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# The Cause of Thermal Fatigue Cracking in Metroliner Wheels

United States Steel Corp, Monroeville, Pa Research Lab

Prepared for

Federal Railroad Administration, Washington, D C Office of Research and Development

Mar 77

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## THE CAUSE OF THERMAL FATIGUE CRACKING IN METROLINER WHEELS

i.F. Carpenter

United 1:stes Steel Corporation Research Laboratory Monroeville PA 15146



MARCH 1977 FINAL REPORT

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U.S. DEPARTMENT OF TRANSPORTATION FEDERAL RAILROAD ADMINISTRATION Research and Development Washington OC 20590

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16 Abstract		

One new wheel and two used wheels (one with a thermal crack in the tread) were examined for mechanical properties, macrostructure, microstructure, and residual stresses. Similar examinations were conducted on three new wheels which were first subjected to various braking cycles designed to define the conditions that produce cracking. The braking tests were conducted on the laboratory dynamometer.

The results of this study indicated that the wheel that had developed a thermal crack in service had been intermittently and severely heated around the tread surface and that such heating had altered the microstructure, produced residual tensile stresses and permitted the crack to initiate.

The results further showed that neither altered microstructures nor cracking could be produced by many emergency brakings or speed-reduction brakings with normal brake shoes and forces.

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#### PREFACE

The research and testing studies documented in this report were carried out as part of the component failure prevention program conducted by United States Steel Corporation Research Laboratory, Monroeville PA, under contract DOT-TSC-712 for the Transportation Systems Center, U.S. Department of Transportation, Cambridge MA, under the auspices of the Federal Railroad Administion, U.S. Department of Transportation.

The work at United States Steel Corporation was subcontracted to USS Engineers and Consultants, Inc., and was carried out under the general supervision of G. P. Carpenter with interest and support by the late J. W. Lyons of DOT/TSC.

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#### 1. INTRODUCTION

The Metroliner high-speed passenger train, operated by the Penn Central Railroad for Amtrak between Washington, D.C., and New York City, is equipped with dynamic brakes and on-tread friction brakes actuated by independent brake cylinders on each wheel. In normal operation, the dynamic brakes are used to decelerate from high speeds, and both the dynamic and the tread brakes are used at intermediate speeds. At low speeds, dynamic braking is ineffective; therefore, only tread braking is used. The two braking modes are blended automatically to provide optimum passenger comfort.

In the past, the dynamic-braking system on Metroliner cars has been reported to fail often, making it necessary to use the tread brakes from speeds as high as 120 mph (193 km/hr); such on-tread braking would be used until the dynamic-braking system was repaired. The treads and flanges of Metroliner wheels have been said to develop thermal cracks, even though the wheels have been made from steel with a carbon content of less than 0.47 percent, the most thermal-crack-resistant wheel-steel composition commercially available. Metroliner wheels are inspected frequently; those containing thermal cracks are immediately removed from service, since it is believed that under certain operating conditions such cracks may initiate hazardous sudden wheel failures.

Because of the Metroliner wheel problem and the need for experimental data of general applicability, the Department of Transportation (DOT) contracted with USS Engineers and Consultants, Inc., to determine the effect of various braking conditions on the tendency to develop thermal-fatigue cracks in Metroliner wheels. The original program of study comprised the evaluation of residual stresses, mechanical properties, and metallurgical characteristics of four new and two used wheels after the wheels were subjected to various numbers of emergency stops on the wheel-test machine at U.S. Steel's Research Laboratory. Unexpectedly, the first new wheel, when tested on the wheel-test machine (Appendix A), failed to develop any thermal cracks after being subjected to 300 emergency

stops from 130 mph (209 km/hr), with an inertial load of 24,100 pounds (107.2 kN) and a brakeshoe force of 7,000 pounds (31.14 kN). This is more than three times the number of stops that had been received by the used wheels in service, one of which contained a thermal crack in the tread.

Therefore, by agreement with DOT, the program of study was revised as shown in Table 1. The revised program included determination of residual stresses (1) in the wheel already subjected to 300 emergency stops, (2) in a wheel subjected to speed-reduction tests, and (3) in a wheel subjected to abnormal braking consisting of braking, stop braking with normal brakeshoes, and stop braking with simulated "wornout" brakeshoes. The foregoing represent conditions reportedly encountedly encounted in service.

This final report summarizes the results of the studies conducted on the four new and the two used Metroliner wheels in accordance with the program as agreed upon with DOT.

REVISED PROGRAM FUR EVALUATION OF METROLINER WHEELS TABLE 1.

Wheel No.	Manufacturer's Serial No.	Condition	Dynamometer Tests	Metallurgical Studies
-	1-72-5-2476A	New	300 Emergency stops from 130 mph	•
2	1-72-5-2474A	Nex	2424 Speed reductions (100 to 50 mmb)	•
•	1-72-S-2472A	Ne.	Four 30-minute draq-brake periods at 45 mph plus 110 emergency stops from 130 mph, 10 of which were made with a "worn-out"	•
4	1-72-S-2466A	Man	None	
s.	1-71-5-10191A	Used, with one crack in tread	None	•
و	1-71-S-10190A	Used, with no cracks	None	•

. All wheels were examined for chemical composition, mechanical properties, macrostructure, and microstructure.

Conversion Factor

1 mph = 1.61 km/hr

#### 2. MATERIALS AND EXPERIMENTAL WORK

#### 2.1 MATERIALS

The six rim-treated Section B-36, Class A (0.4% carbon) wrought-steel wheels studied were produced by the Standard Steel Division of Titanium Metals Corporation of America. The chemical composition of the wheels is shown in Table 2, along with the specified range for AAR Specification M-107, Class A wheels. The plate surfaces of the wheels had been contour-machined and shotpeened.

The new wheels (Nos. 1 through 4) were produced in January 1972 and were selected from stock at the Penn Central shops in Wilmington, Delaware. Both used wheels (Nos. 5 and 6) were produced in January 1971 and had been in service on Metroliner axle No. W80829, car No. 864, from November 7, 1971, to June 23, 1972. During this period, the wheels were used for a total of about 100,000 miles (160,000 km), but the dynamic-braking system was inoperative for about 1610 miles (2590 km). It was estimated that the wheels had been subjected to 85 station stops from normal operating speeds and to numerous speed-reduction brake applications without the assistance of the dynamic brakes. Wheel No. 5 exhibited a fatigue-type thermal crack, which measured about 1-1/4 inches (32 mm) long on the tread, but wheel No. 6 had no apparent cracks. The other six wheels on the car were reported to be crackfree, as determined by magnetic-particle tests conducted by Penn Central.

Wheels No. 1, 2, and 3 were subjected to various braking conditions on the wheel-test machine, to metallurgical studies, and to residual-stress studies. Wheels No. 4, 5, and 6 were subjected only to metallurgical and residual-stress studies.

#### 2.2 METALLURGICAL STUDIES

As indicated in Figure 1, the following metallurgical studies were conducted on the wheels:

CHEMICAL COMPOSITION OF METROLINER WHEELS (RESEARCH LABORATORY CHECK ANALYSIS) TABLE 2.

1					Š	mical	Chemical Composition, percent	tion, p	ercent			
Wheel No.	Manufacturer's Number	o	ş	4	40	5.1	5	N1	ช	Ş.	ř	Total A1
-	1-72-S-2476A	0.42	0.80	0.017	0.026	0.31	0.083	0.061	0.051	0.018	0.014	0.005
~	1-72-5-2474A		0.77	0.022	0.036	0.29	0.136	0.075	0.083	0.017	0.013	0.007
•	1-72-S-2472A	0.43	0.76	0.024	0.030	0.30	0.121	0.071	0.075	0.020	0 013	900 0
•	1-72-5-2466A		0.80	0.022	0.035	9.32	0.127	0.076	0.077	0.022	0.014	0.004
v	1-71-5-10191A		0.79	0.022	0.041	0.33	0.179	890.0	0.154	0.033	0.016	
w	1-71-5-10190A	0.43	0.76	0.014	0.032	0.31	0.080	0.041	0.057	0.017	0.012	0.004
	AAR Specifi- cation M-107- 71 for Class A Wheels, Wrought Carbon Steel	0.57* 0.60 max 0.85	0.60	0.05 max	0.05 max	0.15 min	٠	*	•	٠	•	<b>:</b> €::

"The carbon content of Class A wheels normally ranges from 0.47 to 0.57 percent.

<sup>+</sup> Not specified.

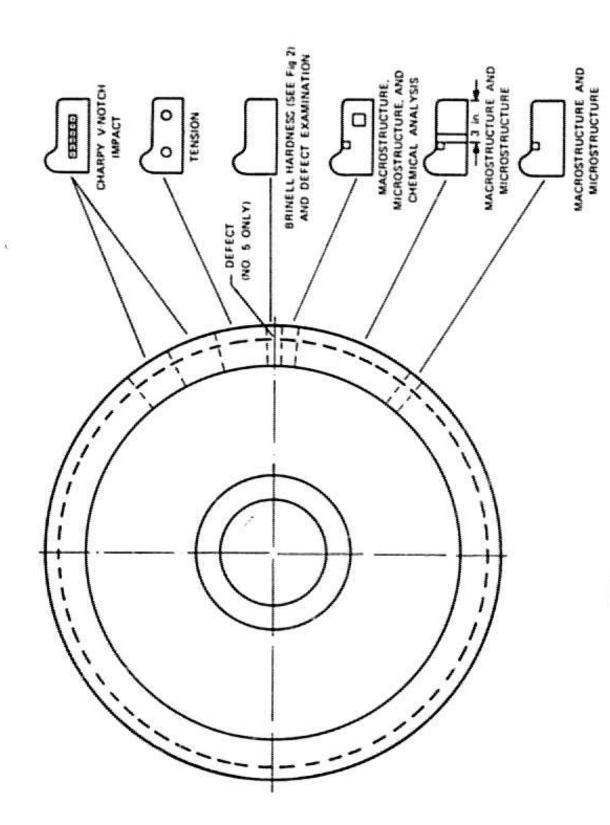


FIGURE 1. LOCATION OF METALLURGICAL SPECIMENS

- The chemical composition of each rim was determined for a specimen from the midwidth-midthickness position.
- The tensile properties of each rim were determined by testing 0.252-inch-diameter (6.4 mm) circumferentially oriented (chordal) tension-test specimens removed from the midthickness-quarterwidth position.
- The impact properties of each rim were determined by testing circumferentially-oriented (chordal) Charpy V-notch specimens removed from the midthickness-quarterwidth position.
- Brinell hardness tests were conducted on a radial-axial section of the rim of each wheel at the locations shown in Figure 2.
- 5. The fatigue-type thermal crack in wheel No. 5 was visually examined after the rim was fractured (in the Laboratory) through the existing crack. In addition, the tread surface of each wheel was examined for surface defects by the fluorescent-magnetic-particle method.
- Several radial and radial-tangential sections of each rim were etched in nital, and their macrostructures were examined.
- Selected specimens from the rim were also etched in picral reagent, and their microstructures were examined.
- The diamond-pyramid hardness of selected metallographic specimens were determined with a Vickers microhardness tester with a load of 2.5 kg.

#### 2.3 DYNAMOMETER TESTS

## 2.3.1 Emergency-Stop-Braking Conditions for Wheel No. 1

The conditions for simulated emergency-stop braking of wheel No. 1 are shown in Table 3. In Metroliner service, the inertial wheel load (gross car weight divided by number of wheels) is 23,100 pounds (102.75 kN). Therefore, the inertial wheel load for these tests on U.S. Steel's wheel-test dynamometer was set at 24,100

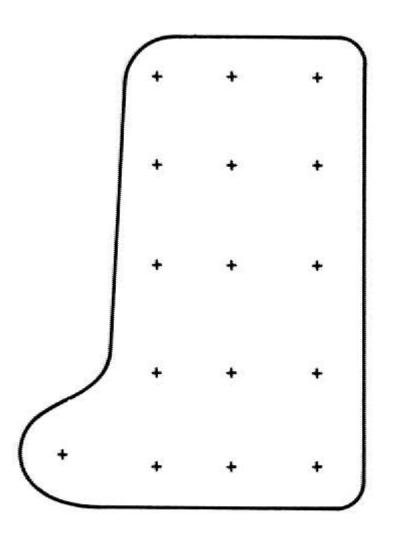


FIGURE 2. LOCATIONS OF BRINELL HARDNESS MEASUREMENTS ON RADIAL-AXIAL SECTIONS OF WHEEL RIMS (FULL SCALE)

## TABLE 3. CONDITIONS FOR EMERGENCY-STOP BRAKING OF WHEEL NO. 1

Inertial wheel load

Simulated car weight

Brakeshoe arrangement and type

Brake force

Initial speed

Kinetic energy per stop

Average stop time

Average stop distance

24,100 lb

192,800 lb

Single Cobra V-202

7000 1b

130 mph (1210 rpm)

13,670,000 ft-1b

87 sec

8660 ft

## Conversion Factors

1 1b = 0.225 N

1 mph = 1.61 km/hr

1 ft-1b = 1.36 J

1 ft = 0.305 m

pounds (107.2 kN), as close as was possible to 23,100 pounds, by using three large flywheels. A single flange-type composition brakeshoe (Cobra Type V-202, the same type used on Metroliner cars) was used for these tests. Brakeshoe force for the tests was 7000 pounds or 31.14 kN (as close as possible to the brake force of 6960 pounds or 30.96 kN used on Metroliner wheels). A translatory speed of 130 mph or 209 km/hr (1210 rpm for a 36-inch-diameter or 91-cm wheel) was selected for this test by DOT because this is the maximum anticipated speed of the Metroliner.

During stop testing, the wheel was initially rotated at 1210 rpm, and the brake was applied until the wheel stopped. Wheel speed, torque, work, stop time, equivalent stop distance, and brake force were continuously recorded and are summarized in Appendix B. Wheel temperature was not determined. After the wheel stopped, it was rotated at 20 rpm in air for 3 minutes, and then water sprays were directed toward the rotating rim for 10 minutes until the temperature of the wheel rim was reduced to ambient temperature. A total of 300 emergency stops were made. The wheel tread and flange were inspected periodically for thermal cracking by the fluorescent-magnetic-particle method.

#### 2.3.2 Speed-Reduction-Braking Conditions for Wheel No. 2

The conditions for simulated speed-reduction braking of wheel No. 2, Table 4, were devised to be comparable to those of service speed-reduction braking. In this test, the single Cobra V-202 brakeshoe was applied with a force of 7000 pounds to slow the wheel from 930 rpm (100 mph or 161 km/hr) to 465 rpm (50 mph or 80 km/hr). After a period of 5 minutes, during which the speed was increased to 930 rpm, the brake was reapplied to reduce the speed to 465 rpm. After an additional 5-minute period and an increase of speed to 930 rpm, the brake was again reapplied to reduce the speed to 465 rpm. The wheel was then stopped dynamically\* and cooled to ambient

An exception was the first cycle in which the wheel was stopped with the brake.

## TABLE 4. CONDITIONS FOR SPEED-REDUCTION BRAKING OF WHEEL NO. 2

Inertial wheel load 24,100 lb

Simulated car weight 192,800 lb

Brakeshoe arrangement and type Single Cobra V-102

Brake force 7000 lb

Test cycle\*

Initial speed 100 mph (930 rpm) Final speed 50 mph (465 rpm)

Kinetic energy per brake application 6,060,000 ft-lb

Average braking time 35 sec

Average braking distance 3800 ft

#### Conversion Factors

1 1b = 0.225 N

1 mph = 1.61 km/hr

1 ft-1b = 1.36 J

1 ft = 0.305 m

<sup>\*</sup> The cycle was repeated three times at 5-minute intervals before the wheel was cooled to ambient temperature with water sprays.

temperature in the same manner as wheel No. 1. This cycle was repeated a total of 808 times (2,424 brake applications). The wheel rim was inspected frequently for thermal cracks by the fluorescent-magnetic-particle method, but none had developed. Data on torque, work, braking time, and wheel revolutions during braking are listed in Appendix C.

To assist in monitoring the simulated speed-reduction braking procedure, thermocouples were imbedded at the midwidth location of the wheel tread at depths of 1/16 and 1/4 inch (1.6 and 6 mm), as shown in Figure 3. In addition, duplicate strain gages (Micro-Measurements type WK-06-250BG-350) suitable for accurate strain measurements at temperatures up to 550°F (290°C) were oriented radially and cemented to the surface of the back-rim-plate fillet and front-hub-plate fillet, as shown in Figure 3. The thermocouples and strain gages were connected to appropriate measuring devices through a slip ring mounted on the stub axle. Selected temperatures and strains recorded during testing are listed in Appendix D.

#### 2.3.3 Conditions for Braking of Wheel No. 3

Inasmuch as the results (to be discussed later) of the studies of the aforementioned dynamometer-tested wheels (Nos. 1 and 2) showed that thermal cracking and microstructural damage, such as observed in one of the used wheels (wheel No. 5), could not be produced by 300 simulated emergency-stop brakings or by 2424 simulated speed-reduction brakings, it was decided to subject wheel No. 3 to severe abnormal braking conditions, Table 5. First, the wheel was subjected to two drag-braking cycles to change the rim residual-stress condition from overall compression to overall tension. Each drag-braking cycle consisted of operation at a nominal speed of 419 rpm (45 mph or 72 km/hr) for 30 minutes and application of the brakeshoe with a force of 1900 pounds (8.45 kN) for 50 seconds of each minute. The rotating wheel was then cooled with water sprays until the temperature of the rim was at ambient. Next, the wheel was subjected to 10 emergency-stop braking cycles in an attempt to produce cracking, but none developed. (The

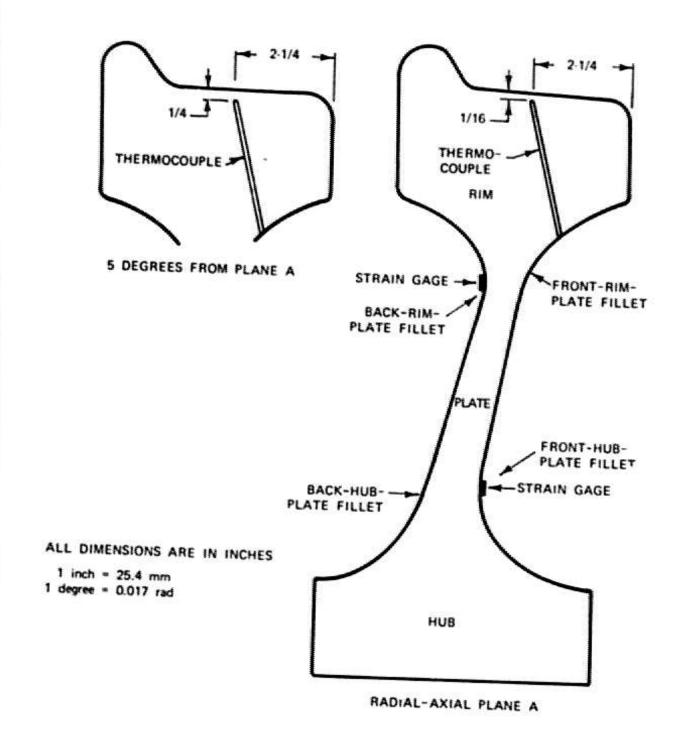


FIGURE 3. LOCATIONS OF THERMOCOUPLES AND STRAIN GAGES FOR DYNAMOMETER TESTS OF WHEEL NO. 2

#### TABLE 5. BRAKING CONDITIONS FOR WHEEL NO. 3

#### All Tests

Inertial wheel load

24,100 lb

Simulated car weight

192,800 lb

Brakeshoe arrangement and type

Single Cobra V-202

#### Drag Braking

Brake force\*

1900 lb

Nominal speed \*\*

419 rpm (45 mph)

Cooling

Water sprays on rim and plate

#### Emergency-Stop Braking\*\*\*

Brake force

7000 lb

Initial speed

130 mph (1210 rpm)

Kinetic energy per stop

13,670,000 ft-1b

Average stop time

87 sec

Average stop distance

8660 feet

Cooling

Water sprays on rim and plate

- \* Braking for 50 seconds of each minute was continued for 30 minutes.
- \*\* Actually consisted of 100 rpm for 4 minutes, 200 rpm for 4 minutes, and 419 rpm for 22 minutes.
- \*\*\* Seven stops were made with "wornout" shoes.

#### Conversion Factors

1 1b = 0.225 N

1 mph = 1.61 km/hr

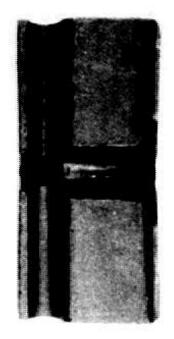
1 ft-1b = 1.36 J

1 ft = 0.305 m

emergency-stop braking cycles were identical with those of wheel No. 1 except that water-spray cooling was started immediately after the wheel was stopped.) Two additional drag-braking cycles were then applied to produce higher hoop tensile stresses in the rim. Emergency-stop braking was then continued for 45 stops.

Because thermal cracks still had not developed, it was postulated that even more severe conditions, such as would occur by braking with a wornout brakeshoe, would be necessary to produce thermal cracking in the wheel. Accordingly, two brakeshoes were machined, as shown in Figure 4, so that only 1/4- and 1/8-inch (6 and 3 mm) thicknesses of composition material remained on the tread portion of the shoes. (The flange portion was cut off to facilitate machining of the tread portion.) The wheel was then subjected to seven emergency-stop braking cycles by using one of the machined brakeshoes. This resulted in a wearing away of about 80 percent of the composition material comprising the brakeshoe, Figure 4D. Using the second simulated wornout brakeshoe, the wheel was subjected to three additional emergency stops. As a result, the second brakeshoe exhibited worn-through areas and severe distortion, Figure 4E. These 10 emergency stops with simulated wornout brakeshoes produced discolored spots and deposits of backing-plate metal on the wheel tread surface. After the wornout brakeshoes were replaced with new shoes, emergency-stop braking was resumed for 45 additional stops. Small thermal cracks developed on the wheel tread surface during this period.

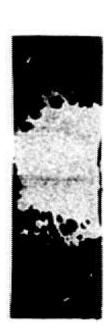
To assist in monitoring the tests on wheel No. 3 a thermocouple was placed in the rim, as shown in Figure 5. In addition,
high-temperature strain gages (Micro-Measurements type WK-06-250WT350) were cemented to the wheel surface at the back-rim-plate and
front-hub-plate fillets, Figure 5. Static strain-gage measurements were taken after the wheel was cooled to determine whether
the residual stresses at these locations were changed by the
thermal cycles. Selected strain-gage and temperature data are
listed in Appendix E, along with other pertinent data for this
wheel.



A. Front view-new shoe.



Side view-new shoe and simulated wornout shoe.



C. Front view-simulated D. Front view-First E. Front view-Second wornout shoe.



simulated wornout shoe after 7 stops.



simulated wornout shoe after 3 stops.

FIGURE 4. BRAKESHOE COMPOSITION (ABOUT 1/4X)

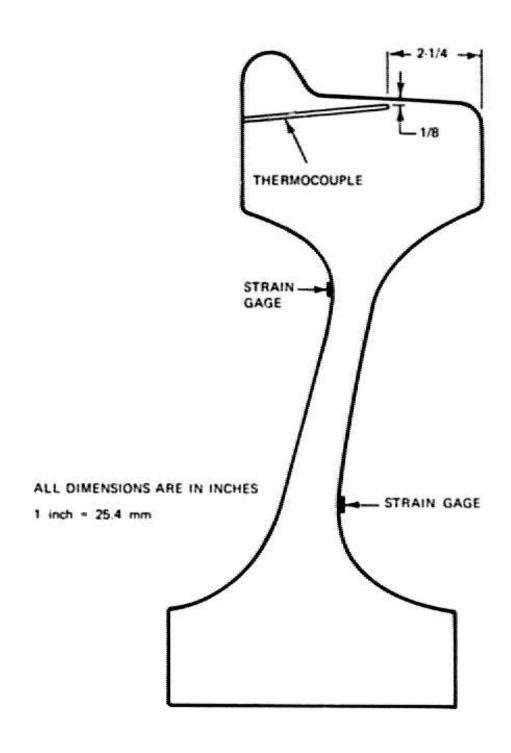


FIGURE 5. LOCATIONS OF THERMOCOUPLE AND STRAIN GAGES FOR DYNAMOMETER TESTS OF WHEEL NO. 3

#### 2.4 RESIDUAL-STRESS STUDIES

Two stress-relaxation methods were used to determine residual stresses in the wheels. The overall residual hoop stresses in the rim were determined by the split-deflection method. The residual stresses at the rim surfaces and at the back-rim-plate fillet and the front-hub-plate fillet were determined by the dissection method. Figures 6 and 7 show the locations of gage marks and strain gages.

#### 2.4.1 Split-Deflection Method

To measure the overall residual hoop stresses in the rim, pairs of gage marks, 2 inches (51 mm) apart on the tangential line 3/4 inch (19 mm) below the tread, were punched at axially opposite positions on the front and back faces of the rim (see Figure 6). The rim was then separated as a concentric section from the rest of the wheel by machining the inside circumference of the rim. A 1-to 1-1/2-inch-thick (25.4 to 38.1 mm) section was removed from between the gage marks. The change in distance between the gage marks was used to indicate the relative magnitude of the overall hoop stresses in the rim; an increase in gage distance (rim opening) indicated overall hoop tension, and a drcrease in gage distance (rim closure) indicated overall hoop compression.

#### 2.4.2 Dissection Method

To measure the distribution of tangential residual stresses in the outer 1-inch-thick layer of the rim, single-element electrical-resistance foil strain gages (Micro-Measurements type EA-06-120AD-120 with 1/8-inch gage length) were cemented to the wheel-rim surfaces.\* Several coatings (polyurethane varnish,

Because the tread surface of wheel No. 2 was "grooved" to a depth of about 0.03 inch (0.76 mm) from brakeshoe action, the grooves were filled with Micro-Measurements AE-10, a 100-percent-solids epoxy system, before the gages were cemented. See Figures 10 and 20 for illustration of the grooves.

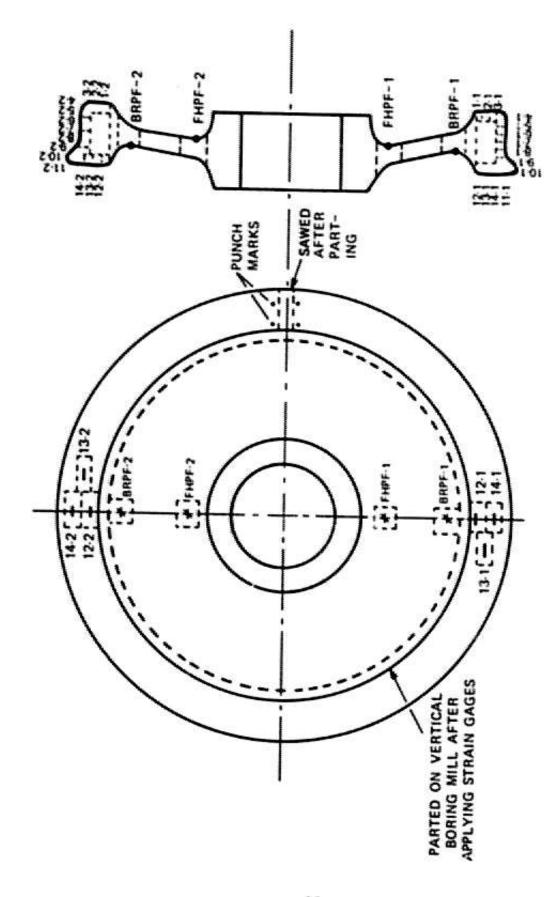


FIGURE 6. LOCATION OF STRAIN GAGES ON SECTION B-36 WHEEL

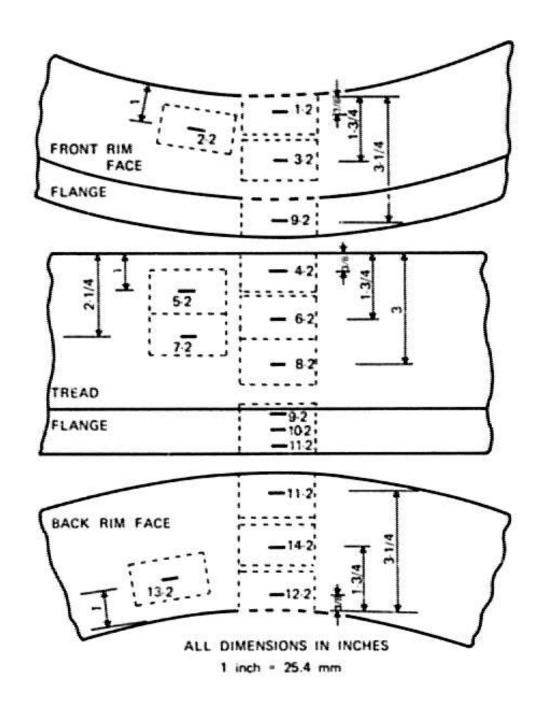


FIGURE 7. LOCATION OF STRAIN GAGES ON RIM OF SECTION B-36 WHEEL

silicone rubber, and steel sheet) were applied to the outside gage surface and to the adjacent steel surface to protect the gages from environmental and mechanical damage. After reference measurements of the strain gages were made, the wheel rim was dissected. Sections I inch square by 2 inches long with the strain gage on one surface were then saw-cut from the rim, and final measurements of the strain gages were made. Using the accepted value of 30 million psi (207 GPa) for Young's modulus, the tangential residual stresses were calculated from the relaxation strains (differences between the reference and final measurements of the strain gages).

The residual stresses at the bases of the back-rim-plate fillet and front-hub-plate fillet\* were measured by means of 45-degree (\*/4 rad) rectangular stacked-rosette electrical-resistance foil strain gages (Micro-Measurements type WA-06-250WR-120 with 1/4-inch gage length). The strain gages were cemented to the wheel plate and coated for environmental protection in the same manner as the rim gages. Before dissecting the wheel, reference measurements of the strain gages were made. Sections 2-1/2-inches (64 mm) square by plate thickness with the strain gages cemented to one surface were then sawed from the wheel, and the final measurements of the strain gages were made. The principal residual stresses were calculated from the measured relaxation strains, which are the differences between the reference and final strain-gage measurements.

The formulas for calculating the principal stresses are as follows:

These locations are defined as the points on the plate-surfaces that have the greatest lateral distance from the back-rim face and the front-hub face.

$$S_{max} = \frac{E}{1-\mu} \left( \frac{e_r \cdot e_t}{2} \right) \cdot \frac{\sqrt{2}}{2} \left( \frac{E}{1+\mu} \right) \sqrt{\left(e_r \cdot e_d\right)^2 \cdot \left(e_d \cdot e_t\right)^2}$$

$$S_{min} = \frac{E}{1-\mu} \left( \frac{e_r \cdot e_t}{2} \right) \cdot \frac{\sqrt{2}}{2} \left( \frac{E}{1+\mu} \right) \sqrt{\left(e_r \cdot e_d\right)^2 \cdot \left(e_d \cdot e_t\right)^2}$$

#### where

e, . strain in radial direction

e, . strain in tangential direction

ed = strain in 45° direction

E \* Young's modulus (30 x 106 psi)

μ = Poisson's ratio (0.3)

#### 3. RESULTS AND DISCUSSION

#### 3.1 METALLURGICAL STUDIES

#### 3.1.1 Chemical Composition

The chemical compositions of the six wheels are shown in Table 2, along with the specified range for AAR Specification M-107, Class A wheels. The chemical compositions of the wheels conformed to the specified ranges; however, their carbon contents (about 0.421) were deliberately lower than typical (about 0.521) for Class A wheels in order to improve their thermal-crack resistance and toughness.

#### 3.1.2 Tensile Properties

The results of tension tests of 0.252-inch-diameter (6.4 mm) chordal specimens removed from the midthickness of the rims of the wheels are shown in Table 6. The tensile properties of the six wheels were similar to one another and were at levels that would be expected for rim-treated 0.4 percent carbon-steel wheels.

### 3.1.3 Impact Properties

The results of impact tests of cherdal Charpy V-notch specimens from the midthickness of the rims of the wheels, Table 6, showed that the six wheels had essentially equivalent 50 percent shear-fracture-appearance temperatures, in the range 110 to 130°F (43 to 54°C). These values are normal for rim-toughened 0.4 percent carbon-steel sheels.

#### 3.1.4 Hardness

The results of Brinell hardness tests of a cross section of each wheel rim, Table 7, showed that the hardnesses of the wheel rims were in the range expected for rim-treated 0.4 percent carbonsteel wheels.

TABLE 6. MECHANICAL PROPERTIES OF METROLINER WHEELS (CIRCUMFERENTIALLY ORIENTED SPECIMENS FROM THE RIMS)

		Tensile Properties	erties		Impact
Wheel No.	Yield Strength (0.2% Offset), ksi	Tensile Strength, ksi	Elongation in   Inch, percent	Reduction of Area, percent	Properties 50% Shear Temperature,"
• •	9.49	110	20.0	48.2	+115
	67.6	112	21.5	45.4	+130
	70.5	110	21.0	45.3	+130
•••	11.3	1112	20.0	46.2	+125
25	73.6	111	20.0	47.0	•110
	66.3	108	21.5	49.9	+130

. After dynamometer tests.

.. NGW.

... Used - with thermal crack.

Conversion Factors

1 ksi = 6.895 MPa 1 inch = 25.4 mm "C = 5/9(\*F = 32)

TABLE 7. BRINELL HARDNESS\* OF RIMS OF METROLINER WHEELS

	Wheel	No.	1			Wheel	No.	2	_
255					269	-			
255	241	241	241	255	262	248	248	255	255
229	217	217	217	241	248	235	229	229	235
207	212	207	217	207	229	223	229	217	217
	Wheel	No.	3			Wheel	No.	4	
269					262				
225	241	269	269	269	248	241	235	235	241
241	229	223	223	229	229	229	224	224	235
223	212	207	207	207	217	229	224	229	217
	Wheel	No.	5		2000-120	Wheel	No.	6	_
255					255				
241	241	285	285	285	235	235	248	255	269
229	229	229	235	255	217	217	217	223	235
21	7 217	229	229	235	207	217	217	212	21

<sup>\*</sup> Brinell hardness (Bhn) values are arranged as shown in Figure 2.

#### 3.1.5 Visual Examination of Wheels

The tread of the untested new wheel (No. 4) exhibited no surface imperfections, as would be expected. The treads of both used wheels (No. 5 and 6) showed only a slight amount of wear, as would be expected from their short service period. Fluorescent magnetic-particle inspection, Figure 8, showed that wheel No. 5 had a 1-1/4-inch-long (32 mm) crack in the tread. The crack was oriented in the radial-axial plane. The appearance of the crack after being broken open in shown in Figure 9. It is apparent from the beach marks and oxidation that the crack was a fatigue-type thermal crack. The crack had a depth of about 3/16 inch (5 mm). Cracks were not present in the tread of wheel No. 6; neither wheel No. 5 nor wheel No. 6 exhibited cracks in the flange.

The tread of wheel No. 1 exhibited only slight grooving from brakeshoe-wheel interactions and no thermal cracks after 300 simulated emergency stops. As shown in Figure 10, wheel No. 2 had deep grooving on the tread but no thermal cracks on the tread or flange after 2,424 simulated speed-reduction brake applications. Wheel No. 3 exhibited slight grooving and several small thermal-fatigue cracks, Figure 11. As noted previously, these cracks developed during the emergency-stop braking which followed the braking with wornout brakeshoes.

### 3.1.6 Macrostructure and Microstructure

As shown in Figure 12, the macrostructure of the rim of the new wheel (No. 4) was uniform, and the microstructure consisted of fine pearlite with proeutectoid ferrite, principally at the prior-austenite grain boundaries, a normal microstructure for a new rimtreated 0.4 percent carbon-steel wheel.

Wheel No. 5, which exhibited a thermal crack developed in service, showed evidence of thermal damage from brakeshoe action. As shown in Figures 13 and 14, the tread of this wheel had been intermittently overheated along the circumference. The overheated regions had widths of about 3/4 inch (19 mm), lengths of from 1/2 to 1 inch (12 to 25 mm), and thicknesses of up to 0.06 inch (1.5 mm).



Marks made by brakeshoe

Thermal crack

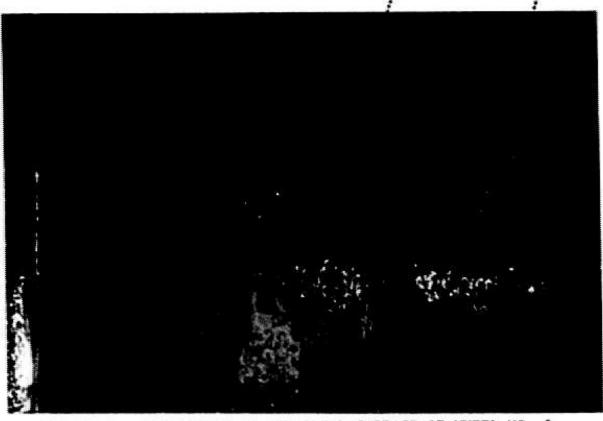


FIGURE 8. APPEARANCE OF THE TREAD SURFACE OF WHEFL NO. 5, SHOWING A THERMAL CRACK AS REVEALED BY FLUORESCENT MAGNETIC-PARTICLE INSPECTION. SCRIBNER MARKS ON CORNER OF TREAD WERE USED FOR REFERENCE FOR NONDESTRUCTIVE TESTS CONDUCTED BY OTHER CONTRACTORS (ABOUT 2X)

Marks made by brakeshoe



FIGURE 9. THERMAL-FATIGUE CRACK IN WHEEL NO. 5 AFTER BEING BROKEN OPEN (ABOUT 3X)

Area with deep grooves



FIGURE 10. APPEARANCE OF TREAD AND FLANGE OF WHEEL NO. 2 AFTER SPEED-REDUCTION BRAKE TESTING. NOTE GROOVES CAUSED BY BRAKE-SHOE-WHEEL INTERACTION (ACTUAL SIZE - SEE FIG. 20 FOR ILLUSTRATION OF GROOVE DEPTH)

Steel deposit from brakeshoe backing plate

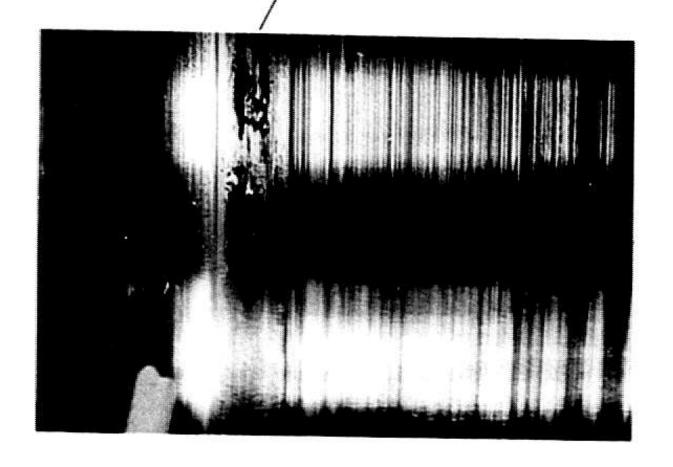
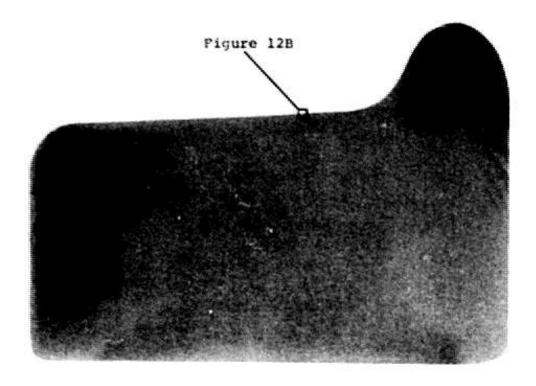
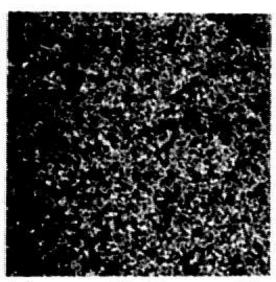


FIGURE 11. APPEARANCE OF TREAD OF WHEEL NO. 3 AFTER TESTING ON WHEEL-TEST MACHINE (DRAG BRAKING AND EMERGENCY-STOP BRAKING WITH WORNOUT BRAKESHOES). THE FENCIL POINTS TO SEVERAL SMALL THERMAL-FATIGUE CRACKS (ABOUT 2X)



A. Macrostructure of radial-axial section. Nital etch. Actual size



B-1. 300 dph. X50.



B-2. 300 dph.

X500.

B. Microstructure and microhardness of tread. Superpicral etch.

FIGURE 12. MACROSTRUCTURE, MICROSTRUCTURE, AND MICROHARD-NESS OF RADIAL-AXIAL SECTION OF RIM OF WHEEL NO. 4 (NEW AND UNTESTED)

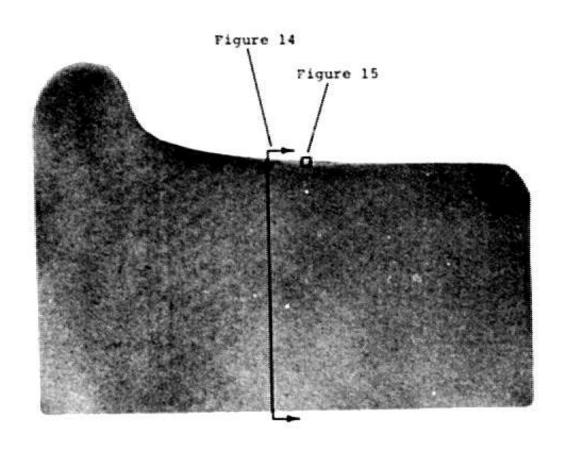
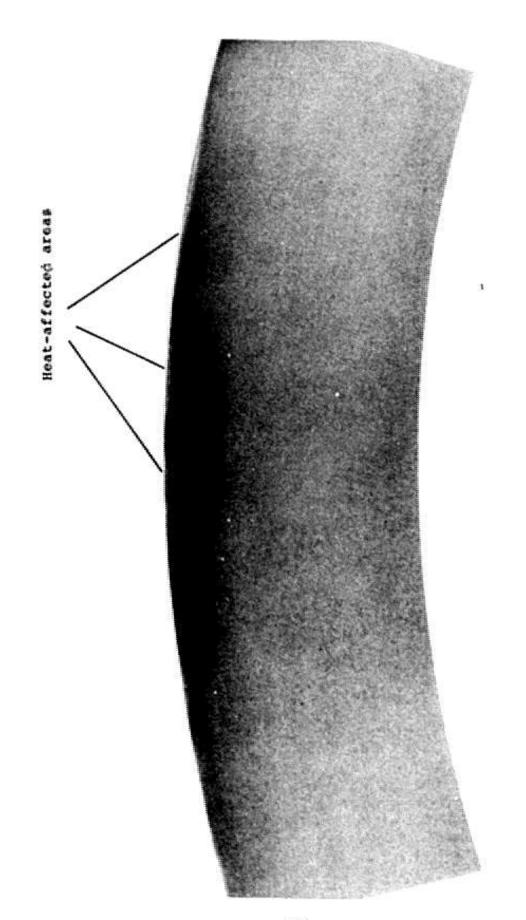


FIGURE 13. MACROSTRUCTURE OF RADIAL-AXIAL SECTION OF RIM OF WHEEL NO. 5, SHOWING LIGHT-ETCHING STRUCTURE RESULTING FROM TREAD OVERHEATING. NITAL ETCH (ACTUAL SIZE)



14. MACROSTRUCTURE OF RADIAL-TANGENTIAL SECTION OF RIM OF WHEEL SHOWING LIGHT-ETCHING STRUCTURE RESULTING FROM OVERHEATING OF NITAL ETCH (ACTUAL SIZE) FIGURE NO. 5. TREAD.

These regions were randomly spaced around the circumference of the tread at intervals of from 1 to 4 inches (25 to 100 mm) at a distance of 2 to 3 inches (50 to 75 mm) from the front face of the rim. Figure 15 shows the range of microstructures obtained in one such heat-affected area. Most of the heat-affected area consisted of a very-fine-grained gerrite-pearlite microstructure. The transsition zone, which had a thickness of about 0.005 inch (0.13 mm), consisted of ferrite and partially spheroidized pearlite with the same grain size as the unaltered ferrite-pearlite of the remainder of the rim. The microhardnesses of these microstructures were all about 300 dph. The thermal-fatigue crack (Figure 9) had initiated in one of these overheated areas. On the basis of the microstructural studies of wheel No. 5, it is estimated that the tread surface was heated to a high enough temperature (above 1333°F or 723°C) to cause reaustenitization, but the cooling rate was insufficient to result in the formation of hard and brittle martensite. In contrast, Figures 16 and 17 show that the macrostructure of the rim of used wheel No. 6 was uniform and devoid of "heataffected" areas. Furthermore, microstructure examinations, Figure 18, of areas near the tread surface of wheel No. 6 showed ferrite and pearlite; there was no evidence of overheating.

On the basis of these studies, it is concluded that wheel No. 5 had been overheated by brakeshoe action while in service and that the mate wheel (No. 6) had not. Such unequal braking conditions are possible inasmuch as the two wheels were independently braked. Because it was suspected that wheel No. 5 had been braked with wornout brakeshoes, causing the overheating and "hot" spots observed, simulated wornout brakeshoes were used in the final series of dynamometer tests performed on wheel No. 3 to test this hypothesis.

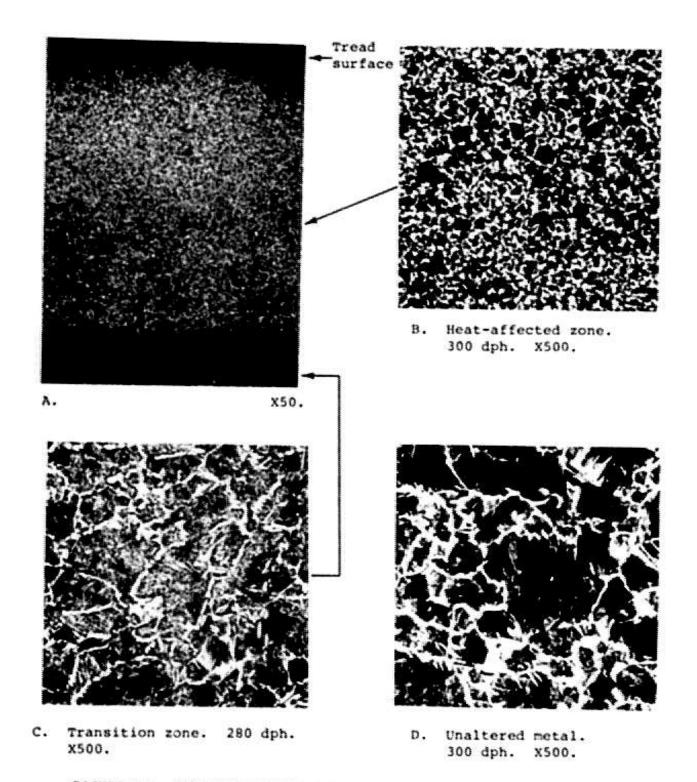


FIGURE 15. MICROSTRUCTURE AND MICROHARDNESS OF RADIAL-AXIAL SECTION OF RIM OF WHEEL NO. 5 THROUGH HEAT-AFFECTED AREA. SUPER-PICRAL ETCH

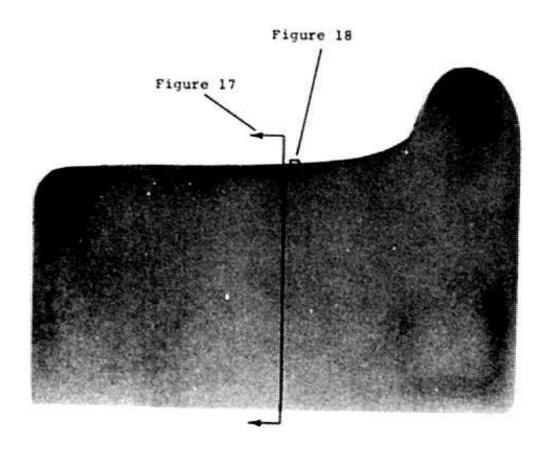


FIGURE 16. MACROSTRUCTURE OF RADIAL-AXIAL SECTION OF RIM OF WHEEL NO. 6. NITAL ETCH (ACTUAL SIZE)

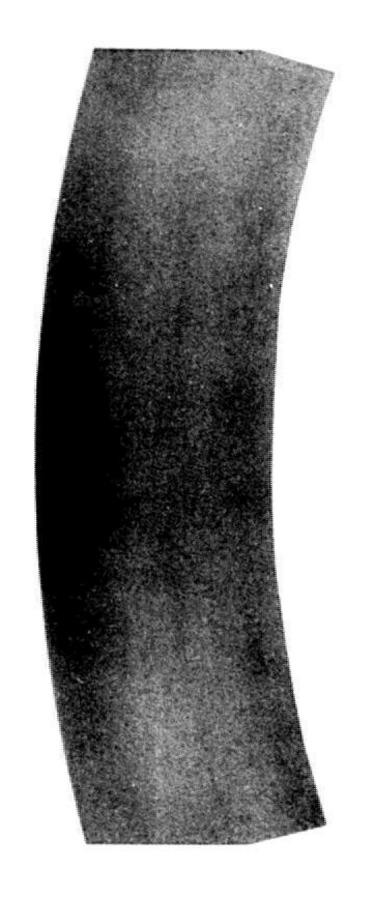


FIGURE 17. MACROSTRUCTURE OF RADIAL-TANGENTIAL SECTION OF RIM OF WHEEL NO. 6. NITAL ETCH (ACTUAL SIZE)

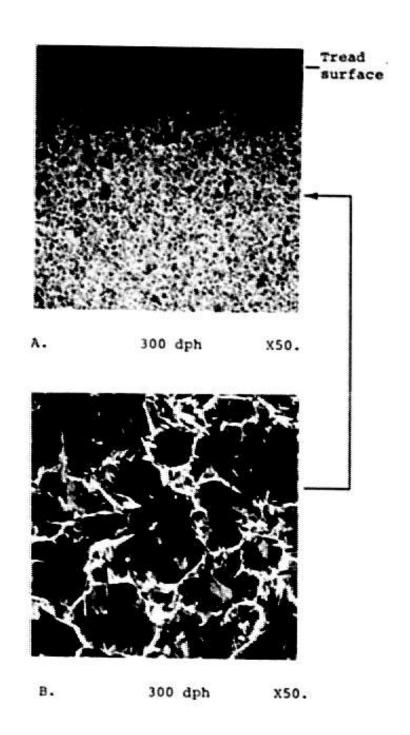
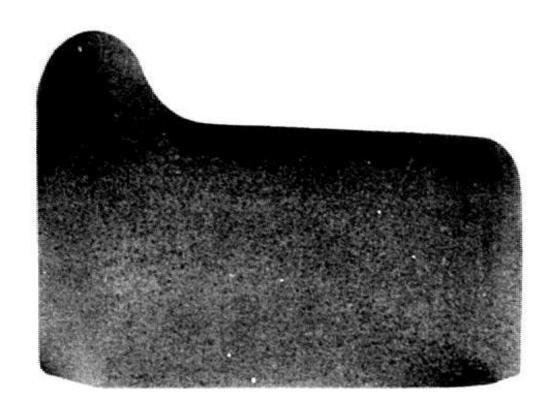
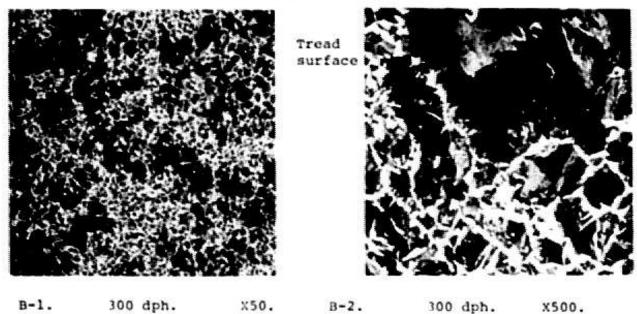


FIGURE 18. MICROSTRUCTURE AND MICROHARDNESS OF RADIAL-AXIAL SECTION OF TREAD OF WHEEL NO. 6. SUPER-PICRAL ETCH

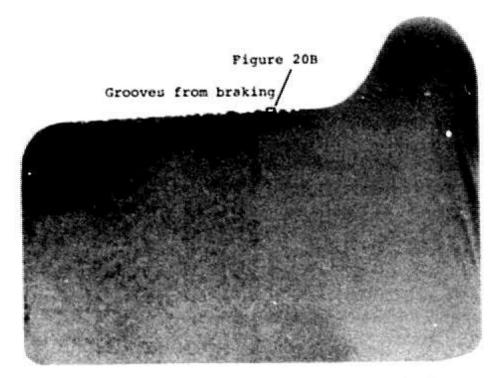


A. Macrostructure of radial-axial section. Nital etch. Actual size.

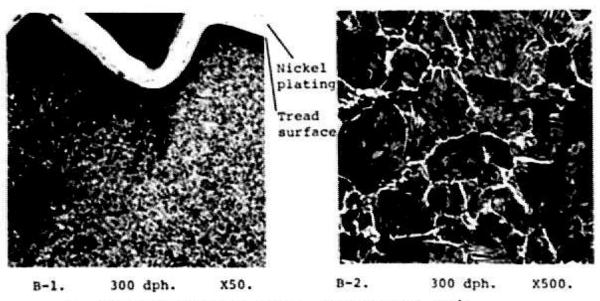


B. Microstructure of radial-axial section at tread center. Super-picral etch.

FIGURE 19. MACROSTRUCTURE, MICROSTRUCTURE, AND MICROHARDNESS OF RADIAL-AXIAL SECTION OF RIM OF WHEEL NO. 1



 Macrostructure of radial-axial section. Nital etch. Actual size.



B. Microstructure of tread. Super-picral etch

FIGURE 20. MACROSTRUCTURE, MICROSTRUCTURE, AND MICROHARDNESS OF RADIAL-AXIAL SECTION OF RIM OF WHEEL NO. 2

Figure 19 shows the macrostructure and microstructure of the rim of wheel No. 1 after 300 simulated emergency stops. No evidence of thermal damage caused by braking was detected in the macrostructure. The microstructure near the tread consisted of fine pearlite with proeutectoid ferrite at the prior-austenite grain boundaries. Figure 20 shows the macrostructure and microstructure of the rim of wheel No. 2, the wheel that had been subjected to speed-reduction braking on the dynamometer. The macrostructure was uniform with no evidence of overheating near the tread and flange. In addition, the grooves resulting from braking are visible. Examination of the microstructure confirmed that there was no overheating.

The macrostructure and microstructure of wheel No. 3, which had been subjected to drag braking, emergency-stop braking, and braking with wornout brakeshoes on the wheel-test machine, are shown in Figures 21, 22, and 23. The results show that this wheel was overheated intermittently around the tread circumference in much the same manner as wheel No. 5. The microstructure and size of these overheated areas. Figure 23, were similar to those in wheel No. 5, Figure 15.

#### 3.2 RESIDUAL STRESSES

Data from the residual-stress studies are listed in Appendixes F and G and summarized in Figures 24 through 29.

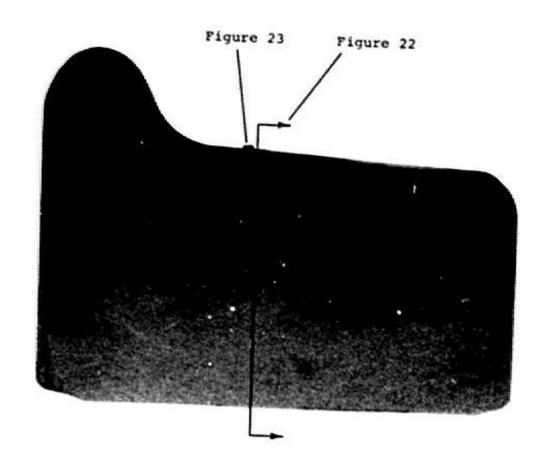


FIGURE 21. MACROSTRUCTURE OF RADIAL-AXIAL SECTION OF RIM OF WHEEL NO. 3, SHOWING LIGHT-ETCHING "HEAT-AFFECTED" AREA ON TREAD. NITAL ETCH (ACTUAL SIZE)

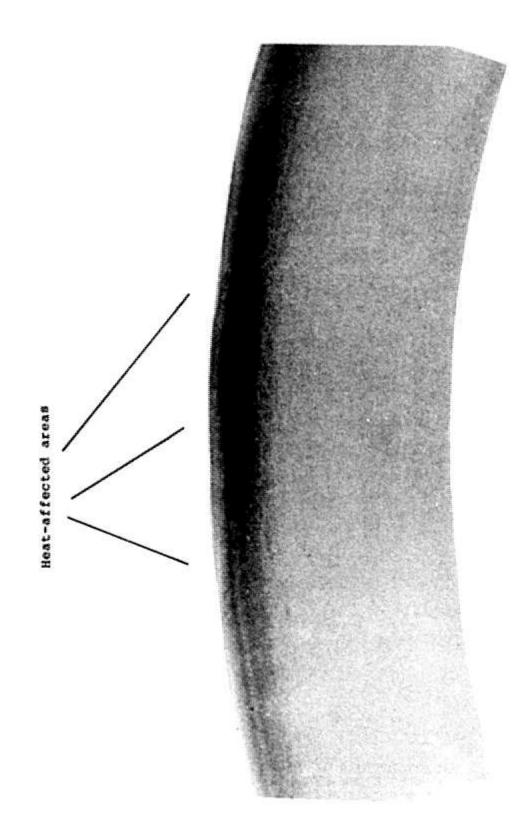


FIGURE 22. MACROSTRUCTURE OF RADIAL-TANGENTIAL SECTION OF RIM OF WHEEL NO. 3, SHOWING LIGHT-ETCHING "HEAT-AFFECTED" AREA ON TREAD. NITAL ETCH (ACTUAL SIZE)

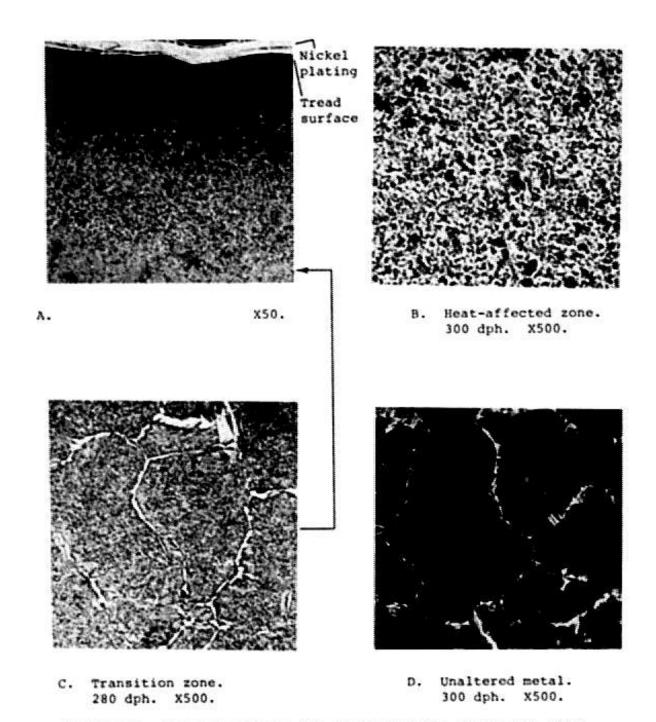
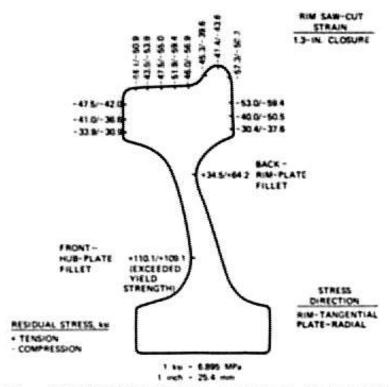


FIGURE 23. MICROSTRUCTURE AND MICROHARDNESS OF RADIAL-AXIAL SECTION OF RIM OF WHEEL NO. 3 THROUGH HEAT-AFFECTED AREA. SUPER-PICRAL ETCH



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FIGURE 24. INDICATED RESIDUAL STRESSES IN NEW WHEEL NO. 4 (1-72-S-2466-A)

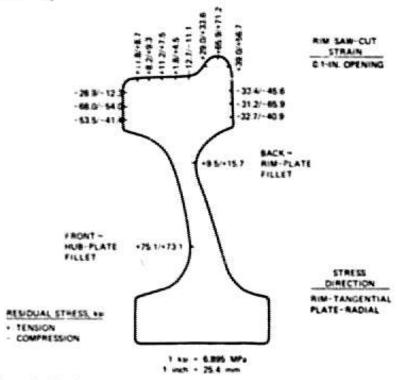


FIGURE 25. INDICATED RESIDUAL STRESSES IN USED WHEEL NO. 5 (1-71-S-10191-A) WHICH CONTAINED FATIGUE-TYPE THERMAL CRACK

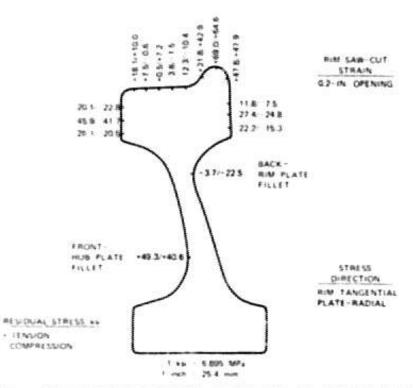


FIGURE 26. INDICATED RESIDUAL STRESSES IN USED WHEEL NO. 6 (1-71-S-10190-A)

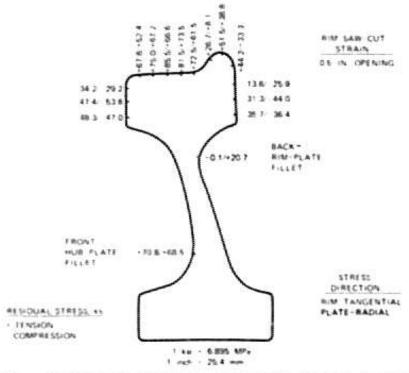


FIGURE 27. INDICATED RESIDUAL STRESSES IN NEW WHEEL NO. 1 (1-72-S-2476-A)

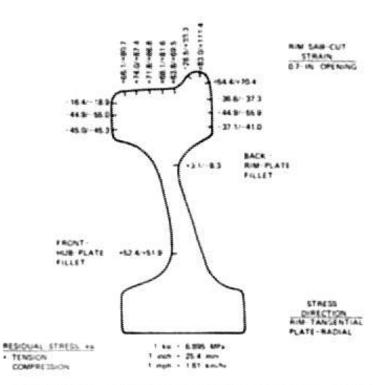


FIGURE 28. INDICATED RESIDUAL STRESSES IN NEW SHEEL NO. 2 (1-72-S-2474-A) AFTER 2424 SPEED-CONTROL BRAKING FROM 100 TO 50 MPH

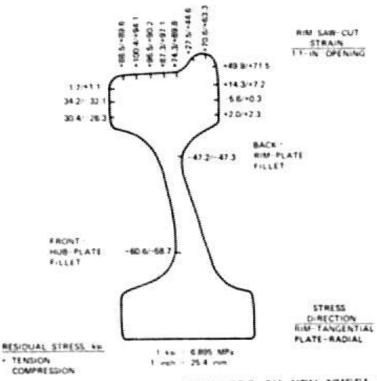


FIGURE 29. INDICATED RESIDUAL STRESSES IN NEW WHEEL NO. 3 (1-72-S-2472-A) AFTER COMBINATION OF DRAG BRAKING EMERGENCY-STOP BRAKING AND EMERGENCY-STOP BRAKING WITH WORNOUT BRAKESHOE

## 3.2.1 New, Untested Wheel

As part of the manufacturing process, rim-treated\* railroad wheels are subjected to a heat treatment that increases the hardness of the rim for improved wear resistance. As a result, overall residual hoop compressive stresses are produced in the rim, and residual radial tensile stresses are produced in the critical plate fillets. The results of residual-stress studies of the new, untested wheel (No. 4) showed that the residual-stress pattern of this wheel was normal (see Figure 24), with compressive stresses in the outer 1-inch-thick (25.4 pm) layer of the rim. The rim saw-cut strain value of 1.3-inch (33 mm) closure also indicated that the overall residual hoop stress was compressive.

## 3.2.2 Used Wheels

During service, the as-manufactured wheel stresses are altered by the thermal effects of on-tread braking and by cold working of the tread surface by rolling loads. With regard to the residual. stress changes caused by braking, repeated severe of prolonged braking converts the rim stress from compression to tension and lowers (and sometimes reverses) the tensile stresses at the critcal plate fillets. Conversely, rail contact loading induces compressive stresses into the tread surface. Because braking and rolling loads have opposite effects on the stresses in the tread areas, the i lependent magnitude of each effect cannot be readily determined. Vevertheless, the results of the residual-stress studies of the two used wheels showed that both wheels had been subjected to severe braking (see Figures 25 and 26). Evidence for this conclusion is the high residual tensile stress in the flange and the relatively low residual tensile stresses in the back-rimplate fillets. (The back rim-plate fillet of wheel No. 6 was

Essentially the same rim-treating practice is followed by all wheel manufacturers to increase the strength and hardness of the wheel rim. In this heat-treatment process, the wheel is heated to the austenitizing temperature, and only the rim portion is quenched. The rim and hub portions are then permitted to aircool below the critical temperature before the wheel is tempered.

actually in compression.) The rim saw-cut strain values of 0.1and 0.2-inch (2.5 and 5.1 mm) openings for wheels No. 5 and 6, respectively, indicate that both wheels had low overall residual hoop tensile stresses.

#### 3.2.3 Dynamometer-Tested Wheels

The results of studies of the residual stresses in wheel No. 1, the wheel that had been subjected to 300 simulated emergency stops on the wheel-test machine, indicated that this wheel had developed high tensile stresses in the tread as well as in the flange (see Figure 27). These tensile stresses were much higher than those in the treads of the two used wheels because the wheel had not been subjected to rolling loads. The residual stresses in the critical plate fillets of this wheel were similar in magnitude to those in the used wheels. The rim saw-cut strain value for this wheel was 0.5-inch (12.7 mm) opening, indicating that the rim was in overall residual hoop tension.

The residual stresses in wheel No. 2, subjected to 2,424 speedreduction braking on the wheel-test machine, were similar to those
of wheel No. 1; high tensile stresses were present in the tread
and flange, and the rim saw-cut strain velue was 0.7-inch (17.8 mm)
opening, as shown in Figure 28.

wheel No. 3 also had high residual hoop tensile stresses in the tread and flange (see Figure 29), but it had compressive stresses in the plate fillets. This wheel had been subjected to drag and emergency-stop braking with a wornout brakeshoe on the wheel-test machine. It is believed that the residual compressive stresses in the critical plate fillets were caused by the dragbraking cycles. In addition, the rim saw-cut strain value of 1.1-inch (27.9 mm) opening indicates that this wheel had the highest overall rim tensile stress of the wheels tested on the dynamometer. In wheel No. 3, the fine tread-surface cracks were initiated by the thermal alteration resulting from abnormal braking (that is, with wornout shoes). It is believed that stress concentrations at these cracks during subsequent brakings would cause the cracks to propagate by fatigue.

### 4. SUMMARY AND CONCLUSIONS

The wheels used on Metroliner cars reportedly develop thermal cracks during service even though the wheels are made from wheel-steel with a high resistance to thermal cracking. Therefore, the present study was conducted to determine the cause of the cracking.

One new wheel and two used wheels (one with a thermal crack developed in the tread during service) were examined for mechanical properties, macrostructure, microstructure, and residual stresses. Similar examinations were conducted on three new wheels which were first subjected to various braking cycles (on the laboratory dynamometer) designed to define the conditions that produce cracking.

The results of the study indicated that the wheel that had developed a thermal crack in service had been intermittently and severely heated around the tread surface and that such heating had altered the microstructure, produced residual tensile stresses, and permitted the crack to initiate. On the basis of this finding, it was hypothesized that the wheel had been braked with a wornout brakeshoe, a condition that could produce hot spots similar to those observed. This hypothesis was confirmed by results of studies of the dynamometer-tested wheels, which showed that emergency-stop braking with a simulated wornout brakeshoe could produce an altered microstructure and small thermal cracks similar in type to the crack in the service-cracked wheel. The results further indicated that neither altered microstructure nor cracking could be produced by numerous emergency or speed-reduction breakings with normal brakeshoes and forces.

# APPENDIX A - U.S.S. RAILROAD WHEEL TESTING MACHINE

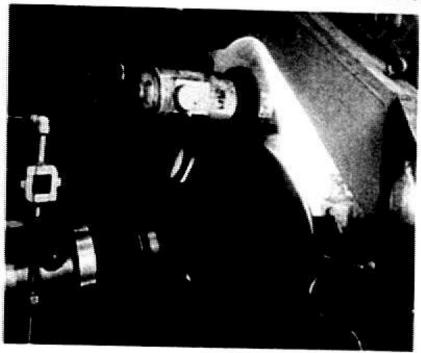


FIGURE A-1. SIMULATED EMERGENCY STOP OF PASSENGER CAR WHEEL FROM INITIAL SPEED OF 120 MPH SHOWS FLAMING OF BRAKESHOES

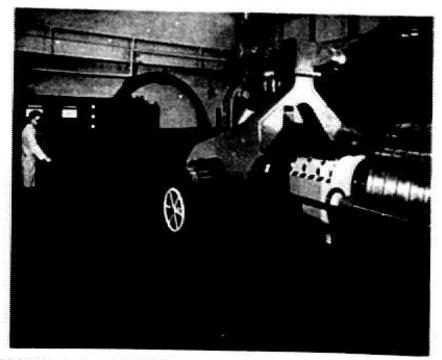


FIGURE A-2. GENERAL VIEW OF WHEEL TESTING MACHINE

APPENDIX B

DATA OBTAINED DURING STOP TEST OF NEW WHEEL NO. 1

(1-72-S-2476A)

Stop No.	Braking Time, sec	Braking Distance, ft	Total Work, 106 ft-1b	Stop No.	Braking Time, sec	Braking Distance, ft	Total Work, 10 <sup>6</sup> ft-li
1.	90	9125	NR	41	87	8630	11.2
2**	25	NR	NR	42	87	8560	11.2
3	96	8922	NR	43	87	8504	11.0
4	93	8851	11.9	44	87	8598	11.2
5	91	8824	NR	45	88	8813	11.6
5	91	8735	NR	46	89	8868	12.4
7	89	8600	13.3	47	91	8854	12.5
8	90	8654	12.3	48	91	8921	12.8
9	94	8257	NR	49	9.	8774	12.8
10	93	8526	10.6	50	95	9501	13.4
11.	112	10756	11.7	51*	132	13135	12.1
12	97	9551	11.9	52	113	11341	11.6
13	95	9265	NR	53	107	10840	11.9
14	93	8980	NR	54	99	10065	11.7
15	93	9188	NR	55	98	9949	11.6
16**	24	3910	NR	56	97	9787	11.6
17	85	8159	11.5	57	93	9292	NR
18	87	8427	10.1	58	88	8930	NR
19	88	8575	10.6	5.9	87	8660	NR
20	90	8665	9.8	60	89	8727	NR
21	91	8997	NR	61	86	8230	13.2
22	86	8945	11.6	62	85	8183	NR
23	83	8331	9.8	63	83	8120	12.2
24	87	8701	9.8	64	83	8276	11.6
25	86	8519	NR	65	86	8353	13.1
26	87	8717	11.3	66	85	8212	11.5
27	85	8549	9.2	67	84	8103	13.2
28	84	8560	9.8	68	84	8291	12.5
29	86	8396	10.8	69	86	8498	13.3
30	87	8234	11.4	70	86	8342	12.9
31	88	8485	11.3	71	85	8424	12.5
32	89	8615	11.6	72	86	8325	12.4
33	87	8671	11.5	73	87	8384	11.1
34	89	8813	11.6	74	85	8238	13.3
35	87	8614	11.3	75	82	7970	13.5
36	87	8722	11.7	76	80	7668	13.5
37	88	P654	12.0	77	82	8049	13.4
38	88	8803	11.3	78	80	7867	13.6
39	86	8738	11.2	79	80	7829	11.9
40	87	8666	11.3	80	89	9539	13.0

Stop No.	Braking Time, sec	Braking Distance, ft	Total Work, 10 <sup>6</sup> ft-lb	Stop No.	Braking Time, sec	Braking Distance, ft	Total Work, 10 <sup>6</sup> ft-11
81	87	8593	12.0	121		75.40	122212
82	84	8140	12.9	122	82	7949	12.8
83	84	8032	12.9 12.7	123	81 82	7951	12.8
84	86	8200		124	82	7856	13.1
85	87	8654	12.5 13.0	125	80	7893	11.5
86	85	8436	12.7	126	78	7712	11.3
87	87	8462	13.0	127	78	7568	12.8
88	86	8314	13.0	128	77	7613	11.5
89	84	7994	13.2	129	79	7375	11.4
90	83	8156		130	79	7590	10.9
91	82	8070	12.7	131	84	7684	12.6
92	88	8501	12.9	132	81	8264	13.1
93	89	8701	12.6	133		7962	12.8
94	87	8392	12.7	134	81 82	7959	12.9
95	88	8618	12.5	135	79	7842	13.1
96	89	8756	12.9	136	79	7639	12.5
97	86	8674	13.3	136	79	7671	12.9
98	89	■ 00000 LU000	12.4	138	77	7634	12.8
99	90	8809	13.1	139	79	7439	12.9
100	86	8890	11.3			7909	12.7
101	E17 110 100 100 100 100 100 100 100 100 1	8335	12.6	140	80	7824	12.8
	8.6	6771	13.6	141	81	7804	12.6
102	86	8431	11.4	142	79	7578	12.7
103	87	8514	13.1	143	79	7718	12.7
	91	8872	13.0	144	80	7738	12.8
105	89	8642	NR	145	78	7576	12.9
106	96	8330	13.1	146	83	8351	12.2
107	86	8387	12.4	147	83	8182	12.5
108	88	8476	12.9	148	83	8145	13.1
109	89	8716	13.2	149	82	7997	12.5
110	87	8525	13.1	150	82	8011	12.4
111.	138	14091	12.7	151	83	8108	13.7
112	121	12124	13.3	152	85	8480	13.5
113	109	10926	13.6	153	84	8293	12.8
114	99	9966	13.0	154	84	9297	13.1
115	94	9427	13.0	155	83	8057	12.0
116	92	9242	13,1	156	81	7963	12.5
117	90	8918	13.1	157	81	7846	12.7
118	89	8937	12.5	158	82	7962	12.8
119	87	8698	12.9	159	84	8196	13.0
120	84	8206	11.6	160	85	8404	12.4

Stop No.	Braking Time, sec	Braking Distance, ft	Total Work, 106 ft-1b	Stop No.	Braking Time, sec	Braking Distance, ft	Total Work, 106 ft-lb
161	89	8501	13.2	201	81	8078	12.8
162	87	8452	13.1	202	82	8029	13.0
163	85	8326	11.7	203	81	7932	12.8
164	85	8245	12.5	204	82	8043	13.4
165	84	8075	12.2	205	80	7937	13.3
166	80	7799	12.9	206	81	8044	13.7
167	83	8172	12.0	207	80	7942	13.8
168	83	8154	11.9	208	80	7885	13.2
169	84	8198	12.0	209	79	7759	13.5
170	83 8079		12.1	210	79	7807	13.0
171	86 8395		12.5	211	79	7799	NR
172	88	8534	11.6	212	85	8509	13.0
173	89	8682	12.0	213	81	8027	13.1
174	84	8026	12.0	214	83	8025	12.5
175	84	8205	12.5	215	78	7622	12.2
176	84	8213	13.1	216	78	7725	12.4
177	85	8331	11.4	217	78	7685	13.2
178	85	8398	11.4	218	82	8028	13.1
179	84	8361	12.2	219	79	7748	12.2
180	87	8677	12.6	220	82	7951	12.7
181	89	8710	12.4	221	78	7737	11.5
182*	128	13058	12.2	222	80	8006	11.5
183	116	11600	12.8	223	81	8025	13.2
184	113	11333	NR	224	82	8007	12.7
185	103	10351	13.0	225	82	8053	12.2
186	100	10028	12.0	226	80	7911	13.3
187	97	9767	13.7	227	81	7883	13.5
188	94	9549	12.0	228	80	7791	13.2
189	91	9145	12.5	229	79	7664	13.4
190	91	9090	13.4	230	82	8071	12.4
191	90	9057	12.8	231	79	7883	12.9
192	90	9046	12.9	232	82	8052	13.0
193	87	8769	13.0	233	84	8194	13.2
194	86	8556	NR	234	82	8075	NR
195	83	8299	13.1	235	86	8521	12.2
196	81	8014	12.2	236	83	8249	13.3
197	83	8401	12.8	237	85	8306	13.2
198	81	8168	12.4	238	86	8650	12.6
199	79	7744	12.6	239	95	9412	12.9
200	81	7987	12.7	240	85	8513	12.5

Stop No.	Braking Time, sec	Braking Distance, ft	Total Work, 10 <sup>6</sup> ft-lb	Stop No.	Braking Time, sec	Braking Distance, ft	Total Work, 106 ft-11
241	87	8683	14.0	271	117	11833	12.4
242	84	8093	13.2	272	103	10476	12.5
243	85	8304	13.6	273	100	10156	12.1
244	85	8469	12.9	274	98	9974	NR
245	91	8957	12.6	275	99	10155	12.5
246	87	8569	12.1	276	96	9825	12.6
247	85	8446	13.0	277	91	9218	12.8
248	84	8106	12.8	278	89	8909	12.9
249	87	8716	13.0	279	86	8488	13.0
250	87	8610	13.0	280	83	8206	12.7
251	85	8393	13.4	281	88	9069	12.5
252	86	8449	11.7	282	85	8470	12.2
253	86	8538	11.5	283	83	8248	12.2
254	94	9397	13.8	284	83	8052	11.9
255	88	8641	13.7	285***	210	12414	12.6
256	89	8775	12.4	286	84	8430	12.8
257	86	8412	12.2	287	82	8055	12.6
258	90	2840	13.0	288	81	7904	12.7
259	89	8290	13.3	289	81	7959	12.8
260	101	9340	13.0	290	80	7958	12.6
261	89	8644	12.7	291	80	7872	12.6
262	88	8661	12.9	292	83	8353	13.0
263	83	8228	13.3	293	82	8057	12.8
264	85	9247	13.1	294	81	7924	13.0
265	87	8799	13.4	295	79	7624	12.4
266	89	6872	12.0	296	79	7676	12.2
267	89	8756	12.1	297	80	7791	12.2
268	91	9044	12.2	298	78	7666	12.2
269	87	8679	12.4	299	78	7679	12.5
270*	131	13474	11.9	300	78	7765	13.4

<sup>\*</sup> New brake shoe installed.
\*\* Incomplete stop.
\*\*\* Low brake pressure.

NR = Not recorded.

## Conversion Factors

1 ft = 0.3 m 1 ft-1b = 1.36 J

APPENDIX C

DATA OBTAINED DURING SPEED-REDUCTION BRAKED TEST OF NEW WHEEL NO. 2

(1-72-2474A)

Cycle	Bra	king sec	Time,		oluti	ons	Cycle	Bra	king sec	Time,	Rev	oluti	ons
No.	1		3	1		3	No.	1		3	1		3
1	Not	Pac	orded	Not	Recor	dad	36	32	33	34	376	396	404
2	33	34	35	381	394		37	31	32	34	361	379	405
3	33	35	35	375	397	400	38	30	32	33	362	385	397
	32	33	34	405	381	396	39	29	31	33	337	355	386
4 5 6	33	34	33	380	398	390	40	31	34	35	368	395	402
í	32	34	34	352	392	396	41	31	32	33	353	385	391
7	34	34	34	376	403	402	42	30	32	34	355	380	39
8	31	33	35	356	371	398	43	30	32	34	355	383	401
9	32	34	35	373	396	409	44	29	32	34	336	383	410
10	32	33	34	378	387	394	45	29	31	33	363	373	391
11	32	33	35	375	387	401	46	31	33	33	372	388	39
12	32	33	35	370	385	402	46	30	33	33	358	389	390
13	31	34	34	360	395	415	48	32	32	33	369	369	390
14	NR	32	33	NR	382	392	49	29	31	33	333	365	39
15	32	34	35	384	398	403	50	30	34	35	354	394	41
16	30	33	34	358	392	407	51	30	32	34	355	372	39
17	32	32	35	375	378	406	52	30	34	34	356	392	39
18	34	35	31	399	420	367	53	30	32	33	355	377	J8
19	31	32	32	351	383	389	54	31	32	34	355	371	40
20	30	32	33	345	369	393	55	30	31	32	344	367	38
21	32	32	33	372	381	394	56	30	31	34	350	370	40
22	31	33	34	375	391	403	57	32	32	35	381	381	40
23	31	33	34	359	389	406	58	29	31	33	344	370	39
24	31	32	34	362	373	404	59	32	30	34	376	356	39
25	32	33	35	379	400	407	60	28	30	32	324	347	38
26	31	32	34	365	383	410	61	28	31	32	325	382	39
27	32	33	34	378	399	408	62	33	34	35	379	403	41
28	32	33	35	372	190	411	63	31	32	33	366	372	39
29	31	33	34	364	389	409	64	28	29	30	330	340	35
30	31	33	33	370	386	396	63	28	31	31	334	357	36
31	33	33	33	390	391	391	66	28	31	33	324	374	39
32	31	39	34	365	444	408	67	29	31	33	334	361	38
33	34	36	39	404	435	479	68	30	31	32	342	368	37
34	31	33	32	363	385	370	69	29	31	32	343	367	37
35	32	33	32	373	396	386	70	28	31	34	325	361	39

Cycle	Bra	king sec	Time,	Rev	oluti	ons	Cycle	Bra	king	Time,	Rev	oluti	ons
No.	1	2	3	1	2	3	No.	1	2	3	1	2	3
71	29	32	32	339	370	373	106	31	33	34	375	393	١
72	29	30	33	341	350	385	107	31	33	34	371	396	411
73	NR	32	32	NR	365	378	108		5				400
74	29	29	32	341	352	376	108	32	33	33	364	397	410
75	28	30	33	323	348	381		31		34	362	384	401
76	46	48	52	549	580	601	110 111	31	32	36 34	362	386	42
77	31	33	35		2	2		30	32		354	376	40
78	34	34	35	413	404	415	112	30	33	33	353	384	388
79	34			401	403	414	113	29	32	35	348	380	410
80	33	34	35	390	396	405	114	33	34	34	386	401	40
81	•	33	34	390	390	400	115	31	33	34	368	394	40
	33	33	33	383	393	394	116	31	33	34	364	388	40
82	31	32	33	368	379	394	117	31	33	36	365	388	42
83	32	32	33	386	387	394	118	33	33	35	382	391	40
84	29	31	32	340	365	373	119	31	33	34	374	388	40
85	28	31	32	330	356	368	120	31	33	33	362	395	39
86	28	31	34	332	378	397	121	31	34	33	367	401	38
87	32	35	35	379	405	405	122	32	32	33	381	389	40
68	33	35	35	388	407	415	123	30	31	33	344	372	40
89	32	33	34	383	399	397	124	32	33	34	376	394	40
90	32	34	34	379	409	403	125	29	32	33	342	376	38
91	31	34	34	376	403	408	126	30	31	33	349	369	37
92	33	34	34	389	409	408	127	30	33	33	354	380	38
93	32	34	35	384	406	418	128	29	31	32	339	363	37
94	31	33	33	371	397	414	129	29	31	32	343	370	37
95	32	34	35	381	396	419	130	30	31	31	352	361	36
96	34	35	35	397	415	410	131	30	32	33	355	370	39
97	32	34	34	381	401	399	132	31	31	33	360	366	38
98	32	34	34	378	406	417	133	35	32	33	403	369	38
99	31	3.3	36	365	385	429	134	30	32	34	359	380	40
100	33	34	35	390	401	407	135	32	33	36	376	392	42
101	32	34	35	380	391	406	136	31	33	34	370	383	39
102	33	34	34	387	400	407	137	30	30	31	350	363	37
103	32	33	36	380	392	430	138	29	31	32	337	360	37
104	31	33	31	364	402	381	139	29	30	33	343	356	38
105	33	34	34	383	411	411	140	31	31	34	363	370	41

Cycle	Bra	king	Time,	Rev	oluti	ons	Cycle	Bra	king	Time,	Rev	oluti	ons
No.	1	2	3	1	2	3	No.	1	2	3	1		3
											712		
141	31 32	32 32	34 34	358	380	407	176	39	NR	32	485	NR	385
142			100 200	372	378	397	177	30	31	33	349	368	392
143	30	31	32	348	360	372	178	30	31	34	347	360	393
144	28	29	32	330	343	374	179	28	30	33	350	352	382
145	30	31	33	345	369	392	180	30	30	32	349	355	383
146	30	32	31	352	375	370	181	30	30	32	349	355	373
147	29	30	32	344	359	377	182	28	30	31	339	356	371
148	28	30	31	332	355	366	183	36	30	31	451	342	351
149	28	30	31	334	357	358	184	28	30	31	332	354	367
150	29	31	33	338	365	381	185	31	NR	31	358	NR	366
151	62	61	56	703	690	632	186	30	30	30	368	364	343
152	31	33	33	365	384	387	187	27	29	NR	325	348	NR
153	31	31	32	372	377	379	188	29	31	31	333	369	374
154	31	32	32	362	384	382	189	28	30	30	327	348	358
155	31	32	32	367	380	377	190	28	28	31	326	325	355
156	32	32	34	372	381	406	191	27	30	32	314	346	379
157	30	32	35	346	384	404	192	29	31	32	342	365	374
158	32	33	33	379	387	391	193	30	30	31	346	354	368
159	32	32	33	376	374	389	194	28	30	31	328	345	362
160	30	33	33	355	391	390	195	29	29	31	341	349	360
161	31	32	38	337	362	443	196	28	31	33	329	361	380
162	34	36	36	380	417	420	197	27	29	31	306	336	355
163	33	33	34	383	395	398	198	27	31	32	328	362	373
164	33	33	33	384	385	387	199	29	30	32	3.5	354	366
165	32	33	32	374	386	379	200	28	31	32	394	363	373
166	32	33	33	374	384	384	201	28	30	32	334	349	370
167	33	NR	NR	373	NR	NR	202	27	29	32	318	342	378
168	30	30	32	360	364	382	203	28	30	32	328	355	371
169	36	31	32	445	363	374	204	29	31	33	338	364	38
170	31	32	34	356	382	403	205	29	31	33	337	364	388
171	31	32	34	364	379	398	206	28	30	32	336	349	371
172	31	33	33	370	384	397	207	30	33	33	351	358	371
173	31	31	33	364	361	387	208	30	30	32	350	362	382
174	30	34	32	353	383	374	209	29	31	32	324	340	362
175	30	31	32	357	367	385	210	28	29	31	316	347	356
175	30	31	32	357	367	385	210	28	29	31	316	347	L

Cycle	Bra	sec	Time.	Rev	oluti	ons	Cycle	Bra	sec	Time,	Rev	oluti	ons
No.	1	2	3	1	2	3	No.	1	2	3	1	2	3
211	27	30	30	316	357	356	246	26	30	30	310	342	350
212	28	30	31	324	351	372	247	28	28	30	329	328	355
213	27	28	31	323	330	369	248	26	29	30	304	339	360
214	28	30	31	336	347	361	249	27	29	32	311	332	36
215	28	29	29	328	337	349	250	37	41	41	486	490	48
216	27	28	31	316	336	359	251	31	33	34	362	383	39
217	27	30	30	376	349	359	252	30	31	21	352	356	361
218	28	29	30	321	346	350	253	27	30	33	335	349	38
219	27	31	32	313	362	383	254	28	29	31	332	344	36
220	27	30	32	318	345	375	255	28	29	31	329	338	35
221	30	32	33	351	374	385	256	28	29	31	332	344	36
222	29	31	31	347	358	368	257	28	29	31	329	358	37
223	28	30	12	328	357	377	258	30	31		431	361	36
224	27	30	33	317	354	388	259	30	31	31 32	349	355	37
225	29	30	32	336	353	369	260	29	30	32	339	355	38
226	28	30	32	329	358	374	261	30	30	30	359	351	35
227	28	29	32	330	347	373	262	29	29	32	338	341	37
228	27	31	33	321	363	389	263	30	29	NR	360	338	N
229	27	29	31	317	336	358	264	31	32	34	370	374	39
230	28	30	32	332	356	371	265	30	31	32	354	329	37
231	28	31	33	329	371	398	266	30	27	31	359	376	38
232	28	28	32	324	333	384	267	30	32	33	356	375	38
233	27	30	31	326	357	380	268	31	32	32	358	355	37
234	29	30	32	341	361	382	269	31	31	32	358	367	37
235	29	30	31	341	353	370	270	29	31	32	347	363	37
236	26	29	29	304	337	345	271	29	30	31	336	356	36
237	29	30	32	337	347	378	272	28	30	32	334	355	38
238	38	26	30	472	311	352	273	27	30	31	324	354	36
239	30	31	31	350	364	371	274	28	30	32	330	349	38
240	28	29	28	317	334	330	275	28	29	31	324	335	35
241	26	29	30	313	344	353	276	28	30	32	338	349	37
242	NR	28	29	NR	325	338	277	28	30	31	328	353	36
243	25	29	30	296	339	352	278	28	30	31	329	351	36
244	NR	29	30	NR	339	397	279	29	30	32	339	359	37
245	25	28	29	297	330	322	280	29	30	31	347	358	37

Cycle	Bra	king	Time,	Rev	oluti	ons	Cycle	Bra	king sec	Time,		olutio	
No.	-31	2	3	1	2	3	No.	1	2	3	1	2	3
281	32	34	37	376	396	413	316	28	30	31	326	353	362
282	27	29	30	320	343	356	317	28	29	30	329	347	362
283	26	29	29	304	331	336	318	27	28	29	314	331	348
284	28	30	32	323	356	370	319	26	28	30	309	326	350
285	21	29	31	324	337	374	320	NR	33	31	NR	381	363
286	26	29	33	304	333	379	321	27	29	29	312	332	341
287	26	29	33	304	333	379	322	26	27	29	301	303	33
288	25	30	33	342	351	380	323	26	29	29	299	334	334
289	28	30	32	321	352	375	324	28	29	29	324	337	33
290	NR	26	28	NR	302	320	325	27	28	29	310	328	33
291	25	26	27	300	301	317	326	42	43	43	492	493	49
292	25	27	28	297	321	356	327	34	35	36	396	404	42
293	27	27	29	312	314	343	328	32	34	34	370	392	40
294	26	27	29	306	320	344	329	30	32	33	347	367	38
295	26	28	29	308	330	341	330	31	31	33	364	367	38
296	20	29	32	330	348	382	331	30	31	32	358	364	37
297	29	30	32	330	348	376	332	NR	33	31	NR	382	37
298	28	30	32	332	347	365	333	30	31	33	356	371	38
299	26	28	30	311	331	353	334	30	31	31	349	357	36
300	29	33	32	337	388	368	335	31	31	31	361	360	36
301	27	28	29	312	328	340	336	32	30	33	374	352	38
302	27	29	30	311	335	346	337	30	31	31	354	359	36
303	27	29	30	321	341	355	338	30	31	32	350	363	37
304	26	29	29	30€	339	346	339	31	31	33	366	360	38
305	27	29	30	321	34	341	340	31	31	31	360	364	36
306	27	29	30	320	43	349	341	32	30	32	351	320	36
307	2"	29	30	31	34	351	342	31	30	31	357	346	35
308	28	29	30	32	7.41	351	343	28	30	32	330	353	36
309	28	29	31	32	34	359	344	30	30	31	351	353	36
310	28	30	31	32	148	364	345	30	30	32	345	352	37
311	28	29	32	324	339	369	346	30	31	31	349	370	36
312	27	29	27	313	1346	329	347	30	30	32	345	350	36
313	28	29	32	327	347	374	348	31	31	33	364	366	38
314	28	29	33	330	344	387	349	32	30	33	367	346	37
315	28	30	32	333	359	381	350	29	29	30	339	325	34

Cycle	Bra	king sec	Time,	Revolutions			Cycle	Bra	king	Time,	Rev	oluti	ons
No.	1	2	3	1	2	3	No.	1	2	3	1	2	3
351	33	33	32	386	378	374	386	27	28	29	312	327	33
352	31	31	32	357	360	369	387	27	28	29	313	324	33
353	31	31	31	362	352	362	388	28	29	32	323	334	36
354	29	31	32	342	366	371	389	NR	29	31	NR	334	35
355	32	30	33	377	346	381	390	27	29	30	325	346	35
356	30	30	31	344	350	364	391	49	49	51	576	575	59
357	30	30	30	345	344	350	392	34	36	35	397	422	42
358	32	28	33	363	308	292	393	37	37	38	364	360	37
359	28	30	31	325	356	362	394	28	28	28	329	329	32
360	28	30	31	328	351	364	395	31	27	30	355	324	35
361	30	31	32	346	361	377	396	27	29	32	307	341	36
362	30	30	31	344	350	363	397	28	30	31	331	347	36
363	30	30	31	351	349	365	398	32	30	31	373	353	36
364	31	30	31	365	352	367	399	30	29	31	350	343	35
365	29	30	30	335	349	352	400	30	30	31	356	360	37
366	29	30	31	336	350	367	401	30	31	30	356	366	35
367	30	30	31	340	344	365	402	28	29	32	328	337	37
368	28	29	30	326	341	347	403	29	30	31	334	356	36
369	28	29	30	323	339	351	404	29	31	32	341	356	37
370	30	29	30	347	338	354	405	30	31	33	347	358	38
371	29	29	31	331	338	353	406	29	29	33	337	342	38
372	30	29	32	347	339	380	407	31	31	31	360	367	36
373	28	29	30	322	341	353	408	27	29	31	315	339	36
374	29	31	30	341	354	347	409	27	28	32	311	333	38
375	31	29	31	353	339	355	410	27	29	29	311	334	34
376	27	29	30	314	334	346	411	26	28	29	307	335	33
377	31	29	32	362	340	370	412	28	28	31	322	325	36
378	27	29	29	317	337	337	413	25	27	31	301	317	35
379	27	29	29	320	338	340	414	31	29	32	355	343	37
380	29	29	31	336	334	356	415	28	32	33	333	367	38
381	27	29	30	314	335	342	416	29	28	31	335	328	35
382	30	29	30	349	337	342	417	29	31	32	341	366	37
383	29	29	30	343	334	344	418	29	31	32	338	361	37
384	27	29	30	315	334	346	419	27	29	32	317	334	37
385	29	28	30	331	359	355	420	28	31	32	325	361	38

Cycle	Bra	king	Time,	Rev	oluti	ons	Cycle	Bra	king	Time,	Rev	oluti	ons
No.	1	2	3	1	2	3	No.	1		3	1	2	3
102.0							02.0						
421	28	30	31	322	356	363	456	27	29	29	310	336	340
422	29	32	34	339	378	394	457	26	28	29	298	325	339
423	28	30	32	340	358	379	458	27	29	30	314	331	349
424	28	30	30	334	352	358	459	26	28	31	306	325	356
425	27	30	30	319	350	346	460	28	28	29	329	324	332
426	26	29	30	311	340	364	461	26	28	29	299	320	338
427	27	29	30	316	332	349	462	27	28	29	316	323	337
428	28	30	30	322	345	360	463	26	27	29	301	317	332
429	28	29	32	323	351	373	464	26	28	30	308	331	355
430	27	29	30	311	335	350	465	28	29	30	319	330	348
431	27	31	32	319	355	371	466	27	29	29	310	335	344
432	27	28	28	319	332	332	467	27	29	29	307	337	336
433	28	30	31	318	345	360	468	27	28	31	312	324	361
434	28	31	31	326	357	365	469	27	29	30	311	338	352
435	28	29	31	322	339	358	470	27	28	29	309	322	339
436	28	31	32	328	358	361	471	27	28	29	311	317	332
437	27	30	31	317	348	365	472	25	26	27	300	302	315
438	28	30	32	330	355	379	473	27	27	27	318	311	31
439	28	30	32	328	354	369	474	26	26	27	302	298	31:
440	29	31	33	343	370	383	475	37	42	41	432	483	47
441	28	30	32	327	347	373	476	34	35	35	393	406	40
442	27	31	35	310	362	408	477	33	33	33	379	384	38
443	29	31	33	339	373	382	478	30	32	33	353	370	38
444	29	32	31	340	369	362	479	31	33	33	362	386	38
445	29	30	31	337	355	361	480	31	31	33	368	364	380
446	40	29	30	488	334	352	481	32	31	33	375	362	38
447	27	29	30	309	336	351	482	30	32	33	354	375	38
448	26	29	30	320	347	362	483	30	31	33	360	367	38
449	27	31	31	319	352	362	484	31	32	34	368	371	39
450	28	29	30	323	345	353	485	31	31	32	361	369	38
451	27	29	31	318	338	362	486	31	32	33	367	375	39
452	27	27	28	313	315	325	487	31	32	33	362	375	38
453	27	28	30	317	327	349	488	31	32	33	373	383	39
454	27	28	29	313	327	340	489	33	31	33	385	370	38
455	27	28	29	317	328	336	490	31	31	33	366	357	37
•33	1 . ,	2.0	4.7	31.	320	330	1 430	1 34	34	33	166	337	1 37

Cycle		sec			oluti	ons	Cycle	Bra	king	Time,	Rev	oluti	ons
No.	1	2	3	1	2	3	No.	1	2	3	1	2	3
491	29	31	33	358	367	387	526						
492	33	31	34	389	363	413	527	30 30	30	34 32	357	353	403
493	30	31	33	343	359	390	528	35	2.00		343	369	370
494	33	30	34	370	348	402	529	28	30	30 30	415	352	35
495	32	33	35	379	390	412	530	27	29	30	323	339	34
496	30	30	31	349	356	379	531	27	29	29	314	339	35
497	29	NR	31	338	NR	370	532	25			313	339	34
498	34	29	30	402	335	351	533	29	27 29	28 30	298 327	314	33
499	31	31	34	364	359	383	534	26	29	30		338	34
500	29	30	31	343	352	364	535	27	30	30	306 317	343	35
501	30	30	32	385	356	371	536	27	27	29	316	319	34:
502	29	30	30	343	348	355	537	27	29	30	315	341	34
503	29	28	29	332	342	343	538	27	30	30	319	343	35
504	30	NR	33	359	NR	381	539	28	29	30	322	343	35
505	28	32	32	329	359	351	540	27	29	31	325	341	36
506	28	30	30	304	321	341	541	28	30	31	320	347	36
507	26	28	30	340	369	372	542	28	29	32	326	345	360
508	29	30	31	340	353	361	543	28	29	30	323	339	36
509	29	30	31	340	362	380	544	27	29	31	323	345	35
510	29	30	30	342	351	356	545	28	29	32	316	342	35
511	29	30	31	327	359	363	546	27	30	32	327	345	37
512	29	31	32	338	358	373	547	28	30	32	318	345	37
513	29	31	31	339	360	363	548	28	30	30	320	339	33
514	27	30	31	315	353	364	549	27	30	31	316	345	34
515	26	27	31	322	323	356	550	27	30	30	318	338	34
516	27	31	30	323	359	359	551	28	31	32	330	345	36
517	29	30	31	339	355	357	552	29	30	33	335	357	379
518	27	31	31	323	359	365	553	35	31	32	413	356	376
519	27	30	32	322	352	361	554	29	31	34	332	363	39:
520	30	31	33	350	358	368	555	30	29	33	344	336	411
521	29	31	32	337	364	373	556	29	31	33	328	363	389
522	30	30	32	348	353	370	557	26	25	27	304	299	314
523	29	30	31	435	368	370	558	24	26	28	284	310	326
524	29	29	30	346	343	361	559	38	39	40	438	448	460
525	28	31	32	330	366	372	560	32	33	34	360	380	38:

Cycle	Bra	king	Time,	Rev	oluti	ons	Cycle	Bra	king	Time,	Rev	Revolutions		
No.	1	2	3	1	2	3	No.	1	2	3	1	2	3	
561	31	32	31	367	372	369	596	26	28	29	309	329	33	
562	29	30	30	334	356	355	597	26	28	28	307	329	33	
563	34	30	30	380	348	353	598	26	27	29	305	314	33	
564	29	29	31	334	342	362	599	26	27	29	303	322	33	
565	30	31	32	352	357	375	600	27	29	29	311	333	34	
566	30	30	31	351	359	361	601	28	28	32	329	327	37	
567	28	30	31	328	348	368	602	29	30	30	338	353	35	
568	28	30	32	330	358	364	603	28	30	30	327	348	34	
569	28	29	31	330	346	364	604	27	28	28	319	338	33	
570	36	29	30	438	346	355	605	27	27	28	319	316	33	
571	29	30	31	332	357	363	606	26	28	27	304	328	31	
572	27	29	30	317	342	355	607	26	27	NR	307	319	N	
573	28	29	31	327	342	363	608	26	28	28	307	323	32	
574	28	29	30	324	344	356	609	26	28	28	310	323	32	
575	28	28	29	321	326	343	610	25	27	29	292	314	32	
576	28	29	30	324	345	356	611	65	40	40	527	492	48	
577	27	30	30	326	350	354	612	32	32	32	373	373	37	
578	28	29	31	332	336	363	613	29	30	30	339	352	35	
579	29	30	30	328	350	349	614	27	29	31	313	345	36	
580	28	30	30	333	350	356	615	29	29	29	342	337	34	
581	28	29	31	330	345	359	616	28	29	30	326	345	35	
582	28	30	31	328	347	371	617	31	31	32	374	365	38	
583	28	30	32	334	356	377	618	31	33	31	360	390	35	
584	29	30	32	341	360	379	619	31	31	32	368	359	37	
585	26	28	29	306	324	346	620	37	34	30	459	397	34	
586	31	30	31	344	337	359	621	30	29	31	351	334	35	
587	40	29	34	492	344	391	622	31	31	31	372	369	36	
588	25	29	30	310	336	355	623	31	30	32	361	353	36	
589	26	29	31	309	336	358	624	29	30	31	347	353	37	
590	38	28	29	418	333	357	625	28	31	32	338	351	37	
591	26	28	30	310	334	356	626	28	31	30	331	364	34	
592	27	28	31	311	328	354	627	26	28	30	313	333	34	
593	26	27	30	309	326	357	628	27	28	31	316	334	36	
594	26	28	31	315	338	363	629	29	29	30	337	337	35	
595	26	27	30	310	326	349	630	26	29	31	310	346	36	

Cycle	Bra	king	Time,	Rev	oluti	ons	Cycle	Bra	king	Time,	Revolutions			
No.	1		3	1	2	3	No.	1	2	3	1	2	3	
						241								
631	27	29	31	324	346	361 371	666	26	27	30	316	339	364	
632	25	26	28	292	303		667	30	28	30	343	328	350	
633	27	28	29	320	325	340	668	30	28	30	320	346	352	
634	25	27	29	295	323	336	669	28	29	31	325	334	360	
635	26	28	28	310	328	330	670	26	27	30	308	317	34	
636	27	28	29	316	327	338	671	28	31	32	320	362	374	
637	28	27	29	325	311	338	672	25	26	28	291	308	32	
638	26	27	29	304	320	337	673	26	26	30	303	304	34	
639	28	29	28	328	340	332	674	32	27	30	389	319	340	
640	27	28	28	327	336	338	675	26	27	31	301	320	351	
641	30	29	43	356	337	487	676	41	45	41	494	527	48	
642	27	28	31	315	337	364	677	31	34	32	373	397	37	
643	30	28	39	346	326	484	678	28	29	30	330	333	35	
644	28	28	30	332	332	352	679	29	30	29	336	354	331	
645	27	30	28	310	348	323	680	29	30	30	338	348	34	
646	27	28	30	313	333	349	681	28	29	31	327	338	35	
647	26	27	30	305	318	341	682	29	29	29	348	344	34	
648	26	26	28	306	306	326	683	32	29	29	377	335	33	
649	26	27	30	310	320	353	684	33	33	30	388	386	35	
650	26	27	29	305	315	336	686	26	26	28	306	306	33	
651	25	27	27	296	314	314	686	26	29	30	312	350	35	
652	23	25	28	258	292	328	687	24	25	27	249	294	34	
653	27	27	28	309	315	324	688	26	26	27	308	303	33	
654	25	26	27	291	303	318	689	29	30	30	334	348	35	
655	25	26	26	293	302	312	690	27	28	30	313	327	34	
656	25	26	27	297	306	316	691	29	30	30	333	352	35	
657	27	26	28	309	305	323	692	28	30	31	323	345	36	
658	26	26	26	307	303	308	693	25	28	30	310	329	34	
659	26	26	27	300	304	324	694	27	30	30	319	354	35	
660	26	26	28	300	312	324	695	27	27	28	314	315	32	
661	29	29	31	341	341	360	696	26	28	28	308	323	33	
662	39	30	31	441	334	360	697	27	27	29	319	325	33	
663	28	31	31	327	368	354	698	27	28	29	316	323	33	
664	28	29	30	326	347	353	699	29	27	28	327	320	-	
665	28	29	31	326	334	360	700	28	30	32	325	303	34	

Cycle	Bra	king	Time,	Rev	oluti	ons	Cycle	Braking Time, sec			Revolutions		
No.	1	2	3	1		3	No.	1	2	3	1	2	3
701	27	27	28	312	315	330	736	25	26	28	296	303	22/
702	26	28	28	310	327	331	737	26	27	28	314		330
703	26	28	30	307	328	348	738	27	27	29		317	32
704	27	28	29	311	332	343	739		27		320	311	34
705	26	29	30	300	336	347	740	26 40	40	29 39	307 471	316	32
706	26	28	29	304	318							477	45
707	25					331	741	31	NR	35	358	NR	41.
708	26	28	30	290	324	344 NR	742	32	28	28	376	327	32
709		28	NR	309	330		743	26	28	29	308	124	336
	28	30	30	326	340	341	744	25	28	30	295	326	34
710	27	29	30	310	332	344	745	27	28	33	329	321	399
711	28	28	31	320	327	366	746	27	30	31	321	348	36
712	27	29	33	318	337	390	747	26	30	31	308	341	350
713	28	28	29	325	326	336	748	27	30	31	322	342	35
714	29	29	29	338	335	337	749	26	30	29	311	340	33
715	28	29	30	324	332	351	750	26	27	29	305	318	33
716	27	25	25	307	291	300	751	30	28	29	349	330	34
717	25	26	28	299	306	328	752	28	27	29	331	321	33.
718	26	26	28	305	305	324	753	28	28	29	322	329	33
719	28	26	29	323	305	333	754	28	29	30	327	337	35
720	26	27	29	304	319	342	755	28	28	30	328	330	34
721	28	30	32	321	357	385	756	29	31	30	332	369	34
722	27	30	29	309	353	346	757	30	28	29	342	331	34
723	26	28	29	305	319	337	758	27	29	30	309	334	34
724	26	28	31	306	325	363	759	27	28	29	320	327	34
725	25	27	31	295	311	352	760	26	27	27	302	318	31
726	28	27	28	317	309	322	761	26	28	32	303	327	37
727	27	29	30	313	343	349	762	26	28	31	306	328	36
728	25	29	29	298	334	334	763	26	28	30	303	329	34
729	26	28	28	299	328	330	764	27	28	29	310	322	33
730	26	28	29	301	327	339	765	27	27	29	312	317	33
731	25	25	26	296	291	301	766	29	30	33	339	351	38
732	23	25	27	274	288	315	767	26	31	30	304	361	35
733	25	26	27	295	305	315	768	28	30	29	323	351	33
734	25	26	29	291	308	332	769	26	25	25	303	306	30
735	26	27	28	310	314	320	770	24	27	27	287	312	31

Cycle		sec		Rev	oluti	ons	Cycle	Bra	king	Time,	Revolutions			
No.	1	2	3	1	2	3	No.	1		3	1	2	3	
771	26	26	28	303	309		***							
772	27	29	27	317	336	331 322	806	27	27	28	313	313	32	
773	25	27	29	298	313		807	27	27	27	118	311	32	
774	28	27	29	328	317	340 339	808	29	26	27	322	300	31	
775	25	28	28	298	327	351								
776	28	29	29	325	341	338								
777	27	29	29	321	344	352								
778	27	27	28	313	312	323								
779	26	28	28	308	326	328							l	
780	27	27	30	316	313	340								
781	27	29	31	325	337	363								
782	28	28	30	321	322	352								
783	38	29	28	478	329	330								
784	26	27	28	306	319	327								
785	26	28	28	307	323	331								
786	27	28	28	317	330	337								
787	26	28	28	302	328	330								
788	28	28	31	327	330	346								
789	25	24	26	291	271	299								
790	24	25	27	280	290	313								
791	35	36	38	405	420	443		8						
792	30	29	31	335	339	362	3	9						
793	27	28	29	311	337	337		\$ 7						
794	37	29	31	455	346	357								
795	26	28	29	322	328	343								
796	28	30	33	326	346	365								
797	27	28	31	325	344	365				- 1				
798	26	28	30	315	322	352								
799	27	30	31	317	342	352	*			- 1				
800	27	28	30	312	336	344				- 1			ŝ	
801	26	28	30	311	328	342				- 1		8		
802	26	28	30	312	327	345						(8)		
803	26	28	29	310	328	344						1		
804	26	26	28	307	323	335				- 1				
805	26	25	26	303	290	298								

Note:

NR = not recorded.

APPENDIX D STRAINS AND TEMPERATURE DEVELOPED DURING SPEED-REDUCTION BRAKING TESTS OF WHEEL NO. 2

						Strain,	Temperat	rure of
	Part o	f Cy	ele		FHPF**	BRPF**	1/16-in. Depth	1/4-in. Depth
			Cycle	No. 1				1000
After	first bre	te a	pelicatio	n**	640	480	440	270
•	second		•	••	1860	1080	500	356
•	third	•	•	***	2830	1480	530	400
•	fourth	•	•	****	4440	2220	600	500
•	cooling				1880	90	#0	*0
			Cycle	No. 2				
After	first bra	ke a	pplicatio	n*	1070	840	400	
•	second	*			1750	1490	550	1
	third	•			2270	1900	590	450
•	cooling				0	0	80	80
			Cycle	No. 13				
After	first bra	ke a	pplicatio	6.4	1000	990		
•	second	•	•		1750	1590	***	***
	third	•	•	•	2260	1980	590	410
•	cooling				0	0	*0	*0
			Cycle	No. 80				
After	first bra	ke at	pplicatio	n+	880	710		
•	second	•	•		1590	1390		***
:	third				2080	1810	190	445
•	cooling				0	0	80	#0
				No. 220				
After	first bra	he as	pplicatio	a•	920	1030	***	
•	second	•			1620	1650		***
•	third		0.00	•	2100	1910		***
•	cooling				0	e	***	***
			Cycle	No. 352				
After	first bra	ke as	pplicatio	n+	1075	1735		
•	second	•	1.00		1090	1450	•••	***
•	third	•	•	•	2210	1955	1	***
•	cooling				0	· ·		***

<sup>\*</sup> The temperatures of the plate fillets did not exceed 250°F; no corrections were required.

# \*C - 3/9(\*F - 32)

<sup>..</sup> Brake only to 565 rpm. ... Brake only to 535 rpm.

<sup>\*\*\*\*</sup> Brake only to 0 (complete stop).

<sup>.</sup> Brake to 465 rpm.

<sup>..</sup> PEPF - Front-hub-plate fillet.

BRPF - Back-rim-plate fillet.

<sup>\*\*\*</sup> Unreliable thermocouple because of moisture.

Note: Cycle No. 1 resulted in plastic deformation of the FNPF. Cycles No. 2 through 808 did not produce plastic deformation of the FMFF.

<sup>1</sup> inch - 25.4 mm

APPENDIX E

DATA OBTAINED DURING DRAG-BRAKING/EMERGENCY-STOP

BRAKING TEST OF NEW WHEEL NO. 3

(1-72-S-2472A)

	Type of	Stop Time,	Max	Partial Redial at Back-Rim-Pla	Residual Stress te Pillet,** Fsi
Cycle	Braking	sec	Temp, **F	Gage No. 1	Gage No. 2
1***	Drag		960	-37	-26
2***			900	-90	-59
3***	Stop	90	680	- 30	-39
4	Beep	85	710		
5		85	720		
6		87	710		
7		86	730	-90	-44
8		88	720	-30	
9		87	720		
10	•	83	630		
11		82	640		
12		84	700	-49	-15
13***	Drag	100000	920	-99	-70
14***			920	-104	-67
15***	Stop	78	820	-201	-07
16		70	700		
17	•	71	700		
18	•	72	750		1
19***		87	600	-54	-30
20	•	72	780		-30
21		71	790		
22		75	700		
23		78	700		
24		76	750	-51	-22
25	:	78	.20	-34	
26	•	83	790		
27		81	800		
28		81	820		
29		83	790	-44	-31
30		77	750	00.707U	
31	:	80	640		
32	22 23	82	720		
33		82	650		
34		81	650	-46	-19
35		85	610	750 <b>3.8</b> 0	(17.4.6)
36	:	87	720		
37		86	600		l
38		83	620		
39		84	740	-31	-4
40		91	700		100.70
41		86	600		
42	:	86	650		
43		90	780		

	Type of	Stop Time,	Мах		Residual Stress te Fillet, ** ksi
Cycle	Braking		Temp, **F	Gage No. 1	Gage No. 2
44	Stop	83	650	-25	-2
45	acop .	86	680	••	. 17. <b>8</b>
46		94	640		
47		88	640		
48		85	750		
49		83	670		
50		93	690		
51		91	870		
52	:	84	650		
53		91	700		
54	ુ•:	88	650	-20	0
55		80	600	(F.50)	125.5
56		86	600		
57		90	660	K	
58		86	640		
59		84	620		
60		125	690		
61		136	630		
62		132	720		
63		105	910		
64		105	680	-17	+8
65		95	880	18.510	3.5
66		55	820		
67****		102	580		
68		94	820		
69		130	600		
70***		84	530		
71		79	510		
72		79	530		
73		76	560		
74		75	600	-20	+14
75		75	600	00. <del>M.</del> M.	S-7-1-1-1
76		76	580		
77		75	600		
78		79	590		
79		78	680		
80		75	620		
81		75	600		
82		77	730		
83		77	600		
84		80	650		
85		79	600		
86	•	77	660		
87		77	800		
88		79	620		
89		76	600		

	Type of	Stop Time,		Partial Radial Residual Stre at Back-Rim-Plate Fillet,**					
Cycle	Braking	sec	Temp, **F	Gage No. 1	Gage No. 2				
90	Stop	78	640						
91 ***	acop .	79	800						
92		90	440						
93		85	520						
94		78	670						
95	:	120	620						
96	-	107	620		1				
97	-	103	610						
98		96	610						
99	•	92	600						
100	-	94	590						
101	•	89	590						
102	-	89	590						
103	-	97	620						
104	- 1	95	640						
105	- 1	84	610						
106	-	84	690		1				
107		84	610						
108	-	83	690		1				
109	•	76	670						
110	.	75	680		I				
111		75	830						
112	:	74	720						
113	-	76	700						
114	-	72	710	-26	-14				

## Conversion Factor

1 ksi = 6.895 MPa

<sup>\* 1/8</sup> Inch (3.2 mm) beneath tread surface at center.

<sup>\*\*</sup> Caused by dynamometer testing. Front-hub-plate-fillet gages were destroyed; no data were obtained.

<sup>\*\*\*</sup> New brake shoe installed.

<sup>\*\*\*\*</sup> Simulated wornout brakeshoe installed.

APPENDIX F
RESIDUAL-STRESS DATA\*

Strain	Strai	in, pir	./in.	Indicated Residual Stress,** Strain S					
Gage	€0	ef	E	ksi	Gage	60	c!	c	ksi
)			Whee	No. 4 (1	-72-S-2466	A)			
Rim 1-1	8410	9440	-1030	-30.9		7550	71.37.45.10	-1130	
2-1	7830	9050	-1220	-36.6	2-2	7785	9150	-1365	-41.0
3-1	8960	10360	-1400	-42.0	3-2	8295	9880	-1585	-47.5
4-1	8660	10130	-1470	-44.1	100000000000000000000000000000000000000	7.5	10640		
5-1	6983	8450	-1465	-43.9	0.000000.000		10820	5 700 - 71 - 51	
6-1	7545	9130	-1585	-47.5	6-2	8675	10510	-1835	-55.0
7-1	8230	9960	-1730	-51.9	7-2	8440	10420	-1980	-59.4
8-1	9225	10760	-1535	-46.0	8-2	8747	10640		
9-1	8360	9680	-1320	-39.6	9-2	8470	9980	-1510	-45.3
10-1	8365	9820	-1455	-43.6	10-2	7180	8760	-1580	-47.4
11-1	8810	10500	-1690	-50.7	11-2	8710	10620	-1910	-57.3
12-1	8415	9440	-1025	-30.4	12-2	8835	10080	-1255	-37.6
13-1	9005	10340	-1335	-40.0	13-2	8825	10510	-1685	-50.5
14-1	8105	9870	-1765	-53.0	14-2	8300	10280	-1980	-59.4
PHPF-1-R	7935	4690	+3245	+110.1	FHPF-2-R	7650	4630	+3020	+109.1
0	9490	7400	+2090		D	8495	6480	+2015	
7	7995	7160	+835	+54.7	Ŧ	7506	6640	+965	+61.7
				a = 90°					a = 90*
BRPF-1-R	7535	6460	+1095	+34.5	BRPT-2-R	7405	5530	+1875	+64.2
D	8970	8490	+480		D	9150	8280	+870	Nec section
1	7510	7640	-130	+6.5 a = 90°	T	7725	7520	+205	+24.8 a = 83*

Strain	Stra	in, pir	./in.	Indicated Residual Stress,**	Strain	Strai	in, vir	./in.		
Gage	€o.	ef.	3	ksi	Gage	ξO	εf	ε	ksi	
			Whee	No. 6 (1	-71-8-1019	0A)				
Rim 1-1	9685	10370	-685	-20.5		7950	8820	-870		
2-1	8990	10380	-1390	-41.7	2-2	9310	10840	-1530	-45.9	
3-1	9060	9820	-760	-22.8	3-2	9240	9910	-670	-20.1	
4-1	8030	7410	+620	+18.1	0.000000	8565		+335		
5-1	8840	8590	+250	+7.5	5-2	8380			410100000000	
6-1	8735	8720	+15	+0.5	6-2	9010	8770	+240	+7.2	
7-1	7325	7450	-125	-3.8	7-2	7740	7790	-50	-1.5	
8-1	9170	9580	-410	-12.3	8-2	8555	8700	-345	-10.4	
9-1	9490	8060	+1430	+42.9	9-2	8860	7800	+1060	+31.8	
10-1	9655	7500	+2155	+64.6	10-2	9080	6780	+2300	+69.0	
11-1	8905	7310	+1595	+47.9	11-2	8870	7280	+1590	+42.8	
12-1	8800	9540	-740	-22.2	12-2	8900	9150	-250	-7.5	
13-1	8635	9550	-915	-27.4	13-2	7775	8570	-795	-24.8	
14-1	8905	9300	-395	-11.8	14-2	8910	9420	-510	-15.3	
FHPF-1-F	7530	6130	+1400	+49.3	FHPF-2-R	7730	6610	+1120	+40 6	
t	ND	ND	ND		D	8920	8160	+760		
1	7075	6760	+315	+24.2	T	7650	7280	+370	+23 2	
			1-50,000	a = ND					0 = -0.	
BRPF-1-F	8270	8370	-100	-3.7	BRPF-2-R	8350	8960	-610	-22 5	
	9155	9200	-45		D	9415	9750	-335		
7	7450	7650	-200	-9.1	T	7675	7880	-205	-12.5	
				a = 87*			217.12.22		a = 10*	

Strain	Strai	n, pir	./in.	Indicated Residual Stress,**	Strain			./in.	Indicated Residual Stress,**
Gage	£0	cf	C	ksi	Gage	£0	εf	ε	ksi
			Whee	No. 1 (1-	72-5-247	(A)			
Rim 1-1	4925	6535	-1610	-48.3		5580		-1565	0.0000000000000000000000000000000000000
100	5480	7060	-1580	-47.4	0.00	4445		-1795	C
COT 1777.0	5350	10.000	-1140			5640	7 (11) (2) (2)	-975	E 0.0 123 224 7 146
4-1	5735	3475	+2260		E 10 000 000 1	5575	574 775 775 775 775	+1745	10/100 VIII (1777 10)
5-1	5050	2550	+2500	+75.0***	201 DECEMBER 1991	5165	Calman Transcore	+2245	
6-1	5455	2600	+2855	+85.5***		5360		+2220	
7-1	5108	2390	+2720	+81.5***	7-2	4520	2070	+2450	+73.5**
8-1	4640	2220	+2420	+72.5***	8-2	5560	3510	+2050	+61.5
9-1	5525	4435	+890	+26.7	9-2	5700	5430	+270	+8.1
10-1	4580	2865	+1715	+51.5	10-2	5850	4555	+1295	+38.2
11-1	4455	2980	+1475	+44.2	11-2	5730	4620	+1110	+33.3
12-1	5980	7170	-1190	-35.7	12-2	6545	7760	-1215	-36.4
	4305	5350	-1045	-31.3	13-2	5125	6590	-1465	-44.0
14-1	5490	5945	-455	-13.6	14-2	5820	6685	-865	-25.9
PHPF-1-R	4010	2025	+1985	+70.8	FHPF-2-R	3930	1980	+1950	+68.6
D	5595	1210	+1385		D	5230	4000	+1230	
T	3955	3350	+605	+40.2	7	3960	3445	+515	+36.9
				a = 90*					a = 90°
BRPF-1-R	3770	3745	-25	-0.1	BRPF-2-R	3730	3080	+650	+20.7
D	6240	6835	-595		D	5885	5540	+345	
T	4270	4780	-510	-20.5	T	3810	3870	-60	++4.1
179	0.00	3.0000000000000000000000000000000000000	11/2/20	a = 108*				200	a = 90*

	Stra	in, pi		Indicated Residual Stress,**		Stra	in, vi	n./in.	Indicated Residual Stress,**
Gage	£0	εť	£	ksi	Gage	r <sub>o</sub>	cf	C	ksi
			Whee	No. 5 (1-	-71-S-101	91A)			
		10230		-41.4	Rim 1-2	8655	10430	-1775	-53.5
		10390		-54.0	2-2	9090	11360	-2270	-68.0
3-1	9980	10390	-410	-12.3			10520		
	9285		+395	+11.8		8080			
5-1	8945	8660	+275	+8.2	5-2	9600	9290	+310	
	8915		+375	+11.2	6-2	8510	8260	+250	
7-1	7410	7350	+60	+1.8	7-2	8070			0.000
8-1	9025	9430	-425	-12.7	8-2	8600	8970	-370	
9-1	8800	7830	+970	+29.0	9-2	9030		+1120	
10-1	8755	6380	+2375	+71.2***	10-2	8185		+2195	
11-1	8800	7500	+1300	+39.0	11-2	8415	6540	+1875	
12-1	8630	9720	-1090	-32.7	12-2	8525		-1365	
10 TO	7920	0.500 51.500 50.500	-1040		13-2	9005	11220	-2195	
14-1	8935	10050	-1115	-33.4	14-2	8730	10270	-1540	-45.5
FHPF-1-R	8055	5920	+2135	+75.1***	PHPF-2-R	7770	5670	+2100	+73.1**
D	9285	7680	+1605			8465		+1365	
T	8240	7930	+310	+39.5		7685			
				a = 81.5	1				a = 88*
BRPF-1-R	The second second second		+235	+9.5	BRPF-2-R	7750	7350	+400	
D	9310	9210	+100	WEIGHT .		9490	9120		
T	7416	7310	+105	+5.1	Ŧ		7840	+235	+11.3
				a = 113.5		ं	123.00		a = 73.5*

Strain	Strain, pin./in.			Indicated Residual Stress,**	Strain	Strain, µin./in.			Indicated Residual Stress,**
Gage	60	ef	ε	ksi	Gage	€0	cf	C	ksi
			Whee	1 No. 2 (1-	72-S-247	IA)			
Rim 1-1	7550	9050	-1500	-45.0	Rim 1-2	6540	8050	-1510	-45.3
2-1	6695	8190	-1495	-44.9	2-2	6970	8800	-1830	-55.0
3-1	6610	7155	-545	-16.4	3-2	6515	7145	-630	-18.9
4-1	6680		+2200		4-2	6295	3600	+2690	+80.7***
5-1	6750	4285	+2465	+74.0***	5-2	6515	3605	+2910	+87.4***
6-1	7320	4925	+2395	+71.8***	6-2	6390	3495	+2895	
7-1	6890	4620	+2270	+68.1	7-2	6450	3730	+2720	+81.6**
8-1	7880	5755	+2125	+63.8	8-2	6185	3870	+2315	
9-1	6520	5570	+950	+28.5	9-2	6710	4595	+1110	+33.3
10-1	7070	4305	+2765	+83.0***	10-2	6955	3240	+3715	+111.4***
11-1	6675	4865	+1815	+54.4	11-2	6380	4030	+2350	+70.4
12-1	6770	8005	-1235	-37.1	12-2	8005	9370	-1365	-41.0
13-1	7495	8995	-1495	-44.9	13-2	6425	8290	-1865	-55.9
14-1	7490	8710	-1220	-36.6	14-2	7545	8785	-1245	
FHPF-1-R	8420	6960	+1460	+52.4	FHPF-2-R	7675	6276	+1400	+51.9
D	7520	6525	+995		D	6985	5945		
T	8210	7767	+445	+29.0	T	8370	7800	+570	+32.7
				a = 87.5	1			1855,051	a = 86.5
BRPF-1-R	8680	8535	+145	+3.1	BRPF-2-R	7315	7480	-165	1 S S S S S S S S S S S S S S S S S S S
D	7785	7975	-190		D	7695	8050	-355	
T	8550	8945	-395	-9.5	S	8030	8395	-365	E. C.
				a = 97*					a = 117*

Strain	Strain, pin./in.			Indicated Residual Stress,**		Strain, pin./in.			Indicated Residual
Gage	E0	ef	c	ksi	Gage	£0	e f	£	ksi
			Whee	No. 3 (1	-72-S-247	2A)			
	7736		-1014		Rim 1-2	6007	6883	-876	-26.3
	6356	7495	-1139	-34.2	EDT 100	7546		-1069	-32.1
	7955	8013	-58	-1.7		7677			+1.1
	7254		+2949			8174		+2987	
	6585		+3345			6963		+3138	
	6950		+3216			5696		+3008	
	6654		+2909	+87.3***		7343		+3238	
	7393		+2478	+74.3***		7232		+2992	+89.8**
	6883	5965	+918	+27.5		7427		+1485	+44.6
	7413		+2353			6606		+2110	
	6630		+1664	+49.9		7330		+2384	+71.5**
	6458	6393	+65	+2.0		6846	6768		+2.3
	6880	7066	-186	-5.6		6504	6495		+0.3
14-1	6810	6332	+478	+14.3		6900	6660		+7.2
PHPF-1-R	7774	9506	-1732	-60.6	FHPF-2-R	7345	2000	-1644	
D	7330	8320	-990	5555		7583	8546	-963	-58.7
T	6825	7185	-360	-29.0	7	0.0000000000000000000000000000000000000	8054		
				a = 89*	1	,614	0034	-440	-30.7
BRPF-1-R	6640	8063	-1423		BRPF-2-R	7032	8445	-1411	0 - 88*
D	7287	8057	-770	5000000000		7530	8270	-1413 -740	-47.3
T	7090	7116	-26	-15.0		7415	7488		
				a = 91*	•	.413	/408	-73	-16.3
									a = 90°

NOTE: co = strain value before cutting.

of = strain value after cutting.

c = co - cf = relaxation strain.

indicates tension.

- indicates compression.

FHPF = Front-hub-plate fillet. BRPF = Back-rim-plate fillet.

- \* A Young's modulus of 30 x 10<sup>6</sup> psi and a Poisson's ratio of 0.3 were used for the calculations.
- \*\* Also listed is a, the angle between the normal strain and the maximum principal stress directions.
- \*\*\* The yield point was possibly exceeded.

### Conversion Factors

- 1 ksi = 6.895 MPa
  - 1° = 0.017 rad
- 1 inch = 25.4 mm

APPENDIX G SAW-CUT STRAIN DATA

	Rim Punch	Distance Batween Punch Marks, inche					
Wheel No.	Mark	Initial	After Saw Cuts	Change			
1	Inner front	2.00	2.50	+0.50			
	Outer front	2.00	2,56	+0.56			
	Inner back	2.00	2.5	+0.50			
	Outer back	2.00	2.56	+0.56			
2	Inner front	2.00	2,71	+0.71			
	Outer front	2.00	2,65	+0.65			
	Inner back	2.00	2,71	+0.71			
	Outer back	2.00	2.65	+0.65			
3	Inner front	2.00	3.06	+1.06			
	Outer front	2.00	3.12	+1.12			
	Inner back	2.00	3.06	+1.06			
	Outer back	2.00	3.12	+1.12			
4	Inner front	2.00	0.71	-1.29			
	Outer front	2.00	0.59	-1.41			
	Inner back	2.00	0.71	-1.29			
	Outer back	2.00	0.56	-1.44			
5	Inner front	2.00	2.07	+0.07			
	Outer front	2.00	2.08	+0.08			
	Inner back	2.00	2.08	+0.08			
	Outer back	2.00	2.09	+0.09			
6	Inner front	2.00	2.24	+0.24			
	Outer front	2.00	2.27	+0.27			
	Inner back	2.00	2.25	+0.25			
	Outer back	2.00	2.29	+0.29			

# Conversion Factor 1 inch = 25.4 mm

<sup>+</sup> indicates opening (tension)
- indicates closing (compression)

### APPENDIX H

#### REPORT OF INVENTIONS

No inventions or discoveries were made or conceived, or for the first time actually reduced to practice by U.S. Steel, its employees, its subcontractors, or their employees, in the course of, in connection with, or under the terms of this contract. However, this study contract does represent an improved understanding of the propagation of microstructure cracking to thermal cracking.

The results further indicated that neither altered microstructures nor cracking could be produced by many emergency brakings or speed-reduction brakings with normal brakeshoes and forces.

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