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LABORATORY EVALUATION OF FECKER AND LORAL OPTICAL IR **PWI SYSTEMS** CASE FILE COPY

DETECTION SYSTEMS BRANCH

TRANSPORTATION SYSTEMS CENTER 55 BROADWAY CAMBRIDGE, MA 02142

FEBRUARY 1971 **TECHNICAL REPORT**

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LABORATORY EVALUATION

I. INTRODUCTION

The closure of ERC in June 1970 brought into focus several programs which would be completed by the Dept. of Transportation Systems Center. Pilot Warning Indicator systems using a flashing xenon strobe and silicon detectors as cooperative elements was one of these programs. Flight hardware has been delivered and a flight test evaluation on two Electro-Optical Pilot Warning indicators has been prepared in FY 1970 by the previous NASA group.*

The flight test results clearly pointed out several design deficiencies which prevented a complete evaluation of the equipment on hand. The present laboratory evaluation program has corrected these faults which prevented the equipment from operating, and has calibrated the sensitivity of both systems in azimuth, elevation and range. These tests serve as a basis for the flight test simulation and plans. The measurement data will be used to refine the models of the alarm and hazard envelopes of the equipment.

The laboratory provides the ideal environment for performing the most detailed studies of the PWI system. The temperature and humidity can be controlled or held constant and atmospheric scintillation effects on the strobe emission are reduced to zero.

The laboratory tests were performed on an optical bench and consisted of three basic components:

- A xenon strobe lamp whose output is monitored at the PWI detector by a separate calibrated detector to give pulse to pulse information on energy content in the .8 to 1.1 µ region at the receiver.
- (2) A strobe light attenuating optics which is calibrated photometrically to provide simulated range.

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(3) A positioning table on which the PWI system under study is mounted. This table provides spatial location coordinates for all data points. A detector is used to monitor the pulse amplitude of the received IR energy.

The detectors are scanned to determine the sensitivity structure as a function of detector position and simulated range. *Reference:

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 NASA: Flight-Test Evaluation, Two Electro-Optical Pilot Warning Indicators, National Aeronautics and Space Administration, Electronics Research Center, Cambridge, Mass., June 30, 1970.

II. LABORATORY EQUIPMENT MODIFICATION

Previous testing of the two PWI systems disclosed several problem areas which limited system performance. No changes to the optical designs were made because only a short time was available for modification. The modification program focused on "fixes" to the electronics and the following changes were made:

(1) <u>Radio Frequency Interference (RFI)</u>.- One prevalent problem that existed in past tests was the activation of the indicator lights whenever the radio transmitter was turned on. It was determined that the problem was due to RFI interference. The problem was corrected in the lab by incorporating EMI line filters and appropriate shielding and grounding which will also be implemented into flight units. These modifications permit operation of a transmitter or chattering relay at any distance from the PWI without producing an alarm indication.

(2) <u>Photo Diode Biasing Arrangement</u>. - The original Fecker design called for a silicon photo diode back biased to about 150V. It was found that some of the diodes exhibited breakdown noise when operated in the dark. This behavior was found to be intermittent and was dependent upon the value of bias voltage. In every case, the bias voltage could be reduced to a value at which this effect did not occur. In order to avoid breakdown effects over a long period, the bias voltage was changed to 15 volts and the input circuitry was scaled down appropriately so that the same sensitivity was obtained over the same background range.

The reduction of bias voltage will allow the use of a photodiode with a smaller guard ring area so that the size of dead zones in the lobe pattern will be reduced considerably.

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The new bias and Input circuitry is shown in Fig. 1. The reduced bias voltage specification on the diode will increase the reliability and will decrease the cost of the system substantially.

Frequency Response

The low frequency cutoff was raised to about 600 Hz for the Fecker unit to provide better rejection signals below 100 Hz such as propeller effects. This was achieved by changing the capacitors in the second stage of the amplifier as shown in Fig. 2.

The frequency response of the first and second stage before and after these changes are plotted in Fig. 3.

Cross Coupling Instabilities

It was found that the "turn on" and "turn off" of the display lamp of one channel would turn on a different channel so that, once started, the whole system would oscillate with a period of several seconds of no light input. This trouble was traced to the fact that the lamp currents were driven at logic level speeds. By placing a large capacitator (15 MF) across the input of the lamp driver, the higher frequencies of the current surge were surpressed and the system no longer responded to lamp "turn on" and "turn off".

Discriminator Level Setting

The last stage of the Fecker amplifier chain feeds directly into a level discriminator shown in Fig. 4.

Injecting a current pulse into the first stage of the amplifier will produce a voltage pulse at the input of the discriminator similar to the signal from the xenon strobe pulse. With the pulse height fixed at a given value, the level potentiometer is adjusted until the discriminator just triggers.

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Figure 3.

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Figure 4

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e . . For the laboratory tests the levels were set for 4, 6, 8 and 10 volts, so that the lobe pattern which was taken at a discriminator level of 6 volts could be scaled if it were found necessary to change the level settings later in either the field test or the flight test.

Two changes were made to facilitate the level setting procedure. The last amplifier stage coupling to the discriminator was changed from Dc to Ac. This change avoided drift of the second stage which would be amplified by 200 times the gain of the third stage.

Video Test Points

Two video test points from each channel were brought out to a connector in the front of the Fecker Electronic processing unit — the input to the discriminator and the diode load resistor. The first point gives the signal and the second is a measure of the background.

Mechanical Modifications

To protect the Fecker Systems ball lens assembly from dirt and debris, a Plexiglas shield was designed and molded.

III. ROOF TOP TEST

As a first step in the PWI Test Program a simple roof top test was made from the roof of the 13 story Systems Management Building at T.S.C. The purpose of this test was to provide a realistic operating environment for the PWI so that bounds could be placed on the range of discrimination levels used for the laboratory test. The roof top test also determined whether there were any glaring deficiencies in the equipment which would have to be corrected before the test program began.

The roof top test was conducted on the roof of the Systems Management Building located near Kendall Square in Cambridge, Mass. From this location there are four different xenon strobe lamps in view. These lamps are affixed on buildings in Boston and all are within 2.5 miles of T.S.C. One Whalen strobe was placed on the roof of the State Street Bank Building in Boston by T.S.C. personnel. The State Street Bank Building is located about 1.8 miles from T.S.C.

Direct exposure to the sun would cause a number of channels to alarm. The noise level at the discriminator rose rapidly from about 3 volts P-P to 15 volts (the amplifier saturation voltage) as any channel was scanned from 30° off sun to full sun illumination. It was found that a discriminator setting of 6 volts would be enough to cause false alarms. At this level, detection of the Whalen lamp at 1.8 nm was marginal. There exists the possibility that the discriminator level may have to be lowered in the field tests.

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IV. SPECIAL TEST EQUIPMENT AND CALIBRATION

Both the Fecker and Loral PWI systems use ball lenses coupled to discrete detectors and thus the range sensitivity of these PWI's are strongly dependent on the azimuth and elevation of the flashing Xenon source. To map out the lobed sensitivity patterns, a special test facility was constructed and calibrated. Under NASA/ERC the sensitivity tests consisted of simulating a <u>single</u> range (between 1300 and 1700 feet for the Fecker tests and between 8700 and 11500 feet for the Loral tests) for each azimuth or elevation scan and inferring an overall range capability from the single scan. The DOT/TSC laboratory tests were designed so that <u>any</u> range from 200 feet to about 6 nautical miles can be simulated. Conventional radiometers could not be used because of the extreme variation in range.

The output from the range simulator is range if one makes the following assumptions: (1) an inverse square fall-off of intensity, (2) no aerosol scattering or water-vapor absorption or scintillation, and (3) a value for the number of joules/steradian emitted by a Whalen lamp fitted with a Fresnel lens. The first two assumptions will be removed in the field tests when actual ranges and atmospheric paths are used. The third assumption will be removed in the near future when the lamps are calibrated. With the three restrictive assumptions, the laboratory measurement of the lobed sensitivity patterns is really a measurement of the fixed geometry or configuration of the optical sensors of the two PWI systems.

The major components of the Range Simulator are shown in the block diagram. Light from the Xenon strobe passes through a diffuser which was mounted in the lamp housing and then through a pinhole aperture. After passing through a series of irises to eliminate stray light, the slightly divergent beam enters the variable neutral density filter which is used to control the light flux. The attenuated beam then either fills the PWI optics

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or is monitored with a calibrated diode. To standardize all the measurements, the diode current is amplified until a one volt output signal is attained and displayed on an oscilloscope. The gain of the amplifier required to deliver a one volt signal is proportional to the square of the range.

A range measurement consists of the following operations: The PWI system is set to a given azimuth and elevation using two indexing tables. The intensity of the flashing light is varied until the PWI just registers an intruder; a slightly lower intensity would not be sensed by the PWI. This setting represents the maximum range of the PWI for the given orientation. The calibrated diode is then placed into the optical train. It generates a current proportional to the number of watts falling onto the detector surface. (Note that the face of the diode is covered with a Schott RG-780 filter and that the diode is made of Silicon - the wavelength region being measured is the 0.8 to 1.1 micron region.) Determining the number of amps/watt delivered by the diode in the 0.8 to 1.1 micron region is the prime objective of the calibration. The signal from the diode is fed to an amplifier whose gain is adjusted so that a one volt signal is produced and displayed on a fast rise-time oscilloscope. From the relationship of the amplifier gain as a function of the incident current required to produce a one volt signal, the total charge collected is determined by measuring the area under the voltage-time curve displayed on the oscilloscope. Thus from the number of amps/watt or equivalently the number of coulombs/joule, the total energy collected by the diode can be determined. Knowing the amount of energy emitted per steradian by the lamp and assuming an inverse square fall-off of energy with distance, the measurement of the range capability of the PWI is reduced to the recording of the gain of the amplifier.

The following sections discuss each of the components of the range simulator and their calibration and possible introduction of error.



Figure 6.- Pulse Shape of the Delta Products Lamp The Time Base is 0.2 milliseconds/cm.

A. <u>Xenon Lamp</u>. - A Delta Products "Sky Strobe" Xenon flash lamp was used for the laboratory tests even though the Whelan lamp will be used for the flight tests. A Whelan lamp was not available; however, the Delta lamp has roughly the same pulse width (about 400 microseconds at the half power points) and it emits about 1/4 the energy of the Whelan. (See accompanying photograph.)

B. <u>Variable Neutral Density Filter</u>. - An Optical Coating Laboratory Inc. variable neutral density filter was used to attenuate the beam from the flash lamp. The device consists of two counter-rotating filter wheels allowing a continuous variation in neutral density.

It was not necessary to calibrate the filters (although it is expected that they would be linear in the 0.8 to 1.1 micron region).

C. <u>Calibrated Diode</u>.- A United Detector Technology Pin-10 Large area (1.25 cm²) photodetector was calibrated and used as the range measurement standard. It has a Schottky barrier on silicon construction, a response time of less than 10 nanoseconds and a dark current of less than a microampere. With the Schott RG-780 filter mounted on the face of the active area, the spectral response is from 0.8 to 1.1 microns. The linearity is rated at better than 2% from 10^{-11} watts to 10^{-4} watts. It remains to determine the sensitivity in the desired spectral range and this was the object of the calibration procedure. A nine volt bias was put on the detector in order to be able to operate in a DC-light environment.

To calibrate the Pin-10 diode, a Reeder RHL-7c thermopile was used. This thermopile is traceable to NBS and has a rated sensitivity of 2.38 microvolts per microwatt. Light entered the device through a 20 square-millimeter quartz window and the output voltage was read on a nanovoltmeter. The procedure

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consisted of aligning a GaAs laser on the optical rail and first reading the diode-amplifier-oscilloscope combination and then the thermopile-nanovoltmeter combination. A value of 0.498 microamps/ microwatt was obtained which should be good to 10%. As a double check, the GaAs laser was replaced by a projection lamp and only the (roughly) 0.8 to 1.1 micron radiation selected by passing the light through a Schoeffel monochrometer. Using the detectors as described above, a value of 0.51 microamps/microns was found which at least increased the confidence in the calibration procedure.

One problem that became apparent midway in the range sensitivity measurements is that the Pin-10 diodes age rapidly. However, as they age the sensitivity (and hence the calibration) does not change significantly, but the dark current increases by a large amount. Fortunately, this did not affect the measurements but it did complicate the data taking.

D. <u>Amplifier</u>.- A Dymec amplifier was used to amplify the diode current sufficiently to produce a one volt output signal (obviously a function of the impedance). The linearity of the Dymec was checked by applying a known current across a load and tabulating the gain required to produce a one volt signal and the result is shown in an accompanying graph. In addition, since one is detecting a signal which rises up to a peak value in about 100 microseconds, the frequency response of the Dymec was checked. On the times 100 potentiometer setting (see the accompanying figure) the frequency response could severely affect the measurements. Fortunately, such a high gain was not required more than a couple of times.

E. <u>Oscilloscope</u>.- A Tektronics 547 oscilloscope was used to determine when one had a one volt output signal. The unit and its associated plug-in were calibrated using a square-wave generator prior to the tests and the diode calibration.

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V. LABORATORY TESTS

A. Introduction and Range Calculation

In the range simulation section it was emphasized that the lobed sensitivity patterns represented the geometry of the PWI systems. The data presented in this section are in two forms: (1) in the arbitrary "gain" units previously explained which allow one to appropriately change the data if (say) the Fecker discriminator level is changed during field or flight tests, and (2) in units of range (nautical miles) without atmospheric effects.

To assign a range performance capability due to the lobed sensitivity patterns, a "standard" Xenon strobe must be defined. The range values given in the report assumed a Whelan lamp which emits one joule in the 0.8 to 1.1 micron region and is equipped with a Fresnel lens so that the radiation is emitted in 2Π radians of azimuth but only $\pm \Pi/18$ radians in elevation. This gives $(2\pi^2/_9)$ joules/steradian emitted horizontally. Using these factors and a simple inverse-square law for intensity:

Range (n.m.) = 0.0339 (Gain)^{1/2}

B. Fecker Systems PWI

The Fecker Systems PWI uses a ball lens of 1.5-inch radius and is formed out of SF-18 glass. SF-18 glass (sometimes denoted 722293) has an index of refraction of about 1.72 at the Sodium doublet and n = 1.69756 at the infrared Mercury line (1014.0 nm). It has a density of 4.49 g/cm³ and is rated in group 2 for resistance to climatic variations (i.e., under normal humidity conditions during use or storage, only a small change in the surface of an uncoated glass may occur.)

Because of the configuration for mounting detectors on a facetted surface, a lobed structure sensitivity should be expected. Surrounding each detector is a guard ring 0.05 inches wide and

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each detector-guard ring unit is separated by about 0.02 inches to prevent electrical problems. The image of a strobe will be a circle of 0.11 inch radius when normal to a detector. Portions of the image are not detected when the image falls between the detectors on the guard-ring or electrical dead space. A rough calculation shows that as one scans along zero degrees elevation, a condition can be attained between detectors such that only 25% of the signal will be sensed by each of two detectors (the detector area sensitivity is ignored). A 25% figure indicates that a 16:1 variation is possible in the range sensitivity due to geometry alone. If the image is positioned at a junction of four detectors, the signal may fall to less than 5% of full intensity, thus indicating a severe lobed structure - a 50:1 variation in range!

The data to be presented in this section are for two Fecker systems: #19336A, TSC's modified version of the delivered system, and #19335B, a working version of the system delivered and tested for NASA/ERC.

As a guide for the laboratory (and field and flight) tests, the centers of peak sensitivity for each channel were located. (Figure 9 shows for reference the indicator for the Fecker and the convention used in this write-up to indicate a particular channel.) Table I lists the centers of peak sensitivity in a coordinate system in which the center of sensitivity of diode or channel 7 is defined as the origin for both azimuth and elevation. Note that channels 5 and 9 have two sensitivity centers as two detectors are tied to each of these channels.

Data were collected along seven scans or traversals. Referring to Table I, the scans were:

(1)	1-3-7-11-13	Constant	Azimuth, V	Varying E	levation
(2)	5-6-7-8-9	Constant	Elevation,	Varying	Azimuth
(3)	2-3-4	Constant	Elevation,	Varying	Azimuth

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Figure 9.- Fecker Indicator and Numbering Convention Used in the Data Collection

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Channel	el Fecker #19336A		Fecker	#19335B
Number	Azimuth	Elevation	Azimuth	Elevation
1	0°	44°	0 °	44°
2	-21	24	-22	20
3	0	25	0	19
4	21	23	23	22
5	-44	0	-41	0
	-65	0	-52	-3
6	-22	· 0	-23	0
7	0	· 0	0	0
8	21	0	23	0
9	44	0	43	-1
	63	0	63	0
10	-20	-14	-22	-23
11	0	-22	3	-25
12	21	-23	24	-25
13	-3	-42	3	-42

TABLE I.- CENTERS OF PEAK SENSITIVITY

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- (4) 10-11-12 Constant Elevation, Varying Azimuth
- (5) 2-6-10 Constant Azimuth, Varying Elevation
- (6) 4-8-12 Constant Azimuth, Varying Elevation
- (7) 7-12 or 7-4 Varying Azimuth, Varying Elevation

The range capability of the Fecker PWI's depends upon the level or discriminator voltage selected. The roof tests (Section IV) determined that a 6-volt setting will adequately discriminate against background noise and all the ensuing data were collected for this 6-volt value. However, as was experienced in the previous flight tests, one may wish to change this value; so a conversion factor was found for operating with a different threshold voltage. Table XVIII lists a multiplicative factor by which the gain values (to be presented) must be multiplied by in order to predict the range capability for a threshold value of other than 6 volts.

Tables II through XVII consist of the laboratory data describing the lobed sensitivity patterns for a level of 6 volts. Each table consists of azimuth, elevation, Rl gain, Rl range, R2 gain, and R2 range. Polar plots are given for the Rl range (in nautical miles) as a function of either azimuth or elevation. The lobes have been identified by indicating the channel which registered an event.

The data for each scan were consistent in that any value could be repeated within two percent. However, there existed a small discrepancy from day-to-day due perhaps to the slowly increasing dark current in the diode. Since the effect of the atmosphere has not as yet been incorporated into the range values, the small inconsistancies will be removed when the data are fitted to the field test results. The aerosol scattering and water vapor absorption effects will be determined as a function of temperature, humidity, and visibility so that given these parameters, one may (hopefully) accurately predict the range performance of the Fecker PWI.

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MODIFIED FECKER - 19336A						
AZ IMUTH	ELEVATION	DIODE	NO ATMOS. Rl GAIN	NO ATMOS. Rl RANGE	NO ATMOS. R2 GAIN	NO ATMOS. R2 RANGE
0°	0°	7	426.	.700	165.	.436
· · ·	- 1	7	426.	.700	165.	. 436
	- 2	. 7	426.	.700	165.	.436
	- 3	7	387.	.667	150.	.415
	- 5	7	294.	.581	114.	.362
	- 8	7	169.	.441	65.3	.274
	- 9	7	125.	. 379	48.5	.236
	-10	11	100.	.339	38.7	.211
	-12	11	188.	.465	72.9	.289
	-15	11	286.	.573	111.	. 357
	-18	11	358.	.641	139.	. 400
	-20	11	348.	.632	135.	. 394
	-21	11	361.	.644	140.	.401
	-22	11	358.	.641	139.	.400
	-23	11	354.	.638	137.	. 397
	-24	11	354.	.638	137.	. 397
	-25	11	356.	.640	138.	. 398
	-26	11	327.	.613	127.	. 382
	-29	11	198.	.477	76.8	.297
	-32	11	101.	.341	39.2	.212
·	-33	11	70.0	.284	27.2	.177
	-34	13	83.6	.310	32.4	.193
	-37	13	172.	.445	66.9	.277
	-39	13	203.	.483	78.9	.301
	-40	13	207.	.488	80.6	. 304
•	-41	13	217.	.500	84.3	.311
	-42	13	223.	.506	. 86.4	.315
	-43	13	223.	.506	86.4	.315
	-44	13	218.	.501	84.9	. 312
	-45	13	211.	. 492	81.7	.306
	-46	13	206.	.487	80.0	. 303
	-47	13	202.	. 482	78.2	.300
	-48	13	167.	.438	64.8	.273
	-50	13 .	122.	.374	47.6	.234
•	-52	13	74.5	.293	28.9	.182
	-53	13	50.0	.240	19.4	.149
	-54	13	34.0	.198	13.2	.123
•	-56	13	10.2	.108	3.94	.067

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Table II

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		MODIFI	ED FECKER	- 19336A		
AZIMUTH	ELEVATION	DIODE	NO ATMOS. Rl GAIN	NO ATMOS. Rl RANGE	NO ATMOS. R2 GAIN	NO ATMOS. R2 RANGE
0°	59°	1	7.11	.090	2.76	.056
1	58	1	27.6	.178	10.9	.112
	57	1	48.2	.235	18.5	.146
	56.5	1	51.9	.244	20.7	.154
	56	1	68.9	.281	26.7	.175
	55.5	1	79.9	.303	31.7	.191
	<u>5</u> 5	1	106.	.349	39.0	.212
	52	1	174.	.447	68.2	.280
	49	1	297.	.584	115.	.365
	46	1	328.	.614	127.	. 382
	45	1	328.	.614	127.	. 382
	44	1	326.	.612	126.	.381
	43	1	310	.597	120.	.371
	42	1	276.	.563	107.	.351
	39	1	161.	.430	62.2	.267
	37	1	94.2	.329	36.5	.205
	36	3	70.3	.284	27.2	.177
	34	3	159.	.427	61.8	.267
	31	3	274.	.561	103.	.344
	28	3	334.	.619	129.	.385
	. 27	.3	344.	.629	133.	.391
	26	3	344.	.629	133.	.391
	25	3	351.	.635	136.	. 395
	24	3	372.	.654	144.	.407
	23	3	351.	.635	136.	.395
	22	3	357.	.641	138.	.398
	21	3	357.	.641	138.	.398
	20	3	331.	.617	128.	.712
	19	3	308.	.595	119.	.370
	17	3	218.	.501	84.6	.312
	15	3	124.	.378	48.0	.235
	14	3	85.2	.313	33.0	.195
	<u> 13 </u>	7	88.8	.320	39.4	.213
	12	7	142.	.404		.252
	10	7	232.	.516	90.0	. 322
	7	7	331.	.617	128.	. 384
	5		426.	./00	1 105.	430
 	3		401.	.0/9	155.	422
	2		434.	706	168.	439
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Table III

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Figure 10

	.	MODIFI	ED FECKER	- 19336A		
AZ IMUTH	ELEVATION	DIODE	NO ATMOS. Rl GAIN	NO ATMOS. Rl RANGE	NO ATMOS. R2 GAIN	NO ATMOS. R2 RANGE
-20°	38°	2	3.39	.062	1.31	.039
	37	2	31.3	.190	12.2	.118
	36	2	71.4	.286	27.7	.178
	34	2	172.	.445	66.8	.277
	32	2	231.	.515	89.4	. 321
	30	2	353.	.637	137.	.397
	28	2	402.	.680	156.	.423
	26	2	405.	.682	157.	.425
	24	2	413.	.689	160.	. 429
	22	2	413.	.689	160.	.429
·	20	2	391.	.670	152.	.418
	18	2	293.	.580	114.	.362
· ·	16	2	202.	.482	78.3	.300
	14	2	101.	.341	39.1	.212
	12	2	45.7	.229	17.7	.143
	11	6	142.	.404	55.0	.251
	10	6	284.	.571	110.0	. 356
	8	6	385.	.665	149.	.414
	6	6	495.	.754	192.	.470
	4	6	608.	.836	235.	.520
	2	6	612.	.839	238.	.523
	0 1	6	640.	.858	248.	.534
	- 2	6	652.	.866	253.	.539
	- 4	6	499.	.757	194.	.472
	- 6	6	374.	.656	145.	.408
	- 8	6	216.	. 498	83.7	.310
	-10	6	92.8	.327	36.0	.203
	-11	10	124.	.378	48.1	.235
	-12	10	130.	.387	50.4	.241
	-14	10	138.	. 398	53.5	.248
	-16	10	107.	.351	41.5	.218
	-18	10	100.	.339	38.8	.211
	-20	10	79.0	.301	30.6	.186
·····	-22	10	14.9	.131	5.78	.081
	-24	10	11.1	.113	4.31	.070
	-26	10	6.87	.089	2.66	.055

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Table IV

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Figure 11

MODIFIED FECKER - 19336A							
AZIMUTH	ELEVATION	DIODE	NO ATMOS. Rl GAIN	NO ATMOS. Rl RANGE	NO ATMOS. R2 GAIN	NO ATMOS. R2 RANGE	
-77	0°	5.	16.1	.136	6.25	.085	
-76		5	26.4	.174	10.3	.109	
-74		5	52.8	.246	20.4	.153	
-72		5	70.0	.284	27.1	.177	
-70		5	73.9	.291	28.6	.181	
-68		5	77.7	.299	30.1	.186	
-67		5	80.0	. 303	31.0	.189	
-66		5	84.0	. 320	32.6	.194	
-65		5	86.8	.316	33.6	.197	
-64		5	84.0	.311	32.5	.193	
-63		5	88.2	.318	34.2	.198	
-62		5	77.4	.298	30.0	.186	
-60		5	55.0	.251	21.3	.157	
- 58		5	39.7	.214	15.0	.131	
-57		5	48.5	.236	18.8	.147	
-56		5	64.6	.273	25.0	.170	
-54	· · · · · · · · · · · · · · · · · · ·	5	138.	. 398	53.9	.249	
-52		5	208.	.489	80.8	.305	
-50	•	5	292.	.579	113.	.360	
-48		5	286.	.573	111.	.357	
-46		5		.594	119.	.370	
-45	· ·	5	318.	.605	123.	.376	
-44		5	318.	.605	123.	.376	
-43		<u> </u>		.592	118.	.368	
-41		5	282.	.569	109.	.354	
-38		5	174.	.447	67.4	.278	
- 36	ļ	5	106.	.349	41.3	.218	
- 35	L	6	87.6	.317	34.0	.198	
-33	L	6	177.	.451	68.6	.281	
-29		6	398.	.676	154.	.421	
-26		6	522.	.775	202.	.482	
-24		6	541.	.789	210.	.491	
-23	L	6	540.	.788	209.	.490	
-22		6	580.	.816	225.	.509	
-21		6	522.	.775	202.	.482	
-19		6	529.	.780	205.	.485	
-16	L	6	358.	.641	139.	.400	
-13		6	164.	.434	63.9	.271	
-12	<u> </u>	7	129.	.385	50.0	.240	
-11		7	204.	.484	79.2	.302	

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 σ_{ij}

		MODIFI	ED FECKER	- 19336A		
AZIMUTH	ELEVATION	DIODE	NO ATMOS. R1 GAIN	NO ATMOS. R1 RANGE	NO ATMOS. R2 GAIN	NO ATMOS. R2 RANGI
-8°	0°	7	328.	.614	127.	. 382
-6	1	7	388.	.668	150.	.415
-3		7	426.	.700	165.	.436
-2		7	429.	.702	166.	.437
-1		7	434.	.706	168.	.439
0	1	7	444.	.714	172.	.445
1		7	426.	.700	165.	.436
2		7	429.	.702	166.	.437
3		7	429.	.702	166.	.437
4		7	382.	.663	148.	.412
. 7		-7	279.	.566	108.	.352
10		7	122.	. 374	47.1	.233
11		8	157.	. 425	60.7	.264
14		8	310.	. 597	120.	.371
17		8 ·	475.	.739	184.	.460
19		8	504.	.761	197.	.476
20		8	500.	.758	194.	.472
21		8	506.	.763	198.	.477
22		8	508.	.764	200.	. 479
23		8	483.	.745	187.	.464
25		8	449.	.718	174.	.447
27	·····	8	398.	.676	154.	.421
30		8	228.	.512	88.2	.318
32		8	119.	.370	46.2	.230
33	·····	8	82.3	.308	31.9	.192
34		8	45.8	.229	17.6	.142
35	<u>↓ · · · · · · · · · · · · · · · · · · ·</u>	9	51.8	244	18.2	.145
38	+	9	81.6	.306	31.6	.191
41		9	104.	.346	40.3	.215
43	1 1	9	106.	.349	41.0	.217
44	<u> </u>	. 9	106.	.349	41.0	.217
45		9	103.	.344	40.0	.214
46	1	9	100.	.339	38.7	.211
48		9	97.0	.334	37.6	.208
51	<u> </u>	9	56.1	.254	21.7	.158
.52	<u>† ··· · † ····</u>	9	45.5	.229	17.6	.142
53		9	30.5	.187	11.8	.117
54	1	9	42.6	.221	16.5	.138
57		9	101.	.341	39.4	.213
60	1	9	177.	.451	68.7	.281



Table VI

 ϕ_{ij}

MODIFIED FECKER - 19336A								
AZIMUTH	ELEVATION	DIODE	NO ATMOS. Rl GAIN	NO ATMOS. Rl RANGE	NO ATMOS. R2 GAIN	NO ATMOS. R2 RANGE		
61°	0°	9	202.	.482	78.2	.300		
62	•	9	201.	.481	78.0	.299		
63	•	9	203.	.483	78.6	.301		
64		9	199.	.478	77.5	.298		
65		9	199.	.478	77.5	.298		
66		9	199.	.478	77.5	.298		
67		9	192.	.470	74.4	.292		
68		9	192.	.470	74.4	.292		
69		9	182.	.457	70.5	.285		
71		9	158.	.426	61.4	.266		
73		9	91.4	.324	35.5	.202		
75		9	28.6	.181	11.1	.113		
76		9	17.0	.140	6.60	.087		
77		9	11.7	.116	4.56	.072		
78		9	7.88	.095	3.05	.059		
79		9	6.91	.089	2.68	.056		
80	.	-	-		-			

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Table VI continued

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-31-



Figure 12

MODIFIED FECKER - 19336A							
AZIMUTH	ELEVATION	DIODE	NO ATMOS. Rl GAIN	NO ATMOS. Rl RANGE	NO ATMOS. R2 GAIN	NO ATMOS. R2 RANGE	
21°	36°	4	12.9	.122	5.0	.076	
	34	4	35.5	.202	13.8	.126	
┝━╍──┞────	30	4	95.4	. 331	37.0	206	
	28	4	111.	.357	43.0	.222	
	25	4	124.	.378	48.1	.235	
·····	24	4	132.	. 390	51.1	.242	
	23	4	136.	. 395	52.8	.246	
	22	4	122.	.374	47.3	.233	
	21	4	124.	.378	48.0	.235	
<u> </u>	19	4	100.	.339	38.7	.211	
	16	4	62.9	.269	24.4	.168	
	14	4	37.9	.209	14.7	.130	
	13	8	31.0	.189	12.0	.117	
	10	8	215.	.497	83.3	.309	
	7	8	425.	.699	165.	.436	
	4	8	546.	.792	211.	.492	
	2	8	549.	.794	212.	: 494	
	1	8	538.	.786	208.	.489	
	0	8	505.	.762	196.	.475	
	- 1	8	512.	.767	198.	.477	
	- 2	8	540.	.788	209.	.490	
	- 3	8	469.	.734	182.	.457	
}	- 6	8	348.	.632	135.	.394	
	- 9	8	154.	. 421	59.8	.262	
	-10	8	119.	. 370	46.2	.230	
·	-11	12	90.0	. 322	34.8	.200	
	-14	12	215.	.497	87.3	.317	
	-17	12	335.	.621	130.	.387	
	-20	12	376.	.657	146.	.410	
	-22	12	389.	.669	151.	.417	
<u> </u>	-24	12	389.	.669	151.	.417	
	-26	12	389.	.669	151.	.417	
	-28	12	298.	.585	116.	.365	
<u> </u>	-30	12	288.	.575	112.	.359	
┝─── ┤ ─── [╼]	-32	12	119.	.370	46.2	.230	
	-34	12	54.4	.250	21.6	.158	
	-36	12	10.0	.107	3.71	.065	
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Table VII



Figure 13
		MODIFI	ED FECKER	- 19336A		
AZIMUTH	ELEVATION	DIODE	NO ATMOS. Rl GAIN	NO ATMOS. Rl RANGE	NO ATMOS. R2 GAIN	NO ATMOS. R2 RANGE
-37°	24°	2	19.2	.149	7.43	.092
-36		2	41.5	.218	16.1	.136
-33	····	2	128.	. 384	49.9	.240
-30		2	224.	.507	87.2	.317
-27		2.	300.	.587	116.	.365
-24		2	317.	.604	123.	.376
-22		2	278.	.565	108.	.352
-21		2	332.	.618	129.	385
-20		2	289.	.576	112.	.358
-19	· ·	2	278.	.565	108.	.352
-16	ļ	2	191.	.469	74.0	.292
-13		2	105.	.347	40.9	.217
-12		3	78.7	,301	30.5	.187
- 9		3	1/2.	.445	66.9	.277
- 6		3.	257.	.544	99.9	.339
- 3		3	358.	.641	139.	.400
- 2		3	358.	.641	139.	.400
- 1		3	301.	.644	140.	.401
<u> </u>		3	333.	621	130.	.38/
2	 		335.	621	130.	.38/
<u><u></u></u>	<u> </u>		335	621	130.	. 387
<u> </u>			279	565	109	- 307
	<u> </u>		104	460	71 2	.332
-0			156	400	60.6	.200
10		3	49.2	339	38.5	204
10		3	78 0	299	30.3	197
12	┠───┼────	4	81 1	305	31 4	. 190
13	<u> </u>	4	117.	. 367	45.5	228
<u> </u>	<u> </u>	4	214.	.496	83.0	.309
19	<u> </u>	4	276.	.563	107.	.351
20	tt	4	300.	.587	116.	.365
21	<u> </u>	4	327.	.613	127.	.382
22	<u> </u>	4	341.	.626	132.	.390
23	<u> </u>	4	325.	.611	126.	.381
24	<u> </u>	4	325.	.611	126.	.381
25		4	314.	.601	122.	. 374
28	1	4	278.	.565	108.	.352
-31		4	153.	.419	59.4	.261
33		. 4	72.3	.288	28.0	.179
37		4	8.24	.097	3.20	.061





Table VIII

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	MODIFIED FECKER - 19336A								
AZIMUTH	ELEVATION	DIODE	NO ATMOS. R1 GAIN	NO ATMOS. Rl RANGE	NO ATMOS. R2 GAIN	NO ATMOS. R2 RANGE			
35°	-22°	12	14.0	.127	5.42	.079			
34		12	32.2	.192	12.5	.120			
31	[12	60.0	.263	23.3	.164			
28		12	120.0	.371	46.6	.231			
24		12	186.	.462	72.3	.288			
23		12	182.	. 457	70.3	.284			
22		12	180.	.455	70.0	.284			
21		12	190.	.467	73.3	.290			
. 20		12	190.	.467	73.3	.290			
19		12	186.	.462	72.2	.288			
16		12	148.	.412	57.2	.256			
13		12	102.	.342	39.7	.214			
11		12	60.0	.263	23.3	.164			
10		12	41.0	.217	15.9	.135			
9		11	54.1	.249	21.0	.155			
7.		11	87.0	.316	33.8	.197			
3		11	158.	. 426	61.1	.265			
1		11	188.	.465	73.0	.290			
0		11	190.	.467	73.7	.291			
- 1		11	180.	.455	70.0	.284			
- 2		11	172.	.445	67.0	.278			
5		11	156.	.423	60.8	.264			
- 8		11	127.	. 382	49.1	.238			
11		11	78.0	.299	30.2	.186			
-13		11	43.4	.223	16.8	.139			
14		11	27.0	.176	10.5	.110			
15		10	10.4	.109	4.03	.068			
-16		10	13.3	.124	5.18	.077			
-19	· · ·	10	12.2	.118	4.74	.074			
-20		10	3.90	.067	1.51	.042			



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Table IX

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MODIFIED FECKER - 19336A								
AZIMUTH	ELEVATION	DIODE	NO ATMOS. Rl GAIN	NO ATMOS. Rl RANGE	NO ATMOS. R2 GAIN	NO ATMOS. R2 RANGE		
0 °	0 °	7	216.	. 498	83.8	.310		
2	- 2 .	7	182.	.457	70.8	.285		
4	- 4	7	154.	.421	59.8	.262		
6	- 6	7	114.	.362	44.4	.226		
8	- 8	7	56.8	.256	22.0	.159		
10	-10	7	19.0	.148	7.40	.092		
12	-12	12	21.6	.158	8.39	.098		
14	-14	12	49.0	.237	19.0	.148		
16	-16	12	79.5	.302	30.8	.188		
18	-18	12	119.	.370	46.1	.230		
20	-20	12	166.	.437	64.5	.272		
22	-22	12	157.	.425	61.0	.265		
24	-24	12	157.	.425	61.0	.265		
26	-26	12	118.	.368	45.6	.229		
28	-28	12	72.6	.289	28.1	.180		
30	-30	12	36.6	.205	14.2	.128		
32 .	- 32	12 -	10.7	.111	4.16	.069		
33	-33	12	3.77	.066	1.46	.041		

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Table X

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	UNMODIFIED FECKER - 19335B								
AZIMUTH	ELEVATION	DIODE	NO ATMOS. Rl GAIN	NO ATMOS. Rl RANGE	NO ATMOS. R2 GAIN	NO ATMOS. R2 RANGE			
-37°	20°	2	36.3	.204	12.2	.118			
-36		2	68.3	.280	24.1	.167			
-35		2	105.	.347	37.3	.207			
-32	1	2	172.	.445	66.5	.276			
-30		2	264.	.551	102.	. 342			
-27	1	2	413.	.689	160.	. 429			
-25	1	2	480.	.743	186.	.462			
-23	1	2	475.	.739	184.	.460			
-22	<u> </u>	2	465.	.731	180.	.455			
-20		2	472.	.737	183.	.459			
-18		2	387.	.667	150.	.415			
-14	1	2	204.	.484	79.1	.302			
-13		2	148.	.412	57.3	.257			
-12		2	111.	.357	41.3	.218			
		3	141.	.403	54.7	.251			
- 8		3	284.	.571	110.	.356			
- 5		3	425.	.699	165.	.436			
2		3	492.	.752	191.	.469			
0		3	507.	.763	196.	.475			
3		3	498	.757	193.	.471			
5		3	429.	.702	166.	.437			
. 8		3	264.	.551	102.	.342			
11		3	126.	.381	42.0	.220			
12		4	93.3	.327	32.3	.193			
13		4	_ 121.	.373	46.9	.232			
15		4	194.	.472	75.3	.294			
18		4	289.	.576	112.	.359			
20		4	351.	.635	136.	. 395			
22		4	402.	.680	156.	.423			
24		4	374.	.656	145.	.408			
27		4	359.	.642	139.	.400			
30		4	231.	.515	89.9	.321			
33		4	129.	. 385	42.7	.222			
35		4	69.6	.283	23.4	.164			
37		4	18.2	.145	6.62	.087			
41		. 4	3.06	.059	1.18	.034			

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Table XI

 $a_{\rm V}$

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		UNMODI	FIED FECK	ER - 19335B		- *
AZIMUTH	ELEVATION	DIODE	NO ATMOS. R1 GAIN	NO ATMOS. Rl RANGE	NO ATMOS. R2 GAIN	NO ATMOS. R2 RANGE
2 3 °	39°	4	2.59	.055	1.01	.034
1	37	4	10.3	.109	3,36	.062
	35	4	102.	.342	39.8	.214
	32	4	238.	.523	92.8	.327
	29	4	307.	. 594	119.	.370
	28	4	325.	.611	126.	.381
	27	4	335.	.621	130.	.387
	26	4	337.	.622	131.	.388
	25	4	337.	.622	131.	.388
	24	4	337.	.622	131.	. 388
	23	4	343.	.628	133.	.391
	22	4	345.	.630	134.	. 392
	21	4	345.	.630	134.	. 392
	20	4	327.	.613	127.	.382
	18	4	245.	.531	95.0	.330
	15	4	112.	.359	43.6	.224
	14	4	51.6	.244	20.0	.152
	13	8	51.6	.244	20.0	.152
	9	8	153.	.419	59.3	.261
	5	8	279.	.566	108.	. 352
	2	8	279.	.566	108.	.352
	0	8	279.	.566	108.	.352
	- 1	8	279.	.566	108.	.352
	- 2	8	273.	.560	106.	. 349
	5	8	176.	.450	68.2	.280
1	- 8	8	84.0	.311	32.6	. 194
	- 9	8	49.8	.239	19.3	.149
	-10	12	64.0	.271	24.8	.169
	-13	12	276.	.563	107.	.351
	-16	12	340.	.625	132.	. 390
	-19	12	400.	.678	155.	. 422
	-22	12	400.	.678	155.	.422
	-26	12	400.	.678	155.	.422
	-27	12	366.	.649	142.	.404
	-29	12	279.	.566	108.	.352
	-32	12	113.	.360	43.9	.225
	-35	12	24.8	.167	9.61	.105
	-36	12	10.0	.107	3.79	.066
	-37	12	7.89	.095	2.50	.053
	1 -40	12	2.52	.054	1.00	.034



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Table XII

 $\sim_{\rm V}$

-42-



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		UNMODIF	IED FECKE	R - 19335B		
			NO	NO	NO	NO
AZIMUTH	ELEVATION	DIODE	ATMOS.	ATMOS.	ATMOS.	ATMOS
			RI GAIN	R1 RANGE	R2 GAIN	R2 RANGE
<u>-81°</u>	0°	5	3.71	.065	1.44	.041
-80	1	5	4.97	.076	1.92	.047
- 79		5	8.91	.101	3.45	.063
~78		5	13.7	.126	4.52	.072
-77		5	33.6	.197	11.0	.112
-76		5	82.8	. 309	27.0	.176
-75		5	124.	. 378	44.3	.226
-74		5	146.	. 410	56.9	.256
-70		_5	230.	.514	89.3	.320
-66		5	241.	.526	93.2	.327
-63		5	266.	.553	103.	.344
-61		5	259.	.546	100.	.339
- 59		5	192.	.470	74.7	.293
-57		5	136.	. 395	47.3	.233
-55		_5	141.	.403	47.4	.233
-53		5	187.	. 464	72.6	.289
-51		5	258.	.545	100.	.339
-49		5	317.	.604	123.	.376
-47		5	325.	.611	126.	.381
-45		5	340.	.625	132.	.390
-43		5	356.	.640	138.	.398
-41		5	356.	.640	138.	. 398
- 39		5	284.	.571	110.	.356
- 36		5	159.	.428	62.0	.267
- 35		5	142.	.404	48.6	.236
- 34		66	140.	.401	46.9	.232
-31		6	206.	.487	80.0	.303
-28		6	358.	.641	139.	.398
-26	LL	6	392.	.671	152.	.418
-24		6	407.	.684	158.	.426
-22	L	6	418.	.693	162.	.432
-18		6	403.	.681	156.	.423
-16		6	333.	.619	129.	.385
-13	L	6	186.	.462	72.2	.288
-12		6	135.	. 394	52.6	.246
<u> </u>	├ ─── │ ────	1	161.	.430	62.5	.268
- 9			275.	.562	106.	. 349
- 7	 	<u> </u>	410.	.686	159.	.428
- 4	↓ ₩	<u> </u>	560.	.802	217.	.499
-1	1 🔻	17	600.	.830	233.	.518



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Table XIII

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		UNMODIF	IED FECKE	R - 19335B		
AZIMUTH	ELEVATION	DIODE	NO ATMOS. R1 GAIN	NO ATMOS. R1 RANGE	NO ATMOS. R2 GAIN	NO ATMOS. R2 RANGE
0°	00	7	600.	. 830	233.	.518
1	1	7	600.	.830	233.	.518
2		7	600.	.830	233.	.518
3		7	600.	.830	233.	.518
4		7	600.	.830	233.	.518
6		7	521.	.774	202.	. 482
10		7	262.	.549	102.	. 342
12		7	143.	.405	48.6	. 236
13		8	145.	.408	44.4	.226
14		8	154.	.421	59.9	.262
18		8	284.	.571	110.	. 356
21		8	335.	.621	130.	. 387
23		8	· 333.	.619	129.	. 385
25		8	333.	.619	129.	. 385
27		8	289.	.576	112.	. 359
29		8	238.	.523	92.7	.326
32		8	137.	.397	39.2	.212
33		9	97.4	.335	31.0	189
34		9	136.	.395	44.2	.225
	·		227.	.511	88.2	. 310
40		<u> </u>	304.	.591	126	391
41	<u> </u>		325.	.011	120.	391
45			325.	611	126	381
43	<u> </u>		310	597	120	371
47			323	609	125	379
50			323.	609	125.	. 379
			323.	609	125.	.379
52	<u> </u>	<u> </u>	260.	.547	101.	.341
56	<u>├──</u>	9	175.	.448	68.0	.280
57		9	170.	.442	66.0	.275
58	tt	9	180.	.455	70.0	.284
61		9	263.	.550	102.	.342
64	<u> </u>	9	296.	.583	115.	. 364
67	h	9	276.	.563	107.	.351
70		9	271.	.558	105.	.347
73		9	192.	.470	74.2	. 292
75		. 9	128.	. 384	43.6	.224
77		9	32.1	.192	11.0	.112
78		9	19.2	.149	7.15	.091
80		9	10.4	.109	3.38	.062
84		9	3.01	.059	1.16	.037

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Table XIII continued

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		UNMODIF	IED FECKEF	R - 19335B		
AZIMUTH	ELEVATION	DIODE	NO ATMOS. R1 GAIN	NO ATMOS. Rl RANGE	NO ATMOS. R2 GAIN	NO ATMOS. R2 RANGE
-23°	35	2	6.06	.084	2.34	.052
	34	2	58.1	.258	22.6	. 161
	33	2	76.2	.296	29.7	.185
	31	2	208.	.489	81.0	.305
	28	2	331.	.617	129.	.385
	25	_ 2	375.	.657	146.	.410
	•23	2	385.	.665	150.	.415
	21	2	390.	.670	152.	.418
	20	2	398.	.676	155.	.422
	19 _	2	372.	.654	145.	.408
	16	· ·2	233.	.518	91.0	. 323
	13	2	93.0	.327	36.2	.204
	12	6	134.	. 392	52.1	.245
	11	6	180.	.455	70.0	.284
	8	6	277.	.564	108.	. 352
	4	6	329.	.615	128.	. 384
	2	6	326.	.612	127.	. 382
	1	6	326.	.612	127.	. 382
	0	6	326.	.612	127.	. 382
	- 1	6	329.	.615	128.	.384
	- 2	6	321.	.607	125.	.379
	- 5 .	6	295.	582	115.	.364
	- 8	6	178.	.452	69.6	.283
_	-10	6	104.	.346	40.5	.216
	-11	10	85.9	. 314	33.4	.196
	-14	10	238.	.523	92.8	. 327
	<u> </u>	10	340.	.625	132.	.390
	-20	10	378.	.659	147.	.411
	-22	10	378.	.659	147.	.411
	-23	10	380.	.661	148.	.412
	-24	10	388.	.668	151.	.417
	-25	10	388.	.668	151.	.417
	-27	10	380.	.661	148.	.412
	-28	10	380.	.661	148.	.412
	-29	10	290.	.577	113.	.360
	-31	10	194.	.472	75.8	.295
	-33	10	105.	.347	35.0	.201
	-34	10	56.9	.256	22.1	.159
	-35	10	34.0	.198	13.2	.123
	-36	10	17.4	.141	6.09	.084
V] -37	10	8.08	.096	2.48	.053



Table XIV

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		UNMODIF	IED FECKER	- 19335B		
AZIMUTH	ELEVATION	DIODE	NO ATMOS. Rl GAIN	NO ATMOS. R1 RANGE	NO ATMOS. R2 GAIN	NO ATMOS. R2 RANGE
-37°	-25°	10	2.80	.057	1.04	.035
-36	1	10	15.1	.132	4.97	.076
-33		10	108.	.352	41.8	.219
-30		10	239.	.524	92.9	.327
-27		10	354.	.638	137.	.397
-24		10	390.	.669	151.	.417
-23		10	403.	.681	156.	.423
-22		10	403.	.681	156.	. 423
-21		10	403.	.681	156.	.423
		10	390.	.670	151.	.417
-16		10	282.	.569	109.	.354
-13		10	156.	.423	60.2	.263
-11		10	79.3	. 302	30.7	.188
-10		11	104.	.346	40.5	.216
- 9		11	145.	.408	56.2	.254
- 7		11 _	230.	.514	89.2	.320
- 5		11	302.	.589	117.	.367
- 2		11	392.	.671	152.	.418
0		11	398.	.676	154.	.421
1		11	398.	.676	154.	.421
2		11	408.	.685	158.	.426
3		11	392.	.671	152.	.418
5		11	354.	.638	137.	. 397
8		11	256.	.542	99.8	.339
10		11	154.	.421	59.8	.262
11		11	110.	.356	42.5	.221
12		12	, 104.	.346	40.4	.216
16		12	274.	.561	106.	.349
19		12	364.	.647	141.	.403
20		12	372.	.654	144.	.407
21	 	.12	408.	.685	158.	.426
22		12	408.	.685	158.	.426
23		12	408.	.685	158.	.426
24		$\frac{12}{12}$	388.	.668	150.	415
27		12	354.	.638	13/.	. 397
30	·	12	255.	.541	99.0	.337
33	<u> </u>	$\frac{12}{10}$	122.	.374	<u>4/.L</u>	.233
35	↓	12	55.0	.251	21.3	.15/
37	l V	12	9.38	.104	3.22	.061

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	·	UNMODIF	IED FECKE	R - 19335B		
AZIMUTH	ELEVATION	DIODE	NO ATMOS. Rl GAIN	NO ATMOS. Rl RANGE	NO ATMOS. R2 GAIN	NO ATMOS. R2 RANGE
1°	59°	1	3,61	.064	1,40	. 040
F	58	1	18.1	.144	5.54	.080
	57	1	39.2	.212	15.2	.132
	56	1	75.9	.295	14.7	.130
	54	1.	161.	.430	62.3	.268
	51	1	247.	.533	96.0	. 332
	47	· 1	296.	.583	115.	.364
	45	1	330.	.616	128.	.384
	43	1	330.	.616	128.	. 384
	42	1	322.	.608	125.	.379
	39	1	204.	.484	79.0	. 301
	36	1	85.6	. 314	33.2	.195
	35	1	54.8	.251	10.6	.110
	34	· 3	99.0	.337	19.2	.149
·	31	3	256.	.542	99.1	.338
	28	3	369.	.651	143.	.405
	25	3	429.	.702	166.	.437
	22	· 3	429.	.702	166.	.437
	19	3	429.	.702	166.	437
		3	390.	.669	151.	.417
	14	3	194.	.4/2	/5.4	.294
	13		196	. 387	50.2	.240
	12		180.	.402	12.1	.288
	9		3/4.	750	145.	.408
			505	752	100.	.407
			508	764	200	.4//
┝━━━╋╋	<u>- 1</u>	7	508.	.764	200.	479
┝	- 4	7	508.	.764	200.	479
┝ <u>╍──</u> ┨────	- 5		472	.737	183.	459
	- 8	7	266.	.553	103.	.344
	-10	7	136.	. 395	52.7	.246
	-11	7	90.2	. 322	35.0	.201
	-12	11	108.	.352	42.1	.220
<u> </u>	-15	11	222.	.505	86.0	. 314
	-18	11	364.	.647	141.	.403
t	-21	11	364.	.647	141.	.403
	-24	11	375.	.657	145.	.408
<u> </u>	-25	11	375.	.657	145.	.408
	-27	11	351.	.635	136.	. 395
	-29	11	268.	.555	104.	.346



Table XVI

UNMODIFIED FECKER - 19335B								
AZIMUTH	ELEVATION	DIODE	NO ATMOS. Rl GAIN	NO ATMOS. Rl RANGE	NO ATMOS. R2 GAIN	NO ATMOS. R2 RANGE		
1°	-31°	11	181.	.456	70.2	.284		
	-32	11	128.	.384	49.6	.239		
	-33	13	90.0	.322	34.8	.200		
	-36	13	244.	.530	94.9	.330		
	-39	13	356.	.640	138.	. 398		
	-42	13	408.	.685	158.	.426		
	-44	13	397.	.676	154.	.421		
	-46	13	377.	.658	146.	.410		
	-48	13	377.	.658	146.	.410		
	-51	13	214.	.496	83.0	.309		
	-54	13	78.3	.300	28.2	.180		
	-55	13	49.0	.237	19.0	.148		
▼	-57	13	10.0	.107	3.83	.066		

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Level: 6 Volts

Table XVI continued



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UNMODIFIED FECKER - 19335B							
AZIMUTH	ELEVATION	DIODE	NO ATMOS. Rl GAIN	NO ATMOS. Rl RANGE	NO ATMOS. R2 GAIN	NO ATMOS. R2 RANGE	
1°	0°	7	493.	.753	191.	.469	
3	2	7	493.	.753	191.	.469	
5	4	7	480.	.743	186.	.462	
7	6	7	356.	.640	138.	.398	
9	8	7	247.	.533	96.0	.332	
11	10	7	84.3	.311	32.7	.194	
12	11	7	40.7	.216	15.8	.135	
15	14	4	38.2	.210	14.8	.130	
17	16	4	107.	.351	41.6	.219	
19	18	4	284.	.571	110.	.356	
21	20	4	356.	.640	138.	.398	
23	22	4	354.	.638	137.	.397	
25	24	4	349.	.633	135.	.394	
27	26	4	313.	.600	121.	.373	
29	28	4	219.	.502	85.0	.313	
31	30	4	126.	.381	49.1	.238	
33	32	4	52.2	.245	20.3	.153	
35	34	4	7.27	.091	2.59	.055	



Table XVII

TABLE XVIII.- MULTIPLICATIVE FACTORS FOR DIFFERENT THRESHOLD VOLTAGE LEVELS

Voltage Level	Fecker #19336A	Fecker #19335B
10 volts	0.61	0.59
8	0.73	0.72
6	1.00	1.00
4	·	1.60

C. Loral PWI

The Loral PWI optical system consists of spherical optics, an aperture stop, filter, and light "pipes". Incident light is collected by a hemispheric lens and is directed through an aperture stop to reduce spherical aberration. A Schott RG-780 filter cuts off the radiation below 0.8 microns. After the filter, the light enters a second hemispheric lens which forms a spherical focal surface. To utilize the smallest detectors, to reduce the danger of sun damage, and to minimize the reduced contour sensitivity at the junctions between detectors, the optical system employs light "pipes" between the focal hemisphere and the photoconductors.

(As an aside, it was found expeditious in the data collection to monitor the video take-out points. A 565-Techtronix dual-beam oscilloscope was used to trigger the time base and measure the pulse size just as the audio alarm sounded. This technique allowed one to see the relative size of the signal in the Loral PWI and, hence, to know how to adjust the variable neutral density filter.)

The Loral system consists of two pods — one for each wing. Figure 22 shows the coordinate system and the location of the centers of sensitivity for each photoconductor. Although the system is equipped with a down-looker, it was not checked in the laboratory tests.

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LEFT POD

RIGHT POD

Figure 22.- Coordinate system used for the Loral Laboratory tests. (Positive elevation is into the plane of the paper. The numbers indicate the centers of sensitivity of the channels or detectors, System 19329B.)

The setting of the sensitivity control (located on the Loral display panel) controls the range capability of the PWI. The potentiometer control, unfortunately, has no indicator markings so the position of the setting was determined by measuring the voltage across the potentiometer. It is imperative that this voltage be known during the field and flight tests (as the data will show), and it is recommended that either a multiposition calibrated switch or a single voltage be selected for future tests.

In addition to the sensitivity voltage, the background environment in which the photoconductors operate is a critical parameter. The laboratory tests used three backgrounds: dark, daylight illumination, and a headlight inclined 15° to the

Loral 1932	29B					Lef	t Pod		
Photo-	Sensitivity	Dark	Dark Test Room Daylight				Headlight		
conductor	Voltage	Gain	Range	Gain	Range	Gain	Range		
8	2	139.	.340	97.0	.334				
	3	4602.	2.30	140.	.401	9.72	.106		
	4	11259.	3.60	3534.	2.02	9.91	.107		
	5	12231.	3.75	5442.	2.50	16.5	.138		
	6	29322.	5.80	7980.	3.03	17.9	.143		
	7	138510.	12.6	9072.	3.23	21.8	.158		
	8	146205.	13.0	13122.	3.88	21.2	.156		
· •	9	137295.	12.6	10287.	3.44	18.2	.145		
9	2	75.2	. 294	45.9	.230	5.77	.081		
	3	2278.	1.62	1224.	1.19	53.8	.249		
	4	3062.	1.88	2090.	1.55	77.8	.299		
	5	5565.	2.53	2538.	1.71	93.2	.327		
	. 6	5767.	2.57	3564.	2.02	117.	.367		
	7	5605.	2.54	3709.	2.06	147.	.411		
	8	8270.	3.08	4074.	2.16	157.	.425		
	9	5905.	2.61	4590.	2.30	157.	.425		
10	2	131.	.388	70.2	.284	9.49	.104		
	3	5836.	2.59	1645.	1.37	131.	.388		
	4	8060.	3.04	2446.	1.68	189.	.466		
	5	11583.	3.65	2584.	1.72	261.	.548		
	6	40905.	6.86	5426.	2.50	319.	.606		
	7	46332	7.30	6688.	2.77	364.	.647		
	8	51840.	7.72	9196.	3.25	398.	.676		
	9	42282.	6.97	6445.	2.72	444.	.714		
11	2	102.	. 342	43.7	.224	6.21	.085		
	3	2926.	1.83	255.	.541	85.2	.313		
	4	2105.	1.55	242.	.527	136.	.395		
	5	1018.	1.08	328.	.614	175.	.449		
	6	745.	.925	128.	.384	214.	.496		
· · · ·	7	745.	.925	118.	.368	255.	.541		
├- -	8	684.	.887	58.4	.259	285.	.572		
	9	821.	.971	50.0	.240	275.	.562		

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Loral 1932	9B		ì	Ri	.ght Pod
Photo-	Headlight				
conductor	Voltage -	Gain	Range	Gain	Range
4	2.5	0	0	0	0
	3.0	•	1	Y	
	3.5	10125.	3.41	242.	.527
	4.0	11016.	3.56	291.	.578
	4.5	16200.	4.31	311.	.598
	5.0	12555.	3.80	388.	.668
	5.5	16200.	4.31	416.	.691
	6.0	17253.	4.45	442.	.712
	6.5	26487.	5.52	489.	.750
	7.0	27378.	5.61	439.	.710
	7.5	27702.	5.64	424.	.698
	8.0	29646.	5.84	471.	.736
L	8.5	30618.	5.93	602.	.831
	9.0	30618.	5,93	522.	.775
<u> </u>	9.5	2576.	1.72	522.	.775
3	2.5	0	0	26.8	.175
	3.0	5832.	2,59		
	3.5			160.	.429
	4.0	13365.	3.92	151.	.417
	4.5			201.	.481
	5.0	27540.	5.63	186.	.462
	5.5			186.	.462
· ·	6.0	112428.	11.37	238.	.523
	6.5	_		250.	.536
	7.0	2158650.	49.81	229.	.513
	7.5	•	•	233.	.517
	8.0	· .	• • •	239.	.524
	8.5	-		228.	.512
	9.0			273.	.560
	9.5			225.	.509
2	2.5			16.6	.138
	3.0	1307.			
	3.5			123.	.375
	4.0	2493.	•	135.	.394
	4.5			150.	.415
	5.0	3086.		173.	.446
	5.5		·	192.	.470
	6.0	3747.		227.	.510
	6.5			207.	.488
	7.0	4256.	· · · · · · · · · · · · · · · · · · ·	245.	.530
L	7.5		·	232.	.516
	8.0	4834.		243.	.529
	8.5			264.	.550
	9.0	5411.		264.	.550
V	9.5			288.	.575

Table XX

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Loral 19329B Right Pod								
Photo-	Sensitivity	Room D	aylight	Headl	ight			
conductor	Voltage.	Gain	Range	Gain	Range			
11	2.5			4.47	.072			
	3.0	791.						
	3.5			19.0	.148			
	4.0	1123.		12.0	.122			
	4.5			12.9	.139			
	5.0	1423.		16.4	.137			
	5.5			18.2	.145			
	6.0	1152.		19.7	.150			
	6.5			18.5	.146			
	7.0			25.5	.171			
	7.5			26.1	.173			
	8.0	287.		26.1	.173			
	8.5			24.3	.167			
	9.0	503.		24.7	.168			
V	9.5		· · · · · ·	29.9	.185			

 $\frac{1}{2} \sum_{i=1}^{n} \frac{1}{2} \sum_{i=1}^{n} \frac{1}$

Table XXI

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Xenon strobe. As a rough measure of the IR background, a Luna-Pro light meter was used whose sensitive area was covered with a RG-780 filter. On a typical day the readings were:

dark:	3
room background:	12 1/3
headlight (at 15°):	18 1/3
headlight (directly):	20 1/3

So one can assert that the dark test environment was 1/650 the room background in daylight and that the headlight (at 15°) was 64 times as bright as the room in daylight. Tables XX and XXI list the gain and range of the Loral as functions of the sensitivity voltage and the background.

The data collected with the headlight mounted at 15° must be interpreted with some care. The roughly parallel light from the headlight always flooded the ball lens at an elevation of 15° (for azimuth scans) and at an elevation of 15° larger than the elevation recorded. There will be instances, especially when the warning transfers from one channel to the next, when the headlight will be illuminating one photoconductor and data will be collected on another.

The data were collected by selecting a sensitivity voltage (usually 3 or 4 volts for convenience) and a background, and by varying either azimuth or elevation. Tables XXII thru XLI present the laboratory data for Loral system 19329B. The patterns are not as lobed as in the Fecker PWI, and this is attributable to the light "pipe" design for avoiding inter-detector gaps.

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Left Pod

		•• •			
		. • •			
		× * *	Sensitivity		Range
Azimuth	Elevation	Channel	Voltage	Gain	Nautical Miles
		· · · ·			
0°	24°	11	. 2	3.75	.066
· · · · · · · · · · · ·	20°	11	2	5.67	.081
	16°	11	2 .	11.0	.112
	12°	11	2	43.5	.224
	8° :	11	2	118.	.368
	4°	11	2	82.82	.309
· · · · · · · · · · · · · · · · · · ·	2°	11	2 :	81.28	.306
	.0°.	. 11 .	2	84.74	.312
•	-2°	11	2	72.19	.288
	-6°	: 11	2	58.1	.258
	-10°	11	2	23.4	.164
•	-14°	11	2	7.24	.091
	-18°	11	2	4.39	.071
	-21°	11	2	3.04	.059
40°	-25°	10	2	3.34	.062
	-22°	10	2	4.37	.071
	-18°	10	2	7.02	.090
*	-14°	10	2	12.6	.120
	-10°	10	2	33.7	.197
	-6°	10	2	107.	.351
	-2°	10	2	116.	.366
	<u> </u>	10	2	97.	.334
	2°	10	2	100.	.339
	<u>4</u> °	10	2	137.7	. 398
	8°	10	2	118.	.367
	12°	10	2	35.2	.201
	16°	10	2	11.	.110
	20°	10	2	5.67	.081
	24°	10	2	3.46	.063
67°	19°	9	2	3.5	.063
<u></u>	16°	9	2	5.1	.077
<u></u>	12°	9	2	14.	.127
· · · ·	<u>8°</u>	9	2	36.2	.204
	4°	9	2	59.2	.261
	2°	9	2	51.2	.243
	0°	9	2	56.8	, 256
	-2°	9	2	69.8	.283
	-6°	9	2	84.	.311
	-10°	9	2	39.6	.213
		+	+	+	+

Dark Test

Table XXII

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Loral 19329B

Left Pod

					=010 10a
Azimuth	Elevation	Channel	Sensitivity Voltage	Gain	Range Nautical Miles
67°	-18°	9	2	4.31	.070
	-20°	9	2	3.08	.059
107°	-19°	8	2 · ·	3.36	.062
	-14°	8	2 .	6.97	.089
	-10°	8	2	38.4	.210
	-4°	8	2	91.	. 323
	-2°	8	· · · 2	102.	.343
· · ·	0°	8 -	- 2	101.	.341
	· 2°	8 ·	2	98.7	.337
	4°	8	2	93.8	.328
	8°	8	2	35.	.200
	12°	8	2	21.5	.157
	16°	8	2	5.6	.080
	19°	8	2	3.46	.063

· Dark Test

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Table XXIII

-62-

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Loral 193	Loral 19329B Left Pod							
Azimuth	Elevation	Channel	Sensitivity Voltage	Gain	Range Nautical Miles			
135°	0°	8	3	2.76	056			
132°		8	3	3.55	.064			
129°		8	3	4.0	.068			
125°		8	3	47.8	.234			
121°		8	3	181.	. 455			
117°		8	3	361.	.644			
113°		8	3	806.	.962			
109°		8	3	3791.	2.10			
105°		8	3	5192.	2.44			
101°		8	3	3953.	2.13			
97°		8	3	4188.	2.19			
93°		8	3	1623.	1.37			
89°		8	3	1854.	1.46			
85°		8	3	1718.	1.41			
82°		8	3	1474.	1.3			
81°		9	3	1504.	1.31			
77°		9	3	7120.	2.86			
73°		9	3	3945.	2.13			
69°		9	3	2786.	1.79			
65°		9	3	3807.	2.1			
61°		9	3	2649.	1.74			
57°		9	3	5216.	2.45			
53°		9	3	2244.	1.61			
52°		9	3	1436.	1.28			
51°		10	3	1794.	1.44			
47°		10	. 3	6909.	2.82			
43°		10	3	5208.	2.45			
<u>39</u> °		10	3	5314.	2.47			
35°		10	3	29970.	5.87			
31°		10	3	12231.	3.75			
27°		10	3	10449.	3.47			
24°		10	3	6051.	2.64			
2.3°		11	3	20250.	4.82			
19°		11	3	603450.	26.3			
15°	<u> </u>	11	2	234.8	.519			
190	<u>├</u>	11	2	230.	.514			
110	<u> </u>	11	2	145.	.408			
70		11	2	120.	.372			
30		11	2	98.2	.336			
-1°	T	11	2	78.6	.301			

Dark Test

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Table XXIV

Loral 19329B Left Pod							
Azimuth	Elevation	Channel	Sensitivity Voltage	Gain	Range Nautical Miles		
-5°	0°	11	2	68.2	. 280		
9°		11	2	28.4	.181		
-13°		11	2	10.6	.110		
-17°		11	2	4.26	.070		
-21°		11	3	18.9	.147		
-24°		11	3	6.84	.089		

Dark Test

Table XXV

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Loral 1932	Loral 19329B Left Pod							
Azimuth	Elevation	Channel	Sensitivity Voltage	Gain	Range Nautical Miles			
121°	0°	8	2	2.78	.057			
<u>117°</u>		8	2	5.89	. 082			
113°		8	2	15.3	.133			
109°		8	2	65.	.274			
105°		8	2 ·	70.4	. 284			
101°		8	2	49.	.273			
97°		8	2	51.7	.244			
93°		8	2	54.2	.250			
89°		8	2	71.2	.286			
85°		8	2	69.8	.283			
82°	• •	8	2	25.5	.171			
81°		9	2	26.5	.175			
77°		9	2	69.8	.283			
73°		9	2	42.	. 22			
69°		9	2	37.5	.208			
65°		9	2	39.3	.213			
61°		9	2	40.7	.216			
57°		9	2	75.4	.294			
53°		9	2	37.3	.207			
52°		9	2	27.5	.178			
51°		10	2	25.3	.171			
47°		10	2	73.6	.291			
43°	·	10	2	68.6	.281			
39°		10	2	94.8	.330			
35°		10	2	130.	. 387			
31°		10	2	77.8	.300			
27°		10	2	92.4	. 326			
24°		10	2	66.	.275			
23°		10	2	43.2	.223			
22°		11	2	45.	.230			
<u>19°</u>		11	2	106.	.350			
15°		11	2	103.	.344			
11°		11	2	62.3	.286			
7°		11	2	57.	.256			
3°		11	2	47.	.232			
<u>-1°</u>		11	2	41.8	.22			
-5°_	·	11	2	44.	.225			
-9°		11	2	12.8	.121			
-13°	*	11	2	4.3	.070			
-15°		11	2	2.6	.055			

Daylight Test

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Loral 19329B Left Pod							
Azimuth	Elevation	Channel	Sensitivity Voltage	Gain	Range Nautical Miles		
0°	14°	11	2	3.23	. 061		
	11°	11	2	17.7	.143		
	8°.	11	2	37.	. 206		
	5°	11	2	37.2	.207		
	2°	11	2	35.8	.203		
	0°	11	2	37.7	.208		
	-2°	11	2	36.3	.204		
	-5°	11	2	34.	.200		
	-8°.	11	2	18.	.145		
	-11°	11	2	6.6	.087		
	-14°	11	2	2.7	.056		
40°	-19°	10	2	2.8	.056		
1	-16°	10	2	4.9	.075		
	-14°	10	2	7.4	.092		
	-11°	10	2	15.5	.133		
	-8°	10	2	41.	.217		
	-5°	10	2	59.	.260		
	-2°	10	2	72.	.287		
	0°	10	2	74.	.291		
	2°	10	2	74.	.291		
	5°	10	2	97.	.335		
	8°	10	2	91.	.323		
	11°	10	2	31.	.188		
	14°	10	2	8.5	.099		
	17°	10	2	4.4	.071		
	20°	10	2	2.7	.055		
67°	16°	9	2	2.8	.057		
	14°	9	2	4.3	.070		
· · ·	11°	9	2	12.3	.119		
	8°	9	2	27.	.176		
	5°	9	2	46.	.230		
	2°	9	2	48.	.235		
	0°	9	2	49.	.237		
	-2°	9	2	60.	.263		
	- 5°	9	2	68.	.279		
	-8°	9	2	42.	.220		
	-11°	9	2	24.	.166		
	-14°	9	2	6.99	.090		
	-17°	9	2	3.3	.061		
	-19°	9	2	2,5	.054		

Daylight Test

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Loral 19329B Left Pod					
Azimuth	Elevation	Channel	Sensitivity Voltage	Gain	Range Nautical Miles
107°	-19°	8	2	2.75	.056
	-17°	8	2	3.25	.060
	-14°	8	2	6.95	.089
	-11°	8	2	21.1	.156
	-8°	8	2	52.0	.244
	-5°	8	2	91.4	.324
	-2°	8	2	110.	.354
	0°	8	2	125.	.378
	2°	8	2	141.	. 402
	5°	8	2	121.	.372
	8°	8.	2	75.2	.294
	11°	8	2	29.4	.184
	14°	8	2	6.25	.085
¥	17.º	8	2	3.38	.062
	18°	8	2	2.64	.055

Daylight Test

Table XXVIII

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Loral 19329B Left Pod					
Azimuth	Elevation	Channel	Sensitivity Voltage	Gain	Range Nautical Miles
110°	110	8	2	2.43	.053
	8°	8	2	4.75	.074
	5°	8	2	-8.13	.097
	2°	8	2	7.18	.091
	0°	8	2	6.38	.086
	-2°	8	2	6.53	.087
	-5°	. 8	2	6.18	.084
	-8°	8	2	4.89	.075
	-10°	8	2	2.78	.057
67°	-11°	9	2	3.23	.061
	-8°	9	2	8.77	.100
	-5°	9	2	10.9	.112
	-2°	9	2	10.8	.111
	0°	9	2	9.23	.103
	2°	9	2	11.1	.113
	5°	9	2	11.0	.112
	8°	9	2	8.25	.097
	11°	9	2	3.53	.064
40°	11°	10	2	3.10	.060
	8°	10	2	7.89	.095
	5°	10	2	8.10	.096
	2°	10	2	7.36	.092
	0°	10	2	5.85	.082
	-2°	10	2	6.31	.085
	-5°	10	2	5.42	.079
	-8°	10	2	3.94	.067
0°	-10°	11	3	2.74	.056
1	-8°	11	3	3.44	.063
	-5°	11	3	9.21	.103
	-2°	11	3	9.16	.103
	0°	11	3	9.78	.106
	2°	11	3	11.8	.116
	5°.	11	3	8.58	.099
•	8°	11	3	6.63	.087
V	11°	11	3	3.53	.064
T	12°	11	3	2.69	.056

Headlight Background Test

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Loral 19329B Left Pod						
Azimuth	Elevation	Channel	Sensitivity Voltage	Gain	Range Nautical Miles	
1130	0°	8	2	2.57	. 054	
109°	Ĩ	8	2	6,90	.089	
105°		8	2	5.00	.076	
101°		8	2	5.37	.079	
97°		8	2	8.88	.101	
93°		8	2	8.70	.100	
89°		8	2	18.5	.145	
85°		8	2	15.1	.132	
82°		8	2	2.99	.059	
81°		9	2	2.97	.058	
77°		9	2	6.22	.085	
73°		9	2	4.06	.068	
69°		9	2	9.03	.102	
65°		9	2	8.12	.097	
61°_		9	2	10.9	.112	
57°		9	2	19.7	.150	
53°		9	2	5.91	.082	
52°		9	2	2.87	.057	
51°		10	2	2.71	.056	
47°		10	2	5,52	.080	
43°		10	2	4.41	.071	
39°		10	2	6.80	.088	
35°		10	2	6.63	.087	
31°		10	2	7.11	.090	
<u>27°</u>		10	2	10.5	.110	
23°		10	2	2.83	.057	
22°		11	2			
22°		10	3	16.0	.136	
<u>21°</u>		11	3	8.21	.097	
<u>17°</u>		11	3	8.02	.096	
13°		11	3	7.87	.095	
9°		11	3	13.6	.125	
5°		11	3	11.9	.117	
1°		11	3	8.86	.101	
-3°		11	3	9.50	.104	
<u>-7°</u>	¥	11	3	3.14	.060	
. –8°			-			

Headlight Background Test

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Loral 19329B Right Pod						
			Songitivity		Panga	
Arimuth	Flowstion	Channel	Voltage	Coin	Range	
AZIMULII	Lievalion	Channer	vorlage	Gain	Naucical Miles	
-30°	- 0°	1	4	3.79	.066	
-28	1	1	4	9,69	.106	
-26		1	4	17.6	.142	
-24		1	4	35.7	.203	
-22	•	1	4	43.1	.223	
-20		1	4	66.2	.276	
-18		1	4	111.	.359	
-16		1	4	125.	.379	
-14		1	4	230.	.514	
-12		1	4	323.	.609	
-10		1	4	392.	.671	
-8		1	4	510.	.766	
-6	· · · ·	1	4	511.	.766	
-4		1	4	663.	.873	
-2		1	4	752.	.930	
0		1	4	942.	1.04	
2		1	4	889.	1.01	
4		1	4	866.	.998	
6		1	4	803.	.961	
8		1	4	866.	.998	
10		1	4	874.	1.00	
12		1	4	505.	.762	
14		2	4	632.	.852	
16		2	4	935.	1.04	
18		2	4	1198.	1.17	
20		2	4	1582.	1.35	
22		2	4	1718.	1.41	
24	· · ·	2	4	1763.	1.42	
26		2	4	1672.	1.39	
28		· 2	4	1403.	1.27	
30		2	4	1403.	1.27	
32	·	2	4	1514.	1.32	
34		2	4	1593.	1.35	
36		2	4	2128.	1.56	
38		2	4	2402.	1.66	
40		2	4	2599.	1.73	
42		. 3	4	2113.	1.56	
44		3	4	3614.	2.04	
46		3	4	11096.	3.57	
48		3	4	10640.	3.50	

Dark Test
Loral 19329B Right Pod							
Azimuth	Elevation	Channel	Sensitivity Voltage	Gain	Range Nautical Miles		
50°	0°	3	4	9720.	3.34		
52	1	3	4	14726.	4.11		
54		3	4	10530.	3.48		
56		3	4	11502.	3.64		
58		3	4	8149.	3.06		
60		3	4	13592.	3.95		
62		3	4	16346.	4.33		
64		3	4	10854.	3.53		
66		3	4	9801.	3.36		
68		3	4	6107.	2.65		
70		3	4	8562.	3.14		
72		3	4	5937.	2.61		
74		3	4	2811.	1.80		
76		4	3	2371.	1.65		
78		4	3	5816.	2.59		
80		4	3	5249.	2.46		
82		4	3	6002.	2.63		
84		4	3	5184.	2.44		
86		4	3	5881.	2.60		
88		4	3	6772.	2.79		
90		4	3	6812.	2.80		
92		4	3	6812.	2.80		
94		4	3	4925.	2.38		
96		4	3	4058.	2.16		
98		4	3	5022.	2.40		
100		4	- 3	3848.	2.10		
102		4	3	2365.	1.65		
104		4	3	1663.	1.38		
106		4	3	1079.	1.11		
108_		4	3	645.	.861		
110		4	3	474.	.738		
112		4	3	279.	.566		
114		4	3	143.	.406		
116		4	3	87.7	.317		
118		4	3	15.6	.134		
120		4	3	3.12	.060		

Dark Test

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Loral 19329B Right Pod						
Azimuth	Elevation	Channel	Sensitivity Voltage	Gain	Range Nautical Miles	
-30°	0°	Ĩ	4	3.42	.063	
-27		1	4	11.5	.115	
-24		1	4	29.0	.183	
-21		1	4	50.0	.240	
-18		. 1	4	100.	.339	
-15		1	4	193.	.471	
-12		1	4	291.	.578	
-9		1	4	410.	.686	
-6		1	4	533.	.782	
-3		1	4	924.	1.03	
0		1	4	1111.	1.13	
3		1	4	1096.	1.12	
6		1	4	1042.	1.09	
9		1	4	1042.	1.09	
12		1	4	789.	.952	
16	· · · ·	2	4	689.	.890	
18		2	4	1696.	1.40	
21		2	4	1920.	1.49	
24		2	4	1892.	1.47	
27		2	4	1558.	1.34	
30		2	4	1687.	1.39	
33		2	4	1786.	1.43	
36		2	4	2075.	1.54	
39		2	4	2485.	1.69	
42	·····	2	4	1816.	1.44	
43		2	4	1490.	1.31	
44		3	4	1702.	1.40	
47		3	4	4332.	2.23	
50		3	4	7448.	2.93	
53		3	4	7549.	2.95	
56		3	4	7403.	2.92	
59		3	4	11502.	3.64	
62		3	4	13527.	3.94	
65		3	4	7298.	2.90	
68		3	4	6998.	2.84	
71		3	4	6577.	2.75	
74		3	4	3645.	2.05	
75		4	3	2052.	1.54	
78		4	3	5457.	2.50	
81	Y	4	3	4957.	2.39	

Daylight Test

Loral 19329	ЭB
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Right Pod

Azimuth	Elevation	Channel	Sensitivity Voltage	Gain	Range Nautical Miles	
84°	0°	4	3	4674.	2.32	
87		4	3	5500.	2.51	
90		4	3	4568.	2.29	
93		4	3	4334.	2.23	
96		4	3	4285.	2.22	
99		4	3	4520.	2.28	
102		4	3	2827.	1.80	
105		4	3	1733.	1.41	
108		4	3	836.	.980	
111		4	3	426.	.699	
114		4	3	202.	. 482	
117		4	3	77.0	.297	
120		4	3	8.47	.099	
123		4	3	5.83	.082	
126		4	3	4.65	.073	
129		4	3	3.71	.065	
130	•	4	3	3.34	.062	

Daylight Test

Table XXXIV

Loral 19329B Right Pod							
			Sensitivity		Range		
Azimuth	Elevation	Channel	Voltage	Gain	Nautical Miles		
	· · · · · · · · · · · · · · · · · · ·			······.			
0°	-15°	1	3	64.6	.273		
0	-13	1		114.	.361		
0	-10	1		228	.512		
0	-7	<u> </u>		362.	.645		
0	-4	<u> </u>	·	657.	.869		
0	-1	<u> </u>		657.	.869		
0	2	1		457.	.725		
0	5	1		489.	.750		
. 0	8	1		419.	.694		
0	11	1		214.	. 496		
0	14	1		106.	.349		
0	17	. 1		70.5	.245		
0	20	1		43.8	.224		
0	22	. 1		30.4	.187		
0.	24	1		21.8	.158		
<u>0</u> .	27	1		18.9	.147		
0	30	1		17.8	.143		
. 0	33	1		11.3	.114		
0	-18	1		60.0	.263		
0	-21	1		38.7	.211		
0	-24	1		29.5	184		
0	-27	1		18.2	.145		
0	-30	1		13.5	. 125		
39	-27	2	1	50.1	.240		
39	-24	2		49.3	.238		
39	-21	2		73.4	290		
39	-18	2		99.7	.338		
39	-15	2		186	462		
39	-12	2		325.	611		
39	-9	2		700	897		
39	-6	2		1357	1 25		
39	-3	2		1771	1 43		
39	0	2		1619	1 36		
39	3	2	+	1664	1 38		
39	6	2		1664	1 39		
39	9	2		1224	1 10		
39	12	2	· · · · ·	507	762		
39	15	2		215	602		
20	18	2	 	130	.002		
	21	2	<u> </u>	114	.400		
5,5				1 774.	.302		

Daylight Test

Loral 19329B

Right Pod

			Sensitivity		Range
Azimuth	Elevation	Channel	Voltage -	Gain	Nautical Miles
39°	24°	2	3	76.5	. 297
39	26	2	1	59.6	. 262
62	55	3		11.2	.113
62	51	3	t	16.1	.136
62	47	3	<u> </u>	22.0	.159
62	43	3	<u> </u>	28 7	182
62	39	3	+	20.7	213
62	35		+	58 9	260
62	31			77 4	298
62	27		·	103	314
62	27			172	• 5 4 4
62	10			211	<u>•444</u>
62	16			472	
62	10	$+$ $\frac{3}{3}$	-{	4/2.	./3/
62	13		+	928.	1.03
62	10	3	+	24/8.	1.69
62	1	3		3899.	2.12
62	4	3	<u> </u>	3884.	2.11
62	1	3		3709.	2.06
62	-2	3	·	4773.	2.34
62	-5	3	ļ	4081.	2.17
62	-8	3		1870.	1.47
62	-12	3		853.	.990
62	-16	3		360.	.643
62	-20	3		128.	. 384
62	-24	3		68.0	.280
62	-28	3		49.6	.239
62	-32	3		25.4	.171
62	-36	3		18.0	.144
62	-40	3		11.2	.113
62	-44	3		7.88	.095
62	-48	3		4.70	.073
100	-46	4		5.27	.078
100	-43	4		12.7	.121
100	-40	4		17.3	.141
100	-37	4		30.4	.187
100	-34	4	+	39.5	.213
100	-31	4	1	59.1	.261
100	-28	4	<u>+</u>	67.3	.278
100	-25	4	<u> </u>	109.	.354
100	-22	4	<u>+</u>	157.	.425
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Daylight Test

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Table XXXVI

Loral 193	29B				Right Pod
Azimuth	Elevation	Channel	Sensitivity Voltage	Gain	Range Nautical Miles
100°	-19°	4	3 '	298.	.585
	-16°	4		497.	.755
	-13	4		1042.	1.09
	-10	4		2029.	1.53
	-7	4		4932.	2.38
	- 4	4		5403.	2.49
	-1	4		5873.	2.60
	2	4		6383.	2.71
	5	4		6480.	2.73
	8	4		5063.	2.41
	11	4		3390.	1.97
	14	4		1292.	1.22
	17	4	•	932.	1.03
	20	4		339.	.624
	23	4		245.	.530
	26	4		195.	.473
	29	4		137.	. 397
	32	4		113.	.360
	35	4		48.0	.235
	38	4		41.5	.218
	41	4		36.7	.205
	44	4		31.0	.189
	47	4		21.7	.158
	50	4		14.5	.129
	53	4		8.10	.096
	56	4	· .	7.09	.090
4	59	4	· •	3.45	.063

Daylight Test

Loral 193	Loral 19329B Right Pod						
Azimuth	Elevation	Channel	Sensitivity Voltage	Gain	Range Nautical Miles		
-17°	0°	1	3	2.84	.057		
-13		1		13.6	.125		
-9		1		31.6	.191		
5		1		31.6	.191		
-1		1		31.6	.191		
3		11		36.4	.205		
7		11		43.0	.222		
11		1		46.5	.231		
15		1		42.5	.221		
17		1		33.6	.197		
18		2		38.3	.211		
22		2		73.0	.290		
26		2		80.8	.305		
29		2		106.	.349		
33		2		105.	.347		
37		2		84.7	.312		
41		2		84.7	.312		
45		2		68.1	.280		
46		2		56.6	.255		
47		3		61.8	.266		
51		3		134.	.393		
55		3		132.	.389		
59		3		128.	. 384		
63		3		125.	.379		
67		3		84.7	.312		
71		3	· · · · · · · · · · · · · · · · · · ·	109.	.354		
75		3		85.8	.314		
76		4		95.7	.332		
80		4		202.	.482		
84		4		245.	.530		
88		4		277.	.564		
92		4		257.	.544		
96		4		209.	.490		
100		4		264.	.550		
104		4		283.	.570		
108		4		183.	.459		
112		4		83.6	.310		
116	4	4		29.4	.184		
120		4		2.88	.058		

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Headlight Background Test

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Table XXXVIII

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Loral 19329B Right Pod						
Azimuth	Elevation	Channel	Sensitivity Voltage	Gain	Range Nautical Miles	
0°	28°	1	3	2.91	.058	
0	25	1		3.84	.066	
0	22	1	· · ·	4.48	.072	
0	19	1		6.00	.083	
0	16	1		7.89	.095	
0	12	1		15.8	.135	
0	9	1		26.5	,175	
0	6	1		36.5	.205	
0	3	1		35.5	.202	
0	0	1		37.0	.206	
0	-3	1		36.7	.205	
0	-7	1		27.1	.176	
0	-11	1		11.0	.112	
0	-15	1		7.62	.094	
0	-19	1		5.25	.078	
0	-23	1		4.03	.068	
0	-27	1		2.80	.057	
0	-30	1		2.78	.057	
32	36	2	· · ·	3.69	.065	
32	32	2		4.54	.072	
32	28	2		6.58	.087	
32	24	2		8.77	.100	
32	20	2		14.8	.130	
32	16 [,]	2		24.0	.166	
32	12	2		47.3	.233	
32	8	2		81.8	.307	
32	4	2		124.	.377	
32	2	2		126.	.380	
32	0	2		127.	.382	
32	-2	· 2		114.	.362	
32	-6	2		115.	.364	
32	-10	2		54.5	.250	
32	-14	2		25.0	.170	
32	-18	2		13.7	.125	
32	-22	2		9.95	.107	
32	-26	2		6.96	.089	
32	-30	2		4.66	.073	
32	-34	2		2.70	.056	
62	54	3		2.13	.049	
62	50	3		2.80	.057	

Headlight Background Test

Loral 19329B Right Pod							
			Sensitivity		Pango		
Azimuth	Elevation	Channel	Voltage	Gain	Nautical Miles		
	Dictación	onume r	VOltage	Gain	Huddieur Hiles		
62°	46°	3	3	4.40	.071		
62	42	3	1	5.86	.082		
62	38	3		7.15	.091		
62	34	3		10.2	.108		
62	30	3		10.3	.109		
62	26	3		15.8	.135		
62	22	3		17.5	.142		
62	18	3		25.9	.173		
62	14	3		39.9	.214		
62	10	3		79.2	.302		
62	6	3		93.7	.328		
62	4	3		109.	.354		
62	2	3		130.	.386		
62	0	3		122.	.375		
62	-2	3		131.	.388		
62	-6	3		118.	.369		
62	-10	3		46.7	.232		
62	-14	3		35.3	.201		
62	-18	3		23.7	.165		
62	-22	3		17.3	.141		
62	-26	3		12.0	.117		
62	-30	3		9.45	.104		
62	-34	3		6.00	.083		
62	-38	3		4.94	.075		
62	-42	3		2.96	.058		
100	58	4		2.67	.055		
100	54	4		4.87	.075		
100	50	4		5.76	.081		
100	46	4		11.1	.113		
100	42	4		13.7	.125		
100	38	4		13.0	.122		
100	34	4		16.2	.136		
100	30	4		17.6	.142		
100	26	4		20.0	.152		
100	22	4		23.0	.163		
100	18	4		35.9	.203		
100	14	4		52.8	.246		
100	10	4		131.	.388		
100	6	4		191.	.468		
100	2	4		214.	. 496		

Headlight Background Test

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Loral 193	29B	··.				. e	Right Pod
Azimuth	Elevation	Channel		Sensitivit Channel Voltage		Gain	Range Nautical Miles
100°	.0°	4	4 3 2		201.	.481	
	-2					233.	.517
	-6	-				227.	.510
	-10					130.	.386
	-14					53.4	.248
	-18					35.6	.202
	-22					24.0	.166
	-26					14.8	.130
	-30			· ·	• •	13.7	.125
	-34				·	13.3	.124
	-38					8.61	.099
	-42		1			5.31	.078
	-46					2.40	.053

Headlight Background Test

Table XLI

CONCLUSIONS

The laboratory evaluation has produced three significant results. The equipment (Fecker System) has been made operable and more reliable by the elimination of the RFI problem, the redesign of the photo diode biasing arrangement, frequency' response, lamp drivers, discriminator level setting, and a new design for a plexiglas shield. The second significant accomplishment from the laboratory tests is the "test set up and procedure". The instrumentation set up on the optical bench enables the calibration and testing of any PWI System using a flashing Xenon lamp as the cooperative element.

Finally, laboratory test results have produced lobe patterns for the Fecker and Loral optical heads, which are to be used in the upcoming flight tests. These patterns are presented as functions of azimuth and elevation angles referenced to axes fixed in the optical head.

These results are now being incorporated in a simulation being constructed by S. Ross & Co. This program will utilize as inputs these lobe patterns, along with models of atmospheric effects, etc. The basic function of the program is to calibrate simulation runs to actual flight test runs. This calibration will serve to fix various simulation modelling parameters, and, in general, improve the models. This done, a system parameter analysis may then be undertaken, resulting in a more optimum set of: instrument settings, test flight trajectories, flight parameters for given weather conditions, and so forth.

The severe lobe structures clearly indicate that the flight tests must be conducted in the region of maximum sensitivity of the lobe pattern. The lobe structure also drastically reduces the effectiveness of the present PWl systems. It is anticipated that a completely new redesign will eliminate this problem.

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