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# CIVIL AERONAUTICS MANUALS-Volume VII

U. S. Department of Commerce

Civil Aeronautics Administration

Civil Aeronautics Manuals and supplements thereto are issued by the Office of Flight Operations and Airworthiness, Civil Aeronautics Administration, for the guidance of the public and are published in the Federal Register and the Code of Federal Regulations.

Supplement No. 1

December 1, 1956

Subject: Revisions to Civil Aeronautics Manual 42 dated August 1956.

The purpose of this supplement is (1) to establish en route limitations for the Cessna Model 310 and DeHavilland Dove Series 1A, 2A, 5A, and 6A; and (2) to delete the aircraft designations in the introductory paragraph of section 42.80-6 so that en route limitations may be added or changed without affecting the introductory paragraph.

New material will be indicated by black brackets.

Remove and destroy the following pages: 61 through 64

Insert the following new pages:

61 through 64 72-1 and 72-2

Attachments.

For William B. Davis, Director, Office of Flight

Operation and Airworthiness.

[42.80-6] En route limitations on multiengine aircraft with maximum allowable takeoff weight below 12,500 pounds (CAA rules which apply to sec. 42.80). The following en route limitations data shall be used in determining compliance with section 42.80. These data are presented in tabular and graphic form by aircraft make and model. En route performance data on other aircraft weighing less than 12,500 pounds and operated under section 42.16 will be made available upon application to the Administrator.

**[**(Pu<sup>c</sup>lished in 19 F. R. 5660, Sept. 8, 1954, effective Oct. 1, 1954; amended in 21 F. R. 8420, Nov. 3, 1956, effective Dec. 1, 1956.)]

Table 1 .- En route limitations

Aero Commander 520

Weight in pounds <sup>1</sup>	Terrain clearance 2 in feet and climb speed in miles per hour (TIAS)		
	Feet 3	Miles per hour	
5,500 5,000 4,500	(3, 480) 6, 820 10, 130	94. 8 93. 5 92. 4	

<sup>&</sup>lt;sup>1</sup> The maximum permissible weight under secs. 42.16 and 42.82 is 5,420

pounds.

3 Highest altitude of terrain over which airplane may be operated in compliance with sec. 42.82.

5 The "terrain clearance" in parenthesis is not usable under secs. 42.16 and 42.82 because the minimum terrain altitude is 4,000 feet under sec. 42.82.

Note.-Inoperative propeller windmilling.

BEECH C-18S AND BEECH 18A

Airplane	Weight in	feet s	earance <sup>1</sup> in and climb n miles per IAS)
(		Feet	Miles per hour
Beech C-18S	7, 850	6, 200	102. 5
	7, 500	7, 620	98. 7
	7, 000	9, 630	93. 3
Beech 18A	7, 200	4, 760	91. 7
	7, 000	5, 540	90. 8
	6, 500	7, 460	88. 9
	6, 000	9, 400	86. 9

<sup>&</sup>lt;sup>1</sup> Highest altitude of terrain over which airplane may be operated in compliance with sec. 42.82.

BEECH AT-11

Weight in pounds	Terrain clearance in fee and climb speed in mile per hour (TIAS)		
•	Feet	Miles per hour	
7,850 7,500	6, 200 7, 800	102. 1 100. 9	
7,000		99. 2	
6,500	40' 400	97. 5	

 $<sup>^{\</sup>rm I}$  Highest altitude of terrain over which airplane may be operated in compliance with sec. 42.82.

Note.—Inoperative propeller idling in high pitch. Cowl flaps are closed on inoperative engine. Desicers are not operating.

## BEECH D-18C

Weight in pounds		ance in feet speed in miles (TAS)
	Feet	Miles per hour
9,000 8,500 8,000 7,500	6, 200 7, 300 8, 450 9, 600	121. 0 120. 0 119. 5 119. 0

 $<sup>^1\,\</sup>mathrm{Highest}$  altitude of terrain over which airplane may be operated in compliance with sec. 42.82.

NOTE.-Inoperative propeller feathered.

### TABLE 1 .- En route limitations-Continued

#### Beech D-18S

			in feet a	
Weight in pounds	Propeller feathered		Propeller idling	
	Feet	Miles per hour	Feet	Miles per bour
8,750 8,500 8,000 7,500	7, 100 7, 600 8, 800 9, 900	103, 5 103, 5 102, 5 102, 0	5, 600 6, 700 7, 900	104. 5 104. 0 103. 0

<sup>&</sup>lt;sup>1</sup> Highest altitude of terrain over which airplane may be operated in compliance with sec. 42.82.

### Веесн 50

Weight in pounds	Terrain clears and climb s per hour (7	ance i in feet speed in miles (IAS)
	Feet	Miles per hour
5,500 5,000	4, 140 7, 710	96. 4 94. 6
4,500	11, 340	92. 7

<sup>&</sup>lt;sup>1</sup> Highest altitude of terrain over which airplane may be operated in compliance with sec. 42.82.

Norg.-Inoperative propeller windmilling.

## CESSNA T-50

(WITH LYCOMING R-680-E3 ENGINES AND MAXIMUM GROSS WEIGHT OF 5,700 POUNDS)

	Terrain clearance <sup>1</sup> in feet and cl speed in miles per hour (TIAS				
Weight <sup>3</sup> in pounds	6135A-15 propeller		6135A-9 propeller		
	Feet 2	Miles per hour	Feet 2	Miles per hour	
5,700 5,500 5,250 5,000 4,750	$\lfloor (2, 280) \rfloor \ \lfloor (3, 740) \rfloor$	87. 2 86. 6 86. 0 85. 3	(450) (1, 540) (2, 890) 4, 320 5, 730	87. 4 87. 0 86. 4 85. 7 85. 2	

### GRUMMAN G-21

Weight in pounds <sup>1</sup>	Terrain clearance 2 in and climb speed in r per hour (TIAS)	
	Feet 3	Miles per hour
7,500 7,000 6,500 6,000 5,500	<sup>3</sup> (3, 620) 4, 610 5, 590 6, 550 7, 530	111. 5 111. 1 110. 6 110. 1 109. 6

<sup>&</sup>lt;sup>1</sup> The maximum permissible weight under secs, 42.16 and 42.82 is 7,310

 $\ensuremath{\text{Note}}.$  —Propeller idling in high pitch. Airplaine is equipped with de-leers.

#### LOCKHEED 10A

			in feet ar er hour (T	
Weight in pounds <sup>1</sup>	Propeller feathered		Propeller idling	
	Fect <sup>3</sup>	Miles per hour	Feet 3	Miles per hour
10,500 10,100 10,000 9,500 9,000 8,500 8,000	(3, 600) 4, 580 4, 820 6, 020 7, 200 8, 350 9, 550	93. 8 93. 6 93. 5 93. 1 92. 8 92. 4 92. 0	(3, 120) (3, 350) 4, 660 5, 900 7, 180 8, 420	92. 9 92. 8 92. 5 92. 1 91. 7 91. 4

<sup>&</sup>lt;sup>1</sup> The maximum permissible weights under secs. 42.16 and 42.82 are 9,750 pounds with propeller idling, 19,340 pounds with propeller feathered.

<sup>2</sup> Highest altitude of terrain over which airplane may be operated in compliance with sec. 42.82.

<sup>3</sup> The "terrain Clearances" in parenthesis are not usable under secs. 42.16 and 42.82 because the minimum terrain altitude is 4,000 feet under sec. 42.82.

<sup>1</sup> Highest altitude of terrain over which airplane may be operated in compliance with sec. 42.82.

1 The "terrain clearances" in parentheses are not usable under sec. 42.16 and sec. 42.82 because minimum terrain altitude is 4,000 feet under sec. 42.82.

3 Maximum permissible weights under sec. 42.16 and sec. 42.82 are 4,950 pounds with 6135A-15 propeller, 5,050 pounds with 6135A-9 propeller.

 $<sup>{\</sup>tt Note.-Inoperative}$  propeller windmilling. No leading edge desicers installed.

<sup>1</sup> The maximum permissible weight under sees, 42.16 and 42.82 is 7,310 pounds.
2 Highest attitude of terrain over which airplane may be operated in compliance with sec. 42.82.
3 The "terrain clearance" in parenthesis is not usable under secs. 42.16 and 42.82 because the minimum terrain altitude is 4,000 feet under sec. 42.82.

## TABLE 1 .- En route limitations-Continued

### LOCKHEED 10E

	Terrain clearance <sup>1</sup> in feet and climb speed in miles per hour (TIAS)			
Weight in pounds	Propeller feathered Propeller idlin			
	Feet	Miles per hour	Feet	Miles per hour
10,500 10,000 9,500 9,000 8,500 8,000	9,000 9,600 10,200 10,700 11,300 11,900	96 96 96 96 96 96	7, 500 8, 100 8, 600 9, 200 9, 750 10, 350	96. 5 96. 5 96. 5 96. 5 96. 5

 $<sup>^{\</sup>rm I}$  Highest altitude of terrain over which airplane may be operated in compliance with sec. 42.82.

### [CESSNA MODEL 310

(WITH ORIGINAL PROPELLER DIAMETER LIMITS OF 82 INCHES TO 84 INCHES)

Weight in pounds	Terrain clearance <sup>1</sup> in feet and climb speed in miles per hour (TIAS)		
	Feet	Miles per hour	
4,600 4,500 4,250 4,000 3,750	6, 900	102. 0 101. 8 101. 4 101. 0 100. 6	

<sup>&</sup>lt;sup>1</sup> Highest altitude of terrain over which airplane may be operated in compliance with sec. 42.82.
Note: Inoperative propeller feathered.

## LOCKHEED 12A

Weight in pounds	Terrain clearance in feet and climb speed in miles per hour (TIAS)		
	Feet	Miles per hour	
8,600 8,000 7,500 7,000 6,500	7, 950 8, 500	98. 5 98. 5 98. 5 98. 5	

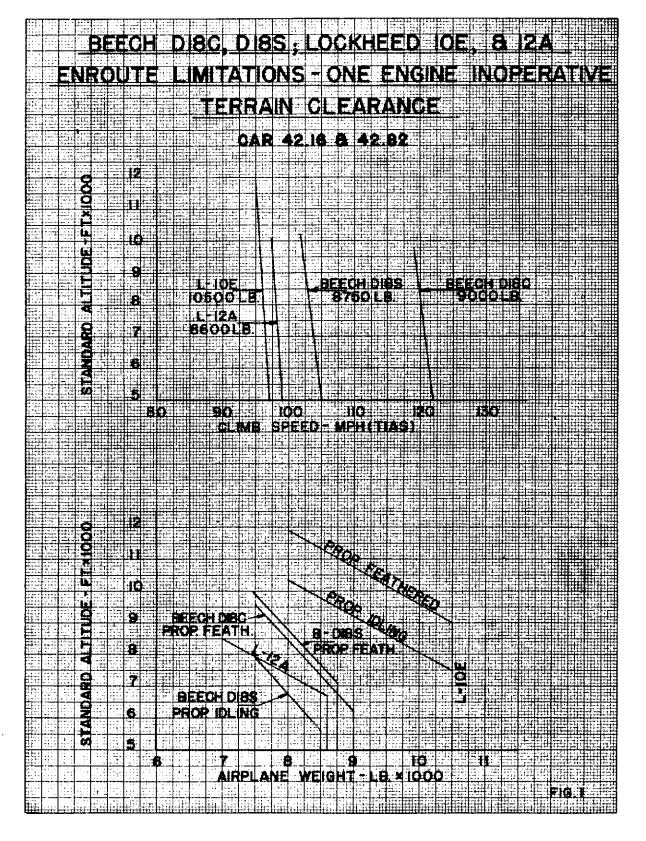
<sup>&</sup>lt;sup>1</sup> Highest altitude of terrain over which airplane may be operated in compliance with sec. 42.82.

### [DE HAVILLAND DOVE SERIES

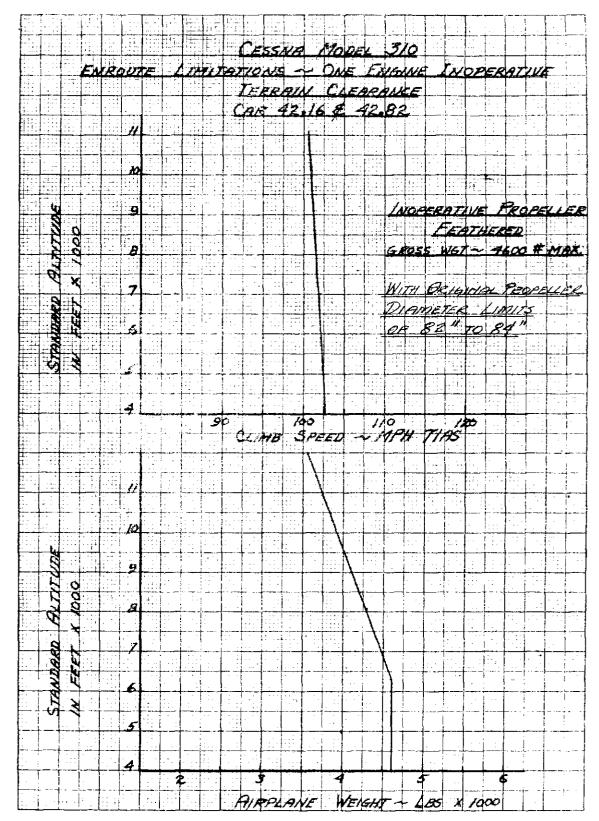
Weight in pounds	Terrain clearance <sup>1</sup> in feet and climb speed in miles per hour (IAS)		
	m. p. h.	Feet	
		1A and 2A Series	5A and 6A Series
8,800 8,500 8,000 7,500	110 108 105 102	6, 600 8, 300 9, 900	5, 600 6, 600 8, 300 9, 900

 $<sup>^{\</sup>rm I}$  Highest altitude of terrain over which airplane may be operated in compliance with sec. 42.82.

Note: Inoperative propeller feathered.]

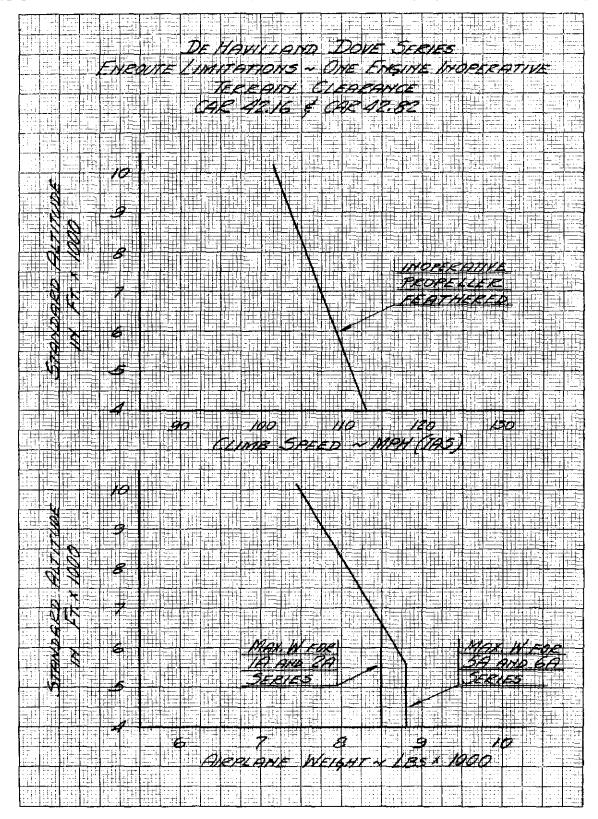


(Rev. 12/1/56)



[Figure 10

(Published in 21 F. R. 8420, Nov. 3, 1956, effective Dec. 1, 1956.)



[Figure 11

(Published in 21 F. R. 8421, Nov. 3, 1956, effective Dec. 1, 1956.)  $\blacksquare$