIMB SPEED" chart (Figure 18, Appendix), read 840 lbs. of fuel used. From the "EN ROUTE CLIMB START CLIMB WT" chart (Figure 19, Appendix) read 3,640 lbs. of fuel used in climb to FL-330. The brake release weight less 4,480 lbs. yields a weight of 146,020 lbs. at FL-330.

34-(1)

True airspeed is 486 knots and fuel flow is 8.86 thousands of pounds per hour, therefore:

NAM/1,000 pounds
$$=\frac{486}{8.86}=54.8$$

35-(1)

An inspection of the Indicated Mach .82 Cruise planning charts for 28,000 feet, 29,000 feet, and 31,000 feet reveals that either increases in altitude or decreases in temperature result in lower fuel flow.

36-(4)

Follow the instructions on your computer for determining Mach number from true airspeed and temperature. Most computers have a Mach index.

37—(2)

Determine 298 knots calibrated airspeed by setting up your computer with the given altitude, temperature, and true airspeed.

38-(1)

In regard to "holding fuel," FAR 121.645 states that a flag air carrier turbojet must carry sufficient fuel to "... fly for 30 minutes at 1,500 feet above the alternate airport ... under standard temperature conditions." On the "HOLDING ALL ENGINES—2 AIR-BLEEDS" chart (Figure 24, Appendix) determine that the fuel flow per engine under the stated conditions is 2,140 pounds per hour. Fuel for the required 30 minutes holding is

$$\frac{3 \times 2,140}{2} = 3,210$$
 pounds.

39-(3)

Referring to the "DESCENT PLAN-NING" chart (Figure 18, Appendix) determine that 15 minutes and 600 lbs. of fuel are required to descend from FL-290. For a 1715Z landing, the descent must start at 1700Z. According to the "CRUISE PLAN-NING" chart (29,000 ft.) (Figure 22, Appendix) the average total fuel flow at an average gross weight of 145,000 lbs. for 1 hour (1600Z to 1700Z) is approximately 9,400 lbs. The landing weight is 150,000 lbs. minus (600 lbs. +9,400 lbs.) or 140,000 lbs.

40-(1)

Enter the referenced chart (Figure 25, Appendix) on the "runway available" line at 7,200 feet on the upper left and use the sample plot as a guide. The uncorrected field length limit is 206,000 pounds. Applying the appropriate "field length weight correction" of -54,000 lbs. results in a field length limit gross weight of 152,000 lbs.

41-(3)

Refer to the En route High Altitude Chart Legend (Figure 26, Appendix) for the the "hemispheric rule".

42-(3)

Choice 1-Incorrect; Precision Approach Radar (PAR) is not available; however, Airport Surveillance Radar (ASR) is installed.

Choice 2-Incorrect; the distance from the final approach fix to the localizer missed approach point is 5.2 NM.

Choice 3-Correct.

Choice 4 – Incorrect; only runways 18, 22, 9R, and 27L have Visual Approach Slope Indicators (VASI) installed.

43—(1)

An aeronautical computer can be used to determine density altitude.

44—(3)

The wind flow is parallel to the heavy black contour line. At the point described (50°N./ 43°W.) the angle this contour makes with the nearest meridian is approximately 230 degrees. The wind velocity can be determined to be approximately 75 knots by noting the relationship of the described point to the 80K dashed isotach.

45—(4)

The reference for this sample test item is Aviation Weather, AC 00-6. Choices A and C are correct.

Choice B-Incorrect; the average height of the tropopause is greater in the tropics than in the polar regions.

Choice D-Incorrect; the boundary between the troposphere and stratosphere is characteristically a region of turbulence.

46—(2)

The reference is Aviation Weather, AC 00-6. Choice 1-Incorrect; the reverse of the statement is true, *e.g.*, the strength of the jet streams is greater in winter than in summer.

Choice 2 - Correct.

Choice 3-Incorrect; the maximum occurrence of clear air turbulence is below and on the cold side of a jet stream.

Choice 4 – Incorrect; the core of strongest winds in jet streams is usually between 25,000 and 40,000 feet.

47-(2)

The Tropopause Vertical Wind Shear Chart, when used in conjunction with the 300-millibar chart, provides wind and temperature information—vertically and horizontally—within the layer from 300 millibars to 150 millibars. The following data is shown on this chart:

- 1. Intersections of the tropopause in 50millibar intervals from 300 to 150 millibars. Standard heights of the pressure surfaces are given in the inset box at the bottom of the chart.
- 2. Mean vertical wind shear for the layer from 300 to 150 millibars at intervals of 2 knots/1,000 feet, shown by dashed lines. The mean vertical wind shear is

an arithmetic mean of the forecast values of the shear below and above the layer of maximum wind. It is not drawn for values less than 2 knots.

3. Tropopause and 150-millibar level temperatures are enclosed in rectangles and circles, respectively.

At the position stated $(52^{\circ}N./30^{\circ}W.)$, the tropopause is at the 250-millibar level (34,000 feet under standard conditions); therefore, at FL-330, the aircraft is below the tropopause.

48—(2)

Choice 1 – Incorrect; the symbol indicates moderate turbulence near LFPO.

Choice 2 – Correct.

Choice 3-Incorrect; in the vicinity of 52°N./30°W., the flight will encounter cirroform layers with tops at 32,000 feet. Choice 4-Incorrect; a cold front is expected in the described area. A low pressure center would be labelled "L".

49---(3)

The true airspeed for cruise is 475 knots and the groundspeed, with the cruise wind factor added, is 555 knots. The time for climb (200 NM) is 30 minutes and that for descent (115 NM) is 20 minutes, totaling 50 minutes for 315 NM. The time required for the cruise portion (2,285 NM) at 555 knots is 4 hours and 8 minutes. The total estimated flight time is, therefore, 4 hours, 58 minutes.

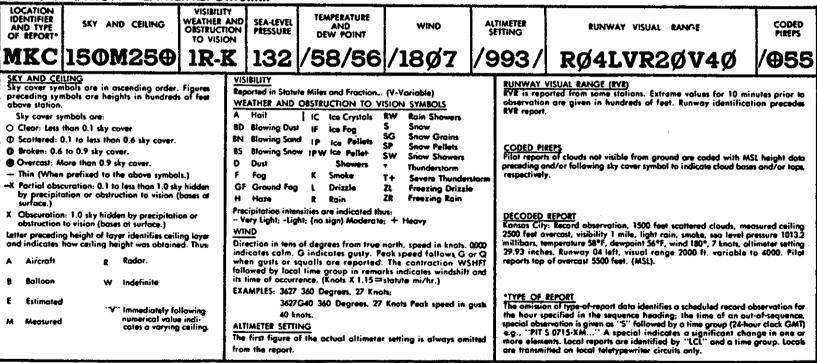
50-(3)

Climb	14,000 lbs./hr.	00 :30	7,000 lbs.
Crulse	11,500 lbs./hr.	04 :08	47,500 lbs.
Descent	6,000 lbs./hr.	00:20	2,000 lbs.
Total	Fuel Burn		56,500 lbs.

APPENDIX

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KEY TO AVIATION WEATHER REPORTS



KEY TO AVIATION WEATHER FORECASTS

SIGMET or AIRMET warn airmen in flight of potentially hazardous weather such as TERMINAL FORECASTS contain information for specific airports on ceiling, cloud heights, cloud squall lines, thunderstorms, fog, icing, and turbulence. SIGMET concerns severe amounts, visibility, weather condition and surface wind. They are written in a form similar to the and extreme conditions of importance to all aircraft. AIRMET concerns less severe AVIATION WEATHER REPORT. conditions which may be hazardous to some aircraft or to relatively inexperienced pilots, Both are broadcast by FAA on NAVAID voice channels. CELLING: Identified by the letter "C" CLOUD HEIGHTS: In hundreds of feet above the station (ground) WINDS AND TEMPERATURES ALOFT (FD) FORECASTS are computer prepared forecasts of wind direction (nearest 10° true N) and speed (knots) for selected CLOUD LAYERS: Stated in ascending order of height Right levels. Temperatures aloft (*C) are included for all levels (2.2500 H. above VISIBILITY: In statute miles, but omitted if over 8 miles station elevation) except the 3000-foot level. SURFACE WIND: In tens of degrees and knots; omitted when less than 10. EXAMPLES OF WINDS AND TEMPERATURES ALOFT (FD) FORECASTS: FD WIC 121745 BASED ON 121200Z DATA EXAMPLE OF TERMINAL FORECASTS VALID 1300002 FOR USE 1800-03002. TEMPS NEG ABY 24000 O11/2GF Clear, visibility one and 6000 9000 12000 18000 24000 30000 34000 39000 C15@ Ceiling 1500', broken clouds FT 3000 one-half miles, ground fog. 805 3127 3425-07 3420-11 3421-16 3516-27 3512-38 311649 292451 283451 Scattered clouds at 2000'. FK 3026 3327-08 3324-12 3322-16 3120-27 2923-38 284248 285150 283749 ceiling 7000' overcast, visibility CSX1/45+ Sky obscured, vertical visibility 500 ft. 20/0C70@6K 3230G 6 miles, smoke, surface wind visibility one-fourth mile, heavy snow. At 6000 feet ASL over JFK wind from 330° at 27 knots and temperature minus 8° C. 320 degrees 30 knots, gusty. AREA FORECASTS are 12-hour forecasts plus 12-hour OUTLOOKS (18 hour autlook in FA valid at 13002) of cloud, weather and frontal conditions for an area the size of several states. Heights of cloud tops, icing, and turbulence are ASOVE SEA LEVEL (ASL); ceiling heights, ABOVE GROUND LEVEL (AGL); bases of cloud loyers are ASL unless indicated. Area Forecasts are amended by SIGMETs PILOTS report in-flight weather to nearest FSS or AIRMET L

FIGURE 1b. Key to aviation weather forecasts.

STATION IDENTIFIERS

ALB ALBANY, N.Y. ALBANY AIRPORT BOS BOSTON, MASS. LOGAN AIRPORT J.F. KENNEDY INTERNATIONAL AIRPORT, N.Y. JFK EWR NEWARK, N.J. NEWARK AI-RPORT PHL PHILADELPHIA, PA. PHILADELPHIA INTERNATIONAL AIRPORT CLE CLEVELAND, OHIO, CLEVELAND-HOPKINS AIRPORT DTW DETROIT, MICH. DETROIT-METROPOLITAN AIRPORT FWA FORT WAYNE, IND. BAER FIELD CHICAGO, ILL. CHICAGO-O'HARE INTERNATIONAL AIRPORT ORD MKE MILWAUKEE, WIS. GENERAL MITCHELL FIELD

FA JFK 081245 132 SAT-012 SUN

ERN PA SERN NY CONN NJ CSTL WTRS

HGTS ASL UNLESS NOTED

SYNS. HI PRES CNTRD UPR HUDSON VLY RDGD SWD OVR N.J. CNTR WILL CONT MOVG EWD.

CLDS AND WX. SERN PA SRN NJ ADJ CSTL WTRS 2500 BCMG BY 17Z 3002000V⊕ AND BY 22Z 25-300120⊕. NERN PA SERN NY W OF THE CTSKLS 250-0V0 LWRG GRDLY TO 120⊕. RMNDR SERN NY CONN NRN NJ RMNDR CSTL WTRS 250-0V0 BCMG 2000V⊕

ICG. FRZLVL AT OR NR SFC. LGT RIME ICGIC.

TURBC. BCMG OCNL LGT BLO 60 BY 16Z

OTLK 01Z-19Z SUN. AREA OF MXD PCPN SPRDG INTO W AND S PTN SERN PA AND EXTRM S NJ 02Z-05Z SPRDG NEWD THRU RMNDR AREA BY 12Z AS SNW THRU EXTRM NERN PA INTO SERN NY AND CONN AND MXD PCPN RMNDR AREA. CIGS LWRG RPDLY IN PCPN TO 3-10 VSBYS 1-3 VRBL BLO 1 IN SNW. AFT 1 TO 3 HRS OF SNW CIGS AND VSBYS FQTLY BLO 1 THSD FT AND 2 MIS.

FIGURE 2. Station identifiers and area forecast (JFK).

FA CHI 081245 132 SAT-012 SUN

WIS MICH ILL IND LK MICH US PTNS OF LKS SUPR AND HURON

SYNS. AT 13Z LOW PRES CNTR OVR SERN MO MOVG EWD ABT 25 KTS ACRS SRN ILL AND RCHG ERN KY BY 01Z.

AIRMET. SNW AND FOG WITH CIGS BLO 1 THSD FT AND VSBY BLO 2 MI OVR SRN THIRD WIS AND NW THIRD ILL SPRDG EWD INTO NERN THIRD ILL BY 18Z NWRN IND BY 19Z AND NERN IND BY OOZ. OVR RMDR ILL CIGS BLO 1 THSD AND VSBY BLO 2 MI IN RAIN AND FOG. SRN THIRD WIS AND NW THIRD ILL FQT MDT MXD RIME AND CLR ICGICIP SPRDG EWD. ELSW IN ILL IND SRN LK MICH AND SRN LWR MICH LCL MDT RIME ICGIC GRDLY INCREASING.

CLDS AND WX. WIS. OVR SRN THIRD WIS C8-15@1-3S-F LCLLY C4X1/2S-F. CONDS IMPVG AFT 21Z TO C15@3-5SW-F. OVR RMDR WIS C60@.

MICH. C20-30 \oplus 7 LCLLY OVR NRN LWR MICH NRN LK MICH AND UPR MICH C10 \oplus 3-55W-. CONDS LWRG OVR SRN LK MICH AND SRN LWR MICH TO C15 \oplus 3-5S- CHC ZL- BY 20Z.

ILL. OVR NWRN THIRD ILL C8-15#1-3S-F LCLLY C4X1/2S-F. THESE CONDS WILL SPRD EWD OVR NERN THIRD ILL BY 18Z PRCDD BY C15-25#3-5HK. TOPS 80-100 INCRG TO 140-160 DURG THE PRD IN MERGG LYRS. CONDS IPVG OVR NWRN ILL AFT 21Z TO C15#3-5SW-F. OVR RMDR ILL C4-8#1-3R-F LCLLY C2-4X1/4-1R-F. OVR CNTRL ILL CHC MXD ZL- OR S-.

IND. C15-25 \oplus 3-6HK LCLLY OVR W CNTRL AND SWRN IND R-. CONDS GRDLY DTRTG FM W AND BCMG OVR NWRN IND C5-8#1/2-2S- BY 19Z AND OVR NERN IND BY 00Z.

ICG. OVR SRN WIS NWRN ILL FQT MDT MXD RIME AND CLR ICGICIP. OVR RMDR ILL IND SRN LK MICH AND SRN LWR MICH LCL MDT RIME ICGIC GRDLY INCRG DURG THE PRD. FRZLVL 40-60 SRN IND SRN ILL SLPG TO 20-40 CNTRL IND CNTRL ILL AND TO SFC-20 NRN IND NRN ILL. FRZLVL AT SFC LK MICH AND MICH.

TURBC. OVR ILL AND IND LCL MDT TURBC DVLPG 8 THSD BY 18Z.

OTLK 01Z-19Z SUN. LOW CNTR CONTG TO MOV EWD AS HIGH PRES SPRCS INTO WIS AND ILL BY 06Z. CONDS GRDLY BCMG $@V \oplus OVR$ ILL BY 06Z AND OVR IND MICH AND LK MICH BY 19Z.

FIGURE 3. Area forecast (CHI).

FTUS 081045
ALB 081123 250-D. 21Z C200D..
BOS 081123 250-D 3610..
JFK 081123 250-D. 21Z C200⊕ 0610..
EWR 081123 250-D4K OCNL 3K. 16Z 250-D7. 21Z C200⊕8 0610..
PHL 081123 250-D4K. 16Z C200D7. 21Z C30D140⊕ 0510..
FWA 081123 C18⊕3HK 0810 CHC S- AFT 14Z. 20Z C8⊕2S-F 0415 OCNL C5X1/2S-F..
ORD 081123 C18⊕4HK 0710 CHC S-. 16Z C8⊕2S-F 0415 OCNL C5X1/2S-F..
MKE 081123 C12⊕5HK 0710 CHC S-. 17Z C8⊕2S-F 0415 CHC C5X1/2S-F..
CLE 081123 250-Ф6HK. 17Z C80⊕7 0812. 22Z C8⊕2S- 0812 VRBL C5X1S..

SA 22081500

ALB 280-020 243/18/11/0000/022 BOS M JFK 300012 227/31/15/0605/019 EWR 300-08 227/32/15/0106/019 PHL E25005HK 217/31/18/0907/016 CLE 250E'00021/2HK 150/29/20/1107/993 DTW M28011/2HK 152/27/21/0508/994/R03LVR60+ FWA M19021/2H 125/29/23/0709/986 ORD M1401S-HK 138/34/27/0409/991/R14RVR60+ MXE M15002806HK 155/32/28/0410/995

SA 22Ø816ØØ

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ALB 28Ø-02Ø 23Ø/22/13/Ø3Ø5/Ø18
BOS 3ØØ-012 215/31/11/25Ø7/Ø16
JFK 3ØØ-012 217/32/15/1ØØ3/Ø16
EWR 3ØØ-06HK 217/34/17/Ø5Ø5/Ø16
PHL E25Ø€6HK 213/34/18/Ø9Ø7/Ø15
CLE M1Ø08Ø€21/2HK 148/32/24/11Ø6/993
DTW S 3Ø0E22Ø€11/2HK 148/29/22/Ø8Ø9/993/RØ3LVR6Ø+
FWA M19€21/2H 117/3Ø/24/Ø71Ø/984
ORD S M11€3S--HK 135/33/28/Ø311/99Ø
MKE M17€6HK 155/32/28/Ø3Ø9/995
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FIGURE 4. Terminal forecasts and surface weather reports.

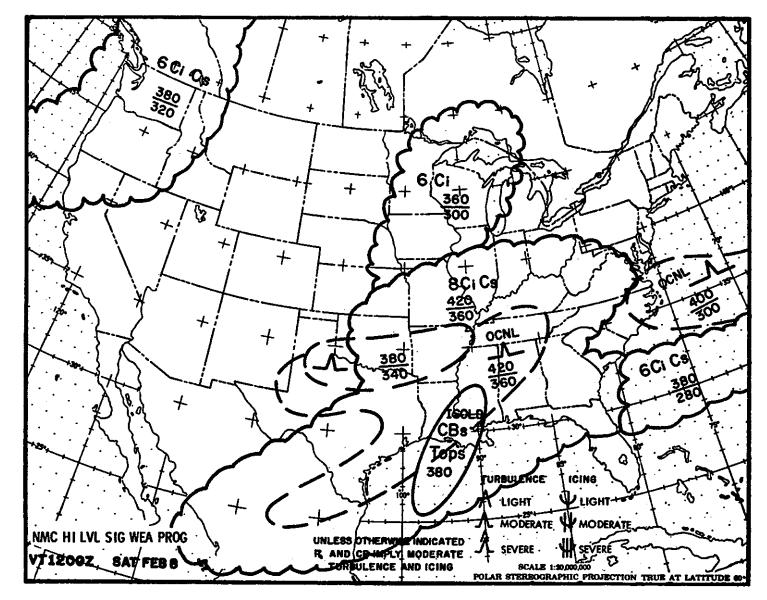


FIGURE 5. NMC high level weather prognostic chart.

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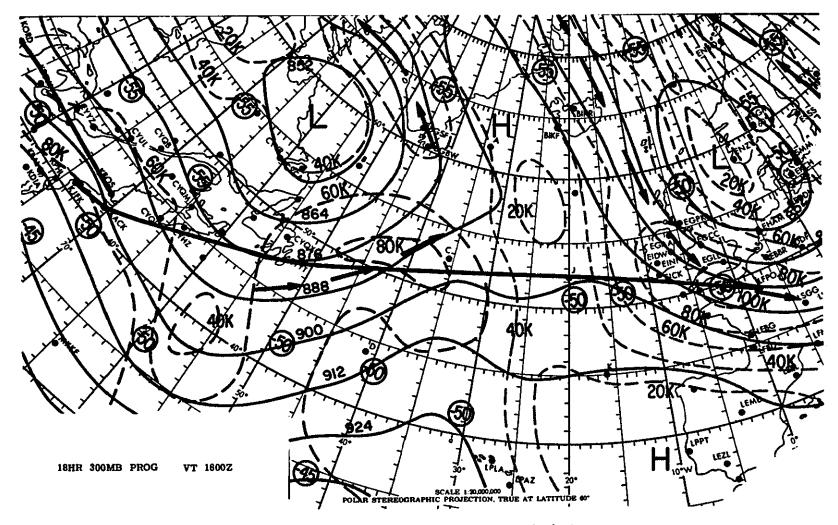


FIGURE 6. 300 millibar prognostic chart.

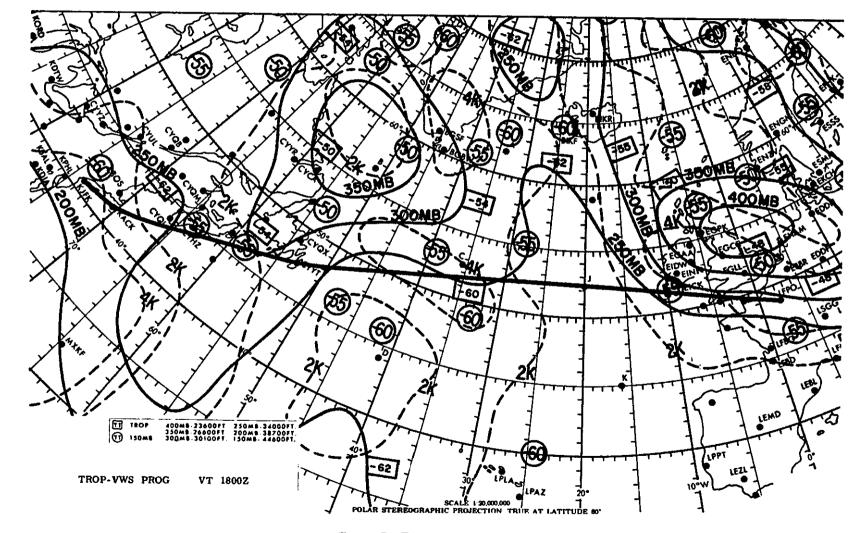


FIGURE 7. Tropopause/vertical wind shear chart.

30

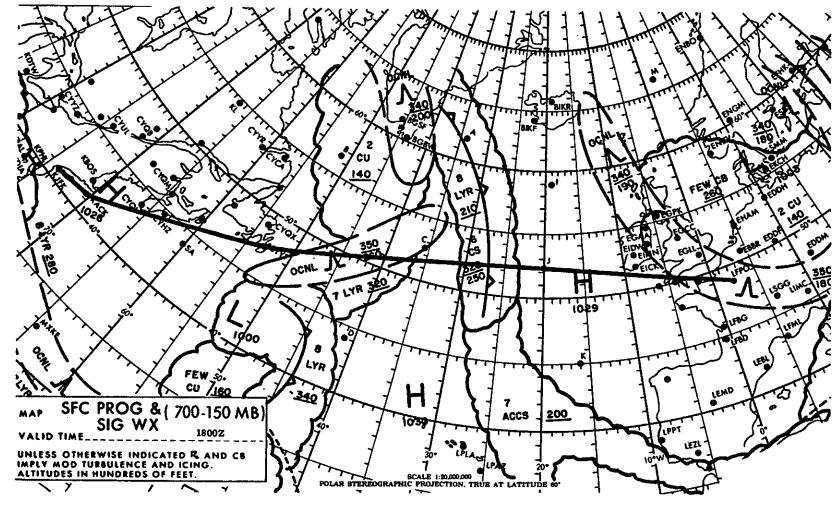
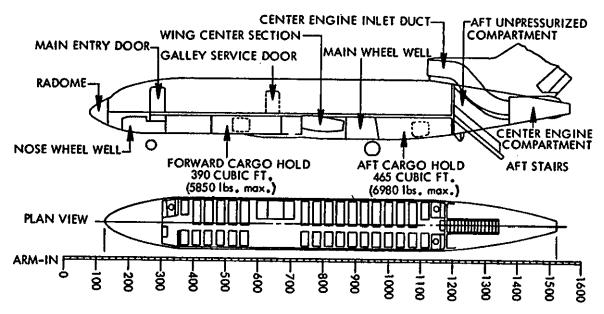


FIGURE 8. Significant weather prognostic chart (700-150 MB).



WEIGHT LIMITATIONS

Basic operating weight	
Maximum taxi weight	
Maximum takeoff weight	
Maximum landing weight	
Maximum zero fuel weight	

FUEL LOADING TABLE									
Weight	Tank	1 and 3		nk 2 Cell)	Weight				
Lb.	Arm	Moment 1000	Агт	Moment 1000	Lb.				
8,500	892.1	7,583	817.5	6,949	8,500				
9,000	893.0	8,037	817.2	7,355	9,000				
9,500	893.9	8,492	817.0	7,762	9,500				
10,000	894.7	8,947	816.8	8,168	10,000				
10,500	895.4	9,402	816.6	8,574	10,500				
11,000	896.1	9,857	816.5	8,982	11,000				
11,500	896.8	10,313	816.3	9,387	11,500				
12,000	897.5	10,770	816.1	9,793	12,000				
18,500	906.8	16,776	815.1	15,079	18,500				
19,000	907.8	17,248	815.0	15,485	19,000				
19,500	908.9	17,724	814.9	15,891	19,500				
20,000	910.1	18,202	814.9	16,298	20,000				
20,500	911.7	18,690	814.8	16,703	20,500				
21,000	913.4	19,181	814.7	17,109	21,000				
21,500	915.5	19,683	814.6	17,514	21,500				

AIRPLANE DATUM CONSTANTS

Mean Aerodynamic Chord applicable to this airplane:

MAC = 180.7 inches

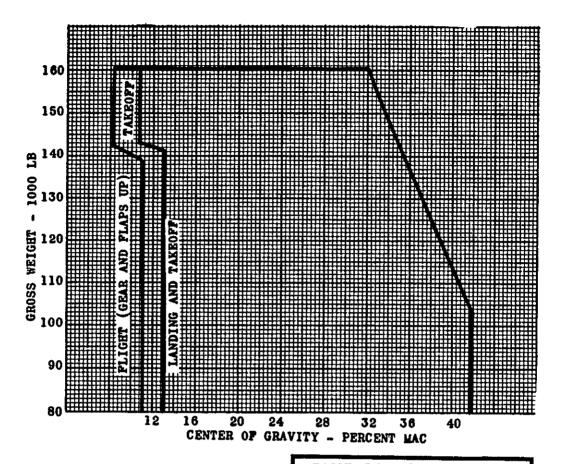
Leading edge of MAC = 860.2 inches

FUEL DUMPING

Fuel dumping rates with all boost pumps on, all dump valves open, and both nozzle valves open are:

Tank	#	1 600	lbs/min
Tank	#	21100	lbs/min
Tank	#	3 600	lbs/min

FIGURE 9. Airplane data.



CARGO LOADING TABLE									
	Moment 1000								
	Forward Hold	Aft Hold							
Weight	Arm	Arm							
<u> </u>	581	1066							
6,000		6,396							
5,000	2,905	5,330							
4,000	2,324	4,264							
3,000	1,743	3,198							
2,000	1,162	2,132							
1,000	581	1,066							
900	523	959							
800	465	853							
700	407	746							
600	349	640							
500	290	533							
400	232	426							
300	174	320							
200	116	213							
100	58	107							

PASSENGER LOADING TABLE									
Number	Weight	Moment							
of	_	1000							
Pass.	Lb.								
FWD.	COMP. CENTR	OID 486.3							
5	850	418							
10	1,700	827							
15	2,550	1,240							
20	3,400	1,653							
25	4,250	2,067							
29	4,930	2,397							
AFT. C	OMP. CENTRO	DID 928.8							
5	850	789							
10	1,700	1,579							
15	2,550	2,368							
20	3,400	3,158							
25	4,250	3,947							
30	5,100	4,736							
35	5,950	5,528							
40	6,800	6,315							
45	7,650	7,105							
50	8,500	7,894							
54	9,180	8,526							

FIGURE 10. CG chart and loading tables.

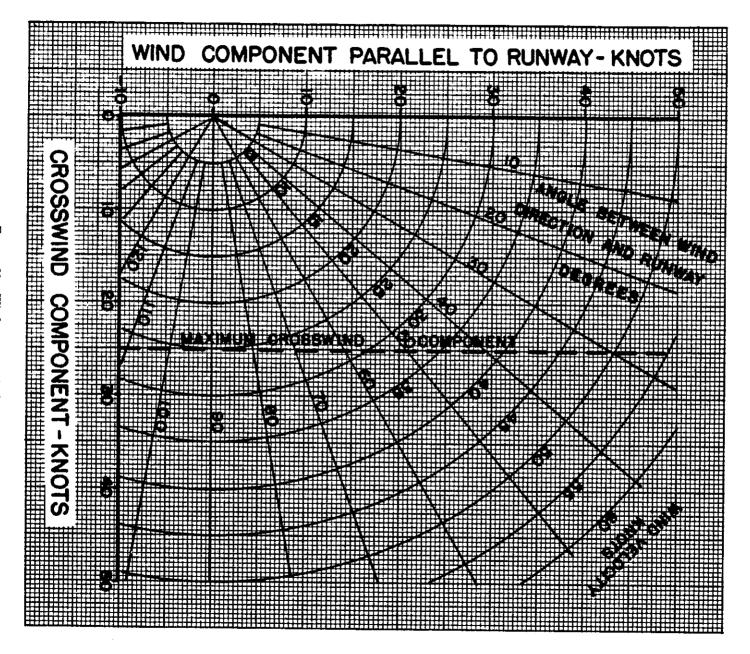


FIGURE 11. Wind component chart.

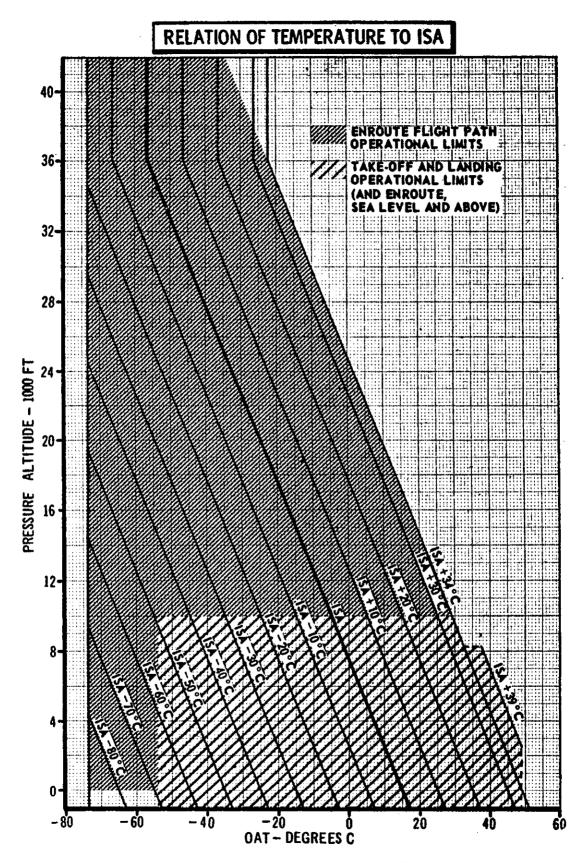


FIGURE 12. Relation of temperature to ISA.

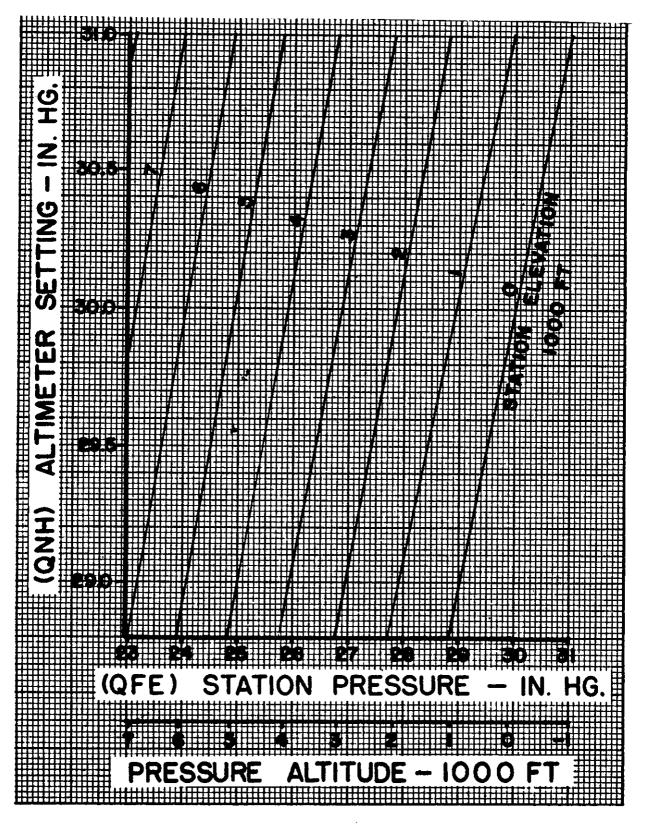


FIGURE 13. Relation of altimeter setting to station pressure.

TAKEOFF

					2IG 1 &	3 - A/	c our	·······	EPR REED CORRECTIONS				ENG 1	63	ENG 2 ON:03	
EPR				3	NG 2 - 1	NO BLE	Ð		ENDINE ANTI-ICE ON						03	
<u> </u>			A.T.		23	STA 24	TION 25	PPES 26	SUPE	INCI 28	ies Ho 29	30	31	32		.T. F.
	ł		F.	22		(((1	1.93	1.93		1.92	1.89	70	· · ·
	ŀ	70		1.97	1.96		<u>1.96</u>	1	1.94				T	1	6	
		• 60	<u>, (</u>		1.96	ii	<u>1.96</u>		1.95		1.95		1.92		1	
		5()	1.98	1		1.98			1.98	1.98		1.92		5	
		4()	2.02	2.02				2.02		4	r i i i i i i i i i i i i i i i i i i i		1	40	
		30)	2.05	2.05			12.05	2.05	2.04	12.00	1.95	11.92	<u>h 89</u>	1 3	<u> </u>
	ſ	VI	= V,	- V	2	PRESSU ALTITU -1000	DE				DAT					
	Ļ				<u> </u>	9 70 1	<u>د</u> ه	č			-60 TO -51 TO		-24 10		21 -6	TO 88 TO 29
			-SKID /			7 10 9	5	r c	-60 10		-30 10 -34 10		11 70	o l		70 100 70 38
	-					5 20 1		ř	-60 10	5	6 TO -14 TO	40	5 TO	75	76	TO 108 TO 42
						3 70 5	- LA '	7	-60 TO	35	36 TC 3 TO	65	66 TO 19 TO	102	103	TO 115 TO 46
AFTE	R TAK	TAFE	NOP			1 70 3	- 4	ř.	-60 10 (50	61 10	95	96 10	120		<u></u>
	NEUVER	RING	SPEE			1 70 1	5	Ť	-51 TO	x	17 70 91 70	120	36 10			
		<u>is ia</u> Hax i	ABOVE	XAX			<u></u>	<u>c</u>	╝╔	8		49				F
PLAPS	LANDING	I VT	LANDIN	THE		FLAPS	WEIGH		V1=VR	V2	V1-VR	V ₂	VI=VR	V2	V1=1	R V2
0 2 5 15 25	200 190 160 150 140		210 200 170 160 150			5*	17 16 15 14 13		$\begin{array}{c} 1^{144} & 10 \\ 139 & 10 \\ 134 & 11 \\ 126 & 10 \\ 122 & 10 \end{array}$	60 56 51 47 42	145 140 136 130 124	160 155 151 146 141	147 142 137 132 126	159 155 150 146 141	138 134 128	149 145 140
AFTER	NAMEUVER	T EXO	DED HIG	15*			12		116 1 109 1	37 32		136 131		136 131	122 116	135 130
V ₂ +	NK, MAINT	IAKF-	OFT FLA			15*	17 16 15 14 13 12 11	ما 10	$\begin{pmatrix} 131 & 1\\ 127 & 1\\ 121 & 1\\ 116 & 1\\ 110 & 1 \end{pmatrix}$	50 47 42 39 34 30 26	133 128 123 117 112	149 146 142 138 133 129 125	129 124 119 113	145 141 137 132 128 123	125 120 115 109	136 131 127 122
N ₁ N ₂ STA	RPM - 10 RPM - 10 ARTING EC 420°C ABC 350°C BEI X CONT EC (E-OFF EC	00.1% 00.0% 7 DVE 19 LOW 19	5°C OAT			25*	16 15 14 13 12 11		$ \begin{array}{c} 118 & 1 \\ 113 & 1 \\ 108 & 1 \\ 103 & 1 \end{array} $	38 35 31 27 22 18	119 115 110 104 99	137 133 130 126 122 117	111 106 100	133 129 125 121 116	117 112 107 102	128 124 120 115

											r n									
40 43	Π	31	14	34	32		30	71	26	24	22	20	10	t6	14	12	10	,]	Co	
1511	- 6	12	12%	25	34	11	14	4.	45	5	55		43		71		iV.			
vi v	Т	12	34	3	15	Т	4	4%	5	55	•	67	7	75	r i	15			13	FLAPS
13 1	T	Σ	25	3	1	Т	34	44	5'+	54	6%	7	75		15	7	Å,	•	125	
	1	2	24 IRE	3	<u>.</u>	J.	<u>.</u>	4	5.	54	674	7	75		8 5		Η.	•	25	r Eart

FIGURE 14. Takeoff data.

Takeoff Performance Flaps 15°---Gross Weight at Brake Release

Given :

Runway length available	8,150 feet
Tailwind component	4 knots
Slope	1% UP
Airport pressure altitude	4,000 feet
Outside air temperature	+88°F.
Average takeoff EPR	1.88

For runway limit: start at runway length available line and follow dotted line and arrows. Answer is 132,000 pounds.

For climb start: start where EPR 1.88 intersects climb limit baseline and follow dotted line and arrows. Answer is 143,500 pounds.

Explanation of Chart

Gross weight at brake release for this aircraft under the conditions specified on the chart is seen to be influenced by either *Runway* or *Climb* limitations. The following explanation of these limits is offered.

Runway Limit. Based on the runway length available, the operating variables (wind, runway slope, pressure altitude, and temperature) together with average takeoff EPR directly influence the gross weight at brake release. In the plotted example, this value is seen to be 132,000 lbs.

Climb Limit. Regulations specify that certain climb gradients or profiles must be met during the takeoff and climb phases. (See FAR Part 23.121.) This chart, therefore, shows the weights for various combinations of power (EPR) and pressure altitude at which the aircraft is able to equal the prescribed climb gradients. In the plotted example, the gross weight at brake release which would meet the climb restrictions imposed by the regulations is 143,500 pounds. This value far exceeds the runway limit of 132,000 pounds. Of these two weights, the smaller value is always used and in most cases, this is the runway limit. Structural weight limitations, of course, must not be exceeded.

Note.--Under certain conditions, the climb limit gross weight at brake release will be *less* than the runway limit value and thus it becomes the limiting value. Plot the example below which shows this relationship:

Given	
Runway length available	9,250 feet
Wind	Calm
Slope	0
Airport pressure altitude	3,000 feet
Outside air temperature	+59°F.
Average takeoff EPR	1.90

Solution

Runway limit gross weight at brake release	156,000 pounds.
Climb limit gross weight at brake release	153,500 pounds.

(The climb limit value is the limiting weight in this case.)

FIGURE 15. Explanation of takeoff performance chart (fig. 16).

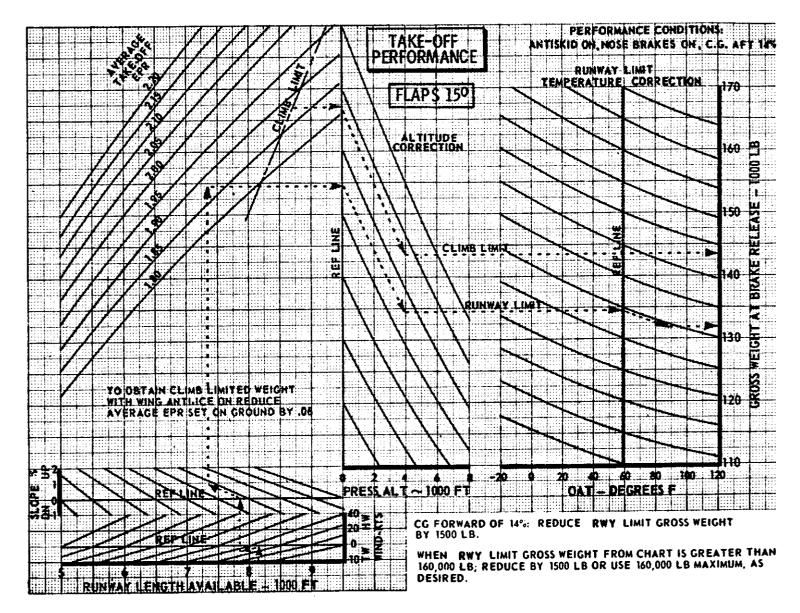


FIGURE 16. Takeoff performance chart.

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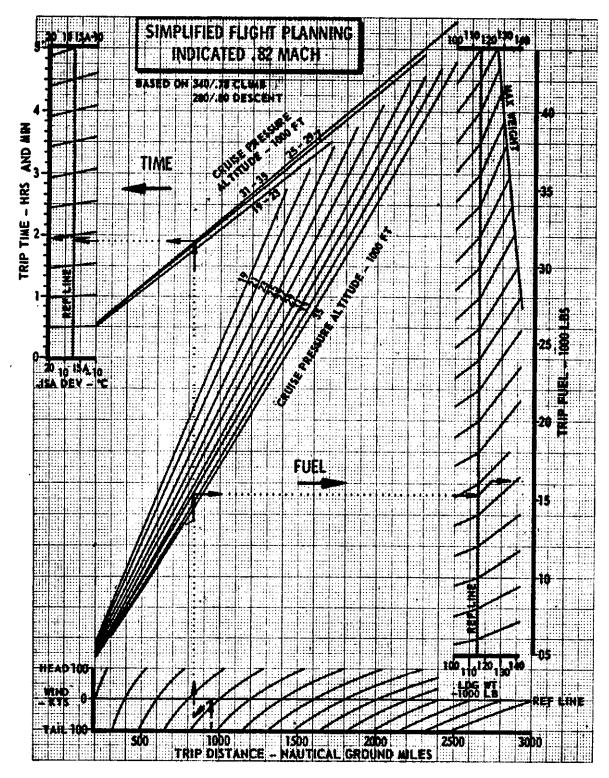


FIGURE 17. Simplified flight planning chart.

	340/	.78 CLIM	В		
GROSS WT		FIELD EL			
-1000 LB	S.L. FUEL LB		4000 FT FUEL LB	6000 FT FUEL LB	
170	1030	1090	1150	1210	TIME AND FUEL
165	970	1020	1080	1140	FROM BRAKE RELEASE
160	920	980	1030	1090	_
155	880	930	980	1030	TO CLIMB SPEED
ſ50	840	890	940	990	
145	800	840	890	940	
140	770	810	850	900	TIME = APPROX 3 MIN
135	740	780	820	870	
130	710	750	790	830	
125	690	720	760	800	
120	660	690	730	770	
115	640	670	700	740	
110	6.20	650	680	710	
105	600	620	650	680	
100	590	610	630	660	

PRESS ALT - 1000 FT	TIME- MIN	FUEL- LBS	DIST- NAM
39	20	850	124
37	19	800	112
35	18	700	101
33	17	650	92
31	16	600	86
29	15	600	80
27	14	550	74
25	13	550	68
23	12	500	63
21	11	500	58
19	10	450	52
17	10	450	46
15	9	400	41
10	6	300	26
5	3	150	13

DESCENT PLANNING

.80/280

FIGURE 18. Climb and descent planning.

PRESS.	CL IMB		0	EVIATI	ON FRO	MISA	- DEGR	FF(C)			
ALT-FT	DATA	- 15	-10	-5	-0	5	10	15	20	25	
40000	TIME MIN FUEL LBS DIST NAM AVTAS KTS							UTE CL]	
39000	TIME MIN FUEL LOS DIST NAM AVTAS KTS	25 5367 177 423	31 6157 218 429				150,				
38000	TIME MIN FIJEL LBS DIST NAM AVTAS KTS	21 4749 145 422	23 5194 167 427	27 5787 197 433	33 6661 243 438						
37000	TIME MIN FUEL LBS DIST NAM AVTAS KTS	18 4397 128 421	20 4752 144 426	23 5190 165 431	26 5748 193 437	31 6506 231 442	39 7663 292 448				
36000	TIME MIN FUEL LBS DIST NAM AVTAS KTS	17 4145 116 420	18 4456 130 425	21 4P30 147 430	23 5289 169 436	27 5871 197 441	32 6647 236 447	39 7775 294 453	52 9742 401 459		
35000	TIME MIN FUEL LBS DIST NAM AVTAS KTS	15 3952 107 420	17 4236 120 425	19 4574 135 430	21 4 983 1 54 4 35	24 5491 178 440	28 6144 209 446	34 7031 253 451	42 8354 321 458		
34000	TIME MIN FUEL LBS DIST NAM AVTAS KTS	14 3788 100 419	16 4053 112 424	18 4366 125 429	20 4740 142 434	27 5199 163 439	26 5777 190 445	30 6536 226 450	37 7592 279 456	47 9217 362 463	
33000	TIMF MIN FUEL L9S DIST NAM AVTAS KTS	13 3640 94 418	15 3889 104 423	16 4181 117 428	18 4529 132 433	21 4952 151 438	24 5478 175 443	28 6153 207 449	33 7059 250 455	41 8360 314 461	
32000	TIME MIN FUEL LRS DIST NAM AVTAS KTS	13 3503 88 417	14 3739 98 422	15 4014 110 426	17 4340 124 432	19 4734 141 437	22 5220 163 442	26 5834 190 448	30 6639 228 453	37 7754 281 460	
31000	TIME MIN FUEL LBS DIST NAM AVTAS KTS	12 3375 83 415	13 3598 92 420	15 3859 103 425	16 4167 116 430	18 4537 132 435	21 4989 152 441	24 5555 177 446	28 6285 210 452	34 7274 256 458	
30000	TIME MIN FUEL LBS DIST NAM AVTAS KTS	11 3253 79 414	12 3466 87 419	14 3713 97 424	15 4005 109 429	17 4354 124 434	19 4778 142 439	22 5304 165 444	26 5976 195 450	31 6873 236 456	
29000	TIME MIN FUEL LBS DIST NAM AVTAS KTS	11 31 37 74 413	12 3340 82 417	13 3576 92 422	14 3853 103 427	16 4184 117 432	18 4583 134 437	21 5075 155 443	24 5700 182 448	29 6527 219 454	
28000	TIME MIN FUFL LBS DIST NAM AVTAS KTS	10 3025 70 411	11 3219 78 416	12 3444 87 420	14 3708 57 425	15 4021 110 430	17 4397 126 435	20 4858 145 441	23 5441 170 446	27 6207 203 452	

Note: 1. Enter chart at cruise flight level.

2. Subtract 150 lbs. fuel and 1/2 minute for each 1,000 feet that departure airport is above sea level.

FIGURE 19. En route climb chart (150,000 lbs.).

PRESS.	CL IMR		08				DEGR				
AL T-FT	DATA	-15	-10	-5	-0	5	10	15	20	25	
40000	TIME MIN FUEL LBS DIST NAM AVTAS KTS						ENROUTE CLIMB				
39000	TIME MIN FUEL LAS DIST NAM AVTAS KTS										
38000	TIME MIN FUEL LBS DIST NAM AVTAS KTS	23 5246 165 423	27 5865 195 428	34 6255 246 434							
37000	TIME MIN FUEL LBS DIST NAM AVTAS KTS	20 4745 140 472	22 5166 160 427	26 5706 186 432	30 6438 222 438	38 7566 281 444					
36000	TIME MIN FUFL LBS DIST NAM AVTAS KTS	18 4430 125 421	20 4784 141 426	22 5217 161 431	26 5761 187 436	30 6480 222 442	37 7506 274 448	48 9252 366 454			
35000	TIME MIN FUEL LBS DIST NAM AVTAS KTS	16 4203 115 420	18 4521 129 425	20 4903 146 430	23 5372 168 436	27 5969 196 441	31 6765 234 447	39 7920 292 453	52 9922 397 460		
34000	TIME MIN FUEL LAS DIST NAM AVTAS KTS	15 4016 107 419	17 4308 119 424	19 4656 135 429	21 5078 153 434	24 5602 177 440	28 6280 209 445	34 7201 254 451	42 8567 322 458	58 10982 448 465	
33000	TIME MIN FUEL LAS DIST NAM AVTAS KTS	14 3849 100 418	16 4121 111 423	1 P 4443 125 42 P	20 4 830 1 42 4 33	22 5306 163 439	26 5909 191 444	30 6701 228 450	37 7806 281 456	47 9505 366 463	
32000	TIME MIN FUEL LRS DIST NAM AVTAS KTS	13 3697 94 417	15 3953 104 422	16 4254 117 427	18 4614 132 432	21 5053 151 437	24 5602 176 443	28 6308 208 448	33 7259 252 455	41 8632 318 461	
31000	TIMF MIN FUEL LAS DIST NAM AVTAS KTS	13 3556 88 416	14 3798 98 421	15 4082 110 426	17 4420 124 431	19 4829 141 436	22 5335 163 441	26 5977 192 447	31 6824 230 453	37 8003 285 459	
30000	TIMF MIN FUEL LBS DIST NAM AVTAS KTS	12 3423 83 414	13 3653 92 419	15 3922 103 424	16 4241 116 429	18 4625 132 434	21 5096 152 440	24 5688 178 445	212	34 7507 260 457	
29000	TIME HIN FUEL LBS DIST NAM AVTAS KTS	11 3298 78 413	12 3517 87 418	14 3772 97 423	15 4074 109 428		143	22 5427 166 443	26 6135 197 449	7092 240	
28000	TIME MIN FUEL LBS DIST NAM AVTAS KTS	11 3177 74 411	12 3386 82 416	13 3629 92 421	15 3915 103 426	117	4671	21 5183 155 441		6715 222	

Note: 1. Enter chart at cruise flight level.

2. Subtract 150 lbs. fuel and 1/2 minute for each 1,000 feet that departure airport is above sea level.

. . .

FIGURE 20. En route climb chart (155,000 lbs.).

PLANNING 3 engines 2 airbleeds

IND. MACH. 82 CRUISE

					ISA=-40.5	DEG C		28	000) FT
GROSS WT	OAT-DEG C	-60	-55	-50	-45	-40	-35	-30	-25	-20
165000 LB	MACH/TAS TOTAL FF	•820/454 9423	.820/469 9558	.820/474 9693	-820/480 9828	-820/485 9963	.820/490 10095	.803/485 9816		
160000 LB	NACH/TAS Total FF	+820/464 9246	.820/469 9381	.820/474 9513	-820/480 9645	.820/485 9777	+820/490 9906	.813/491 9864		
155000 LB	MACH/TAS Total FF	.820/464 9084	.820/469 9216	.820/474 9345	•B20/480 9474	•820/485 9603	.820/490 9732	.820/495 9858		
150000	MACH/TAS	.820/464	.820/469	.820/474	.820/480	.820/485	-820/490	•820/495	-809/493	
18	Total FF	8922	9051	9177	9306	9432	9558	9684	9546	
14500D	MACH/TAS	.820/464	.820/469	.820/474	-820/480	•820/485	.820/490	•820/495	.817/499	
18	Total FF	8775	8901	9027	9150	9276	9399	9522	9585	
140000	MACH/TAS	+820/464	-820/469	.820/474	-820/480	.820/485	-820/490	.820/495	.820/500	•804/495
LB	Total FF	8628	8754	8877	9000	9123	9243	9366	9486	9225
135000	MACH/TAS	.820/454	.020/469	.820/474	-820/480	•820/485	+820/490	.820/495	.820/500	•812/500
L8	Total FF	8490	8610	8733	8853	8973	9093	9213	9333	9258
130000	MACH/TAS	•820/464	•820/469	.820/474	-820/480	.820/485	+820/490	.820/495	.820/500	.819/505
LB	Total FF	8358	8478	8598	8718	8835	8955	9072	9189	9288
125000	4ACH/TAS	.820/464	.820/469	.820/474	-820/480	+820/485	.820/490	.820/495	+820/500	.820/505
LB	Total FF	8226	8346	8463	8580	8697	8814	8928	9045	9159
120000	MACH/TAS	-820/464	-820/469	.820/474	•820/480	.820/485	.820/490	.820/495	.820/500	.820/505
LB	TOTAL FF	8115	8232	8349	8466	8580	8694	8808	8922	9036
115000	MACH/TAS	-820/464	.820/469	.820/474	-820/480	+820/485	.820/490	•820/495	.820/500	.820/505
18	Total FF	8007	8121	8238	8352	8466	8577	8691	8802	8916
110000	MACH/TAS	•820/464	.820/469	.820/474	+820/480	.820/485	.820/490	.820/495	.820/500	.820/505
LB	TOTAL FF	7902	8016	8130	8244	8355	8466	8577	8688	8799
105000	MACH/TAS	.820/454	•820/469	+820/474	.820/480	•820/485	.820/490	.820/495	.820/500	.820/505
LB	TOTAL FF	7806	7917	8031	8142	8253	8364	8472	8583	8691
100000	MACH/TAS	•820/464	•820/469	.820/474	.820/480	.820/485	-820/490	.820/495	.820/500	.820/505
LB	TOTAL FF	7710	7821	7932	8043	8151	8259	.8370	8476	8586

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FIGURE 21. Cruise planning chart (28,000 ft.).

IND. MACH. 82 CRUISE

PLANN ING 3 engines 2 airbleeds

20000 ET

					ISA=-42.5	DEG C		ΖЭ,	JUU	
GRESS WT	DAT-DEG C	-60	-55	-50.	-45	-40	-35	-30	-25	-20
165000 LB	MACH/TAS TOTAL FF	•820/464 9309	+820/469 9444	•820/474 .9576	+520/480 9708	+820/485 9840	-810/484 9723			
160000 LB	MACH/TAS Total FF	.820/464 9120	.820/469 9249	.820/474 9381	.820/480 9510	-820/485 9642	-820/490 9768			
155000 L8	MACH/TAS Total FF	• 820/464 8934	.820/469 9063	.820/474 9189	.820/480 9318	.820/485 9444	.820/490 9570	•808/488 9432		
150000 L9	MACH/TAS TOTAL FF	.820/464 8769	.820/469 8895	+820/474 9021	-820/480 9147	•820/485 9270	.820/490 9393	•818/494 9477		
145000	MACH/TAS	-820/464	.820/469	-820/474 -	-820/480	+820/485	•820/490	•820/495	.8067491	
LB	TOTAL FF	8607	8730	8853	8976	9099	9219	9342	9132	
140900	MACH/TAS	.820/464	.820/469	.820/474	.820/480	.820/485	.820/490	.820/495	.815/497	
La	Total FF	8454	8574	8697	8817	8937	9057	9174	9171	
135000	NACH/TAS	.820/464	*-820/469	.820/474	.820/480	.820/485	-820/490	.820/495	•820/500	.801/493
LB	Total FF	8307	8427	8547	8664	8781	8901	9018	9132	8829
130000	MACH/TAS	.820/464	-820/469	.820/474	-820/480	\$820/485	.820/490	+820/495	-820/500	-810/499
LB	TOTAL FF	8163	8283	8400	8514	8631	8745	8862	8976	8865
125000	MACH/TAS	-820/464	.820/469	.820/474	-820/480	.820/485	.820/490	.820/495	.820/500	.818/504
LB	TOTAL FF	8031	8148	8262	8379	8493	8604	8718	8832	- 8895
120000	MACH/TAS	-820/464	.820/469	.820/474	-820/480	+820/485	.820/490	.820/495	.820/500	•820/505
LB	TOTAL FF	7902	8016	8127	8241	8352	8465	8577	8688	8796
115000	MACH/TAS	.820/464	•820/469	.820/474	.820/480	.820/485	.820/490	-820/495	.820/500	-820/505
LB	Total FF	7788	7899	8013	8124	8232	8343	8454	8562	8670
110000	MÁCH/TAS	-820/464	.820/469	-820/474	.820/480	-820/485	.820/490	#820/495	.820/500	.820/505
L9	TOTAL FF	7677	7788	7899	8007	8118	8226	8334	8442	8547
105000	MACH/TAS	•820/464	.820/469	.820/474	.820/480	-820/485	.820/490	.820/495	.820/500	•820/505
LB	TOTAL FF	•.7575	7683	7791	.7899	8007	8115	8220	8328	8433
100000	MACH/TAS	-820/464	•820/469	.820/474	-820/480	.820/485	.820/490	.820/495	.820/500	.820/505
LB	TOTAL FF	7476	7584	7689	7797	7902	8010	8115	8220	8322

FIGURE 22. Cruise planning chart (29,000 ft.).

IND. MACH. 82 CRUISE

PLANN ING 3 engines 2 airbleeds

					LSA=-40.4	31.	000	FT		
GROSS WT	DAT-DEG C	-65	-60	-55	-50	-45	^c -40	-35	-30	-25
165000 LB	MACH/TAS Tutal FF	•820/458 9015	•820/464 9147	•820/469 9279	-820/474 9411	•820/480 9540	•015/482 9558			
160000 LB	MACH/TAS Total FF	.820/458 8802	.820/464 8931	•820/469 9060	•820/474 9189	-820/480 9315	.820/485 9441	-806/481 9225		
155000 LB	MACH/TAS Total FF	-820/458 8595	•820/464 8721	.820/469 8847	.820/474 8973	•820/480 9096	+820/485 9222	.816/488 9267		
150000	MACH/TAS	-820/458	•820/464	.820/469	.620/474	-820/480	.820/485	.820/490	.804/486	
LB	TOTAL FF	8400	8523	8646	6769	8889	9012	91,32	8904	
145000	NACH/TAS	+820/458	.820/464	•820/469	. 820/474	-820/480	.820/485	•820/490	•815/492	
LB	Total FF	8214	8334	8454	8574	8491	8811	8928	8937	
140000	MACH/TAS	-820/458	.820/464	.820/469	.820/474	.820/480	.820/485	• 820/490	.820/495	
LB	Total FF	8034	8154	8271	8388	8505	8619	8736	8850	
135000	NACH/TAS	•820/458	•820/464	.820/469	.820/474	.820/480	.820/485	.820/490	-820/495	.811/495
LB	Total FF	7875	7992	8106	8220	8334	8448	8562	8673	8595
130000	MACH/TAS	-820/458	+820/464	•820/469	.820/474	•820/480	.820/485	•820/490	-820/495	-820/500
LB	Total FF	7719	7833	7544	8058	8169	828C	8391	8502	8610
125000	HACH/TAS	+820/458	.820/464	•820/469	.820/474	.820/480	.820/485	•820/490	.820/495	.820/500
L8	Total FF	7575	7686	7797	7908	8016	8127	8235	8343	8451
120000	MACH/TAS	+820/458	•820/464	•820/469	.820/474	.820/480	.820/485	820/490	.820/495	.820/500
LB	Total FF	7434	7542	7650	7758	7866	7974	8079	8187	8292
1150 <i>0</i> 0	MACH/TAS	+820/458	.820/464	.820/469	.820/474	.620/480	•820/485	.820/490	.820/495	.820/500
LB	Total ff	7302	7410	7515	7623	7728	7833	7938	8043	8145
110000	MACH/TAS	-820/458	•820/464	.820/469	.820/474	•820/480	.820/485	.820/490	•820/495	.820/500
LB	Total FF	7173	7281	7383	7488	7593	7695	7800	7902	8004
105000	MACH/TAS	.820/458	.820/464	•820/469	.820/474	.020/480	+820/485	•820/490	.#20/495	.820/500
LB	Total FF	7062	7167	7264	7371	7473	7575	7677	7776	7878
100000	MACH/TAS	.820/458	•820/464	.820/469	- 820/474	•820/480	.820/485	.820/490	-820/495	.820/500
LB	TOTAL FF	6954	7056	7158	7257	7359	7458	7557	7656	7755

FIGURE 23. Cruise planning chart (31,000 ft.).

HOLDING ALL ENGINES 2 AIRBLEEDS

EPR IAS - KTS FF PER ENGINE - LBS/HR STD. DAY TAT - °C

MINIMUM DRAG AIRSPEED (200 LOWER LIMITS)

FUEL FLOW BASED ON ISA. ADJUST FUEL FLOW + 1% PER + 5°C ISA DEVIATION

PRESSURE		GROSS	WEIG	нт -1	000 L	В
-FT	150	140	130	120	110	100
40000				2.00 200 2000 -37	200	1.80 200 1610 -37
35000	1.98	1.92	1.86	1.79	1.71	1.64
	223	215	206	200	200	200
	2450	2260	2070	1870	1700	1560
	-35	-36	-37	-38	-38	-38
30000	1.79	1.74	1.69	1.62	1.56	1.51
	220	212	204	200	200	200
	2350	2180	2030	1840	1700	1580
	-28	-29	-31	-31	-31	-31
25000	1.64	1.60	1.55	1.49	1.44	1.40
	218	210	202	200	200	200
	2330	2170	2010	1830	1700	1590
	-21	-22	-23	-23	-23	-23
20000	1.51	1.47	1.43	1.39	1.35	1.32
	217	209	201	200	200	200
	2340	2190	2040	1880	1770	1650
	-13	-14	-15	-15	-15	-15
15000	1.40	1.38	1.35	1.31	1.28	1.25
	216	208	200	200	200	200
	2400	2250	2110	1960	1840	1740
	-6	-6	-6	-6	-6	-6
10000	1.32	1.30	1.27	1.24	1.22	1.20
	215	207	200	200	200	200
	2480	2340	2200	2050	1940	1960
	3	3	2	2	2	2
5000	1.26	1.24	1.22	1.19	1.18	1.16
	214	207	200	200	200	200
	2580	2430	2280	2140	2030	1930
	12	12	11	11	11	11
1500	1.22	1.21	1.19	1.17	1.15	1.14
	214	206	200	200	200	200
	2650	2500	2360	2240	2130	2030
	17	17	17	17	17	17

FIGURE 24. Holding data.

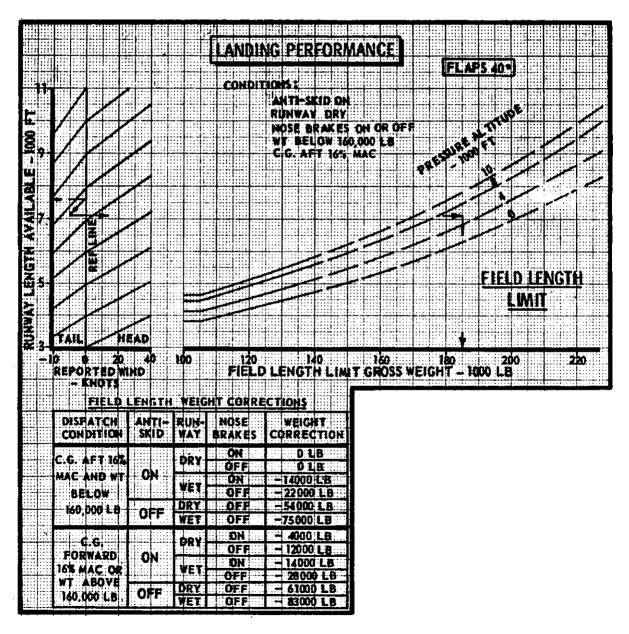


FIGURE 25. Landing performance data.

ENROUTE HIGH ALTITUDE - U.S.

For use at and above 18,000' MSL

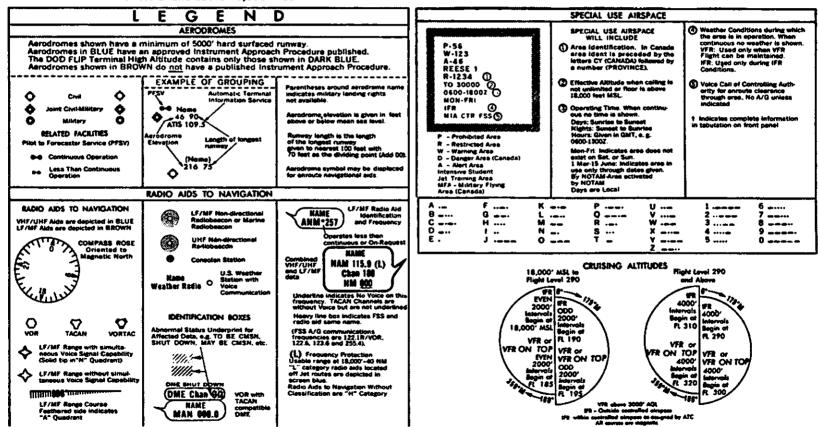


FIGURE 26. En route high altitude chart legend.

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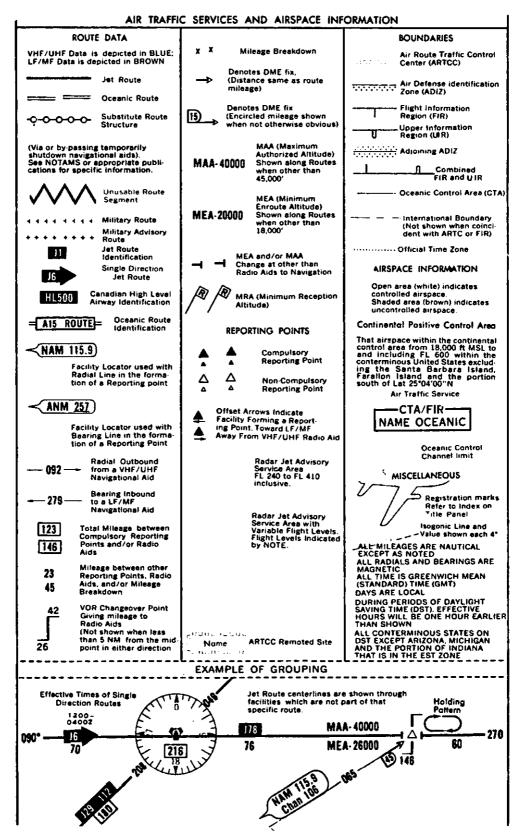


FIGURE 27. En route high altitude chart legend.

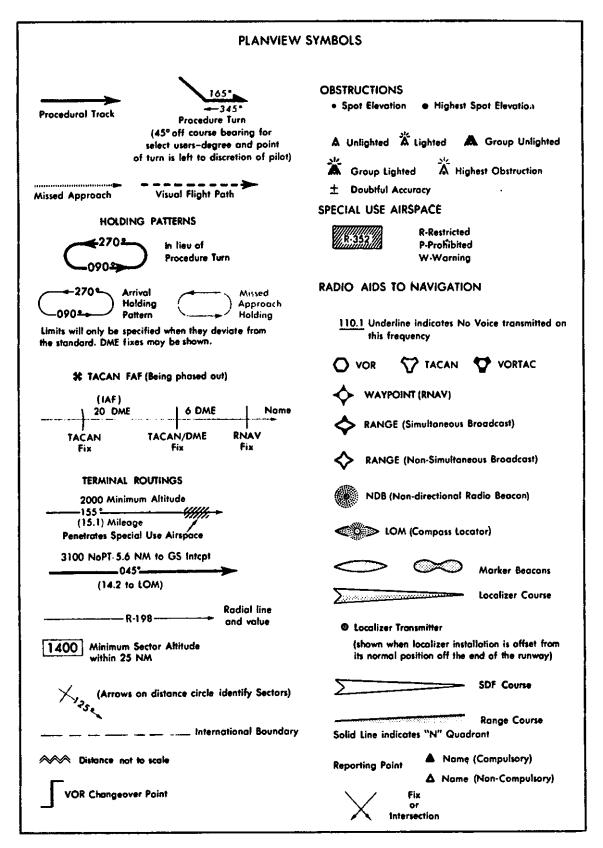


FIGURE 29. Legend for instrument approach procedures charts.

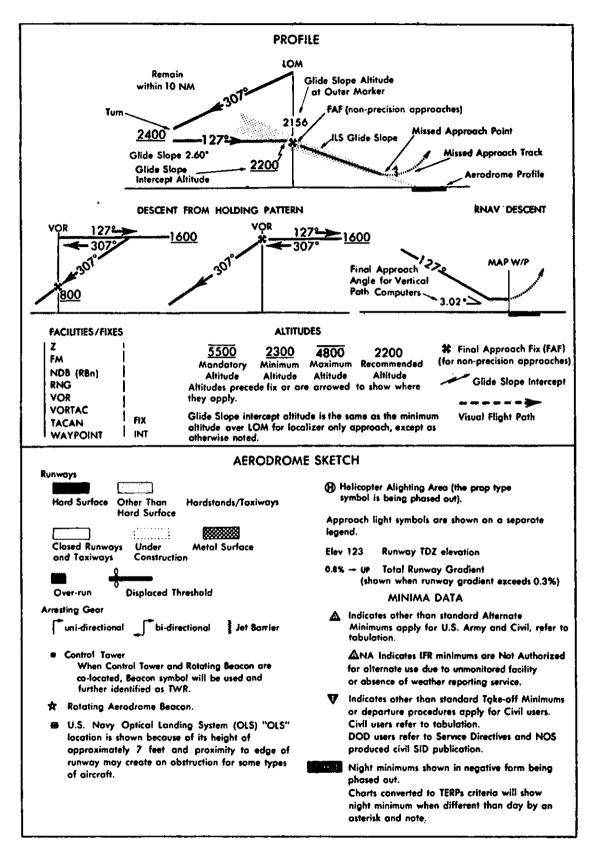


FIGURE 30. Legend for instrument approach procedures charts.

Actual length will be shown on Airport Diagram for any system, or portion thereof, not conforming to standard lengths listed on this page.

Each approach lighting system indicated on Airport Diagrams will bear system identification fetter (A, B, etc.) indicated in legend.

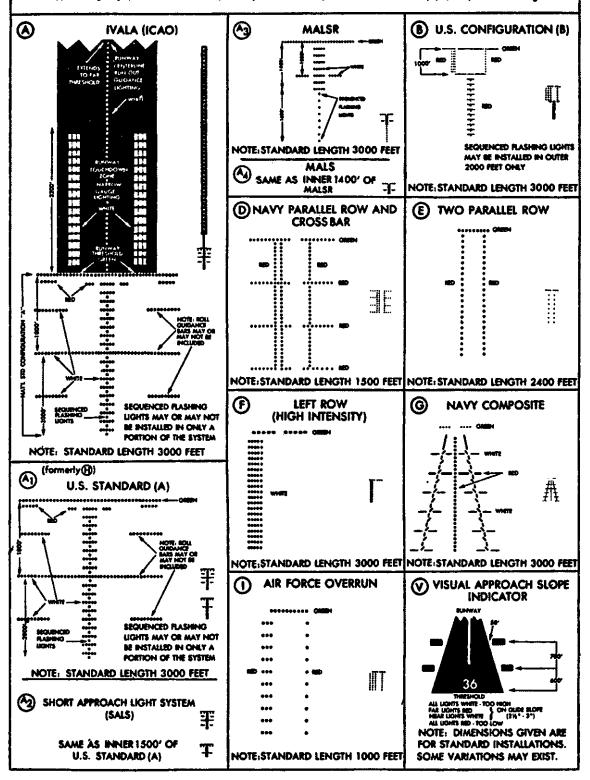


FIGURE 31. Legend for approach lighting systems-U.S.

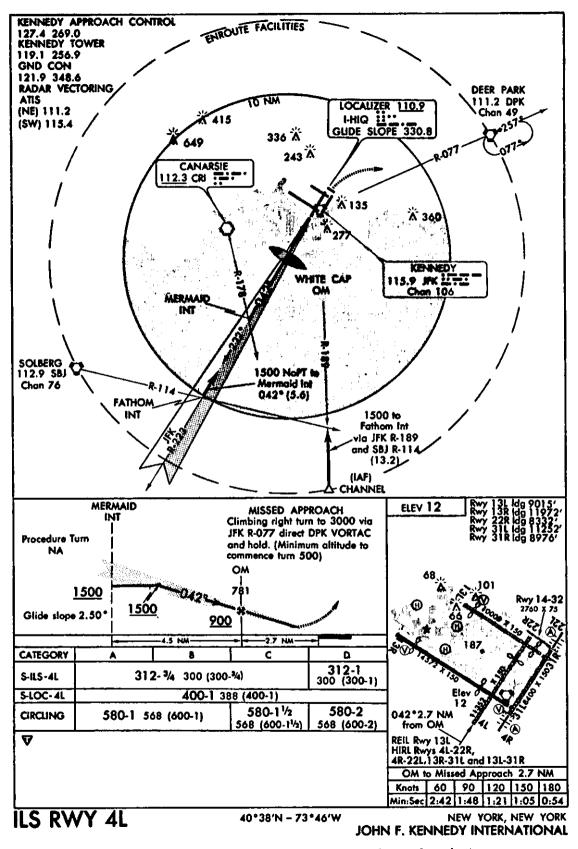


FIGURE 32. John F. Kennedy instrument approach procedure chart.

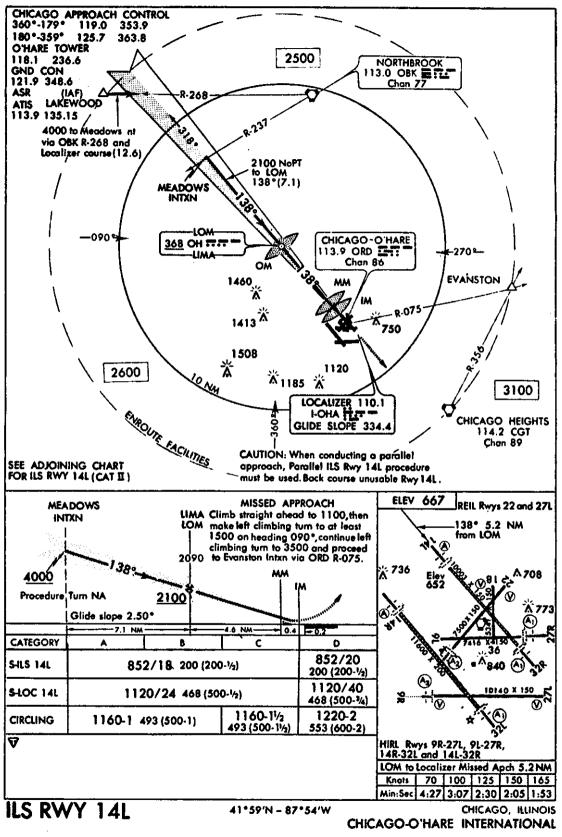


FIGURE 33. Chicago-O'Hare instrument approach procedure chart.

CHECK	POINTS	NOUTE CRUISE		TRUE	AIRSPE	ED-ETS.	WINDS ALOFT	DEIPT	CROUND	DETANCI	Π	345	COMPU	TEL METTON CALS.	×196.
PROM	10	ALT./FLT. LEVEL		COURSE	CE NO.	TAS	VELOCITY TEMPERATURE	COBLE ANCLE	SPEED	N.M.	LEC	TOTAL	LEG	TOTAL	
JFK		Radar Ve 763 FL 3		r								:11		2,800*	*Includes taxi fue
HOU							-60 kts. -40°C.	1							
100	- DAGK	<u> 736 FL 3</u>	10		.82		-70 kts.	<u> </u>							
DKK_	FNT	136 FL 3	20		.82		-40°C.		<u> </u>						
FNT		J94 FL 3			.82		-70 kts. -40°C.								
		Descent,									:20		3,000		
PMM		Approach	_					 			ļ				[
<u></u>		and Land	ing					 							
	ļ		·,									ļ	·		
	†								1						
ALTERNA	TE DATA	<u></u>					<u></u>		L		£			FUEL SUND	LART
ORD	General Airport	Mitchell					1					:15		TDE	LBL/GALS.
UND	MILDUIL					 	1			ŀ	1	- <u>'**</u> ~	ENBOUTE		[
												<u> </u>	ALTERMATE	:15	2,900
		L				<u>.</u>		<u></u>	<u></u>	L	<u> </u>				
Use 8,	900 lbs.,	/hr. fuel	£I	low wh	ere fi	uel i	s not given	•					EXTRA		
													TOTAL		

FLIGHT TIME ANALYSIS

_ _ _

FIGURE 34. Flight time analysis.

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